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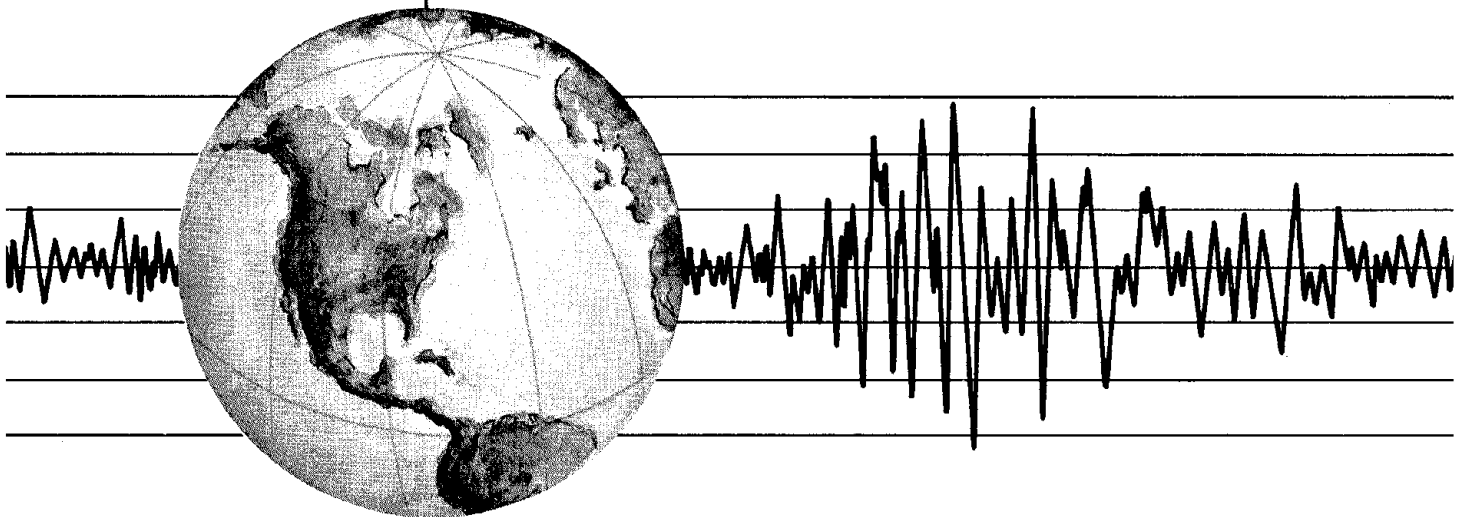
REPORT NO.
UCB/EERC-80/08
APRIL 1980

EARTHQUAKE ENGINEERING RESEARCH CENTER

STUDY OF X-BRACED STEEL FRAME STRUCTURES UNDER EARTHQUAKE SIMULATION

by
YUSOF GHANAAT

Report to
American Iron and Steel Institute
and
National Science Foundation



COLLEGE OF ENGINEERING

UNIVERSITY OF CALIFORNIA · Berkeley, California

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REPORT DOCUMENTATION PAGE		1. REPORT NO. NSF/RA-800194	2.	3. Recipient's Accession No. PB81 122335	
4. Title and Subtitle Study of X-Braced Steel Frame Structures Under Earthquake Simulation				5. Report Date April 1980	
7. Author(s) Yusof Ghanaat				8. Performing Organization Rept. No. UCB/EERC-80/08	
9. Performing Organization Name and Address Earthquake Engineering Research Center University of California, Richmond Field Station 47th and Hoffman Blvd. Richmond, California 94804				10. Project/Task/Work Unit No.	
12. Sponsoring Organization Name and Address National Science Foundation 1800 G Street, N.W. Washington, D. C. 20550				11. Contract(C) or Grant(G) No. (C) (G) PFR-7908257	
15. Supplementary Notes				13. Type of Report & Period Covered	
16. Abstract (Limit: 200 words) Diagonal steel bracing systems are intended to limit lateral displacements of buildings when subjected to wind or other lateral loadings. Many existing buildings with such bracing were designed on the basis of nominal building code requirements for wind, with no consideration of the function of ductility in seismic response. Although the seismic behavior of such structures has been studied analytically, no experiments had been performed previously for verification of the analytical results. This report presents experimental shaking table test results on the seismic performance of a model three-story building frame, both unbraced and with three different wind bracing systems; and correlates these results with analytical predictions. Considerable compression buckling and tension yielding of the diagonal bracing members were observed in the tests, but the bracing provided significant reductions in the lateral displacements when compared with the unbraced frame response. Analytical techniques employing three different hysteresis models to represent the three types of bracing systems are shown to predict the response of braced frames with excellent accuracy. Analytical response predictions for the unbraced frame, employing concentrated bilinear plastic hinges for all members including joint connections, also are shown to be very accurate for the levels of nonlinearity encountered. The results of this study indicate that diagonal bracing systems such as pipe and double angle braces are very effective in reducing lateral displacements of buildings for moderate earthquakes and that their energy dissipation will be significant if their compressive capacity is not less than 50 percent of their tension capacity. Consequently, damage to both the primary structural members as well as non-structural components can be reduced by the use of appropriate light weight diagonal bracing systems.				14.	
18. Availability Statement: Release Unlimited			19. Security Class (This Report)		21. No. of Pages
			20. Security Class (This Page)		22. Price

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ABSTRACT

Diagonal steel bracing systems are intended to limit lateral displacements of buildings when subjected to wind or other lateral loadings. Many existing buildings with such bracing were designed on the basis of nominal building code requirements for wind, with no consideration of the function of ductility in seismic response. Although the seismic behavior of such structures has been studied analytically, no experiments had been performed previously for verification of the analytical results.

This report presents experimental results on the seismic performance of a model three-story building frame, both unbraced and with three different wind bracing systems; and correlates these results with analytical predictions. The experimental investigation was carried out on the shaking table of the U.C. Berkeley Earthquake Simulator Laboratory. Considerable compression buckling and tension yielding of the diagonal bracing members were observed in the tests, but the bracing provided significant reductions in the lateral displacements when compared with the unbraced frame response.

Analytical techniques employing three different hysteresis models to represent the three types of bracing systems are shown to predict the response of braced frames with excellent accuracy. The mathematical model of the rod braces simulated both tension yielding and elastic buckling with tension rod rupture mechanism included; pipe and double angle bracing members included both tension yielding and post-buckling behavior; residual elongation and reduction of compressive capacity with the number of cycles was considered in the double angle model. Analytical response predictions for the unbraced frame, employing concentrated bilinear plastic hinges for all members including joint connections, also are shown to be

very accurate for the levels of nonlinearity encountered.

The results of this study indicate that diagonal bracing systems such as pipe and double angle braces are very effective in reducing lateral displacements of buildings for moderate earthquakes and that their energy dissipation will be significant if their compressive capacity is not less than 50 percent of their tension capacity. Consequently, damage to both the primary structural members as well as non-structural components can be reduced by the use of appropriate light weight diagonal bracing systems.

ACKNOWLEDGMENTS

The support of the American Iron and Steel Institute in fabrication of the test structures and of the National Science Foundation in funding the research program described herein is gratefully acknowledged.

The author is grateful to Professor Graham Powell and his students for developing a very suitable nonlinear program, DRAIN-2D, used extensively in the analytical portion of this research. Thanks are also extended to Professor Egor P. Popov and his former student Dr. Charles W. Roeder, and also to Professor Subhash C. Goel and Dr. Ashok K. Jain for their post-buckling truss elements developed for use in DRAIN-2D program.

The author wishes to express his deep appreciation to Professor Ray W. Clough for his supervision, patience and support throughout the course of this study. In addition, he would like to extend his appreciation to Professor Jack G. Bouwkamp and Professor C. D. Mote for serving as members of his thesis committee. Particularly, Professor Bouwkamp's valuable guidance on design of the bracing members is gratefully acknowledged.

The University of California, Berkeley, Earthquake Engineering Research Center and the Computer Center of the Berkeley Campus provided excellent facilities for carrying out this research program. The assistance of many individuals in the Earthquake Simulator Laboratory and elsewhere is gratefully acknowledged. Special thanks are due to Mr. David Steere whose extensive efforts in instrument calibration and operation of the shaking table itself made a successful test program possible.

The manuscript was typed by Ms. Ruth Horning; numerous figures were prepared by Mr. Larry Bell.

Finally, but most sincerely, the author would like to acknowledge the loyalty, encouragement, and understanding of his fiancée, Shahrzad, which gave this study a purpose beyond its mere academic value.

TABLE OF CONTENTS

	<u>Page</u>
ABSTRACT	i
ACKNOWLEDGMENTS	iii
TABLE OF CONTENTS	v
LIST OF TABLES	viii
LIST OF FIGURES	viii
1. INTRODUCTION	1
1.1 Background	1
1.2 Objectives and Scope	3
2. TEST FACILITIES	5
2.1 Earthquake Simulator Laboratory	5
2.2 Data Acquisition System	6
3. TEST STRUCTURES	11
3.1 Moment Resistant Steel Frame Model	11
3.2 Design Criteria for the Bracing Members	15
4. INSTRUMENTATION	22
4.1 Acceleration Measurement	22
4.2 Displacement Measurement	23
4.3 Force Measurement	23
4.4 Local Deformation Measurement	24
4.5 Noise Level and Accuracy of the Experimental Data	25
5. EXPERIMENTAL STUDY	35
5.1 Rod X-Bracing Tests	35
5.1a El Centro Span 100 Test	38
5.1b El Centro Span 1000 Test	39

	<u>Page</u>
5.1c El Centro Span 950 Test	40
5.1d Pacoima Earthquake (1.129 g) Test	41
5.2 Pipe Diagonal Bracing Tests	45
5.2a El Centro Span 100 Test	49
5.2b El Centro Span 400 Test	50
5.2c Pacoima Span 400 Test	51
5.2d Pacoima Span 600 Test	52
5.3 Double Angle Diagonal Bracing Tests	55
5.3a El Centro Span 100 Test	60
5.3b El Centro Span 900 Test	60
5.3c Pacoima Span 800 Test	62
5.4 Unbraced Frame Tests	65
5.4a El Centro Span 100 Test	67
5.4b Pacoima Span 400 Test	67
6. ANALYTICAL STUDY	139
6.1 Rod Bracing System	141
6.1a Correlation with El Centro Span 100 Test . . .	141
6.1b Correlation with El Centro Span 1000 Test . . .	142
6.1c Correlation with Pacoima Earthquake (1.129 g) Test	143
6.2 Pipe Bracing System	154
6.2a Correlation with El Centro Span 400 Test . . .	155
6.2b Correlation with Pacoima Span 400 Test	155
6.3 Double Angle Bracing System	164
(Correlation with El Centro Span 900 Test)	
6.4 Unbraced Frame Structure	168
6.4a Correlation with El Centro Span 400 Test . . .	168

	<u>Page</u>
6.4b Correlation with Paxoima Span 400 Test	169
7. COMPARISON OF DIFFERENT BRACING SYSTEMS	175
8. CONCLUSIONS	185
REFERENCES	188
APPENDIX A	190

LIST OF TABLES

	<u>Page</u>
Table 5.1.A Rod Bracing Tests.....	37
Table 5.2.A Pipe Bracing Tests	47
Table 5.3.A Double Angle Bracing Tests	57
Table 5.4.A Unbraced Frame Tests	66

LIST OF FIGURES

Figure 2.1.1 Test Facility.....	8
Figure 2.1.2 Shaking Table Motion Capabilities.....	9
Figure 2.1.3 Pitch Frequency Response for Unloaded Table.....	10
Figure 3.1.1 Test Structure with Rod Braces on the Shaking Table	12
Figure 3.1.2 Test Structure with Double Angle Braces on the Shaking Table.....	13
Figure 3.1.3 Front Elevation of the Test Structure with Pipe Braces.....	14
Figure 3.2.1 Details of the Pipe Diagonal Braces.....	18
Figure 3.2.2 Details of the Double Angle Diagonal Braces.....	19
Figure 3.2.3 Details of Cross Beam-to-Column Connection.....	20
Figure 3.2.4 Details of Column Base Connection.....	21
Figure 4.3.1 Elastic Flexural and Axial Strain Gage Stations...	30
Figure 4.4.1 Post-Yield Flexural and Axial Strain Gage Stations (Rod Braced Structure).....	31
Figure 4.4.2 Post-Yield Flexural and Axial Strain Gage Stations (Pipe Braced Structure).....	32
Figure 4.4.3 Post-Yield Flexural and Axial Strain Gage Stations (Angle Braced Structure).....	33
Figure 4.4.4 DCDT Transducer Stations.....	34
Figure 5.2.1 Photographs of Compression Buckling and Tension Yielding of Pipe Diagonal Braces.....	48
Figure 5.3.1 Photographs of the 1st Floor Buckled Double Angle Diagonal Braces.....	58
Figure 5.3.2 Photographs of Buckled and Damaged Double Angle Braces.....	59

EL CENTRO SPAN 100 TEST - ROD BRACED STRUCTURE

	<u>Page</u>
Figure 5.1a.1 El Centro Span 100 Horizontal Table Motion.....	69
Figure 5.1a.2 Table and Floor Accelerations.....	70
Figure 5.1a.3 Table and Floor Displacements.....	71
Figure 5.1a.4 Floor Shear Forces.....	72
Figure 5.1a.5 Strain Time Histories of the 1st Floor Rod Braces.	73

EL CENTRO SPAN 1000 - ROD BRACED STRUCTURE

Figure 5.1b.1 El Centro Span 1000 Horizontal Table Motion.....	74
Figure 5.1b.2 Table and Floor Accelerations.....	75
Figure 5.1b.3 Table and Floor Displacements.....	76
Figure 5.1b.4 Floor Shear Forces.....	77
Figure 5.1b.5 First Floor Shear-Displacement Hysteresis Loops...	78
Figure 5.1b.6 Strain Time Histories of the 1st Floor Rod Braces.	79
Figure 5.1b.7 First Floor Column Moment-Strain Hysteresis Loops.	80

EL CENTRO SPAN 950 - ROD BRACED STRUCTURE

Figure 5.1c.1 El Centro Span 950 Horizontal Table Motion.....	81
Figure 5.1c.2 Table and Floor Accelerations.....	82
Figure 5.1c.3 Table and Floor Displacements.....	83
Figure 5.1c.4 Floor Drifts.....	84
Figure 5.1c.5 Floor Shear Forces.....	85
Figure 5.1c.6 Strain Time Histories of the 1st Floor Rod Braces.	86
Figure 5.1c.7 First Floor Column Moment-Strain Hysteresis Loops.	87

PACOIMA EARTHQUAKE (1.129 G) - ROD BRACED STRUCTURE

Figure 5.1d.1 Pacoima Record (1.129 g) Horizontal Table Motion..	88
Figure 5.1d.2 Table and Floor Accelerations.....	89
Figure 5.1d.3 Table and Floor Displacements.....	90
Figure 5.1d.4 Floor Drifts.....	91
Figure 5.1d.5 Floor Shear Forces.....	92
Figure 5.1d.6 Strain Time Histories of the 1st Floor Rod Braces.	93
Figure 5.1d.7 First Floor Shear-Displacement Hysteresis Loops...	94

	<u>Page</u>
Figure 5.1d.8 First Floor Column Moment-Strain Hysteresis Loops.	95

EL CENTRO SPAN 100 - PIPE BRACED STRUCTURE

Figure 5.2a.1 Table and Floor Accelerations.....	96
Figure 5.2a.2 Table and Floor Displacements.....	97
Figure 5.2a.3 Strain Time-Histories of the 1st Floor Pipe Braces	98
Figure 5.2a.4 Floor Shear Forces.....	99

EL CENTRO SPAN 400 - PIPE BRACED STRUCTURE

Figure 5.2b.1 Table and Floor Accelerations.....	100
Figure 5.2b.2 Table and Floor Displacements.....	101
Figure 5.2b.3 Floor Shear Forces.....	102
Figure 5.2b.4 First Floor Pipe Force-Displacement Hysteresis Loops.....	103
Figure 5.2b.5 First Floor Shear-Displacement Hysteresis Loops...	104

PACOIMA SPAN 400 - PIPE BRACED STRUCTURE

Figure 5.2c.1 Table and Floor Accelerations.....	105
Figure 5.2c.2 Table and Floor Displacements.....	106
Figure 5.2c.3 Floor Drifts.....	107
Figure 5.2c.4 Floor Shear Forces.....	108
Figure 5.2c.5 First Floor Shear-Displacement Hysteresis Loops...	109
Figure 5.2c.6 First Floor Column Moment-Curvature Hysteresis Loops.....	110

PACOIMA SPAN 600 - PIPE BRACED STRUCTURE

Figure 5.2d.1 Table and Floor Accelerations.....	111
Figure 5.2d.2 Table and Floor Displacements.....	112
Figure 5.2d.3 Floor Drifts.....	113
Figure 5.2d.4 Floor Shear Forces.....	114
Figure 5.2d.5 First Floor Shear-Displacement Hysteresis Loops...	115

	<u>Page</u>
Figure 5.2d.6 First Floor Column Moment-Curvature Hysteresis Loops.....	116
EL CENTRO SPAN 100 - DOUBLE ANGLE BRACED STRUCTURE	
Figure 5.3a.1 Table and Floor Accelerations.....	117
Figure 5.3a.2 Table and Floor Displacements.....	118
Figure 5.3a.3 Floor Shear Forces.....	119
EL CENTRO SPAN 900 - DOUBLE ANGLE BRACED STRUCTURE	
Figure 5.3b.1 Table and Floor Accelerations.....	120
Figure 5.3b.2 Table and Floor Displacements.....	121
Figure 5.3b.3 Floor Shear Forces.....	122
Figure 5.3b.4 First Floor Double Angle Force-Displacement Hys- teresis Loops.....	123
Figure 5.3b.5 First Floor Shear-Displacement Hysteresis Loops...	124
PACOIMA SPAN 800 - DOUBLE ANGLE BRACED STRUCTURE	
Figure 5.3c.1 Table and Floor Accelerations.....	125
Figure 5.3c.2 Table and Floor Displacements.....	126
Figure 5.3c.3 Floor Drifts.....	127
Figure 5.3c.4 Floor Shear Forces.....	128
Figure 5.3c.5 First Floor Shear-Displacements Hysteresis Loops..	129
Figure 5.3c.6 First Floor Column Moment-Curvature Hysteresis Loops.....	130
EL CENTRO SPAN 100 - UNBRACED STRUCTURE	
Figure 5.4a.1 Table and Floor Accelerations.....	131
Figure 5.4a.2 Table and Floor Displacements.....	132
Figure 5.4a.3 Floor Drifts.....	133
Figure 5.4a.4 Floor Shear Forces.....	134

	<u>Page</u>
PACOIMA SPAN 400 - UNBRACED STRUCTURE	
Figure 5.4b.1 Table and Floor Accelerations.....	135
Figure 5.4b.2 Table and Floor Displacements.....	136
Figure 5.4b.3 Floor Drifts.....	137
Figure 5.4b.4 First Floor Column Moment-Curvature Hysteresis Loops.....	138
Figure 6.1a.1 Mathematical Model 1 with Rod Diagonal Braces.....	145
Figure 6.1a.2 Force-Displacement for Rod Bracing Members.....	146
Figure 6.1a.3 Correlation of the Floor Displacements - Model 1 (EC 100 - Rod Braced Structure).....	147
Figure 6.1a.4 Calculated Periods and Mode Shapes (Rod Braced Structure).....	148
Figure 6.1b.1 Mathematical Model 2 (Shaking Table-Structure Interaction Included)....	149
Figure 6.1b.2 Correlation of Floor Displacements - Model 2 (EC 1000 - Rod Braced Structure).....	150
Figure 6.1b.3 Correlation of Global and Local Forces-Model 2 (EC 1000 - Rod Braced Structure).....	151
Figure 6.1c.1 Correlation of Floor Displacements - Model 3 (Pacoima Record - Rod Braced Structure).....	152
Figure 6.1c.2 Correlation of Global and Local Forces - Model 3 (Pacoima Record - Rod Braced Structure).....	153
Figure 6.2.1 Force-Displacement of Post-Buckling Brace Element.	157
Figure 6.2.2 Proposed Force-Displacement Model for Pipe Bracing Members.....	158
Figure 6.2.3 Mathematical Model for Pipe and Double Angle Braced Structures.....	159
Figure 6.2.4 Correlation of Floor Displacements (EC 400 - Pipe Braced Structure).....	160
Figure 6.2.5 Correlation of Column Forces (EC 400 - Pipe Braced Structure).....	161
Figure 6.2.6 Correlation of Floor Displacements (PAC 400 - Pipe Braced Structure).....	162
Figure 6.2.7 Correlation of Global and Local Forces (PAC 400 - Pipe Braced Structure).....	163
Figure 6.3.1 Axial Hysteresis Behavior of Double Angle Braces..	165
Figure 6.3.2 Correlation of Floor Displacements (EC 900 - Double Angle Braced Structure).....	166

	<u>Page</u>
Figure 6.3.3 Correlation of Global and Local Forces (EC 900 - Double Angle Braced Structure).....	167
Figure 6.4.1 Mathematical Model for Unbraced Frame.....	171
Figure 6.4.2 Correlation of Floor Displacements (EC 400 - Unbraced Frame).....	172
Figure 6.4.3 Correlation of Floor Displacements (PAC 400 - Unbraced Frame).....	173
Figure 6.4.4 Correlation of Column Forces (PAC 400 - Unbraced Frame).....	174
Figure 7.1 Maximum 1st Floor Drift vs. Peak Table Acc.....	179
Figure 7.2 Maximum 2nd Floor Drift vs. Peak Table Acc.....	180
Figure 7.3 Maximum 3rd Floor Drift vs. Peak Table Acc.....	181
Figure 7.4 Maximum 1st Floor Shear vs. Peak Table Acc.....	182
Figure 7.5 Maximum 2nd Floor Shear vs. Peak Table Acc.....	183
Figure 7.6 Maximum 3rd Floor Shear vs. Peak Table Acc.....	184



1. INTRODUCTION

1.1 Background

Diagonal steel bracing systems are frequently used to control the lateral displacements of buildings that are designed for wind or other lateral loadings. Many existing buildings of this type were designed on the basis of nominal building code requirements, with no consideration for modern concepts of ductility. When such structures are subjected to earthquake motions of even moderate intensity, the bracing members typically yield in tension and/or buckle in compression. Some analytical studies have been made of the seismic behavior of buildings with diagonal bracing designed for wind, but experimental research on this subject has not previously been done.

Assuming an elastic resistance mechanism, Clough and Jenschke⁽¹⁾ used computer procedures to study the seismic behavior of two buildings with supplemental diagonal braces; also some evidence about their dynamic behavior was obtained from observation of actual earthquake performance (see Reference 1). Inelastic behavior of braced frame structures is closely related to the hysteretic behavior of the bracing members. This behavior makes the response of a braced frame more difficult than that of an unbraced frame because of the complicated buckling-yielding mechanism. The earliest analytical studies of the inelastic behavior of braced frames were based on the assumption of a "slip-type" behavior for the bracing members. The slip model assumes the presence of two cross braces with each alternately becoming inactive during the application of cyclic loading. Hanson and Fan⁽²⁾, Workman⁽³⁾, and Goel and Hanson⁽⁴⁾ employed this model in their analyses, in which an elasto-plastic resistance mechanism was assumed in tension only for the slender

bracing members and the compression capacity of these members was neglected. This model is not too unrealistic for extremely slender braces, but it cannot be justified for braces having small to moderate slenderness ratios. Hence obtaining a model with a more realistic hysteresis behavior for the bracing members became the object of many later studies.

The cyclic behavior of individual bracing members with different slenderness ratios has been studied both analytically and experimentally by many investigators, in Japan and in the United States. These studies have been summarized and reported fully by Popov, Takanashi, and Roeder⁽⁵⁾. The results of these studies indicate that the general cyclic behavior of less slender braces differs significantly from the slip model; in particular it exhibits considerable energy dissipation in the inelastic response. These studies also suggest that the plastic rotation due to inelastic behavior in the post-buckling range is concentrated in a region near the middle of the brace. Many analytical studies have been performed to calculate the general cyclic behavior of bracing members, but most of them are either excessively complex or otherwise impractical for analysis of large structures. One simple method of analyzing the inelastic behavior of a braced frame was proposed by Nilforoushan⁽⁶⁾. His analysis was based on a straight line segment approximation of the general hysteretic behavior of a brace, and included the post-buckling behavior. The general force-displacement behavior of the brace is approximated by a series of straight lines selected to get the best fit. Nilforoushan used this model to perform dynamic analyses of several concentrically K-braced structures (in concentrically K-braced frames, the center line intersection of the braces intersects the center line of the beam). A similar hysteresis model

was developed by Roeder and Popov⁽⁷⁾, and was adapted for use in the program DRAIN-2D⁽⁸⁾. They used this model to analyze the dynamic behavior of an eccentric braced structure. Note that the eccentric braces employed by Roeder and Popov were quite heavy and differ greatly from the wind bracing members which are the subject of the present study; the eccentric braces deliberately introduce large eccentricities between the brace-beam connection and the beam-column joint (i.e. the center line of the brace does not intersect the center line intersection of the beam and column), to ensure that the eccentric beam element yields in shear while preventing buckling of the brace. An alternative mathematical hysteresis model was developed later by Singh⁽⁹⁾, which consisted of fewer linear segments. Subsequently, Jain and Goel⁽¹⁰⁾ presented another hysteresis model which represents the post-buckling behavior of bracing members in a more realistic manner, and includes the residual elongation in tension and reduction in compressive capacity as a function of the number of cycles.

All of these studies were limited either to experimental study of the individual members or to analytical studies of complete frames which were not verified by experimental results. The purpose of the present investigation was to perform dynamic tests on a building frame with diagonal wind bracing systems, and to correlate these results with computer analyses.

1.2 Objectives and Scope

The specific objective of this research was to obtain experimental data on the seismic performance of a building frame having three different diagonal wind bracing systems, and to correlate these results with computer analyses. The actual response results were also utilized to

demonstrate the effectiveness of an existing nonlinear structural program in the analysis of diagonal bracing systems. Of particular interest was the adequacy of the available bracing elements in that program. The experimental and analytical responses were then used to compare the performances of the bracing systems.

2. TEST FACILITIES

2.1 Earthquake Simulator

The test program was carried out at the Earthquake Simulator Laboratory, located at the Richmond Field Station of the University of California, Berkeley. The primary facility at this laboratory is a 20-ft square shaking table and its associated control systems. A complete description of this has been reported by Rea and Penzien⁽¹¹⁾. A brief description along with the new modifications is given here.

The reinforced, post-tensioned concrete shaking table shown in Fig. 2.1.1b is able to move independently in one horizontal component and in the vertical direction; and is driven by three 50 kip, and four 20 kip hydraulic actuators, respectively. During the test operation the dead weight of the shaking table (100 kips) and of the test structure are balanced by differential air pressure which frees the vertical actuators from carrying any static load.

The capabilities for both the horizontal and vertical motions of the table are illustrated in Fig. 2.1.2. At frequencies lower than one cps, the intensity of motion is limited by the actuator strokes; at intermediate frequencies from 1 to 4 cps, by the maximum actuator velocity; at frequencies greater than 4 cps, by the maximum actuator force capabilities and the oil column resonance of the drive system.

The command signals of the table are in the form of analog displacement time histories on magnetic tape which are usually obtained through a double integration of acceleration time histories. The shaking table was originally operating with only an active stabilization system to resist overturning moments. Recently a passive

stabilization system consisting of four vertical stabilizers was installed; this is described below.

Passive Stabilizers

The vertical active actuators do not have sufficient force capacity to resist the overturning forces that would be generated by the largest structures that were envisioned for testing on the table. Thus it was planned to install a passive stabilization system for the shaking table that would have a larger overturning moment capacity than the active system provided.

A vertical passive stabilizer system increases the shaking table stiffness and overturning moment capacity in its pitching mode. The effect of the passive stabilizer system on the pitching mode stiffness was determined from frequency response functions for the unloaded table and from the maximum pitch of the table while it was subjecting a three-story steel frame to the El Centro ground motion.

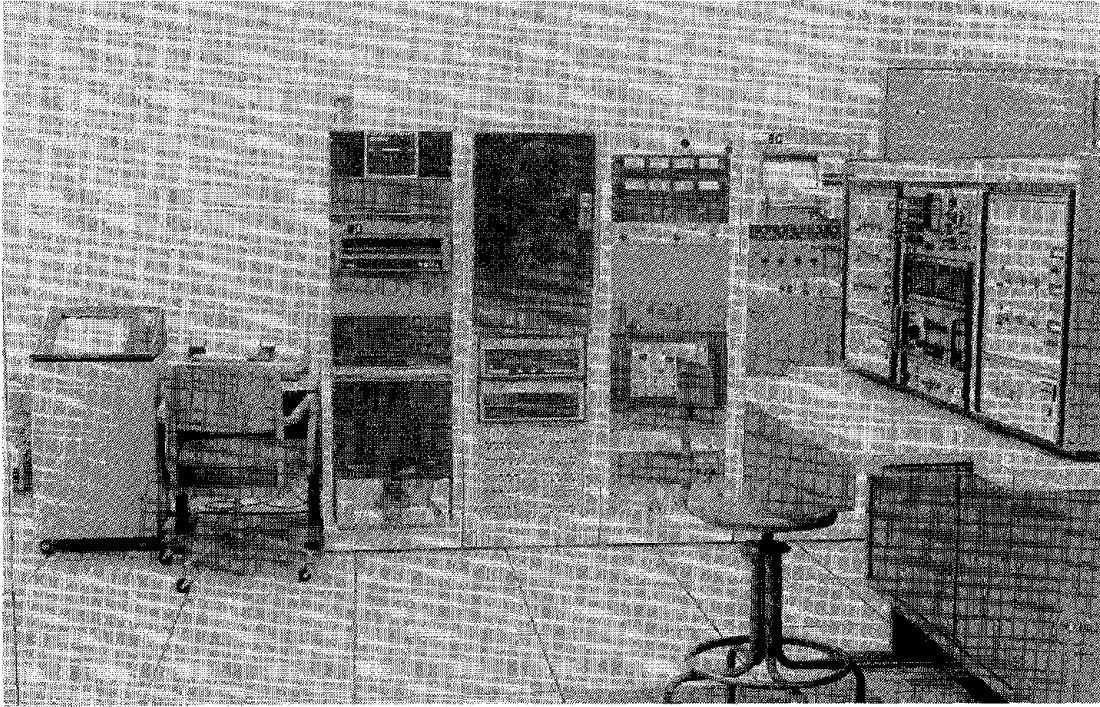
The horizontal motion causes the table to pitch. Pitch frequency response for the table without the passive stabilizers and with the passive stabilizers at operating pressures of 100, 500, 1000, and 1500 psi are shown in Fig. 2.1.3. The pitch resonant frequency of the table before the passive stabilizers were installed was 13.0 cps. The passive stabilizers for operating pressure above 100 psi, increase the pitch resonant frequency to 26 cps.

2.2 Data Acquisition System

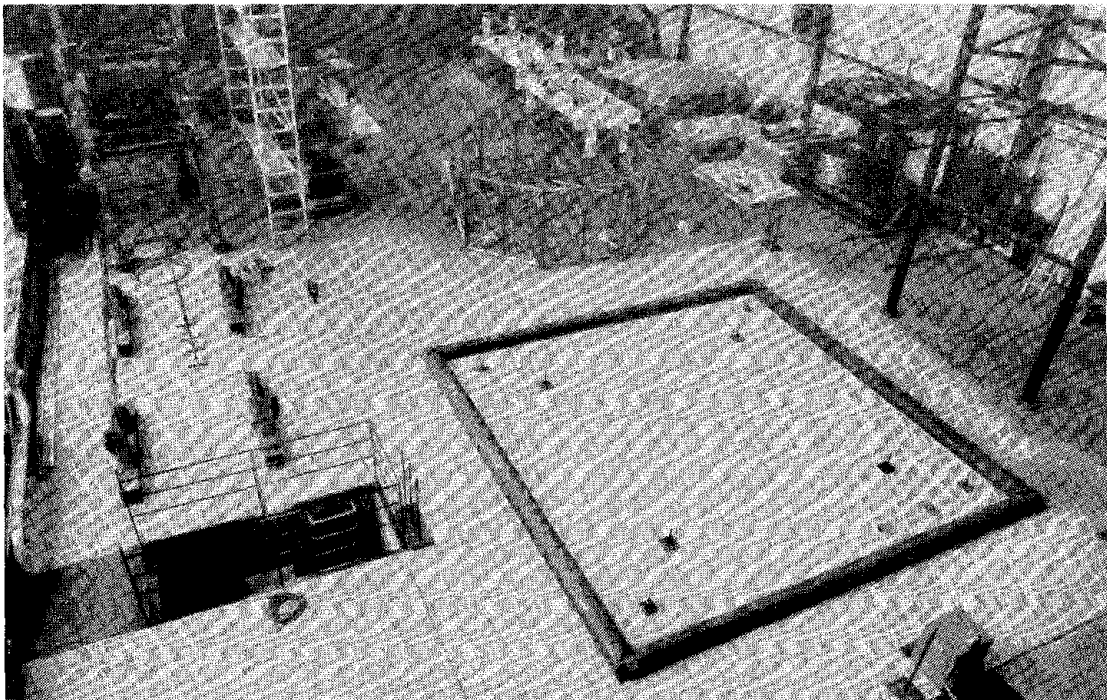
A data acquisition system consists of a NOVA 1200 mini-computer and a NEFF System 620 Analog-Digital processor, whose prime function is the collection of data during a test. The NOVA mini-computer is equipped with a Diablo 31 magnetic disk unit, which is capable of digitally

sampling up to 128 data channels at rates up to 155 samples per second, per channel. Transducer signals in analog form pass through an Analog-Digital processor. The digitized data are then temporarily stored on the magnetic disk before being transferred to tape by a Wang 9-track magnetic tape drive for permanent storage.

Limited data reduction for immediate evaluation of test results can be performed on the mini-computer, but for major data reduction operations the CDC 6400 Computer System at the Berkeley Campus is utilized. In order to be compatible with the CDC system, a conversion to 7-track magnetic tape must be carried out on the data. An extra magnetic tape drive system has recently been installed for the mini-computer to perform this conversion at the Earthquake Simulator Laboratory. The transformed data are then generally displayed in a graphical form using the Calcomp Plotting system at the computer center at the Berkeley campus.



a. Control Room



b. Shaking Table

Fig. 2.1.1 Test Facility

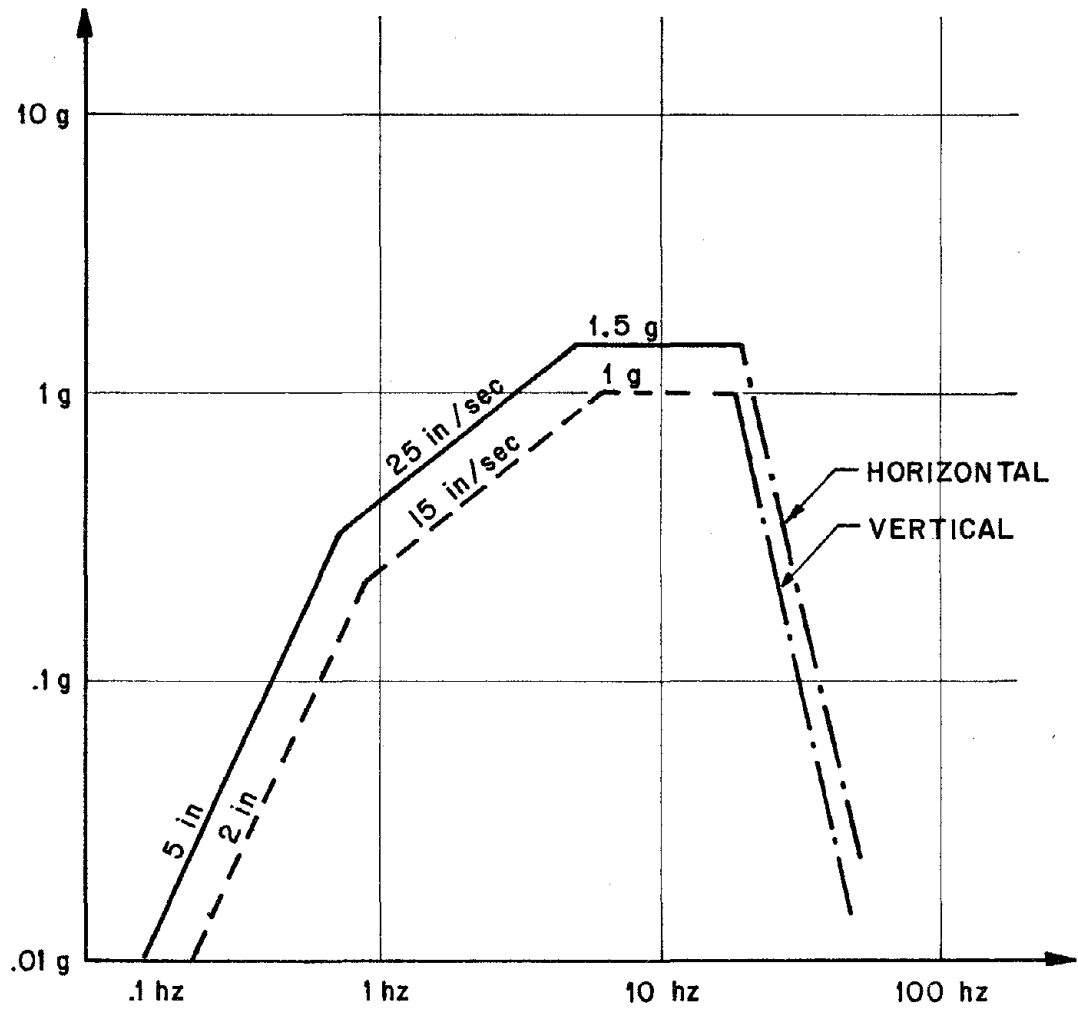


Fig. 2.1.2 Shaking Table Motion Capabilities
(after D. Rea and J. Penzien)

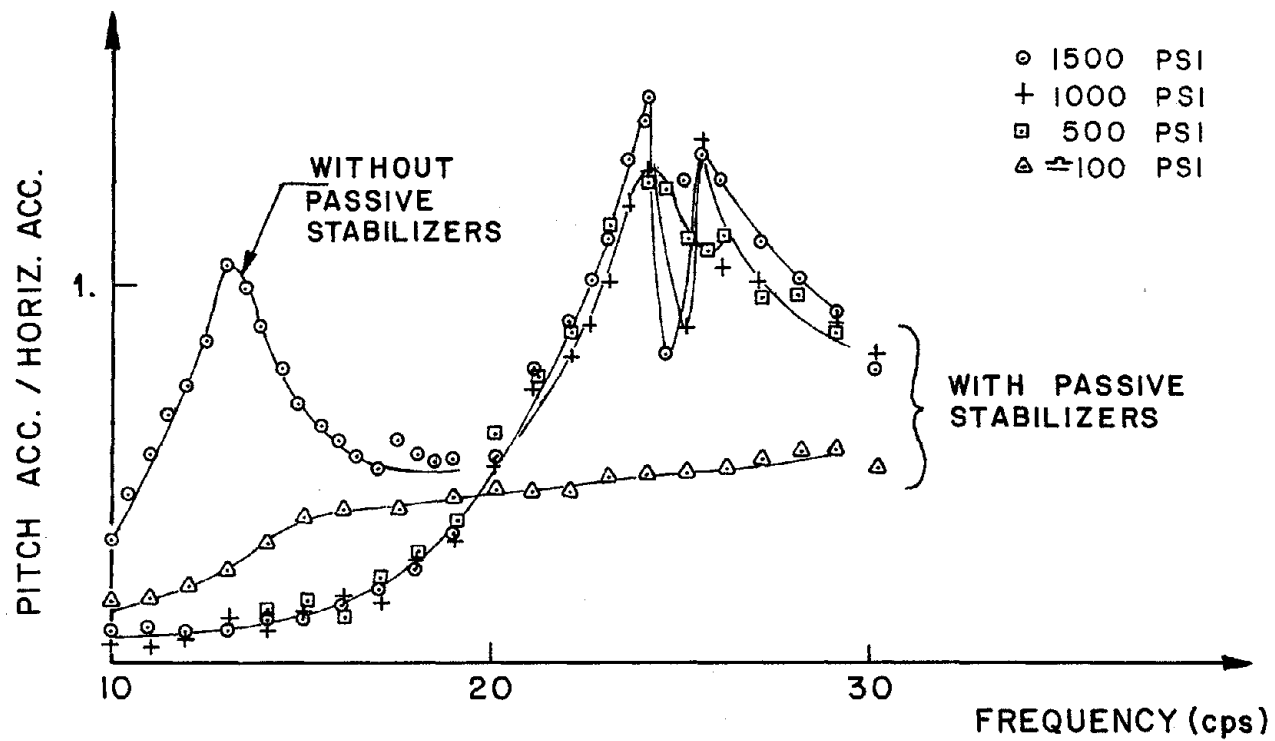


Fig. 2.1.3 Pitch Frequency Responses for Unloaded Table

3. TEST STRUCTURES

3.1 Moment Resistant Steel Frame Model

The three-story moment resistant steel frame test structure of Reference 12 was designed as a 6/10 scale model of typical building construction. This is 6 x 12 ft in plan, 17 ft - 4 in. high, and was fabricated from A36 wide flange sections; W5x16 for columns and W6x12 for girders. The first floor, second floor, and the third floor heights are 6 ft - 8 in., 5 ft - 4 in., and 5 ft - 4 in., respectively. Each floor system was supplemented with enough crossing beams and angle braces to resemble a rigid floor diaphragm. The connections of the cross beams to columns were initially provided by gusset plates and high strength bolts. However, in this investigation the gusset plates were welded to the columns in order to obtain a shear resistant joint in the plane of the columns' weak axes (Fig. 3.2.3).

The original lateral bracing system consisted of 1/2 in. diameter rods with turnbuckles, arranged in an X pattern at each story across the 6 ft dimension of the frame (corresponding to the weak axis of the columns). In the second and third test series, these braces were replaced by 3/4 in. diameter pipe X-braces, and 1 x 1 x 1/8 in. double angle X-braces, respectively. Each pipe or double angle X-brace unit was welded together at the center and to connections at the ends. Figure 3.1.1 and 3.1.2 show the test frame with rod and double angle braces, and Fig. 3.1.3 demonstrates the front elevation of the test frame with pipe braces. Note that concrete blocks were supported at each floor to provide appropriate seismic loads during tests.

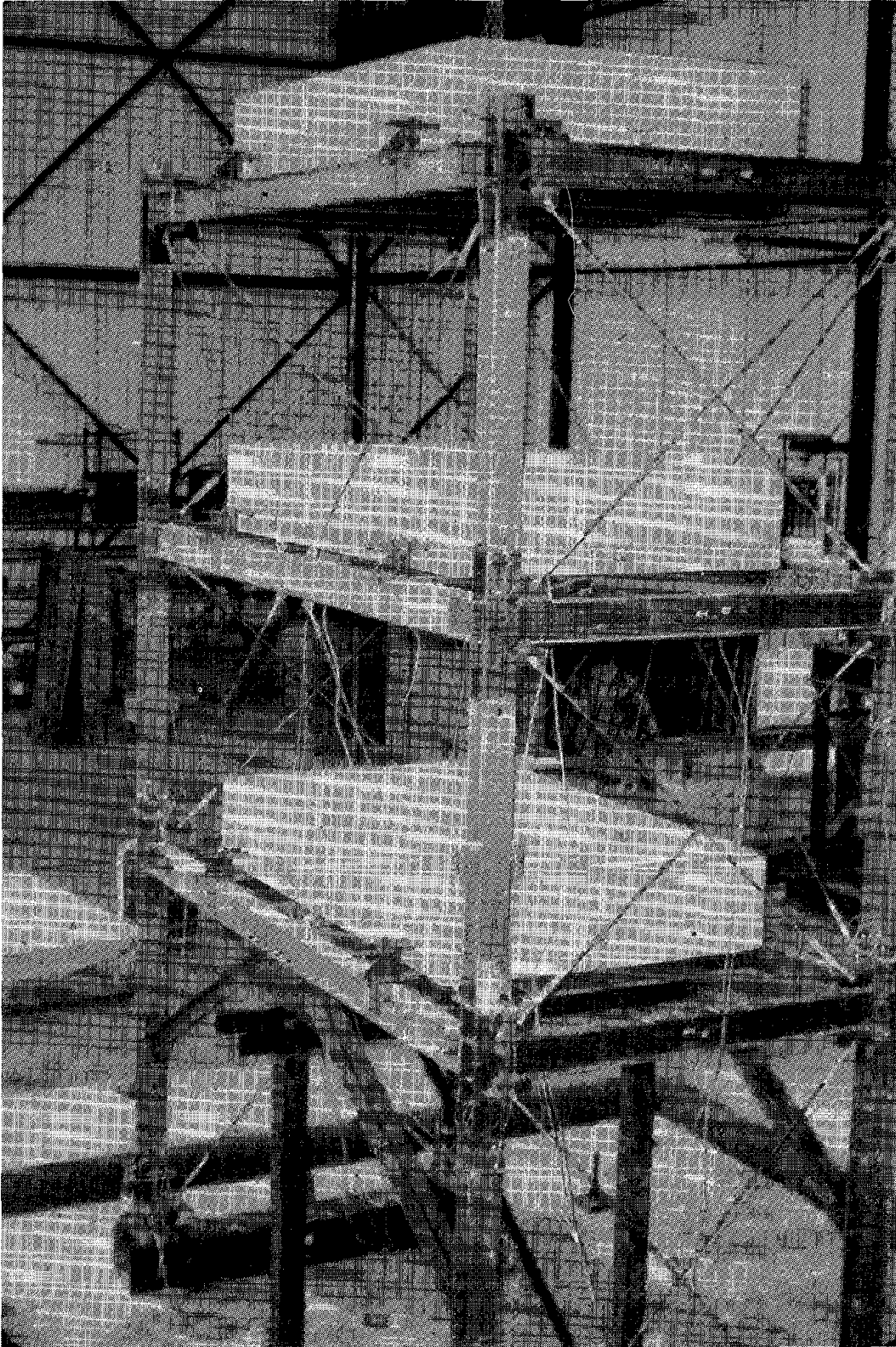


Fig. 3.1.1 Test Structure with Rod Braces on the Shaking Table

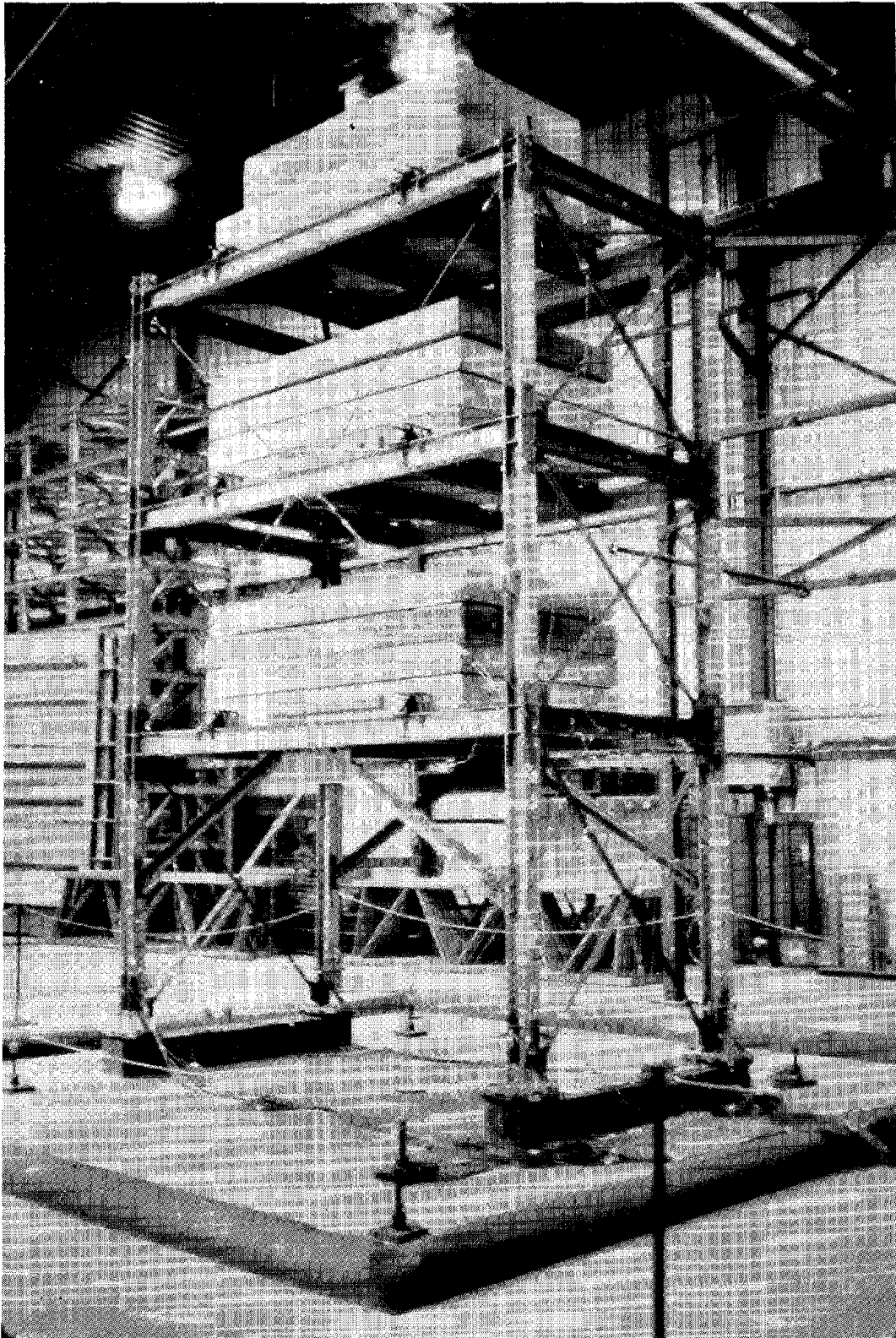


Fig. 3.1.2 Test Structure with Double Angle Braces on the Shaking Table

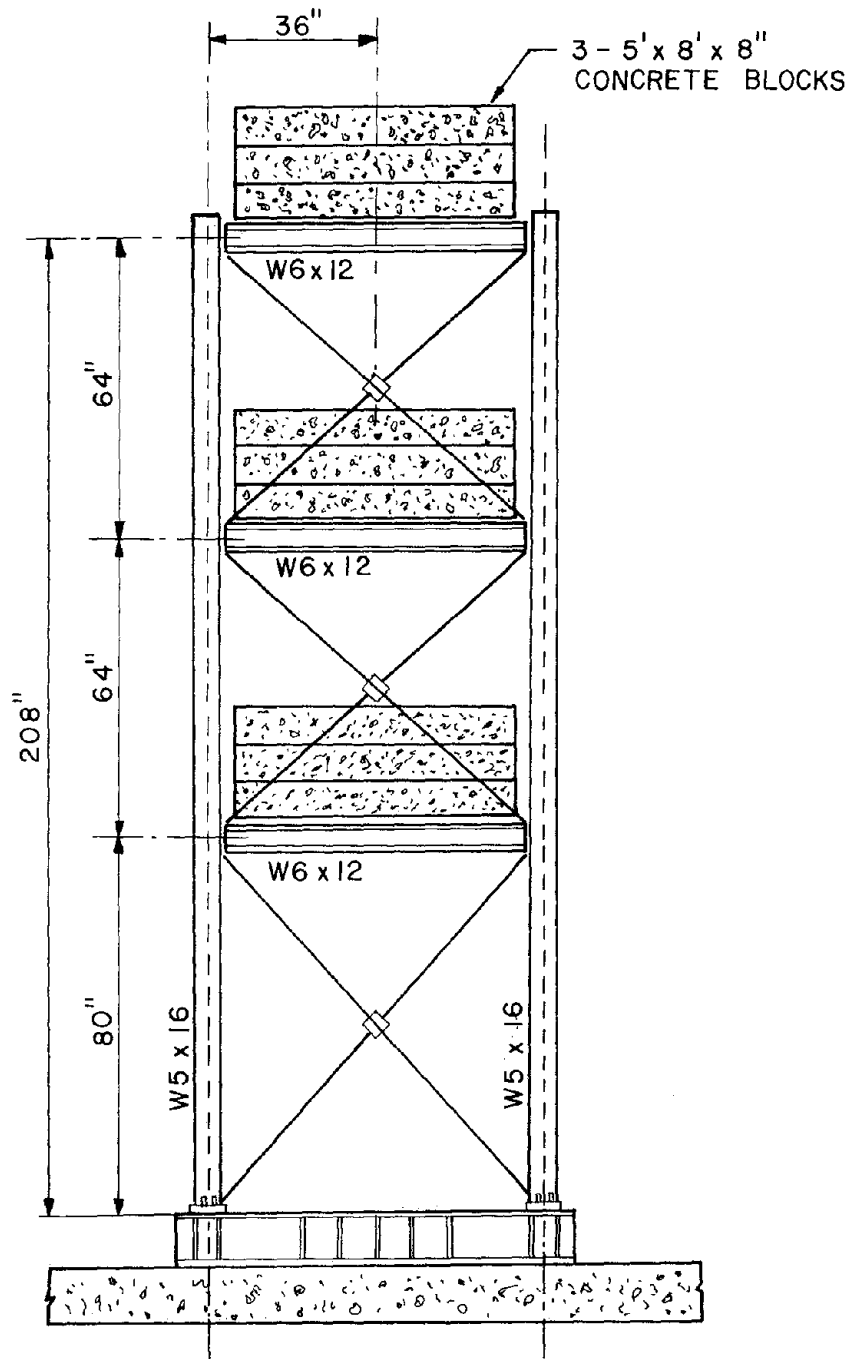


Fig. 3.1.3 Front Elevation of the Test Structure with Pipe Braces

3.2 Design Criteria of the Bracing Members

Bracing members were initially designed for wind loading of a typical steel frame building on the basis of the Uniform Building Code⁽¹³⁾. The designed bracing members were then reduced for use on the scaled prototype test structure using a geometric ratio of 6/10. The wind load pressure was assumed to be 20 lb/ft² constant over the height of building.

Rod X-braces - The diagonal rod X-braces of Reference 12 were supplied originally to control lateral or torsional motions of the frame. These diagonals were made the subject of the present research during test series 1 by mounting the structure on the shaking table at 90° to its previous orientation. The half-inch rod-turnbuckle braces which had a slenderness ratio of $KL/r = 370$ and buckling capacity of $P_e = 80$ lb turned out to satisfy the Uniform Building Code requirements for tension members subjected to wind loading. The rod braces were attached to the steel frame by clevis joints and half-inch diameter pins.

Pipe X-braces - In test series 2, the rod X-braces were replaced by 3/4-in. diameter pipe diagonal X-braces. These diagonal braces were designed as compression members on the basis of the AISC⁽¹⁴⁾ specification (slenderness ratio should be smaller than 200). The maximum experimental compressive load P_{max} was very close to that calculated by the formulas recommended by AISI⁽¹⁵⁾. For tubular sections AISI recommends:

$$P_{max} = \frac{\pi^2 EA}{[KL/r]^2} \quad C_c < KL/r \leq 200 \quad (3.1)$$

$$\text{where } C_c = \sqrt{\frac{2\pi^2 E}{\sigma_y}} \quad (3.2)$$

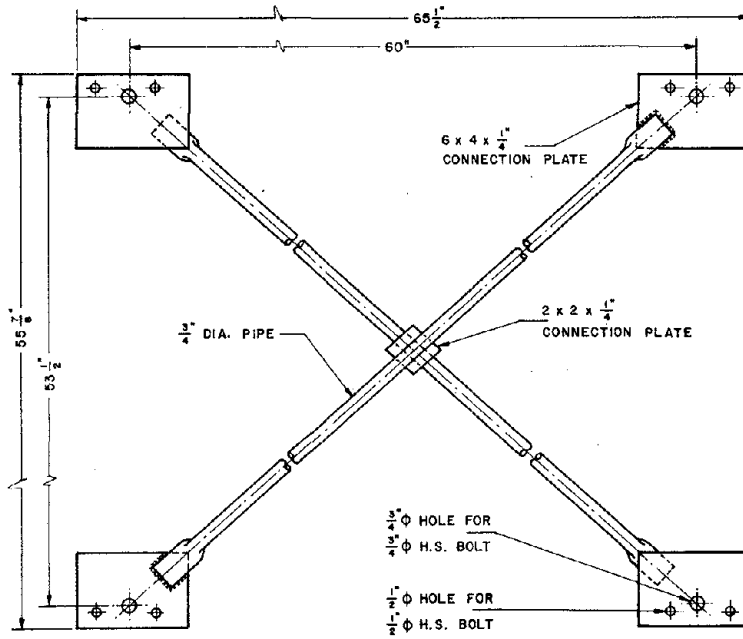
KL/r is effective slenderness ratio and A is area of the cross section. The slenderness ratio of the pipe braces welded at their mid-span intersection was $KL/r = 125$; the theoretical buckling load was 6.3 kips and the tensile yield load was 14 kips. Figure 3.2.1 illustrates the details of the pipe diagonal braces with their connections. The connections and details were designed such that they exceed the elastic capacity of the pipe sections. Thus, ductile performance of the bracing members was possible. For attachment purposes, each pipe was flattened at its ends and was welded to connection plates. The attachment of the connection plates to the steel frame was then accomplished by means of one 3/4 in. and two 1/2 in. diameter high-strength bolts. Figures 3.2.3 and 3.2.4 show these connection attachments to the beam-to-column joint and to the column base joint, respectively.

Double angle X-braces - Double angle (L 1 x 1 x 1/8) diagonal X-braces were tested in test series three. These braces were also designed as compression members according to AISC specifications. The maximum experimental compressive load P_{cr} was closely related to that calculated by the formulas recommended by AISC. For compressive members AISC recommends;

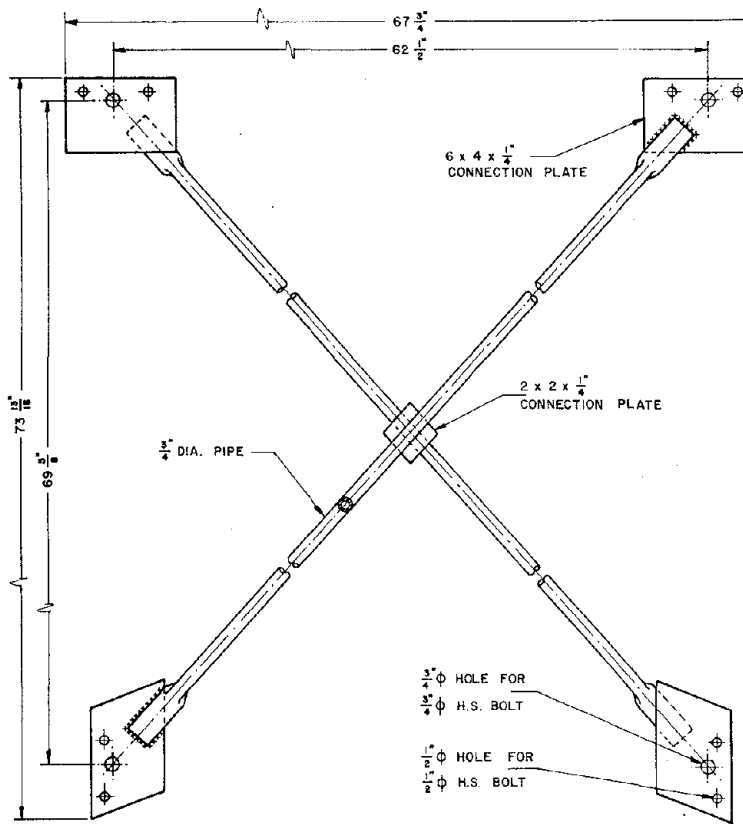
$$P_{cr} = \frac{\left[1 - \frac{(KL/r)^2}{2C_c^2} \right] \sigma_y \cdot A}{\frac{5}{3} + \frac{3(KL/r)}{8C_c} - \frac{(KL/r)^3}{8C_c^3}} \quad KL/r < C_c \quad (3.3)$$

The slenderness ratio of double angle braces welded at their mid-span intersection was $KL/r = 86$; the theoretical buckling load was 8.8 kips and the measured tensile yield load was 24 kips. Figure 3.2.2 shows the details of the double angle braces with their corresponding connections.

Double angle diagonals were welded together at their centers and to the 1/4 in. thick connection plates at the ends. Again, one 3/4 in. and two 1/2 in. diameter high-strength bolts were supplied for the attachment of the braces to the steel frame (See Figs. 3.2.3 and 3.2.4).

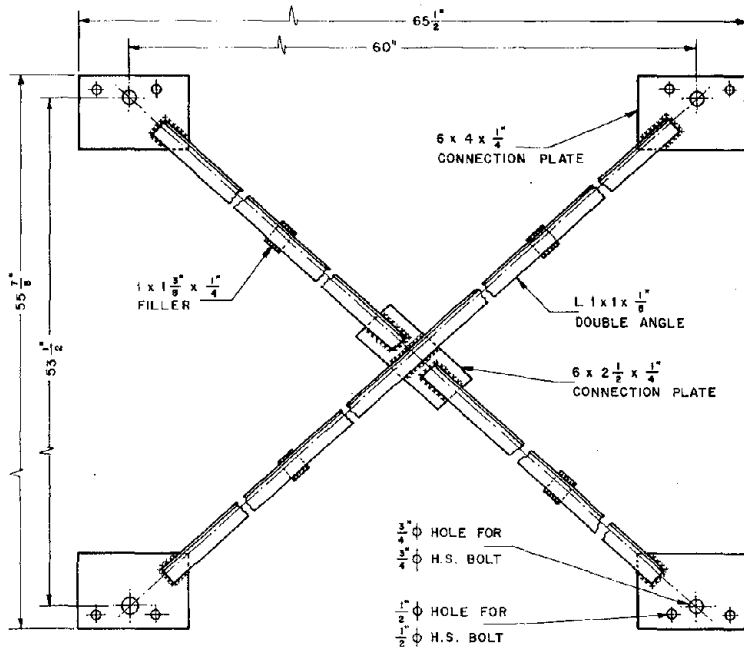


2nd and 3rd Floor Brace

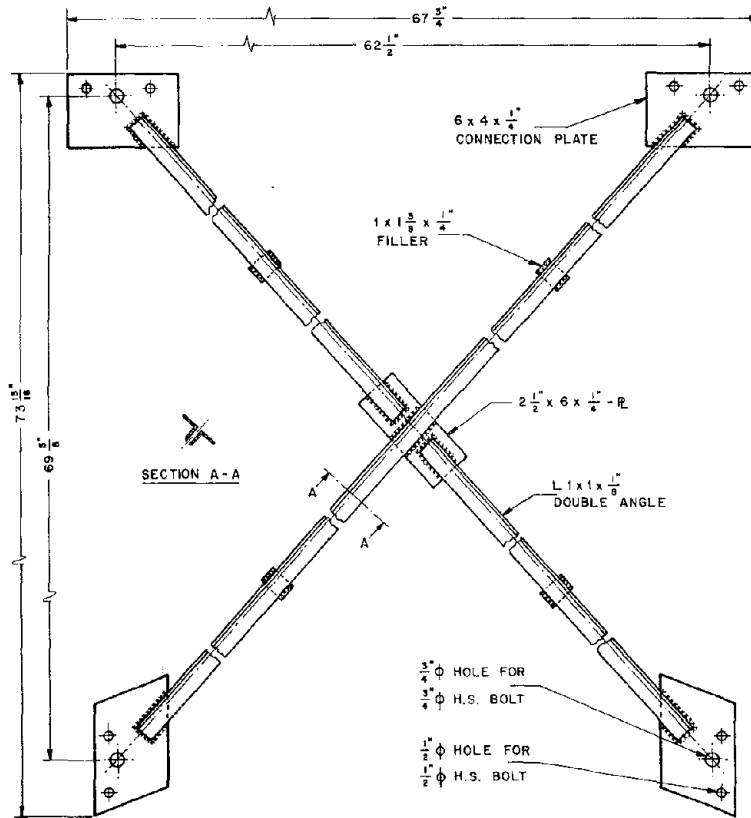


1st Floor Brace

Fig. 3.2.1 Details of the Pipe Diagonal Braces



2nd and 3rd Floor Brace



1st Floor Brace

Fig. 3.2.2 Details of the Double Angle Diagonal Braces

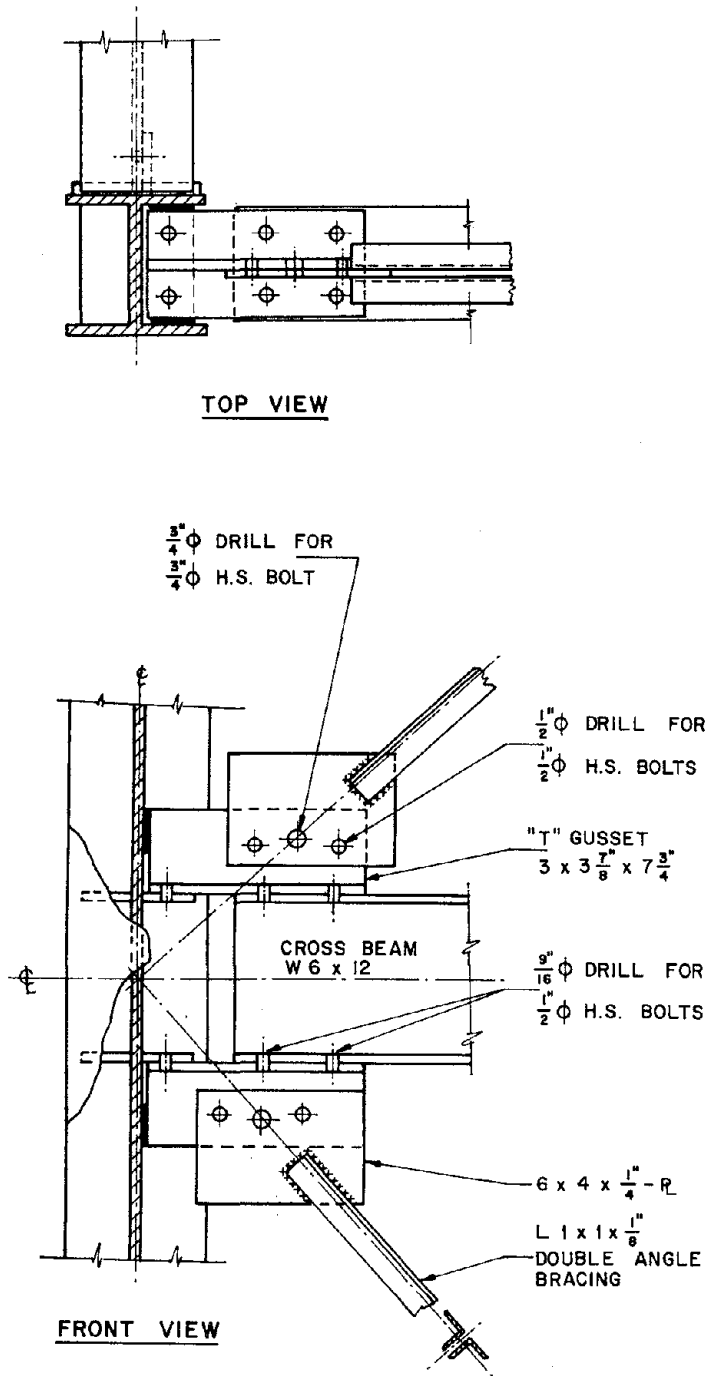
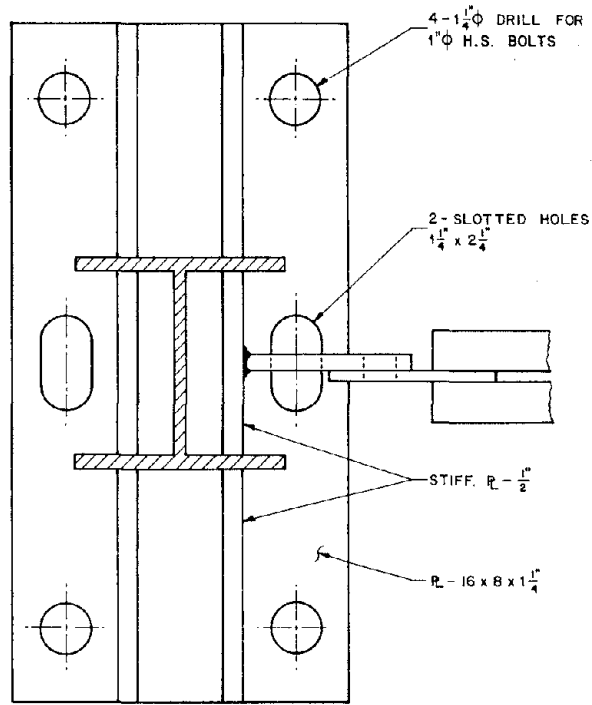
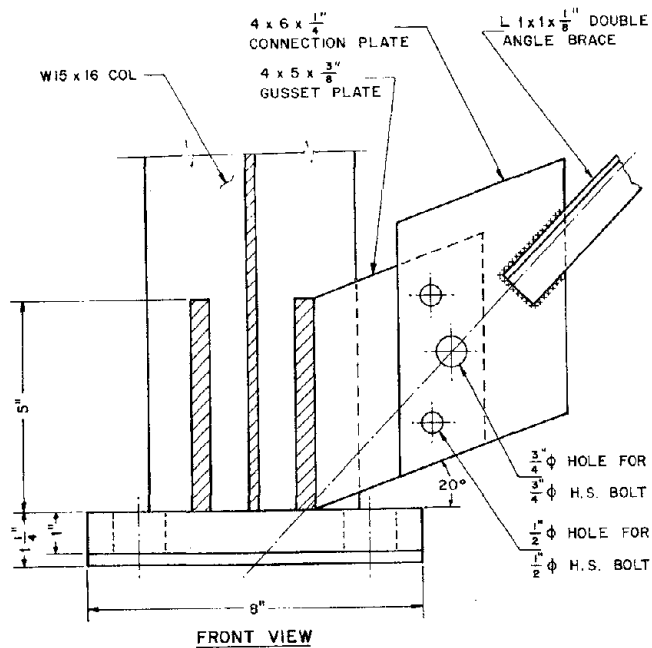


Fig. 3.2.3 Details of Cross Beam-to-Column Connection



TOP VIEW



FRONT VIEW

Fig. 3.2.4 Details of Column Base Connection

4. INSTRUMENTATION

The Earthquake Simulator data acquisition system was described in Chapter 2. Dynamic response measurements of the test structures and the shaking table are discussed here.

The motions of the shaking table and the dynamic response of the test structure were measured by 96 channels of instrumentation during the rod bracing tests, of which 36 channels were devoted to monitoring the shaking table parameters. The scanning rate of the data acquisition system was 52 samples per second, per channel.

During tests of the pipe and double angle bracing system, the total number of instrumentation channels was 104 of which 32 channels were used to measure the shaking table quantities. The sampling rate of the test data was about 50 samples per second, per channel. Finally, in tests of the unbraced frame, a total of 75 channels were monitored to measure dynamic response of the structure, and the scanning rate was the same as before (50 samples/sec /channel).

The measured response quantities included accelerations and displacements of each floor, forces and deformations of selected columns, and also forces and deformations of the bracing members. The measurement procedures used for these quantities are described individually. Complete lists of the data channels used with the different test structures are given in Appendix A.

4.1 Acceleration Measurement

Accelerations in the shaking direction were measured at each floor level. An accelerometer was mounted at the center of the concrete weight on both the first and second floors. The third floor was

provided with two accelerometers which were located at two column ends so as to measure possible twist accelerations as well as the longitudinal acceleration.

Two types of accelerometer were used in testing. One was the Kistler Model 305T non-pendulous, force balance, servo accelerometer, with a Kistler Model 515T servo amplifier attached. The second type was the Statham Model A39TCB-5-500 resistive bridge accelerometer which used strain gage conditioning circuits. Both types of accelerometers were set to measure a data range of ± 5 g.

4.2 Displacement Measurement

Houston Scientific Model 1800-15A potentiometers were employed to measure the total horizontal displacements of each floor of the structure. The potentiometers were mounted on an independent fixed frame, located outside the shaking table, and their wires were attached to the test structure at each floor level. One potentiometer was used for each of the first two floors; for the third floor two potentiometers were utilized to distinguish between twist and horizontal displacements. The travel range of these instruments was ± 15 in. Also, another potentiometer with a travel range of ± 7.5 in. was employed to measure the total displacement of rod braces of the first floor.

4.3 Force Measurement

The global forces such as floor shears and overturning moments were computed from inertia forces at each floor level calculated from the corresponding measured accelerations. All local member axial forces, shears and moments were derived from readings of strain gages mounted in the elastic regions of the various structural members.

Axial strains were determined by averaging measurements from two strain gages attached on opposite faces of a section. Flexural strains were obtained from the differences of readings given by four strain gages placed on flange tips of a column section. Moments at two points within a member were directly computed from the indicated flexural strains, using a nominal section modulus S and a value of 29,600 ksi for modulus of elasticity E . Shear forces were then obtained from the calculated moments, by assuming a linear moment variation within any member.

The locations of axial and flexural strain gages are depicted in Fig. 4.3.1. These elastic gages were manufactured by Micro-Measurements, and the selected model was EA-06-250-BG-120 with option L and W.

The bracing member axial forces were derived from readings of post-yield gages of type YL-10, produced by Tokyo-Sokki Kenkyujo Co., as long as no yielding was indicated at the location of the strain gages. Figures 4.4.1, 4.4.2, and 4.4.3 show the locations of these strain gages for the various bracing members.

4.4 Local Deformation Measurement

It was expected that forces beyond the elastic limits of certain members would develop during moderate and strong shakings of the test structures. Hence, appropriate instruments were installed to measure local deformations of the most critical members which were believed to be the first floor columns and the first floor diagonal cross bracing members.

Two types of local deformation were measured for the columns, both within what can be considered the plastic hinge at the member ends. One type of measurement was the post-yield flexural strain which also was used to compute the average curvature of the member; the other

quantity was the average member rotation, measured over a gage length of 12.5 inches. The post-yield strain gage locations are shown in Figs. 4.4.1, 4.4.2 and 4.4.3 for the various test structures. Flexural strains were obtained by averaging the differential strains from four strain gages (Tokyo-Sokki Kenkujo, Model YL-10) mounted on the flange tips of the column end sections. Curvatures were computed from the flexural strains and the width of the section.

Average column end rotations were measured by pairs of Sanborn Direct Current Displacement Transducers (DCDT) Model 7DCDT-500, mounted in aluminum frames as shown in Fig. 4.4.4. The DCDTs have a travel range of ± 0.5 in. and the distance between the opposed pair was 12.5 in.

Plastic deformations were expected to occur in the mid-section of the pipe and double angle X-braces. These sections were instrumented by four Tokyo-Sokki Kenkujo Model YL-10 post-yield strain gages arranged in the patterns shown in Figs. 4.4.2 and 4.4.3. Axial strains and flexural strains were obtained by averaging and/or differencing the strains measured with specific gages.

4.5 Noise Level and Accuracy of the Experimental Data

The accuracy of response measurement is governed by three prime parameters; input noise, instrument exactness, and the Data Acquisition System (DAS) resolution.

Input noise is caused by mechanical vibrations of the shaking table, which produced a high frequency resonant vibration in the test structure during idle operation of the shaking table, and was superimposed on any dynamic motions applied to the table. The instrument accuracy is characterized by accuracies in gage factor, shunt calibration,

nonlinearity (whichever is applicable), and installation of the instrument in the test structure. The Data Acquisition System, which consists of amplifier, scanner, and analog-to-digital converter, controls the accuracy in the process of sampling experimental data.

These sources of errors are discussed for the various transducers in the following paragraphs. In addition, an example illustration of the overall accuracy for the worst estimate, corresponding to the least intense input signal is presented for each type of response. It should be noticed that the maximum error estimate is computed by the square root of the sum of the square of all the extreme errors. However, the overall error for most cases is much lower than that estimated for the worst case. Thus, the accuracy of experimental data can be considered very good.

Post-yield strain measurement

The strain response due to the input noise was obtained from a one-second zero reading of all post-yield gages during idle operation of the shaking table. The mean amplitude of the strain noise was 0.007 milli in./in., with the extreme amplitude of 0.014 milli in./in. The largest extreme amplitude of strain was observed in the first floor bracing members; this was considered to be due to the sensitivity of these members to high frequency input noise.

In addition, a gage factor tolerance of 0.5 percent and an error of 1.0 percent in the shunt calibration may cause a significant offset of the measurement axis of the strain gage components. Although precise evaluation of the accuracy in the strain gage installation is difficult, it is reasonably assumed that the error of this kind is not more than a few percent.

The El Centro span 100 input signal with a maximum table

acceleration of 0.067 g which produced an extreme strain value of 0.375 milli in./in. was chosen to estimate the maximum error. Based on a 0.2 percent error caused by the DAS, the overall error for these gages is about 5.5 percent; given by 3.7 percent ($= 100 \times .014/.375$), 4 percent (gage error), and 0.2 percent (DAS). It is interesting to note that the error caused by input noise when the structure was subjected to the Pacoima span 300 input motion (max acc = 0.37 g) was only 0.06 percent.

Elastic strain measurement

The mean amplitude strain of 0.007 milli in./in. with the extreme amplitude strain of 0.0015 milli in./in. was computed for sixteen elastic strain gages mounted in the test structure during the application of input noise. The same conservative error of 4 percent is assumed for the error associated with the gage factor tolerance, shunt calibration, and the strain gage installation. The DAS error is also assumed to be 0.2 percent as before.

As an example, the estimate of the greatest overall measurement error for a peak strain of 0.066 milli in./in. corresponding to the El Centro span 100 input signal is 4.6 percent; given by 2.3 percent ($= 100 \times .0015/.066$) (input noise), 4 percent (gage error), and 0.2 percent (DAS).

Acceleration measurement

The accelerometers used are the most accurate instruments available in the Earthquake Simulator Laboratory. The greatest error associated with the input noise, in the acceleration was 1.6 percent.

According to the manufacturer's catalogue, this instrument has a nonlinearity of only 0.01 percent. But, significant error may arise from misalignment of the sensitive axis of the instrument when it is installed. However, this kind of error is not more than a few percent and it can be assumed to be about 1 percent for a carefully installed accelerometer.

As an example, an estimate of the greatest overall measurement error with a 0.2 percent error caused by the DAS, is about 2 percent.

Displacement measurement

Slide wire potentiometers were used to measure the floor displacements of the test structures. The displacement response due to the input noise had a mean amplitude of 0.007 in., with an extreme amplitude of 0.008 in. The associated error caused by input noise for a peak displacement of 0.558 in. during the El Centro excitation with a peak acceleration of 0.067 g was estimated about 1.4 percent.

This transducer (according to the manufacturer's report) has a guaranteed nonlinearity of less than ± 1 percent, but the error resulting from the installation of the transducer, could be significant. For a carefully installed potentiometer this error can be assumed to be not more than 2 percent. Then, the greatest overall displacement error could be about 2.5 percent.

DCDT displacement transducers were employed to measure the column rotations. These transducers have an accuracy of ± 0.5 percent of their total stroke range. Their input noise error was negligible. The only significant error that might arise would be due to poor installation, so that the DCDT is offset with respect to its location and/or direction. A reasonable possible error of this kind

is assumed to be about 2 percent. Therefore, the overall measurement error can be about 2 percent.

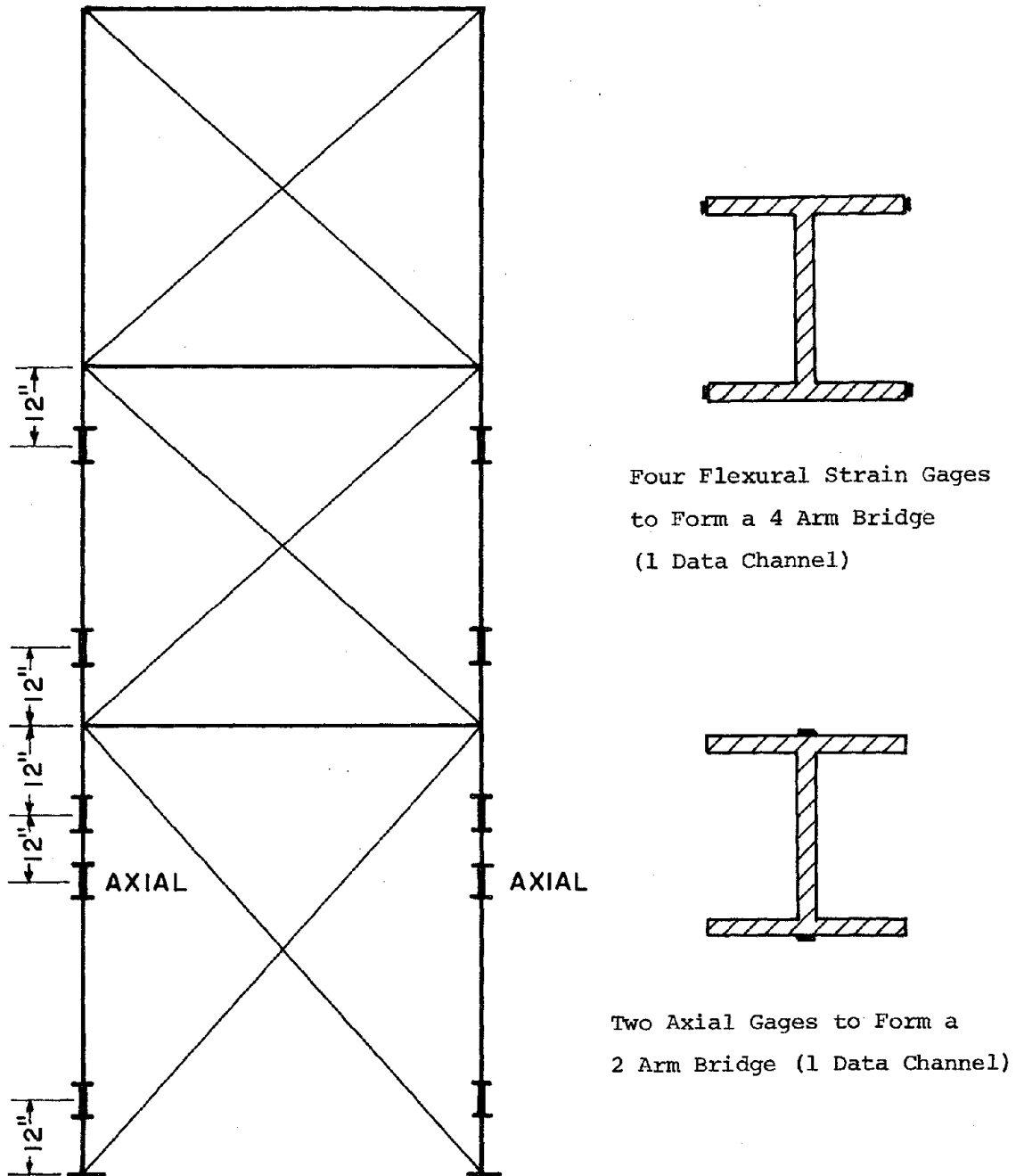


Fig. 4.3.1 Elastic Flexural and Axial Strain Gage Stations

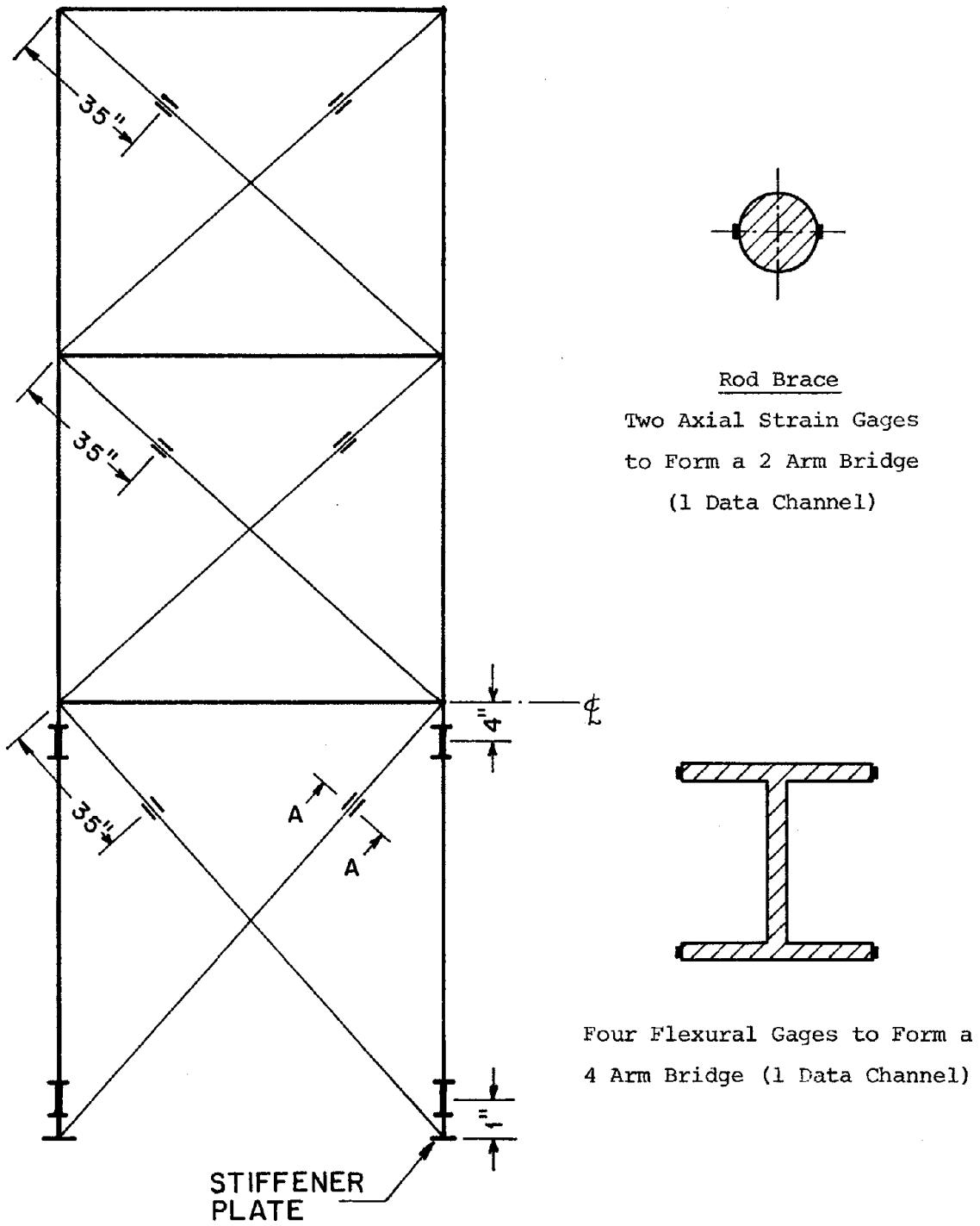


Fig. 4.4.1 Post-Yield Flexural and Axial Strain Gage Stations

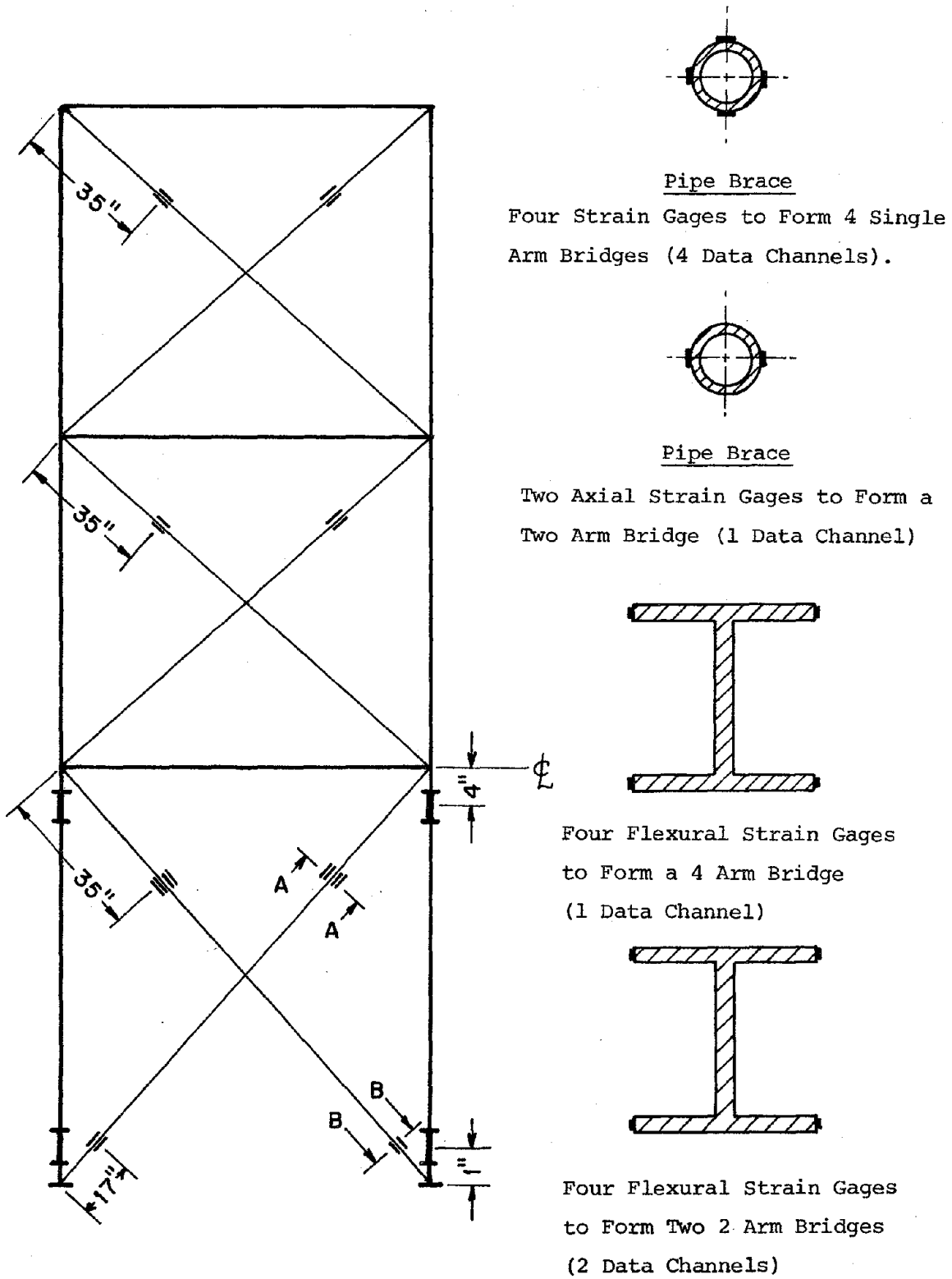


Fig. 4.4.2 Post-Yield Flexural and Axial Strain Gage Stations

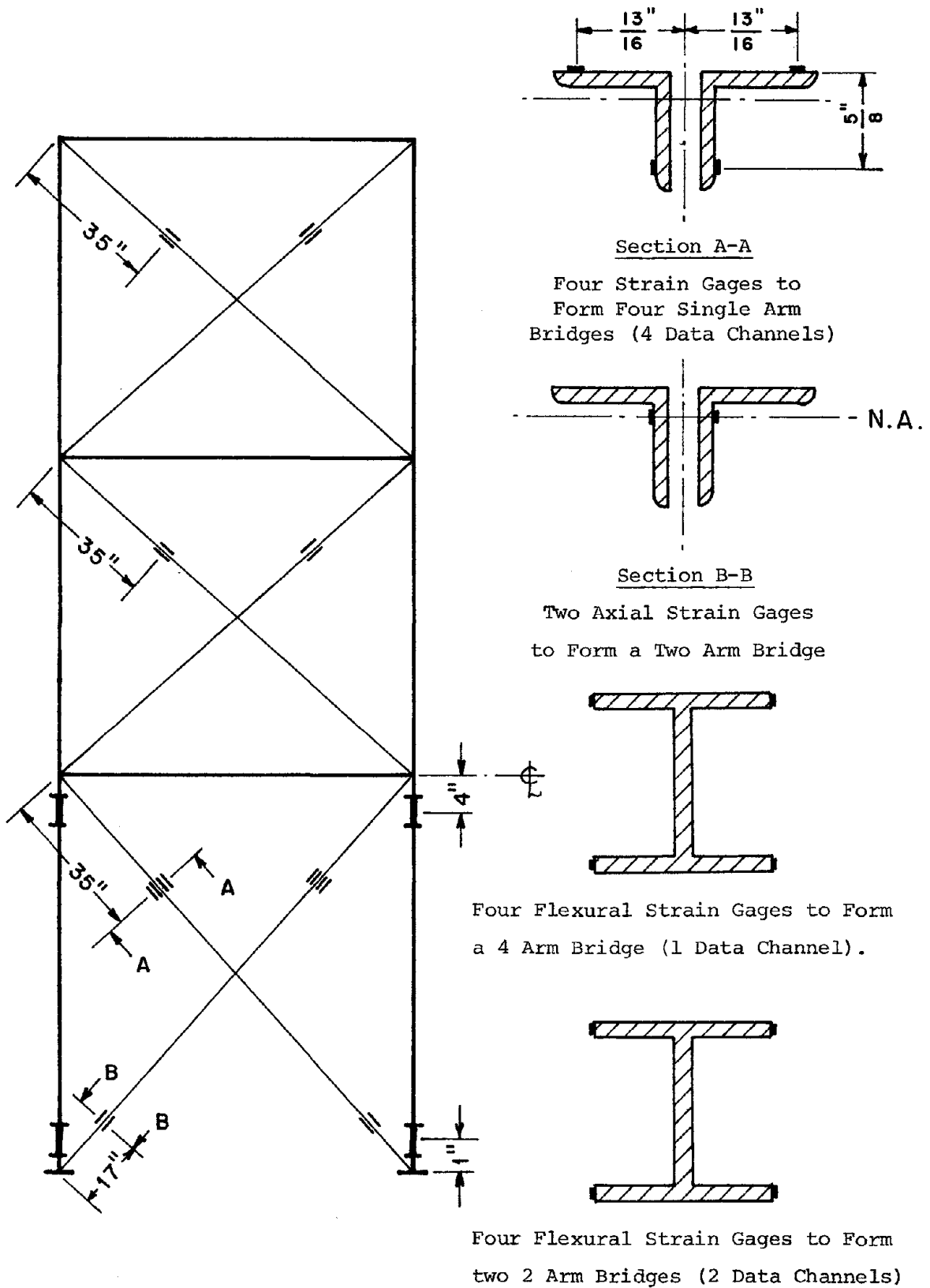


Fig. 4.4.3 Post-Yield Flexural and Axial Strain Gage Stations

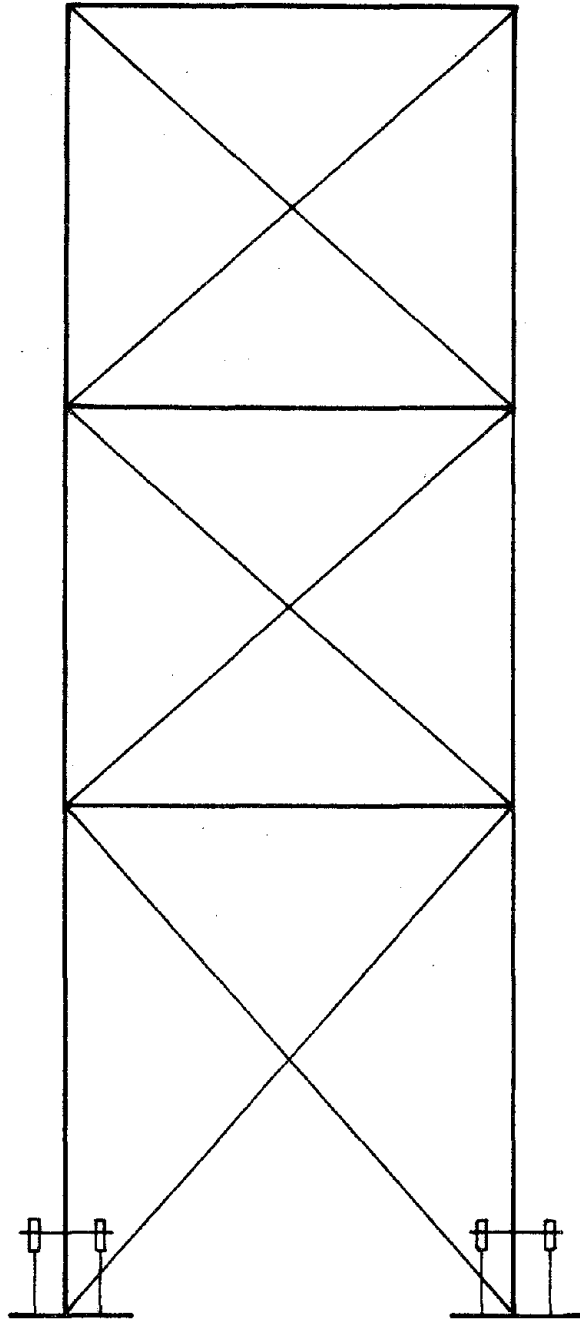


Fig. 4.4.4 DCDT Transducer Stations

5. EXPERIMENTAL STUDY

In this chapter, the shaking table testing of three different bracing systems and the unbraced frame structure is discussed. The principal command signals used during this study were derived from the El Centro 1940 N-S and the Pacoima Dam 1971 S74W earthquake records. Each of these signals was applied to each test structure for a wide range of intensities; results of some specific tests of each structure are reported here.

5.1 Rod X-Bracing Tests

As was mentioned earlier, the slenderness ratio of the half inch rod-turnbuckle braces was very large ($KL/r = 370$) so their buckling capacity was low ($P_e = 80$ lb). Accordingly, all braces were pretensioned to about 2500 lb (40 percent of yield) before testing, to insure initial participation of both members in each panel. The horizontal components of the 1940 El Centro N-S and the 1971 Pacoima Dam S74W records were used for six different test runs (see Table 5.1.A). The first three tests, during which the dead load per floor of the structure was 8 kips, did not introduce any damage in the bracing members. In this series of tests, the maximum table acceleration and displacement were 0.775 g and 5.0 in., respectively, but the induced maximum lateral force was not sufficient to rupture the braces. However, alternate yielding in tension and buckling were observed in the rod braces.

In the second series of test runs one more concrete block was added to each floor, so that the dead weight of each floor was 12 kips. The structure was first subjected to an El Centro span 50 test with a peak acceleration of 0.033 g, which introduced a linear structural response. Then the El Centro span 950 motion, with a peak acceleration of 0.833 g, was applied to the structure. Three of the four rod braces at the first

floor level ruptured and severe "necking" occurred in the fourth brace; also, all of the first floor columns yielded at their bottom ends.

After this test, the first floor braces were replaced by new rods, and it was decided to subject the structure to the scaled Pacoima Dam earthquake motion. The peak table acceleration was about 1.129 g. Although the peak table acceleration of this test and the resulting story shear forces were higher than during the previous test, only two braces ruptured, one in the first floor and another in the second floor. The performance here was better because the new first floor braces had different material properties; the rupture strength was 90 ksi-milli in./in. compared with 45 ksi-milli in./in.

Table 5.1.A

Rod Bracing Tests

	Test No.	Input Signal	Max. Table Acc. (g)	Weight/Flr (Kips)	Comments
Test series 1	1	EC 100	0.063	8	Linear response
	2	EC 400	0.280	8	Linear response
	3	EC 1000	0.775	8	Nonlinear response No damage
	4	EC 50	0.033	12	Linear response
	5	EC 950	0.833	12	Nonlinear response 3 rod braces ruptured
Test series 2	6	PACOIMA	1.129	12	Nonlinear response 2 rod braces ruptured

5.1a Rod Bracing Subjected to El Centro Span 100

The El Centro earthquake signal with a "span" setting of 100 (corresponding to a peak table acceleration of 0.063 g) was applied to the test structure incorporating the rod diagonal braces. The span setting is a control system setting indicating the "intensity" of the input signal. This is linearly proportional to the table displacement. This application of ground motion caused a linear structural response. The table horizontal motion is shown in Fig. 5.1a.1. The story and table accelerations, represented by Fig. 5.1a.2 indicate a dominant first mode vibration. The time histories of the north and south frame accelerations at the third floor level demonstrate a close match which implies a symmetric structural response. The fundamental frequency of the structure, calculated using the FRMSTC program⁽¹⁶⁾, a static load analysis program for multi-story buildings, was about 4.23 cps (see Fig. 6.1a.4).

The story displacements relative to the shaking table and shear forces are shown in Fig. 5.1a.3 and Fig. 5.1a.4, respectively. The maximum first floor shear was about 2.73 kips, and the maximum axial force induced in the first floor braces was estimated to be about ± 1.7 kips, which was lower than the pre-tension load of the braces (2.5 kips). Thus, as was observed during the test, these braces did not lose their pre-tension loads and actively participated in the compression direction as well as in tension.

The strain time history graphs of the first floor rod braces shown in Fig. 5.1a.5 also show the linear behavior of these members. The maximum strain indicated is less than the rod's yielding strain. Therefore, the measured strains were used to compute the axial forces in the rods.

5.1b Rod Bracing Subjected to El Centro Span 1000

In this test, the El Centro signal with a peak acceleration of 0.775 g was employed; also, the structure was loaded with concrete blocks weighing 8 kips per floor. The resulting maximum shear force in each resisting frame was 13 kips. This lateral force was not sufficient to rupture the braces, but caused alternate yielding in tension and buckling in compression. The applied table motion is shown in Fig. 5.1b.1, and the floor absolute acceleration and relative displacement time histories are shown in Fig. 5.1b.2 and Fig. 5.1b.3, respectively. The frequency change in these plots compared with the EC 100 test results was due to the nonlinear nature of the response.

The floor shears are presented in Fig. 5.1b.4, in which the bottom graph displays the first floor shear (solid line) and also the portion of this shear which was resisted by the braces (dashed line) at this floor level. The first floor shear force is plotted versus the first floor displacement in Fig. 5.1b.5, which also represents the combined force-displacement of the first floor rod diagonals. These results show that during the early stage of response, while the braces retained their pre-tension, they provided an efficient dual path for resisting the lateral forces. At this time, they resisted 80 to 85 percent of the lateral loads. After compression buckling and tension yielding occurred, however, the compression diagonal became ineffective; the tension diagonal then supplied only about 50 percent of the resistance with the rest being carried by the moment-resisting frame. During this stage, the diagonals alternately went slack and were subjected to tensile impact which produced additional elongation (see Fig 5.1b.5 and Fig.5.1b.6). At this time, story shear forces less than about 6 kips were carried entirely by

columns while the braces remained slack as may be deduced by comparing the bottom two curves of Fig. 5.1b.4. The column moment-strain loops, presented in Fig. 5.1b.7 show that minor yielding occurred at the bottom end of the first floor columns.

5.1c Rod Bracing Subjected to El Centro Span 950

The El Centro span 1000 excitation described above, which induced a peak lateral force of 13 kips per frame did not cause any rod to rupture. Accordingly, in order to provide a really damaging test, the frame was loaded with additional concrete blocks (to a total weight of 12 kips per floor) and was subjected to the El Centro span 950 excitation with a peak acceleration of 0.833 g (Fig. 5.1c.1). The general behavior during this test resembled the previous El Centro response, but the increased force levels ruptured three rod braces in the first floor, and caused yielding and "necking" of the other first and second story braces. In addition, the first floor columns experienced significant yielding.

The first ruptures occurred simultaneously in two tension rods, similarly oriented in opposite end frames, as shown in the strain time history plots of the first floor rods (Fig. 5.1c.6). At this time the story relative displacement was 0.8 in., but it reached 2.38 inches by the end of this excursion. The first floor column moment-strain plots (Fig. 5.1c.7) demonstrate the significant column yielding which was induced during this test. The permanent strain at the bottom end of the column, according to this hysteresis plot, was estimated to be about 0.33 percent. The next rupture occurred a few cycles after the first, when the frame excursion in the opposite direction reached a displacement of 1.22 in. at the first story level. At this time, only one of the two rods left in the first story ruptured, but the other rod suffered

significant yielding and necking.

The rupture of braces combined with the yielding of the first floor columns reduced the structure frequency considerably. This can be clearly observed in floor acceleration time histories (Fig. 5.1c.2). In general, all floor relative displacements became larger in this test (Fig. 5.1c.3) than in the preceding one. As may be seen, the first floor displacement particularly increased with respect to that of the previous test (Fig. 5.1c.4); this was expected because the column yielding and bracing failure occurred in the first floor.

5.1d Rod Bracing Subjected to the Pacoima Earthquake

After the El Centro earthquake tests, the damaged rod braces were replaced, and the structure was subjected to the Pacoima Dam earthquake with a peak acceleration of 1.129 g (see Fig. 5.1d.1). The dead load of the structure again was supplemented by concrete blocks weighing 12 kips per floor. The general behavior during this test resembled the El Centro response, but the rupture mechanism of rod braces was different. Only two rod braces ruptured, one in the first floor, and the other in the second floor but these ruptures were accompanied by significant yielding and necking of the other first and second floor braces. The first floor rod rupture occurred during a large story displacement of 2 in. During the return swing from this maximum excursion, the tension rod at the second floor level of the opposite end frame ruptured. At this time the displacement of the second floor relative to the first floor and to the shaking table were about 0.8 in. and 2 in., respectively.

The floor acceleration time histories shown in Fig. 5.1d.2 demonstrate the nonlinear behavior of the response. The first and second mode of vibration are present, and the change of frequency due to nonlinear

behavior of the structure is quite evident. The third floor acceleration time history contains a high frequency impulsive type signal which was caused by defective installation of the third floor accelerometers. These accelerometers had been attached to the top ends of the columns and, therefore, picked up the high frequency vibration of the rods installed in the lateral direction.

The floor shear forces associated with inertia forces determined from the masses and accelerations measured at the floors are displayed in Fig. 5.1d.5. The high frequency signal imposed on shear time histories also was caused by the spurious signals recorded by the third floor accelerometers. The bottom graph depicts the first floor shear and the portion of this shear that was resisted by the diagonals of the first floor. This graph shows that the major portion of the first floor shear was resisted by the diagonal. Their resisting shear forces were as high as 13 kips, and the rods were very effective for lateral story shears up to 15 kips. This effective performance was mainly due to the higher strength of these rods in comparison with the diagonals of the previous tests.

The strain time-history plots of the first floor rod braces are shown in Fig. 5.1d.6. Although, the general behavior of the rod braces during this test resembled the El Centro response, they showed some peculiar behavior of their own. As was mentioned before, the first floor rod braces of this test were stronger than those used in the El Centro tests and they had different yield properties. The tension yielding occurred not only in their weak sections (threaded portion of rods), but it also occurred along the half-inch diameter section of the rods. However, only one rod brace of the first floor ruptured. This peculiar behavior of the first floor diagonals introduced a more severe "pinching"

effect in the story displacement hysteresis loops, associated with slack in the braces (see Fig. 5.1d.7).

A detailed examination of the rod diagonal strain time histories shown in Fig. 5.1d.6 may explain the behavior mechanism of the braces. The two upper graphs are the strain time histories of cross braces located in the front frame, and the two lower graphs are the strain time histories of those in the opposite end frame. Three different states can be recognized in the response. During the elastic response, both cross braces efficiently participated in controlling the lateral displacements as well as carrying a substantial portion of the lateral forces; they did not lose their pre-tension loads during this stage. In the second stage, as the response built up and the forces increased, the compression braces lost their pre-tension loads and buckled elastically. During this interval, the cross braces buckled alternately and the slackening mechanism was initiated. Finally, in the last stage, the tension braces yielded as the compression braces buckled. In this stage, the yielded braces remained slack for a longer period of time and the pinching effect was initiated in the hysteresis loops. Also, as the diagonals alternately went slack they were then subjected to tensile impact which produced additional elongation. The first floor rod braces of the opposite frame (two lower graphs) behaved similarly, except that the tension brace in this frame ruptured.

The hysteresis moment - strain plots for the bottom end of the first floor column are illustrated in Fig. 5.1d.8. A significant yielding occurred at the bottom end of this column, and the maximum moment was measured to be about 196 kips-in. The shift of the hysteresis plots to the right was associated with a residual strain of 1.25 milli in./in., and this distortion occurred as the tension rod of the first floor

ruptured.

The floor relative displacements shown in Fig. 5.1d.3 were about 10 percent smaller than those of the El Centro span 950. This was expected because the damage during this test was less. However, the second floor displacement relative to the first floor displacement (drift) shown in Fig. 5.1d.4 was higher, because one rod brace of the second floor ruptured during this test.

5.2 Pipe Bracing Tests

After the rod diagonal bracing tests, the same steel frame structure was equipped with 3/4 in. pipe braces and subjected to a series of simulated earthquakes with the El Centro and the Pacoima Dam earthquake motions. The slenderness ratio of the pipe diagonal bracers welded at their mid-span intersection was $KL/r = 125$ at the first story level, and $KL/r = 107$ at the second and third story levels. The tests performed on the pipe braced structure are listed in Table 5.2.A, in which a summary of the test results and the peak accelerations of the input signals are included.

In the first test series, the structure was subjected to eight tests, with gradually increased input signals. The structural response was linear for input signals up to a peak acceleration of 0.2 g, and no pipe buckling and/or tension yielding was observed. Pipe buckling was initiated during an El Centro span 400 test having a peak acceleration of 0.28 g. The maximum first story shear during this test was about 17 kips, but the columns remained elastic. In subsequent test runs, the peak acceleration of the table motion reached 0.5 g, and this induced compression buckling and tension yielding of all the first floor pipe braces. In addition, yielding occurred in the first floor columns.

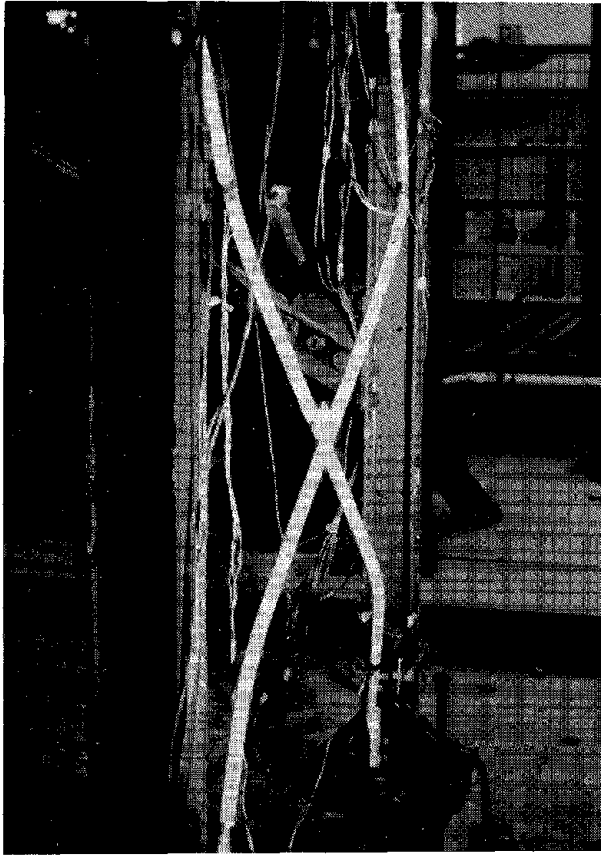
In the second test series, all damaged pipe braces of the first floor were replaced by new members. Then the structure was subjected to the Pacoima earthquake motion with a peak acceleration of 0.068 g which induced a linear response. After this input signal, the Pacoima span 600 signal with a peak acceleration of 0.812 g was applied to the structure. This strong table motion caused significant buckling and

tension yielding in all the first floor pipe diagonals. In addition, the second floor pipe diagonals also buckled and significant yielding was induced in the first floor columns. The maximum first story shear force during this test was about 26.7 kips. Figure 5.2.1 is a photograph of the first and second floor buckled pipe diagonal braces. The tension yielding of the first floor pipes is also shown. In the following subsections the experimental results obtained during some of the tests are discussed.

Table 5.2.A

Pipe Bracing Tests

	Test No.	Input Signal	Max. Table Acc. (g)	Weight/Flr (Kips)	Comments
Test series 1	1	EC 100	0.067	17	Linear response
	2	PAC 50	0.074	17	Linear response
	3	EC 300	0.202	17	Linear response
	4	EC 400	0.283	17	Nonlinear response 1st pipe buckling
	5	PAC 200	0.235	17	Minor pipe buckling
	6	PAC 300	0.373	17	Pipe buckling Minor col. yielding
	7	PAC 400	0.475	17	Pipe buckling & yielding Col. yielding
	8	EC 650	0.503	17	Pipe buckling & yielding Col. yielding
Test series 2	9	PAC 50	0.068	17	Linear response
	10	PAC 600	0.812	17	Pipe buckling & yielding Col. yielding



1st Floor Buckled Pipe



2nd Floor Buckled Pipe

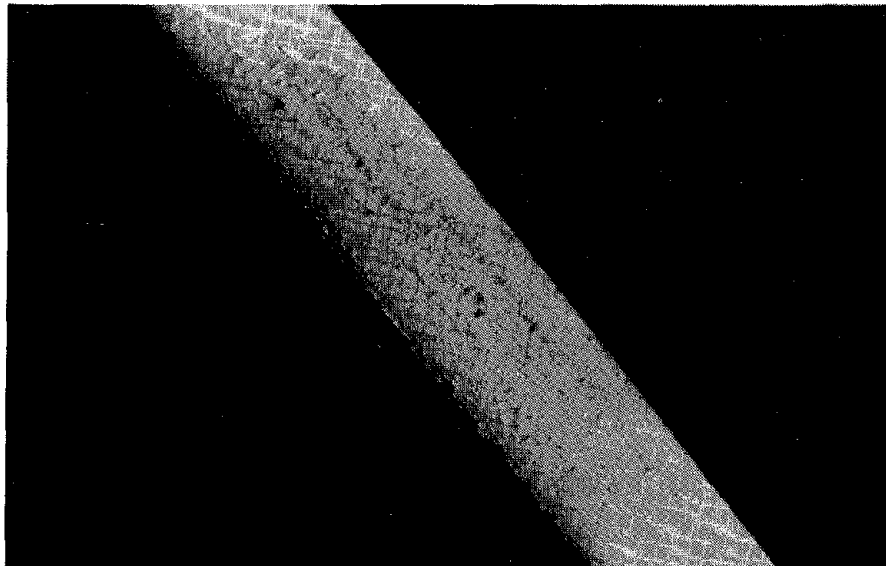


Fig. 5.2.1 Photographs of Compression Buckling and Tension Yielding of Pipe Diagonal Braces

5.2a Pipe Bracing Subjected to El Centro Span 100

The response of the test structure with pipe diagonal braces to the El Centro span 100 test (with a peak acceleration of 0.067 g) was within the elastic range. The fundamental frequency of the structure measured from its free vibration response was 3.63 cps. The typical floor acceleration and relative displacement time histories, shown in Fig. 5.2a.1 and Fig 5.2a.2, respectively, demonstrate that this fundamental frequency dominates the structural response. It is interesting to note that the maximum third floor displacement relative to the table was only 0.13 in. due to the table motion having a peak displacement of 0.5 in.

Figure 5.2a.3 displays the strain time histories of four strain gages mounted at the mid-span section of the first floor pipes. All four time histories are in phase and have the same sign, which indicates that no buckling was induced in the pipe diagonals. The maximum axial force of the first floor pipe diagonals was 2.7 kips which was only 45 percent of its buckling load. The story shear forces are shown in Fig. 5.2a.4, in which the bottom graph displays both the first floor shear and the portion of the first floor shear that was resisted by the first floor pipe diagonals. This graph indicates that more than 90 percent of the first floor shear force was resisted by the diagonals. Therefore, these elastic results show that the pipe diagonals were very efficient in carrying the lateral forces as well as controlling the lateral displacements. Their effectiveness and efficiency in the non-linear cases are discussed in the following paragraphs.

5.2b Pipe Bracing Subjected to El Centro Span 400

The application of the El Centro span 400 signal (with peak acceleration of 0.283 g) to the pipe braced structure caused a nonlinear response. This nonlinear response was initiated when the first floor pipe diagonals buckled in compression and yielded in tension. No column yielding occurred, because the moment induced at the bottom end of the first floor columns was never more than 35 percent of the yielding moment of the columns.

The shaking table and the floor acceleration time histories shown in Fig. 5.2b.1 indicate that the first mode of vibration is dominant. The second mode of vibration which appeared in the first floor acceleration record can be explained by looking at the second mode shape of buildings of this type (see Fig. 6.1a.4). The contribution of the second mode to the structural response is highest at the first floor level.

The time histories of floor displacements relative to the shaking table are shown in Fig. 5.2b.2. The maximum first floor relative displacement is about 19 percent of the maximum table displacement, whereas this was only 13 percent during the El Centro span 100 test. The increase was expected because the first floor stiffness was reduced when the pipe diagonals buckled and/or yielded.

The floor shear forces are shown in Fig. 5.2b.3, in which the bottom frame displays both the first floor shear and the portion of this shear resisted by the pipe diagonals. This graph shows that, despite the pipe buckling, the first floor diagonals resisted about 90 percent of the first floor lateral forces. The figure is not very different from the corresponding one for the El Centro span 100 test, except for the larger amplitudes and a frequency decrease of 9 percent. In

fact, the buckling and yielding of the first floor pipe diagonals as shown in Fig. 5.2b.4 were not drastic, and the change of stiffness at the end of the response was negligible. The buckling and yielding loads were about 6 and 14 kips, respectively. The hysteresis plots of Fig. 5.2b.5 demonstrate the combined force-displacement behavior of the first floor pipe bracing. These hysteresis plots show that the buckling and yielding of the diagonals occurred only during the first four seconds of the response. The pipe braces were quite effective for this intensity of input signal, and no pinching behavior was developed in the hysteresis loops. Therefore, the loss of strength was almost negligible. The second floor shear forces were not sufficient to cause any brace buckling at this level. Note that the buckling capacity of the second and third floor diagonals was higher due to their lower slenderness ratios.

5.2c Pipe Bracing Subjected to Pacoima Span 400

In this test, the frame was loaded with 17 kips of concrete blocks per floor as before and was subjected to the Pacoima earthquake excitation with a peak acceleration of 0.475 g. The theoretical buckling load of the braces as mentioned earlier was 6 kips and the tensile yield load was 14 kips. Before performing this test, however, the braces had suffered buckling and yielding in earlier tests; it is estimated that the residual buckling strength of the damaged first floor braces was only about 3 kips.

The floor accelerations and displacements shown in Fig. 5.2c.1 and Fig. 5.2c.2, respectively, are similar to those of the previous tests. The fundamental frequency of the structure measured at the end of this test was 2 cps (about 55 percent of the elastic case). The floor drift

time histories (story to story displacement) show that the ratio of first floor drift to second floor drift is much higher than that observed in the El Centro span 400 test (see Fig. 5.2c.3). This reduced stiffness was expected, because the buckling and yielding of the diagonals were more severe in this test, especially at the first floor level.

The resulting maximum first story shear force was 28.36 kips, which resulted in tension yielding and compressive buckling of the braces. Despite their alternate buckling during response cycles, the first floor braces resisted 75 to 80 percent of the lateral load in the initial response stage (see Fig. 5.2c.4), and effectively controlled the lateral displacement. However, as tensile yield deformations gradually accumulated, the bracing efficiency diminished and a pinching effect was developed in the force-displacement curves (see Fig. 5.2c.5) due to the slack resulting from tensile yield. The slope of this force-displacement curve at the end of the excitation was only 53 percent of its initial slope; this indicates a significant strength loss of the first floor braces. Also, the resisting force capacity of the braces was as low as 45 to 50 percent of the lateral load at this stage. The pipe bracing behavior became, therefore, somewhat similar to that of the rod braced system, but the much more open hysteresis loops show that the pipe braces continue to absorb significant energy while buckling.

The first floor columns yielded at their bottom ends as the buckling and tension yielding of the braces occurred. The hysteresis loops of the moment-curvature diagram for the bottom ends of the first floor columns are shown in Fig. 5.2c.6.

5.2d Pipe Bracing Subjected to Pacoima Span 600

After the test described above was completed, the damaged pipe braces

were replaced. Then, to provide a more damaging test, the frame was subjected to the Pacoima earthquake excitation with a peak acceleration of 0.812 g (see Fig. 5.2d.1, bottom frame). The general behavior during this test resembled the Pacoima span 400 response, but the increased force levels were such that the second floor pipe braces also buckled and a residual distortion remained at the bottom end of the first floor columns.

The typical floor acceleration and relative displacement time histories are shown in Fig. 5.2d.1 and Fig. 5.2d.2, in which the change of response frequency due to the nonlinear behavior of the structure is evident. The fundamental response frequency measured at the end of this test was about 1.95 cps; this represents a reduction of 46 percent with respect to the elastic case. The floor drifts shown in Fig. 5.2d.3 indicate a larger first and second floor response than was obtained in the Pacoima span 400 response. This was expected, because the plastic deformations at the first floor level were larger and therefore the loss of strength was greater. In addition, larger second floor drifts were caused by the buckling of braces at this level.

The floor shear forces are shown in Fig. 5.2d.4. Although the general features of these plots are similar to those of the Pacoima span 400 test, a careful study of them identifies three distinct phases. Phase one was associated with the elastic response. In this phase the maximum first floor shear was smaller than 15 kips. The pipe braces did not buckle during this interval and effectively resisted up to 90 percent of the lateral forces. In phase two, as the first floor shear forces became larger than 15 kips, the compressive braces buckled. At this time the efficiency of the pipe braces decreased, and they resisted

only about 70 to 80 percent of the lateral forces (see Fig. 5.2d.4). Finally, in phase three as tensile yield deformations accumulated, the bracing efficiency diminished drastically, and a pinching effect was developed in the force-displacement curves (see Fig. 5.2d.5). During this interval the braces resisted about 60 percent of the lateral forces as the pinching initiated, and their contribution decreased to about 35 percent of the total lateral force at the end of this test.

The force-displacement hysteresis loops show that the slope of these curves as the excitation ended was only 35 percent of its initial slope; this indicates the great damage which occurred in the braces during this test. Accordingly, the first floor column yielded significantly, and a residual distortion was developed at their bottom ends (see Fig. 5.2d.6).

5.3 Double Angle Diagonal Bracing Tests

Angle sections are frequently used as diagonal bracing members in many braced steel frame buildings. Therefore, to complete the task of this investigation, the steel frame test structure was provided with 1x1x1/8 in. double angle diagonal braces. The double angle cross braces were welded together at the center and to connections at the ends using 1/4 in. thick gusset plates. The test structure was subjected to motions simulating the El Centro and the Pacoima Dam earthquake records with different intensities. Two series of tests were performed: in series 1, the structure was loaded with 12 kips per floor, and in series 2, the load per floor was increased to 17 kips. The tests performed and a summary of results are shown in Table 5.3.A.

In test series 1, the double angle braces initially had no filler plates between their ends and the middle crossing point. Hence, the double angles did not act together, and the buckling occurred in the direction of each single angle z-axis. The slenderness ratio of the single angles with respect to their z-axis was $KL/r = 230$, and the angle buckling was initiated during the application of the El Centro span 500 test. At this time, the maximum resulting first floor shear was about 13 kips, and the buckling load was estimated to be about 2.5 kips per angle. Then it was decided to modify the braces so that they would act together as a composite member. For this purpose, each pair of angles was welded together through a 1/4 in. thick filler plate at the sections midway between the end and the center crossing (i.e. at quarter-span of the full diagonal length). After this modification, the first buckling of the first floor braces was initiated when the El Centro span 700 signal was applied to the structure with a peak acceleration of

0.485 g. The slenderness ratio of these combined double angle diagonal braces was $KL/r = 86$ as they buckled with respect to their y-axis. At this time, the maximum first floor shear was about 20 kips, and the buckling load was estimated to be about 9 kips. In subsequent tests of this series, the intensity of excitation was increased to a peak acceleration of 0.820 g, which caused significant buckling and tension yielding in the diagonal braces of the first floor level. The first floor columns also yielded during this test, but no buckling occurred in the second floor braces.

After test series 1, all the damaged first floor diagonals were replaced and the structure was loaded with concrete blocks weighting 17 kips per floor, to produce a more destructive test. Note that the new set of diagonal braces was also interconnected using 1/4 in. filler plates at their quarter span sections. The structure was first subjected to the El Centro span 100 signal and then to the Pacoima span 100 to provide elastic response data. Finally, a very intense Pacoima span 800 test with a peak acceleration of 1.314 g was applied to the structure; this produced extensive nonlinear response. During this test, all the first floor double angle diagonals buckled and were damaged significantly. A photograph of the first floor braces is shown in Fig 5.3.1. The intensity of this motion was such that the lateral forces were sufficient to cause buckling in the second floor diagonals as well ($KL/r = 72$). The buckled second floor brace is shown in Fig. 5.3.2. Also a photograph of the local buckling of the angle legs which eventually ruptured during test series 1 is shown in Fig. 5.3.2 (bottom frame). A detailed discussion of the structural behavior during these tests follows.

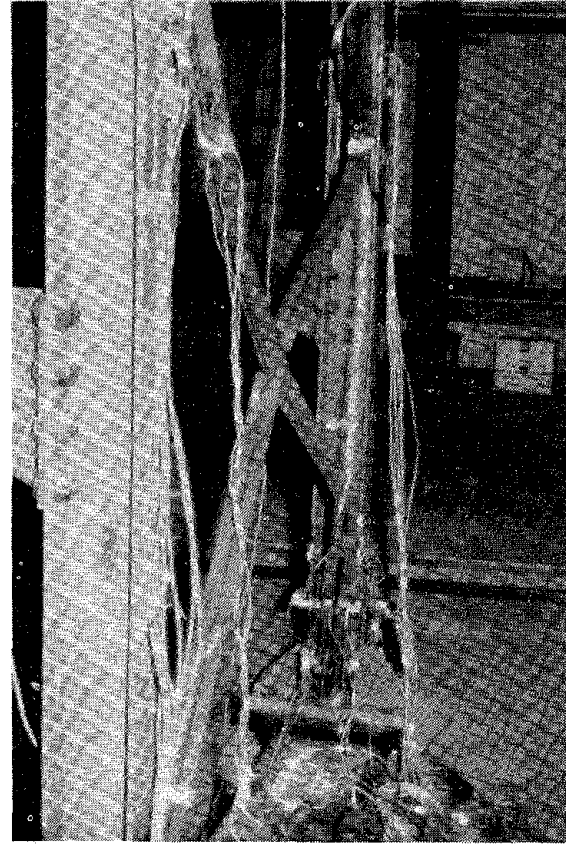
Table 5.3.A

Double Angle Bracing Tests

Test No.	Input Signal	Max. Table Acc. (g)	Wt./Flr (Kips)	Comments
1	EC 100	0.062	12	wo/filler, linear response
2	EC 300	0.192	12	wo/filler, linear response
3	PAC 200	0.244	12	wo/filler, linear response
4	EC 500	0.336	12	wo/filler, angle buckling initiated
5	PAC 350	0.458	12	wo/filler, angle buckling
6	EC 500	0.330	12	w/filler, linear response
7	EC 700	0.485	12	w/filler, angle buckling initiated
8	EC 900	0.689	12	w/filler, angle buckling
9	EC 1000	0.820	12	w/filler, angle buckling minor col. yielding
10	PAC 600	0.772	12	w/filler, angle buckling minor col. yielding
11	EC 100	0.068	17	w/filler, linear response
12	PAC 100	0.127	17	w/filler, linear response
13	PAC 800	1.314	17	w/filler, angle buckling significant col. yielding

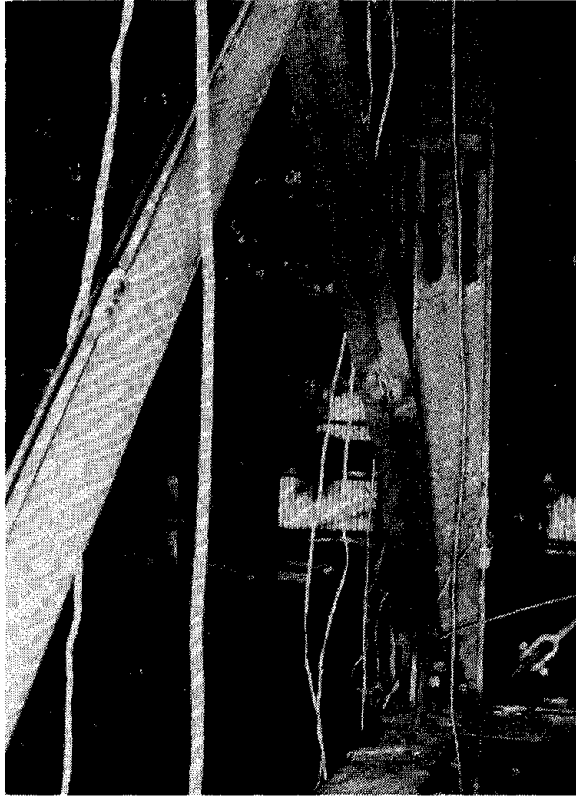


SOUTH FRAME

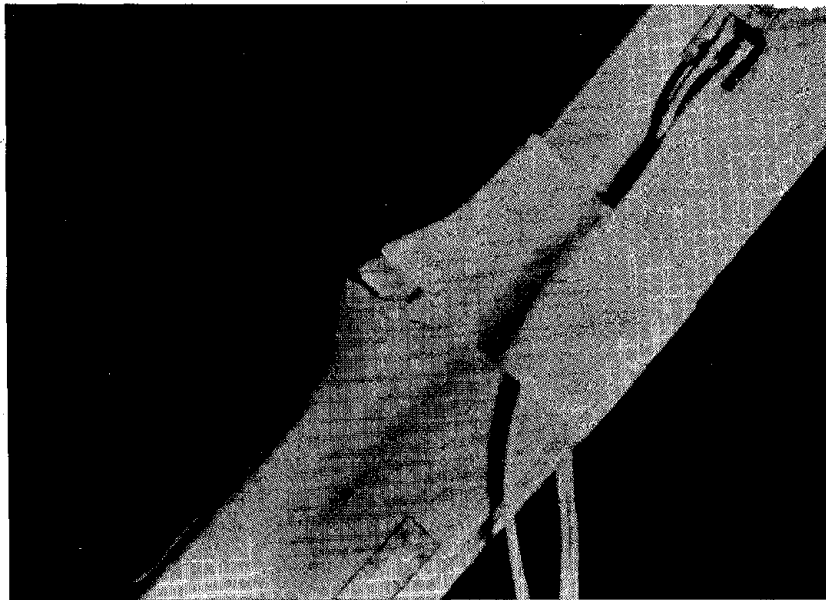


NORTH FRAME

Fig. 5.3.1 Photographs of the 1st Floor Buckled Double Angle Diagonal Braces



2nd Floor Buckled Angle



1st Floor Damaged Angle

Fig. 5.3.2 Photographs of Buckled and Damaged Double Angle Braces

5.3a Double Angle Bracing Subjected to El Centro Span 100

In this test the structure was loaded with concrete blocks weighing 17 kips per floor, and the angle pairs were welded together through 1/4 in. filler plates to insure combined action. The El Centro span 100 signal with a peak acceleration of 0.068 g was applied to the structure, producing a linear structural response. The general behavior during this test was similar to that of the previous braced structure tests using identical excitation. However, the stronger double angle braces reduced the force level developed in the columns and the lateral displacement of the floors.

Time-histories of the floor accelerations and displacements are shown in Fig. 5.3a.1 and Fig. 5.3a.2, respectively. As in the previous tests, the first mode of vibration dominates the structural response. The fundamental frequency during this test was calculated by frequency analysis of the first floor acceleration record using the Fast Fourier Transform (FFT) routine. This frequency was about 3.75 cps which was 15 percent and 20 percent higher, respectively, than the fundamental frequency of the pipe and the rod braced structures under identical loading.

The floor shear time histories shown in Fig. 5.3a.3 indicate that the double angle braces resisted more than 90 percent of the lateral forces (see the bottom frame).

5.3b Double Angle Bracing Subjected to El Centro Span 900

This test was performed during test series 1, while the structure was loaded with 12 kips per floor. The slenderness ratio of the combined double angle diagonals welded at their mid-span intersection was $KL/r = 86$ at the first floor level, and $KL/r = 72$ at the second and third floor levels. The braced frame was subjected to the El Centro span 900 signal with a

peak acceleration of 0.689 g. This excitation induced compression buckling and tension yielding in all the first floor double angle braces. The buckling load for the first cycle was estimated to be about 8.5 kips, and it reduced to about 6 kips in the subsequent cycles. No buckling and/or yielding occurred in the upper floor braces, and all beams and columns responded within their elastic range.

The time histories of the floor accelerations and displacements, in which the first mode of vibration is dominant and the frequency change due to the nonlinear behavior of the braces is not significant, are shown in Fig. 5.3b.1 and Fig. 5.3b.2, respectively. Also it may be noted that the floor relative displacements are quite small.

The floor shear forces are displayed in Fig. 5.3b.3. The maximum first floor shear was about 20 kips which was sufficient to cause brace buckling and yielding. The bottom frame in this figure indicates that about 88 to 93 percent of the first floor shear was resisted by the double angle diagonals. Indeed, the first floor angle braces buckled only during two response cycles as the first floor shear became larger than 16 kips (see Fig. 5.3b.3, bottom frame). The force-displacement hysteresis loops for the diagonal braces are shown in Fig. 5.3b.4 for successive time intervals of 4 seconds. As was mentioned earlier the initial buckling load was estimated to be about 8.5 kips, whereas the buckling load for the subsequent cycles decreased to about 6 kips; in addition, the brace yielded in tension (see the upper left frame). The subsequent lateral force level was such that the bracing diagonals remained within their elastic range. The post-buckling displacement and tension yielding of the first floor diagonals were not severe enough to produce a pinching effect in the force-displacement curves; also the strength loss

of the diagonals due to two induced buckling cycles was negligible. Accordingly, no significant frequency change can be seen in the response time histories.

The hysteresis loops of the first floor shear versus displacement are shown in Fig. 5.3b.5. These curves also include the combined force-displacement contribution of the double angle diagonals. The flat portion of the curve in the first four second time interval (upper left frame) demonstrates the buckling of the diagonals.

These results indicate that despite the buckling of the diagonals, the damage was minor, and the strength loss of the first floor diagonals was negligible. Therefore, the bracing system efficiently resisted the lateral forces and controlled the lateral displacement during this excitation.

5.3c Double Angle Bracing Subjected to Pacoima Span 800

After test series 1, the damaged first floor double angle diagonals were replaced and the structure with a dead load of 17 kips per floor was subjected to the Pacoima Dam earthquake motion with a peak acceleration of 1.314 g. The intensity of this excitation was so great that the resulting lateral forces induced buckling of the first and second floor diagonals, and yielding of the first floor columns. Photographs of the buckled double angle braces are shown in Fig. 5.3.1 and Fig. 5.3.2.

The time histories of table and floor accelerations, displayed in Fig. 5.3c.1, illustrate the frequency change associated with the non-linear structural response. The measured frequency of the damaged structure was 2 cps, a 47 percent reduction from the initial elastic frequency. The high frequency component in the third floor acceleration signal was found to be a disturbance produced by poor installation of the third

floor accelerometers. It is believed that the actual third floor acceleration was not more than 2 g, and the large amplitudes of the peak acceleration in the record were the result of that installation. The three floor displacement time histories shown in Fig. 5.3c.2 are identical in form and have almost the same amplitude. This was expected because the major plastic deformations were concentrated in the first floor level, including plastic hinges formed at the base of the first floor columns (see also the floor drift shown in Fig. 5.3c.3). The two third floor displacement time histories indicate that the displacements of the two end frames were nearly identical and, thus, that structural symmetry was preserved.

The first and second floor double angle diagonals buckled with respect to their Y-axis in a direction perpendicular to the plane of the frame because they were completely restrained by the tensile braces. The initial buckling load was estimated to be about 9.3 and 12 kips for the first and second floor braces, respectively. The maximum compressive force developed in the third floor braces was only 7 kips, so they never buckled. The buckling of the first floor diagonals was repeated in subsequent response cycles at lower force levels, and the post-buckling displacements were such that the plastic hinges developed in the mid-span section of these braces.

The floor shear forces are shown in Fig. 5.3c.4. The bottom frame in this figure displays the first floor shear and the portion of this shear that was resisted by the first floor braces. This graph demonstrates that more than 90 percent of the total lateral force was resisted by the braces before they buckled. As the buckling initiated, the brace force resisting capacity decreased to 70 to 80 percent of the total lateral force, and when they were considerably damaged, this contribution dropped to as low

as 30 to 40 percent of the total. The first floor shear versus displacement curves which are shown in Fig. 5.3c.5, also include force-displacement hysteresis loops of the combined diagonals. The flat portions of these curves are associated with brace buckling and yielding, and the pinching type behavior in the lower left curve is due to the accumulated tensile yield deformations. The bracing efficiency diminished as the pinching developed, and the lateral first floor stiffness decreased to about 14 percent of its initial value (see the lower right curve). Accordingly, the first floor column yielded significantly as the bracing strength diminished, and a residual distortion was developed at the bottom ends of columns. The first floor column moment-curvature hysteresis loops are shown in Fig. 5.3c.6.

5.4 Unbraced Steel Frame Tests

One of the objectives of this investigation was to compare the performance of different diagonal bracing systems and study their benefits and disadvantages with respect to the unbraced steel frame structure. Thus, for the final test series the diagonal braces were removed, and the basic moment-resistant steel frame was subjected to the El Centro and the Pacoima Dam earthquake motions. The dead load of the structure was supplied by concrete blocks weighing 17 kips per floor. The tests performed on the unbraced frame and a summary of the results are given in Table 5.4.A.

The response of the steel frame to excitations with peak accelerations up to 0.2 g was within the elastic range. The resulting lateral forces were much less than those of the braced structures because the unbraced structure was softer. However, the floor displacements and the column force levels were much higher.

Eight tests were performed, ending with the Pacoima span 400 signal with a peak acceleration of 0.481 g. This table motion induced significant yielding in the first floor columns including a residual distortion; the maximum third floor displacement was more than 4 inches. At this stage it was decided to terminate the tests, because a stronger excitation might have caused serious damage to the frame or even collapse. The results of one elastic and one inelastic test are discussed in the following subsections.

Table 5.4.A
Unbraced Frame Tests

Test No.	Input Signal	Max. Table Acc. (g)	Wt./Flr (Kips)	Comments
1	EC 100	0.064	17	Linear response
2	PAC 50	0.063	17	Linear response
3	EC 200	0.123	17	Linear response
4	EC 300	0.193	17	Linear response
5	PAC 200	0.235	17	Minor col. yielding
6	EC 400	0.268	17	Minor col. yielding
7	PAC 300	0.364	17	Col. yielding
8	PAC 400	0.481	17	Col. yielding w/residual distortion

5.4a Unbraced Frame Subjected to El Centro Span 100

The unbraced steel frame was loaded to 17 kips per floor and was subjected to the El Centro earthquake with a peak acceleration of 0.064 g. The response of the structure to this excitation was within the elastic range as had been expected. The fundamental frequency of the unbraced structure measured by a free vibration test was 1.45 cps. This free vibration test was performed when the shaking table was resting on its static supports. However, the response frequency during a simulated earthquake test was smaller due to table-structure interaction. Therefore, to determine this frequency, a Fast Fourier Transform analysis was performed on the first floor acceleration history. The calculated frequency was found to be 1.186 cps, a reduction of 18 percent, which was due to the flexibility of the table actuator system.

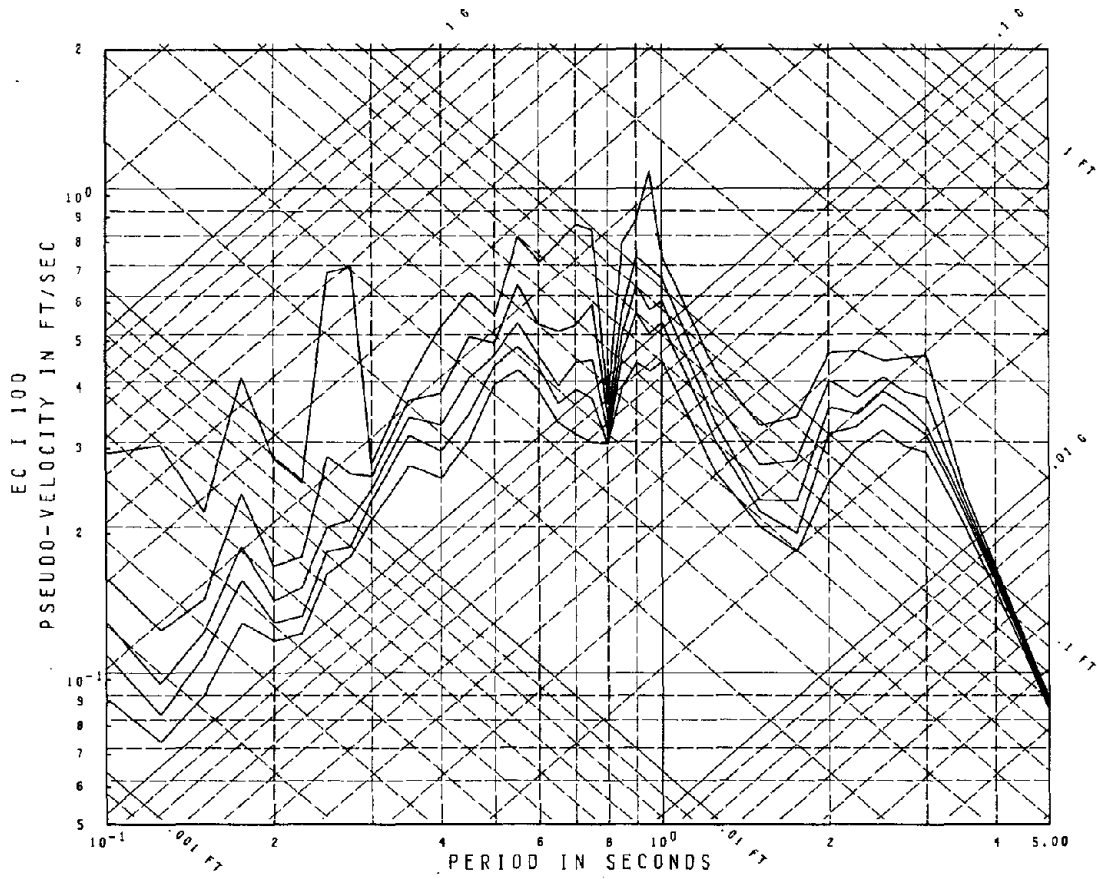
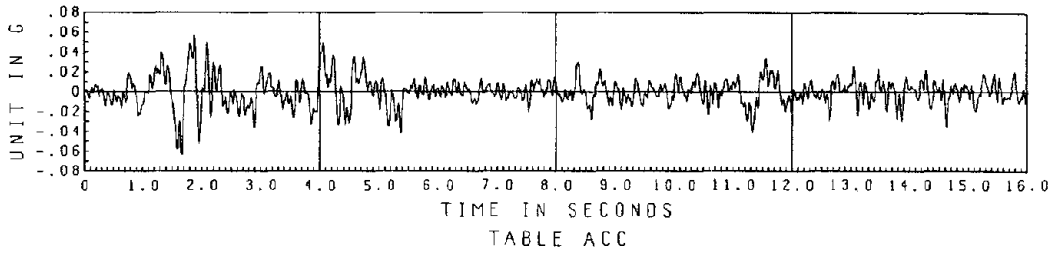
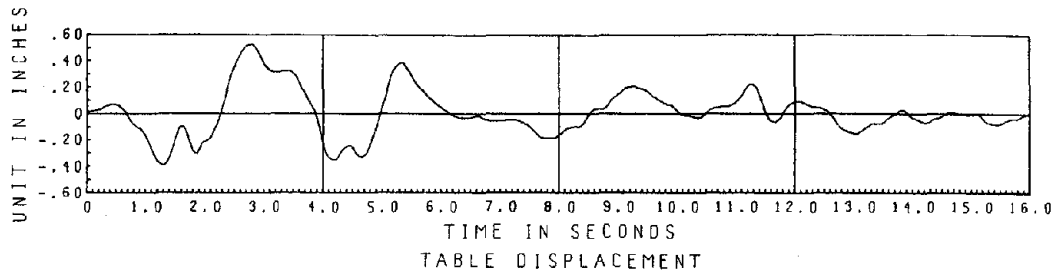
The time histories of floor accelerations, displacements, drifts, and shears are shown in Fig. 5.4a.1 to Fig. 5.4a.4. These results show that the unbraced frame had larger floor displacements and drifts than the braced structures, as would be expected. Also the column force levels were larger. However, the total floor shears were smaller in the unbraced frame because the floor accelerations were not amplified so much in this flexible structure.

5.4b Unbraced Frame Subjected to Pacoima Span 400

The last earthquake motion applied to the unbraced frame was the Pacoima Dam signal with a peak acceleration of 0.481 g. The dead weight of the structure was 17 kips per floor as before. This input motion caused inelastic structural response and induced a residual distortion at the bases of the first floor columns.

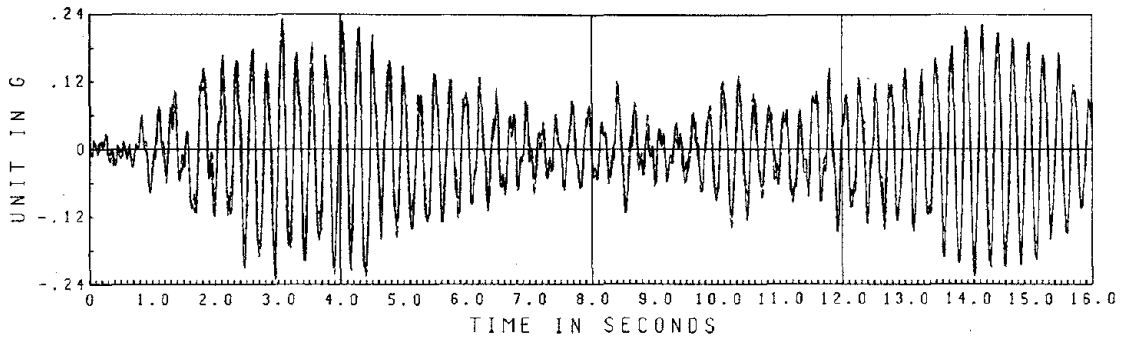
The floor accelerations shown in Fig. 5.4b.1 demonstrate the importance of the second mode of vibration in this structure. The floor displacements were very large in general, especially at the third floor where the peak displacement was twice as large as the maximum of the shaking table motion. The time histories of floor displacements shown in Fig. 5.4b.2 demonstrate that the first mode of vibration is dominant in the displacement response. The fundamental frequency calculated from the response history was about 1.04 cps, a 12 percent reduction with respect to the elastic case. The floor drifts are shown in Fig. 5.4b.3, which demonstrates that the drifts of the higher floors are larger than for the braced structures. The moment-curvature hysteresis loops of the columns show their yielding mechanism; the residual distortion is indicated by the shift of these curves from the center (see Fig. 5.4b.4).

These results show that for the unbraced frame lateral movement is very large because of its insufficient stiffness. The nonlinear behavior of the structure is associated with column yielding. During the application of a strong earthquake signal to an unbraced frame, the formation of plastic hinges at the bases of the columns should be expected; and these plastic hinges could cause the structure to collapse if the dead load were close to its critical buckling load.



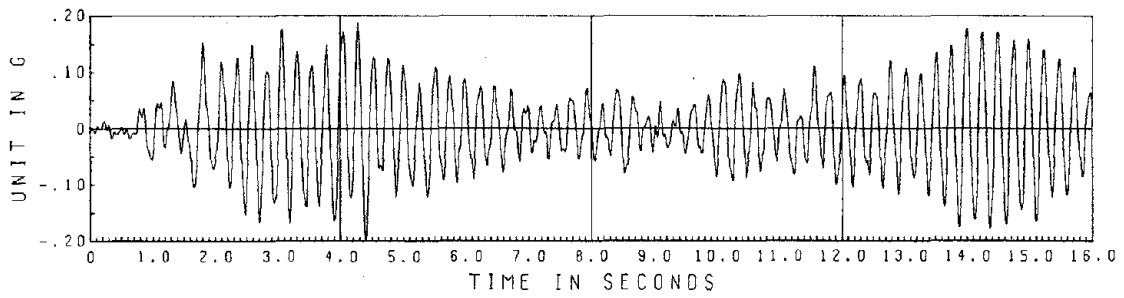
Damping = .00, .01, .02, .03, .05 Critical

Fig. 5.1a.1 El Centro Span 100 Horizontal Table Motion

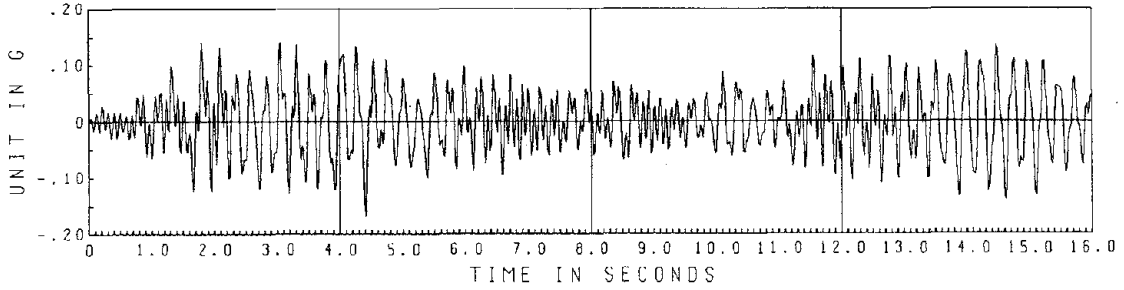


SOLID LINE = 3RD FLOOR ACC. NA

DASHED LINE = 3RD FLR ACC. SA



2ND FL. ABSOLUTE ACCELERATION



1ST FL. ABSOLUTE ACCELERATION

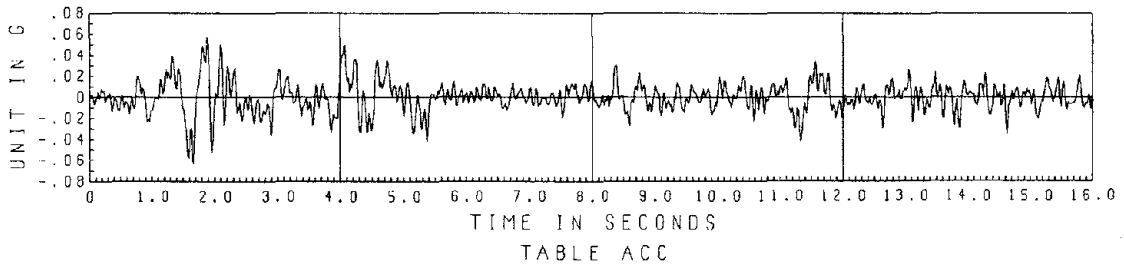
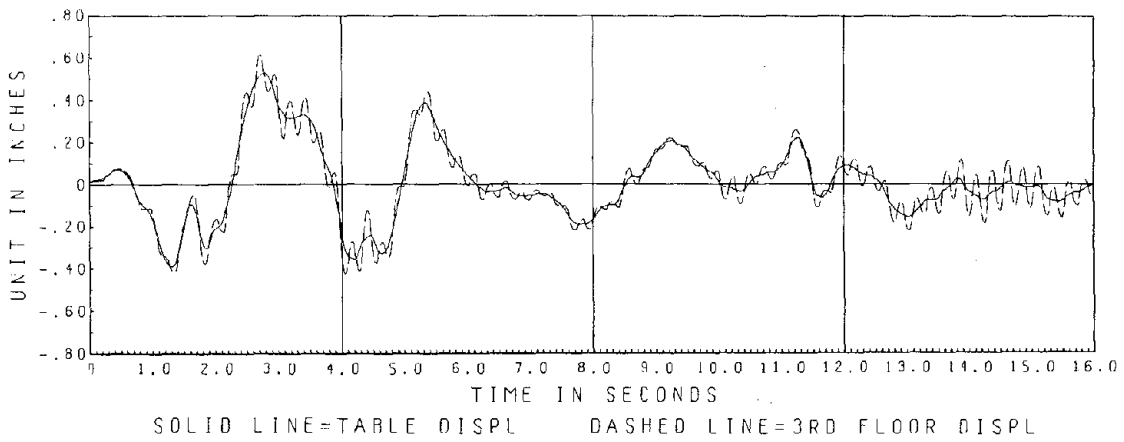
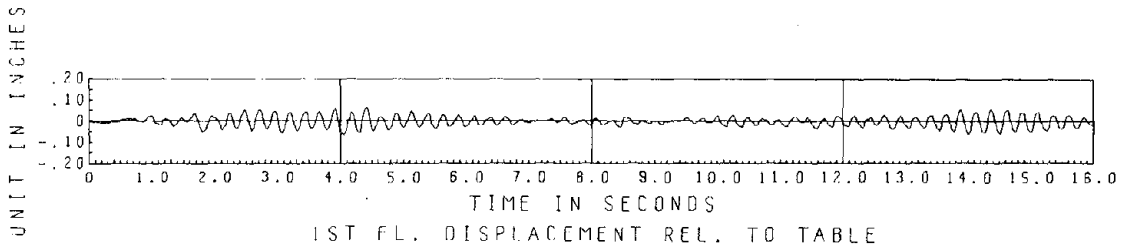
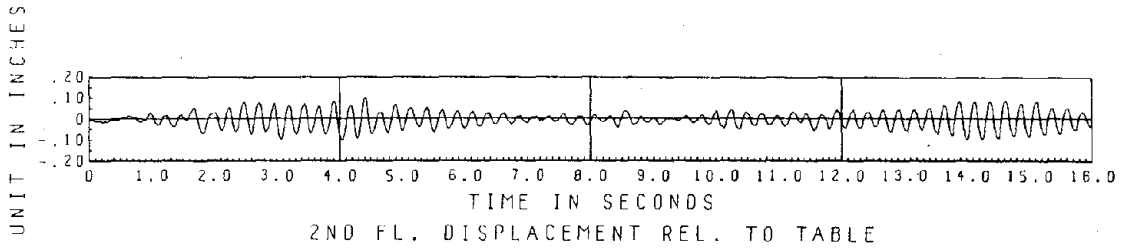
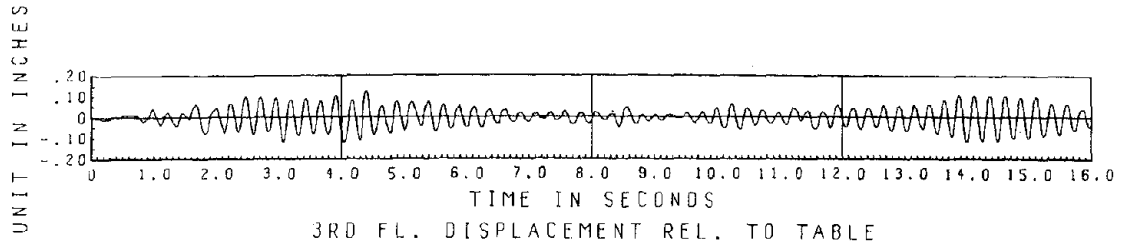


TABLE ACC

EL CENTRO SPAN 100

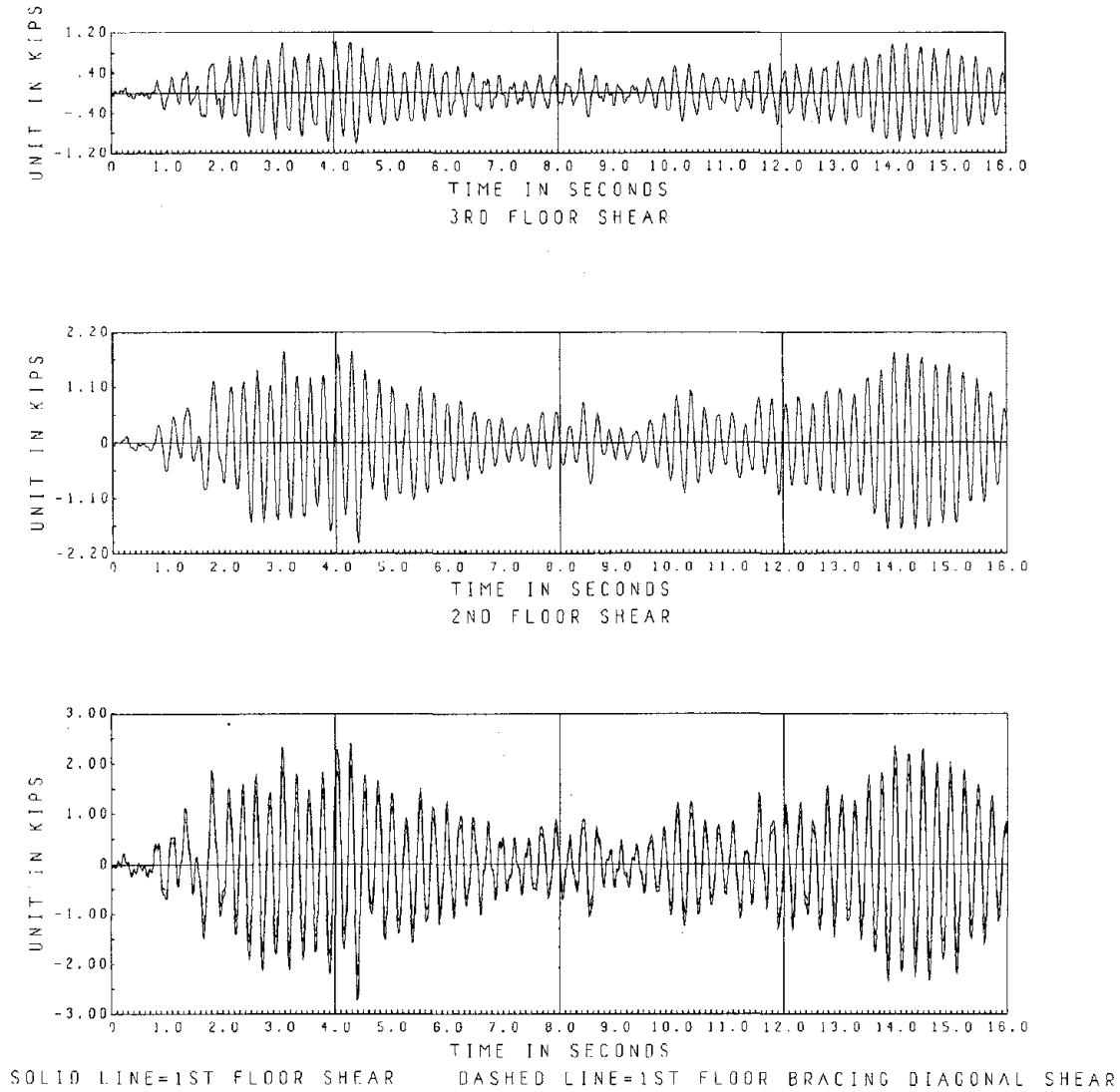
TEST RESULTS OF ROD BRACING

Fig. 5.1a.2 Table and Floor Accelerations



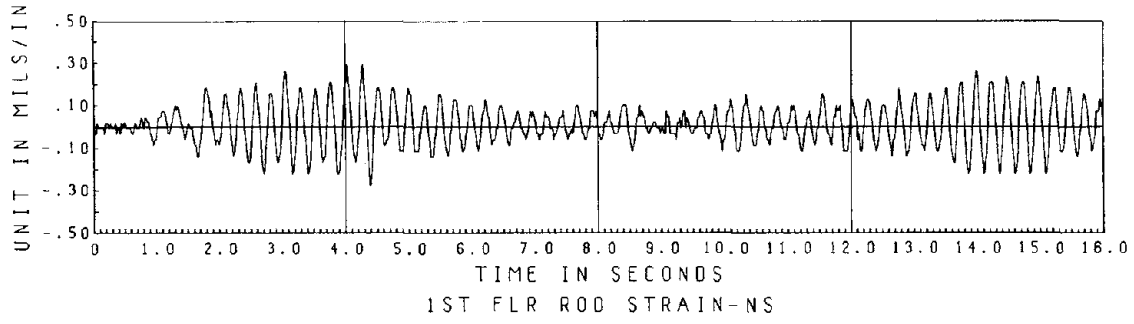
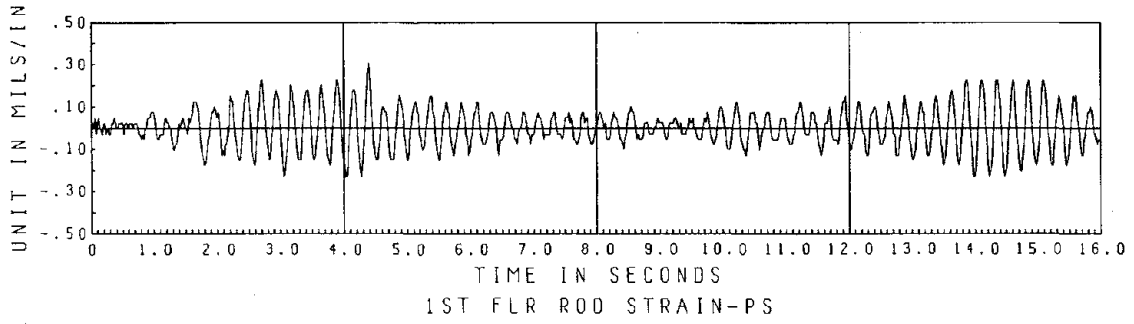
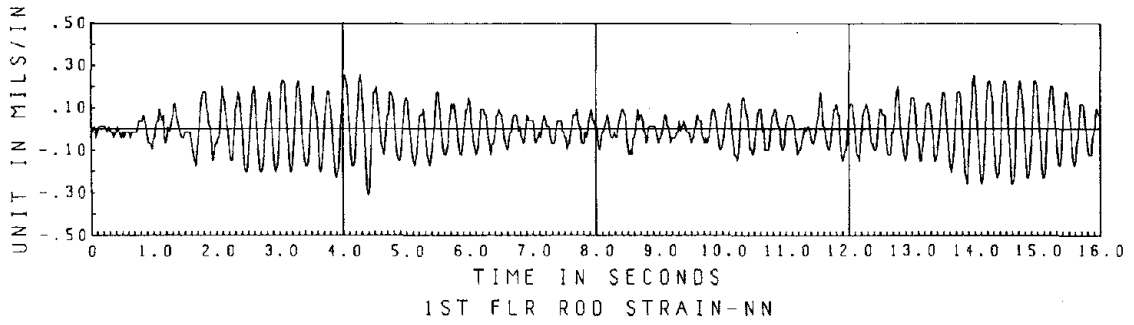
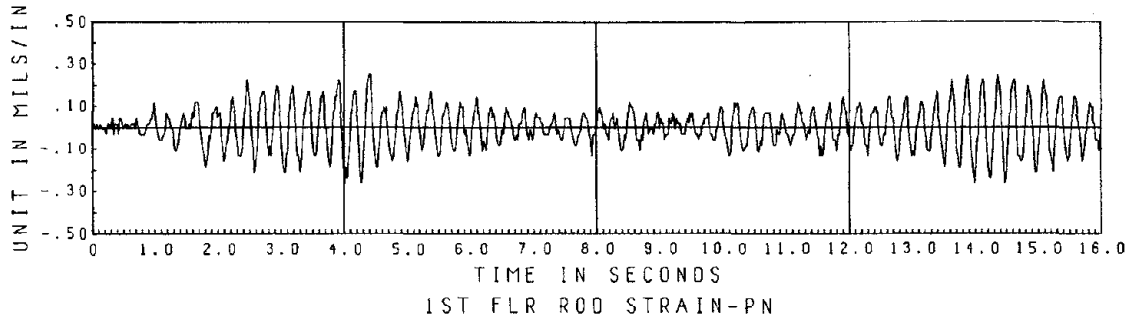
EL CENTRO SPAN 100
TEST RESULTS OF ROD BRACING

Fig. 5.1a.3 Table and Floor Displacements



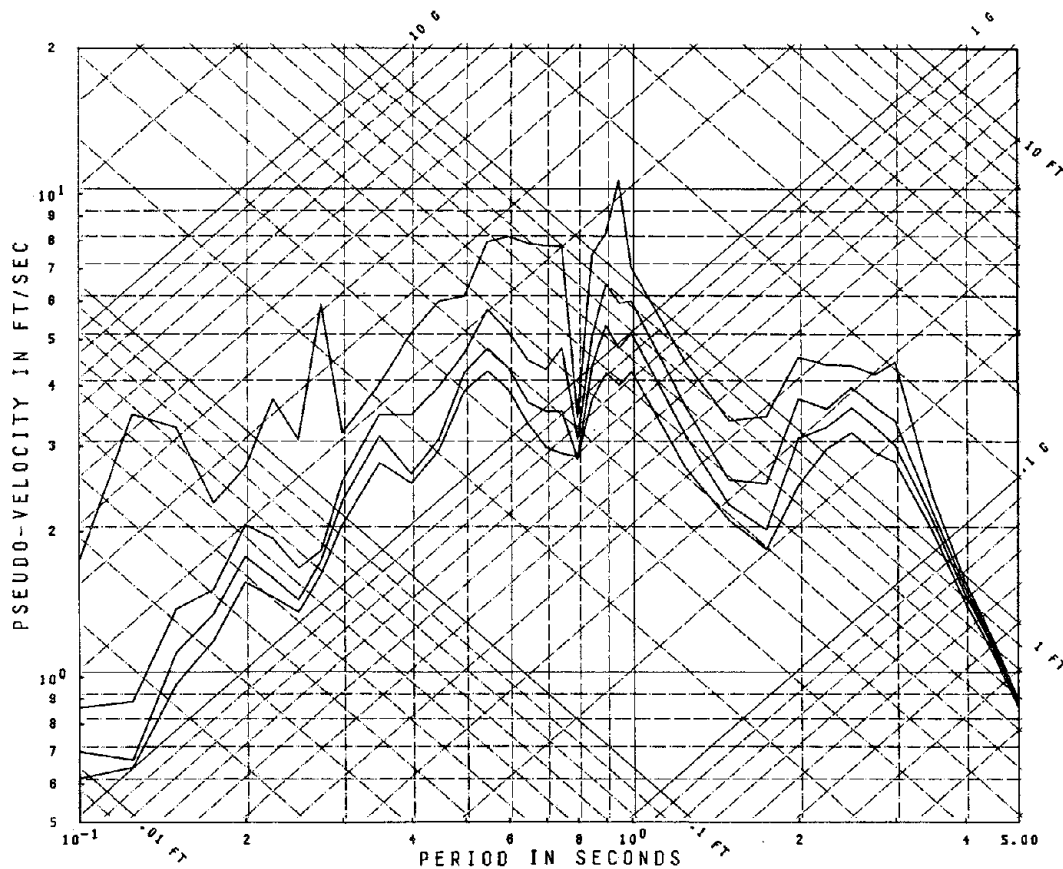
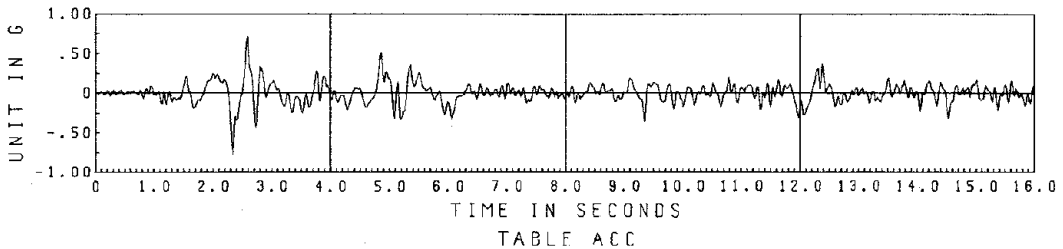
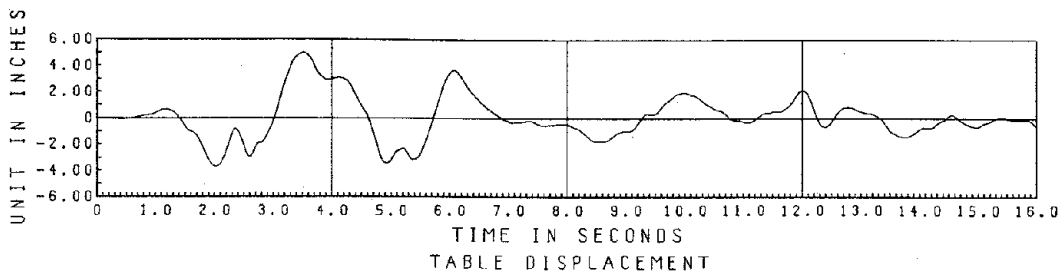
EI. CENTRO SPAN 100
TEST RESULTS

Fig. 5.1a.4 Floor Shear Forces



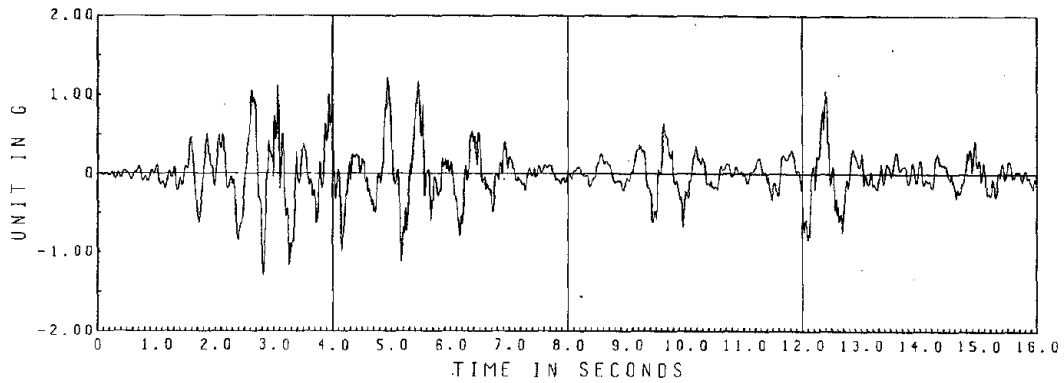
EL CENTRO SPAN 100
TEST RESULTS OF ROD BRACING

Fig. 5.1a.5 Strain Time-Histories of the 1st Floor Rod Braces

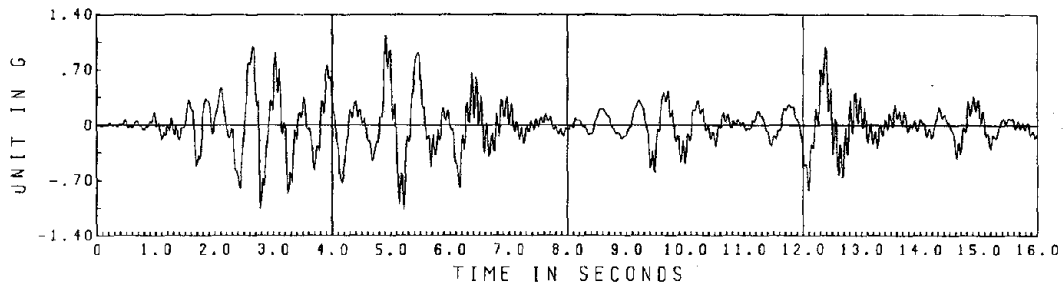


Damping = .00, .015, .03, .05 Critical

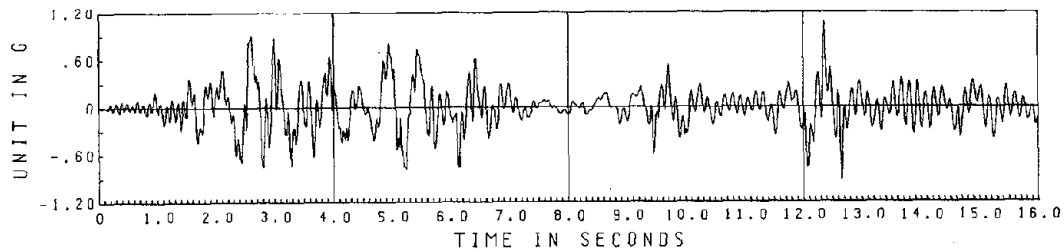
Fig. 5.1b.1 El Centro Span 1000 Horizontal Table Motion



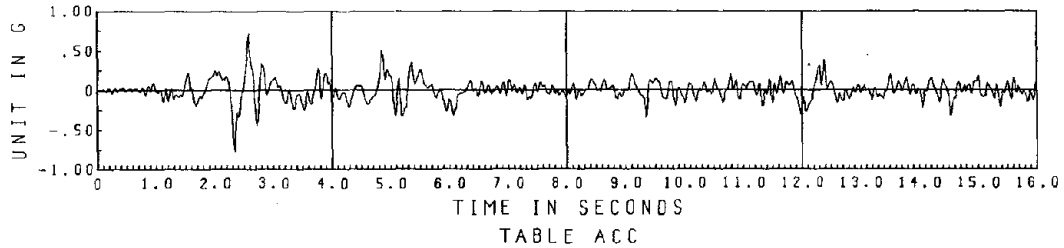
3RD FL. ABSOLUTE ACCELERATION



2ND FL. ABSOLUTE ACCELERATION



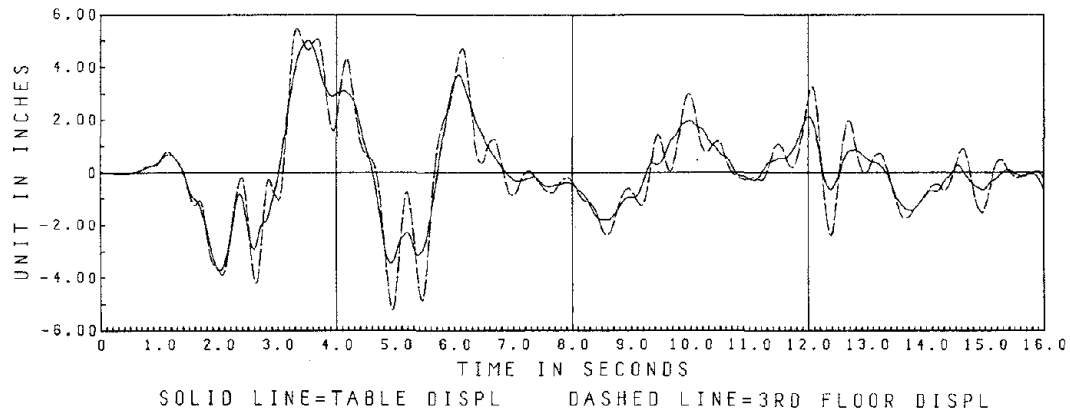
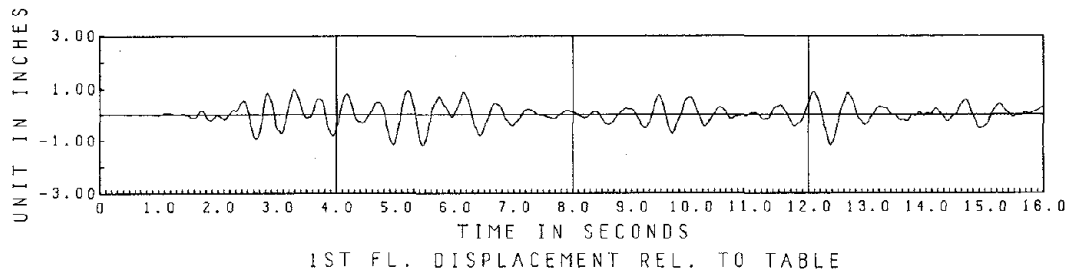
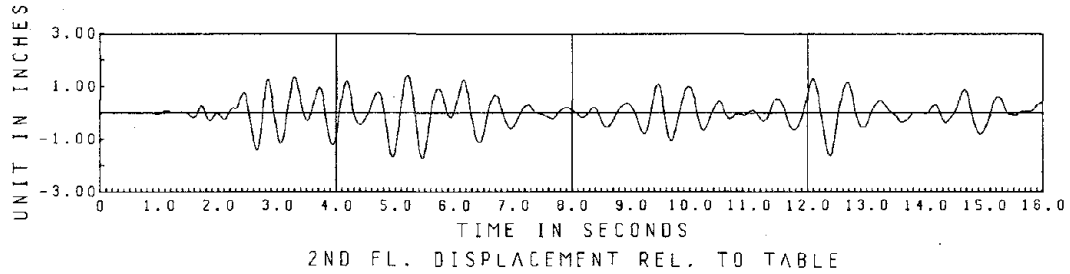
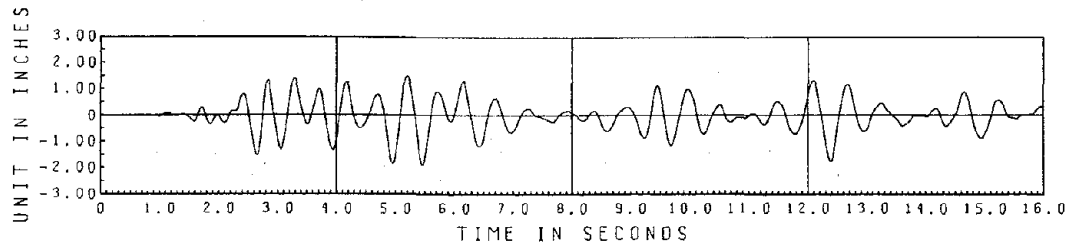
1ST FL. ABSOLUTE ACCELERATION



EL CENTRO SPAN 1000

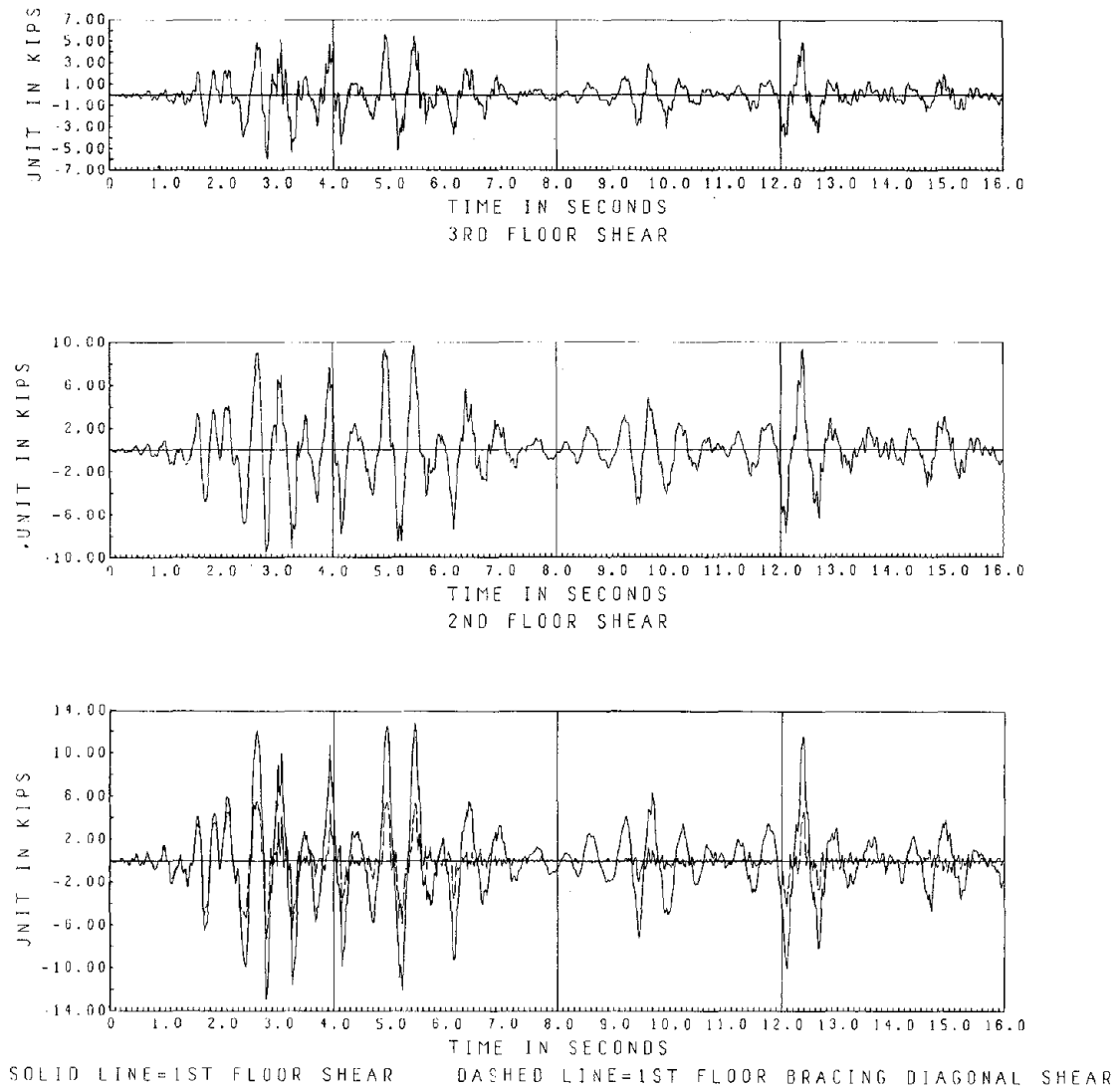
TEST RESULTS OF ROD BRACING

Fig. 5.1b.2 Table and Floor Accelerations



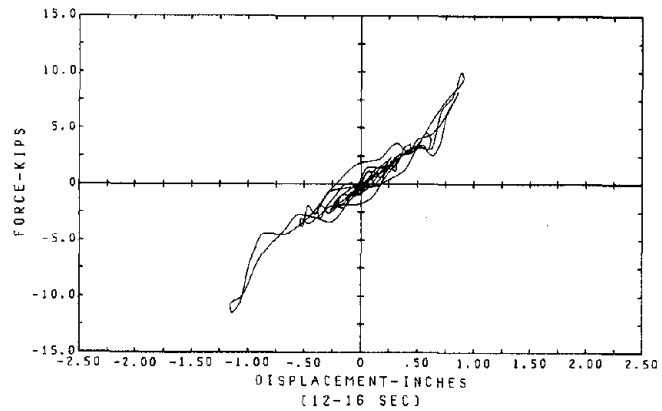
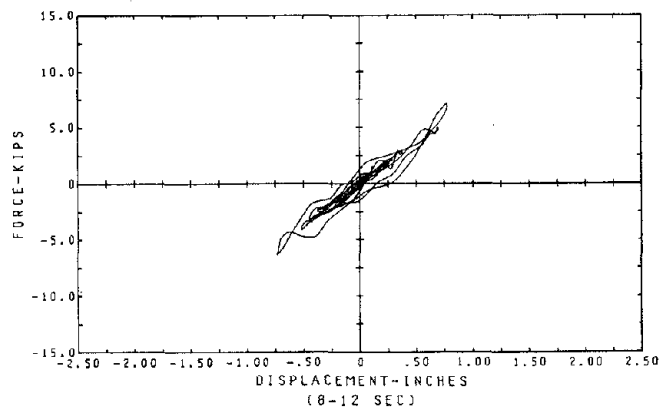
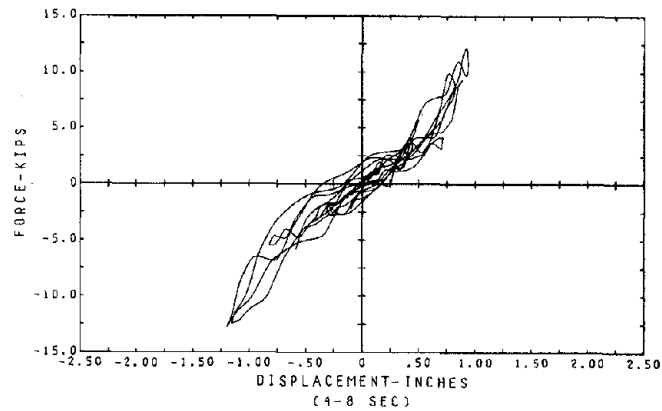
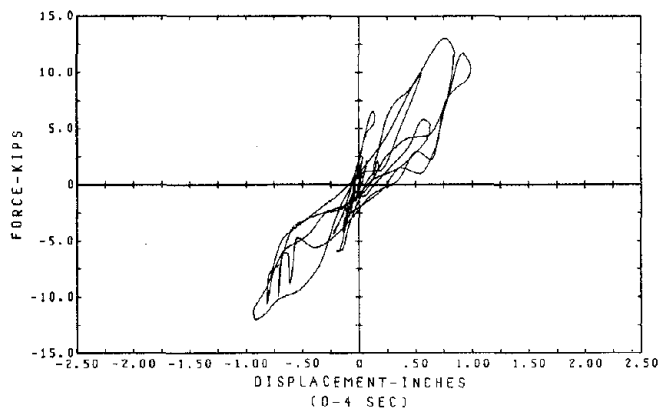
EL CENTRO SPAN 1000
TEST RESULTS OF ROD BRACING

Fig. 5.1b.3 Table and Floor Displacements



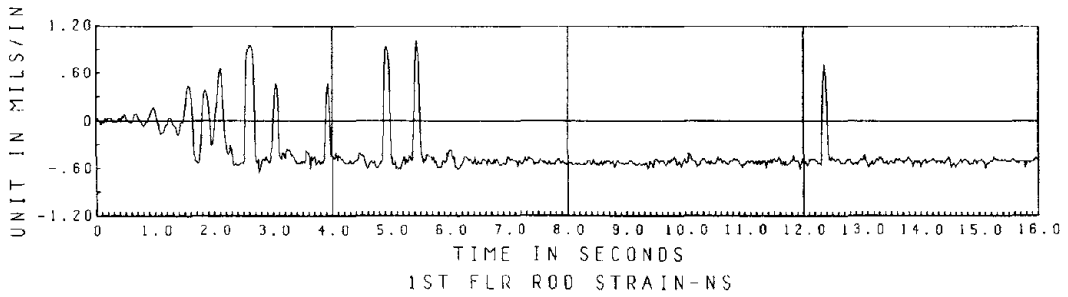
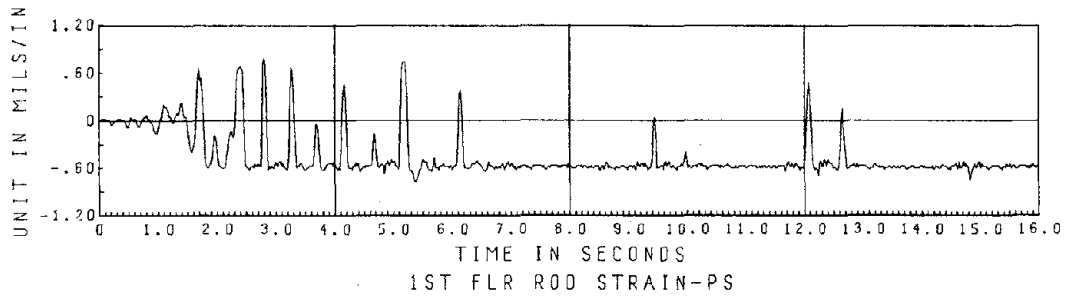
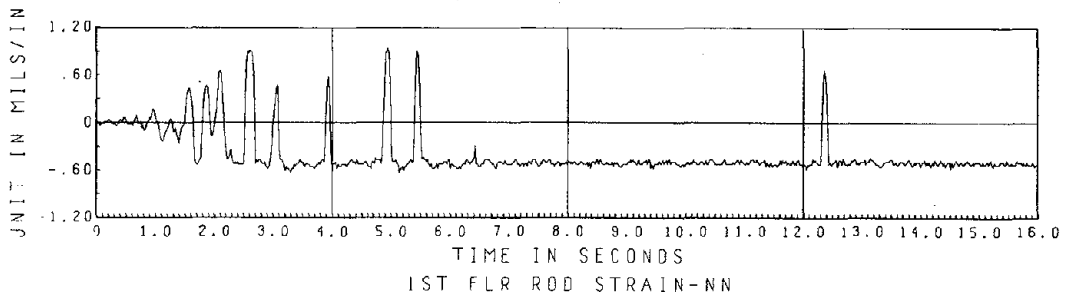
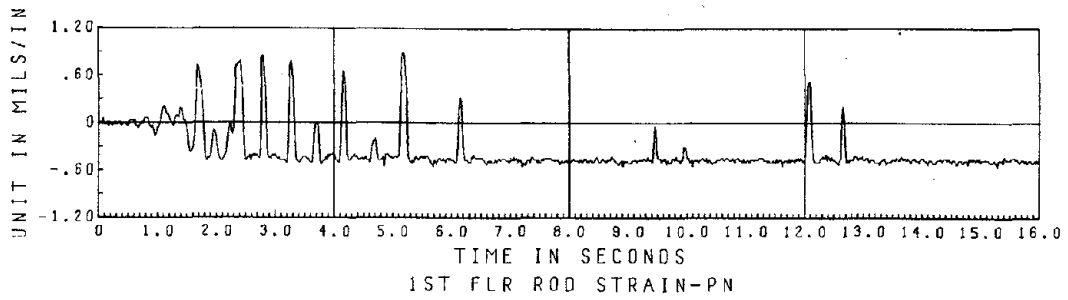
EL CENTRO SPAN 1000
TEST RESULTS

Fig. 5.lb.4 Floor Shear Forces



1ST FLR SHEAR VS. DISPLACEMENT
 REFERENCE FRAME N
 ROD DIAGONALLY BRACED STRUCTURE
 EL CENTRO SPAN 1000

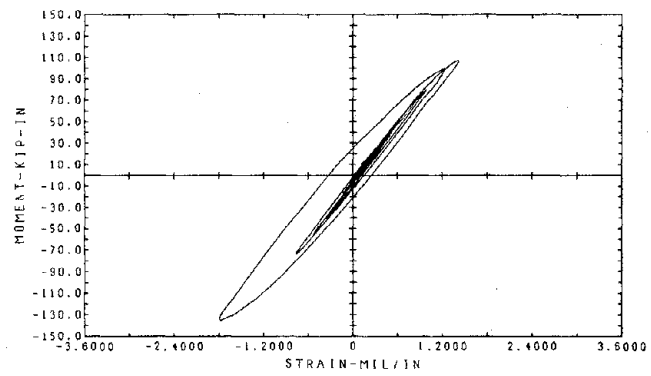
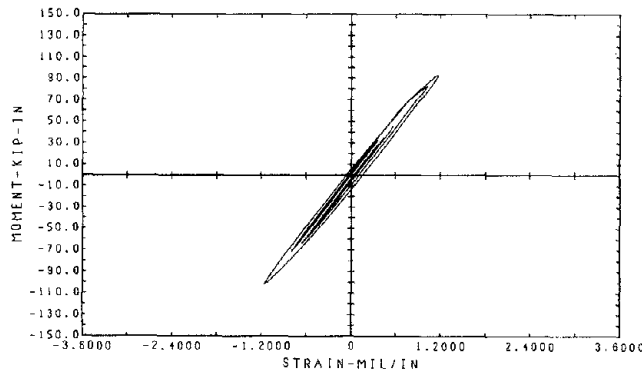
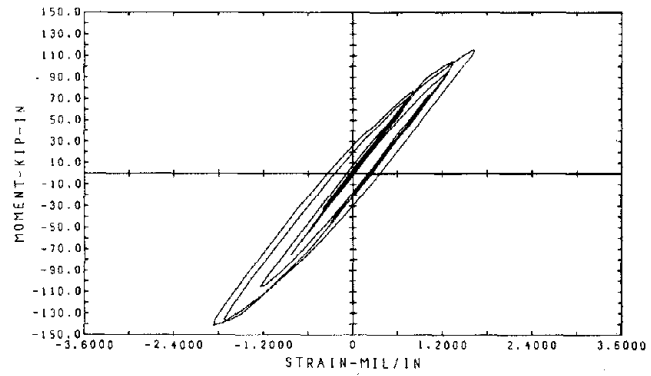
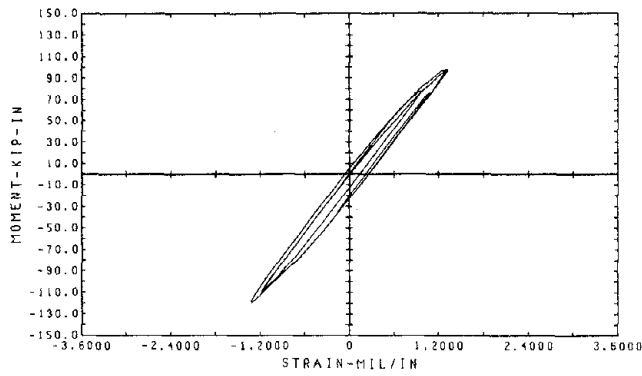
Fig. 5.1b.5 First Floor Shear-Displacement Hysteresis Loops



EL CENTRO SPAN 1000

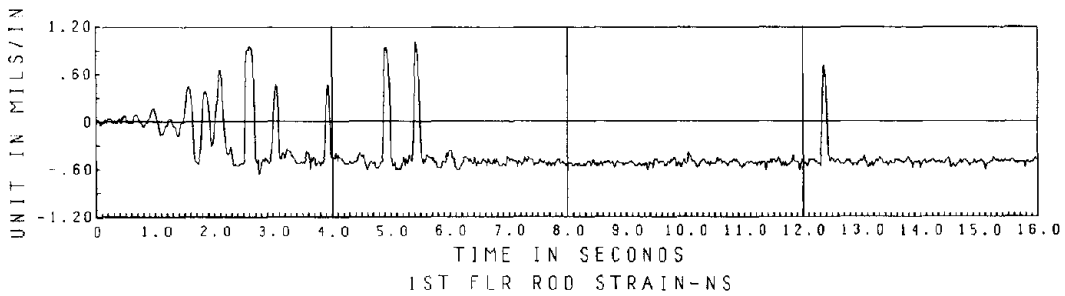
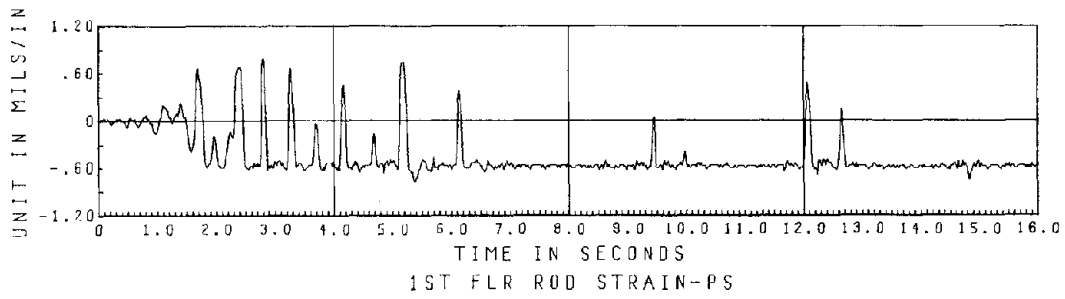
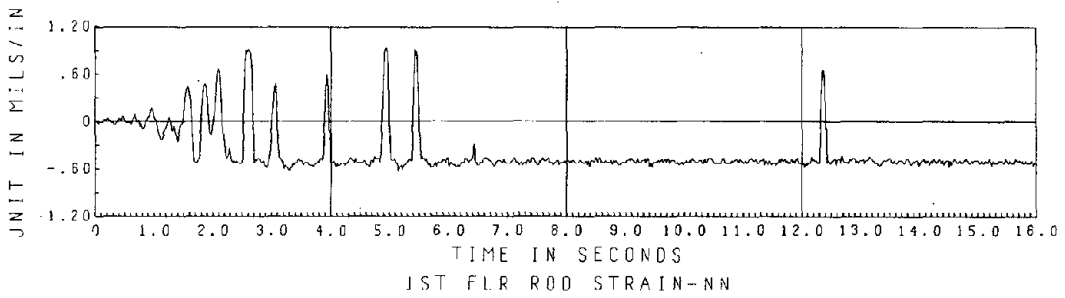
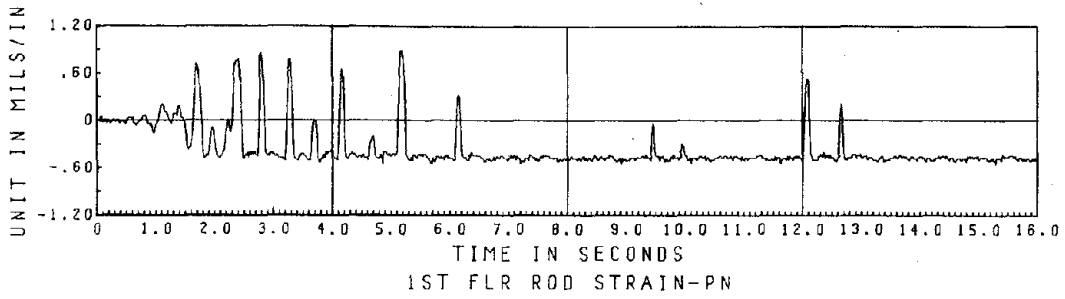
TEST RESULTS

Fig. 5.1b.6 Strain Time-Histories of the 1st Floor Rod Braces



1ST FLR. COL BOTTOM END, GAUGES AT 1 IN. ABOVE STIFF. PLATE
 REFERENCE COLUMN LINE NB
 EC 1 1000

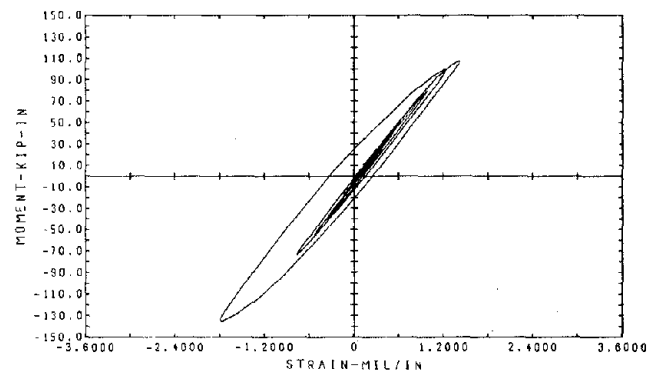
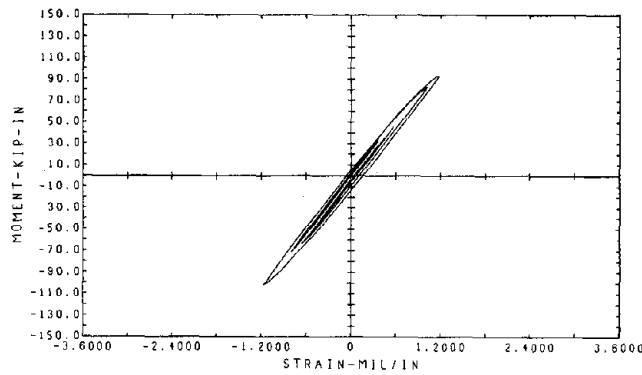
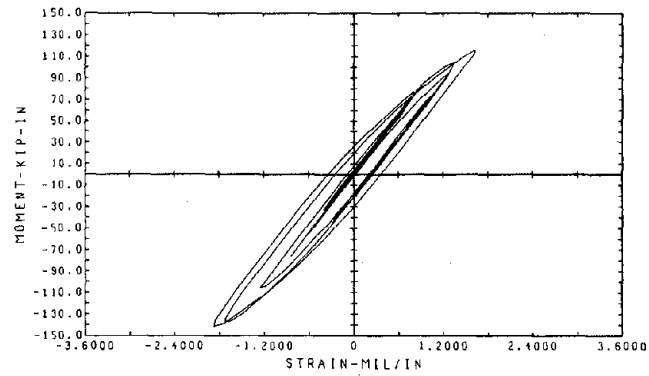
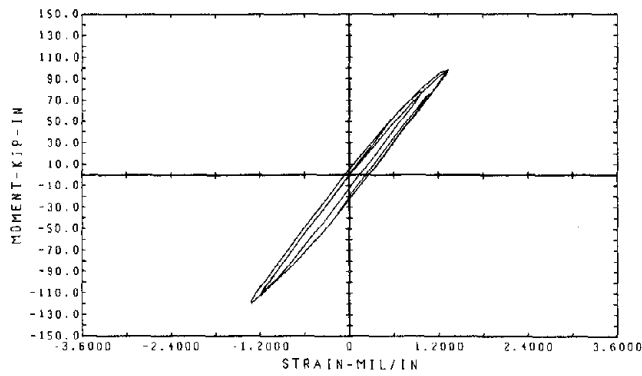
Fig. 5.1b.7 First Floor Column Moment-Strain Hysteresis Loops



EL CENTRO SPAN 1000

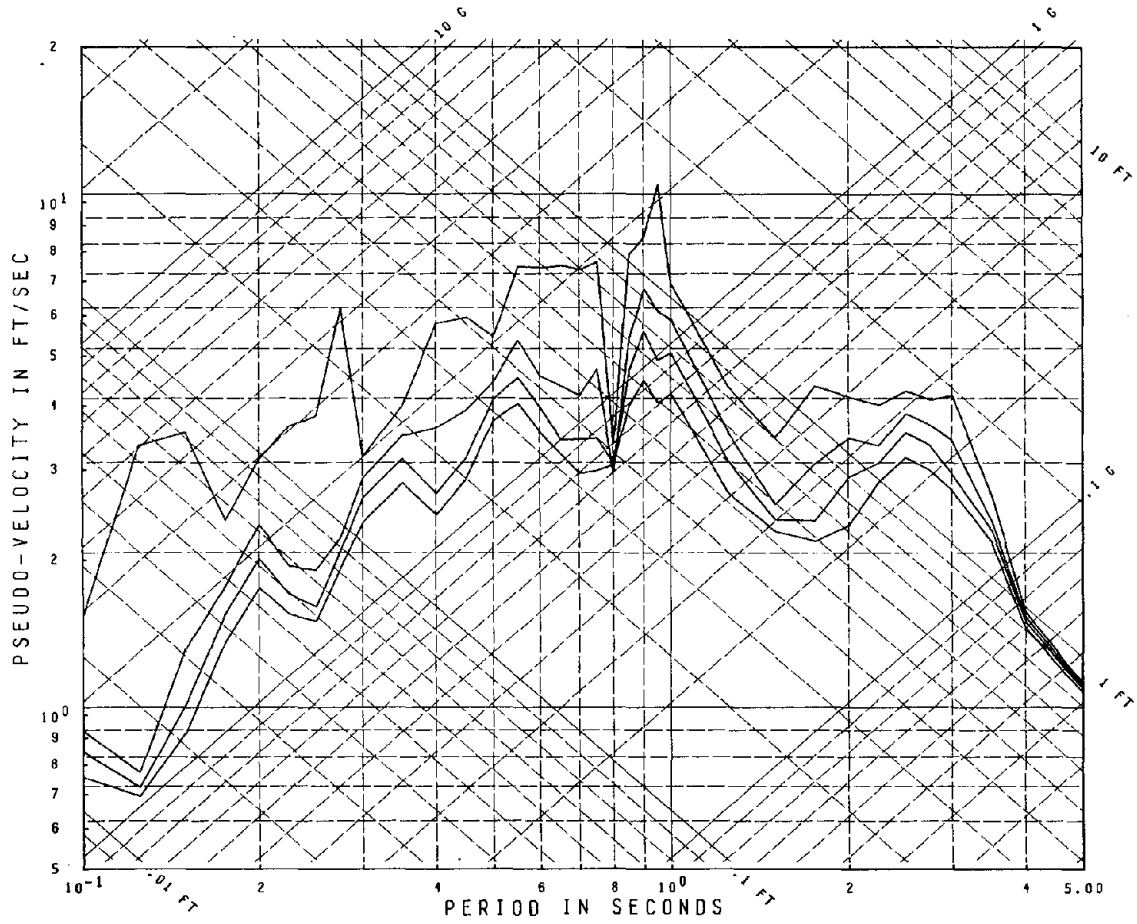
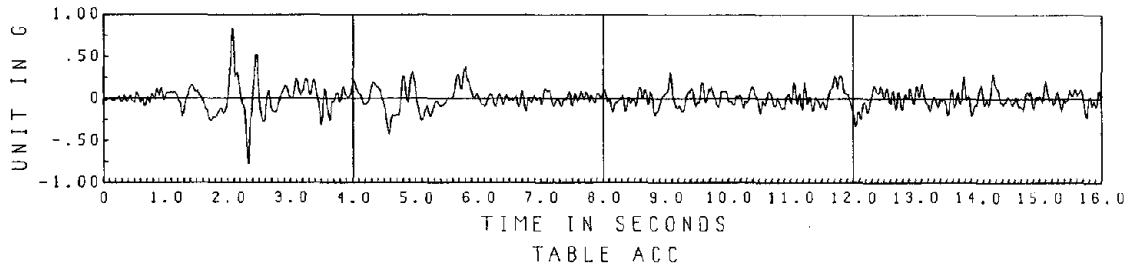
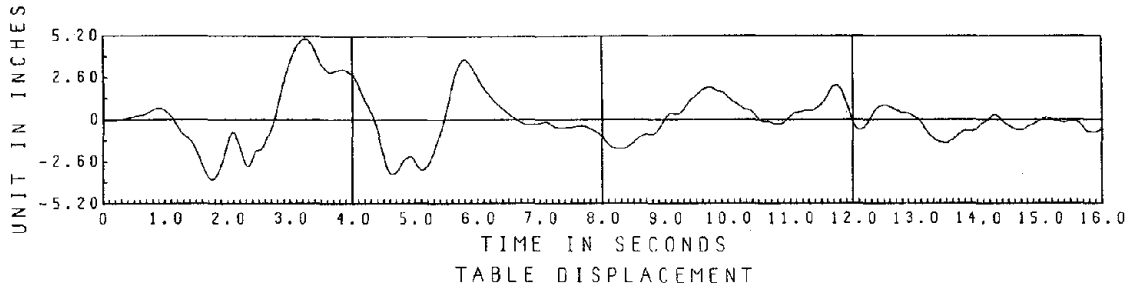
TEST RESULTS

Fig. 5.1b.6 Strain Time-Histories of the 1st Floor Rod Braces



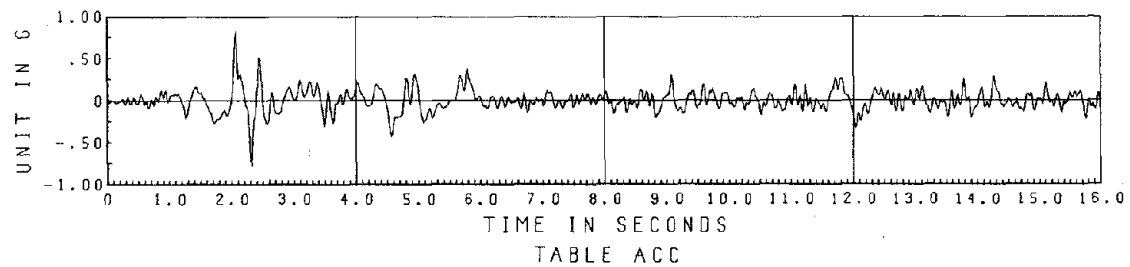
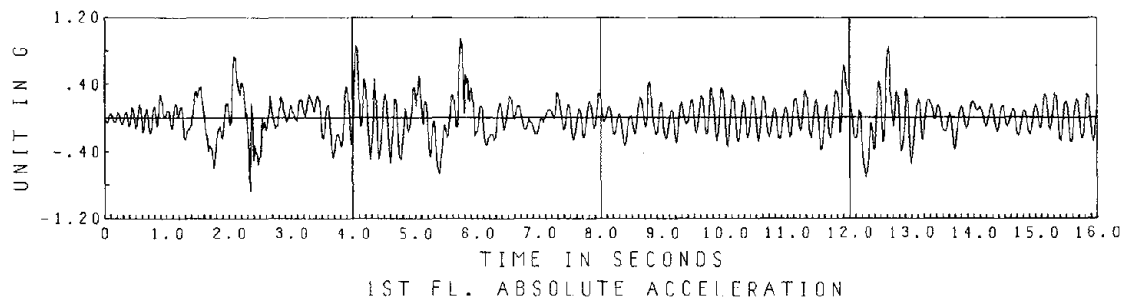
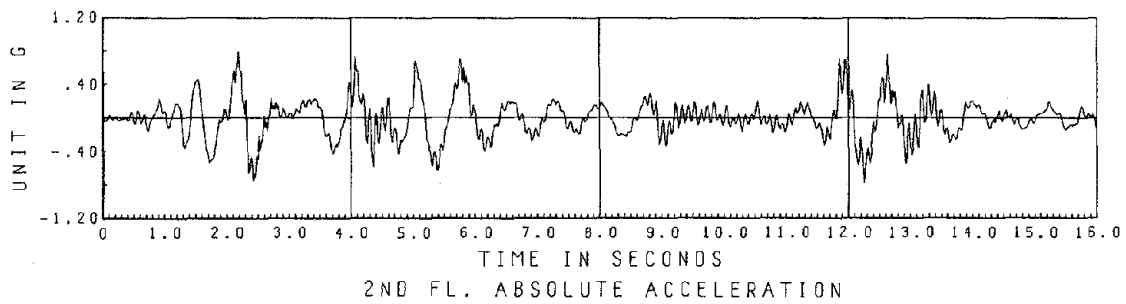
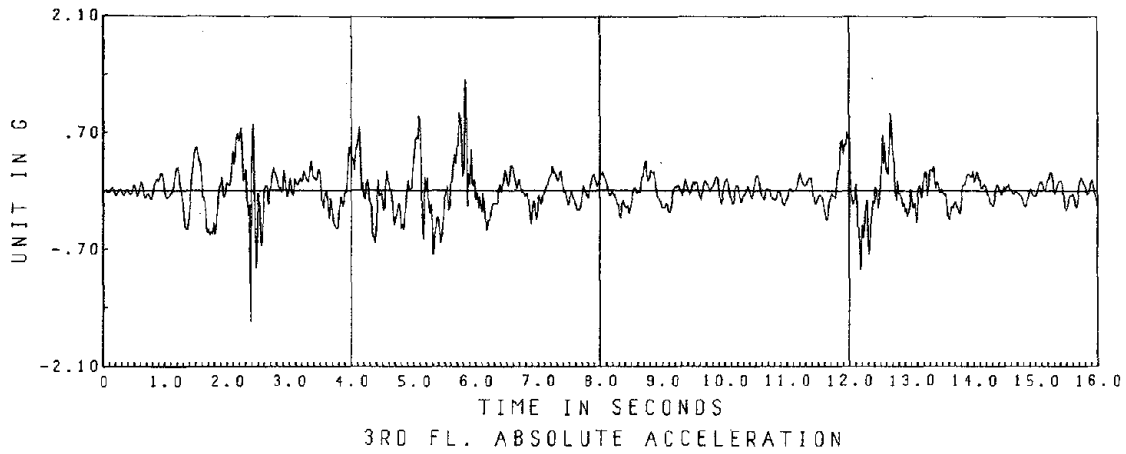
1ST FLR. COL BOTTOM END, GAGES AT 1 IN. ABOVE STIFF. PLATE
 REFERENCE COLUMN LINE N8
 EC 1 1000

Fig. 5.1b.7 First Floor Column Moment-Strain Hysteresis Loops



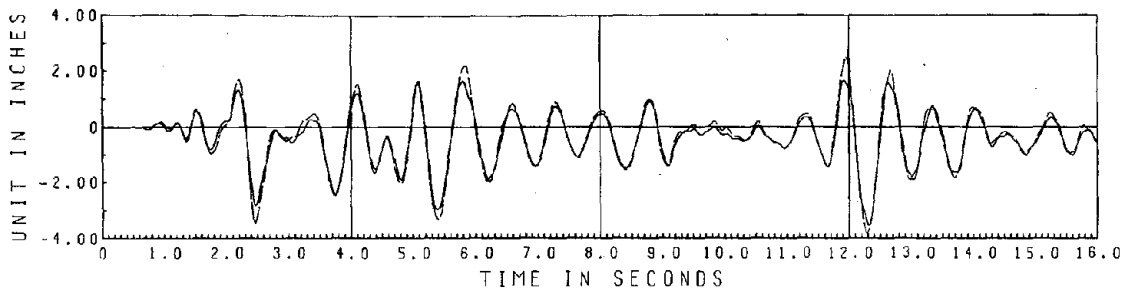
Damping = .00, .015, .03, .05 Critical

Fig. 5.1c.1 El Centro Span 950 Horizontal Table Motion

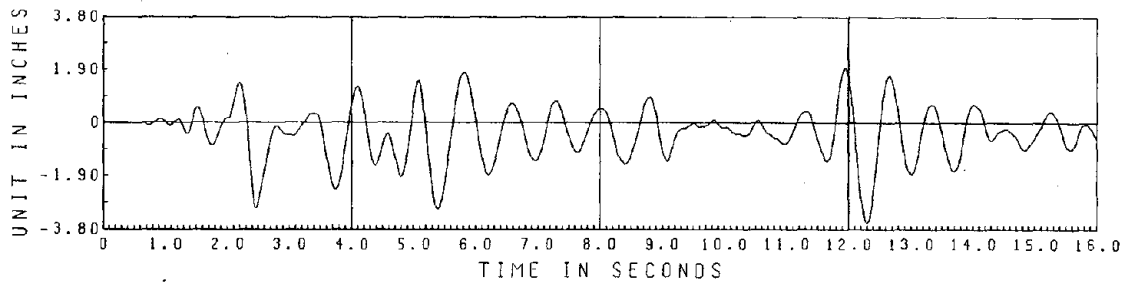


EL CENTRO SPAN 950
TEST RESULTS OF ROD BRACING

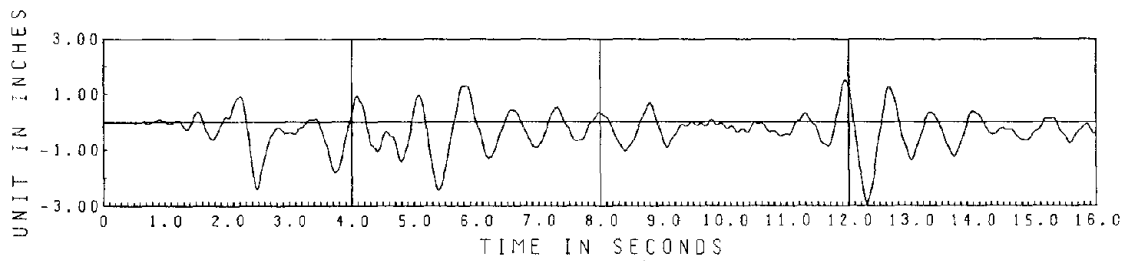
Fig. 5.1c.2 Table and Floor Accelerations



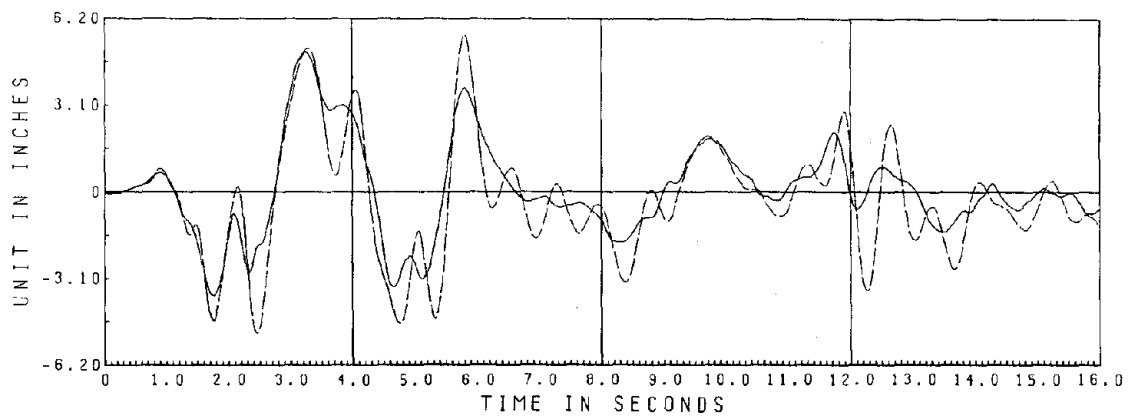
SOLID LINE=3RD FLOOR REL DISPL NA DASHED LINE=3RD FLOOR REL DISPL SA



2ND FL. DISPLACEMENT REL. TO TABLE



1ST FL. DISPLACEMENT REL. TO TABLE

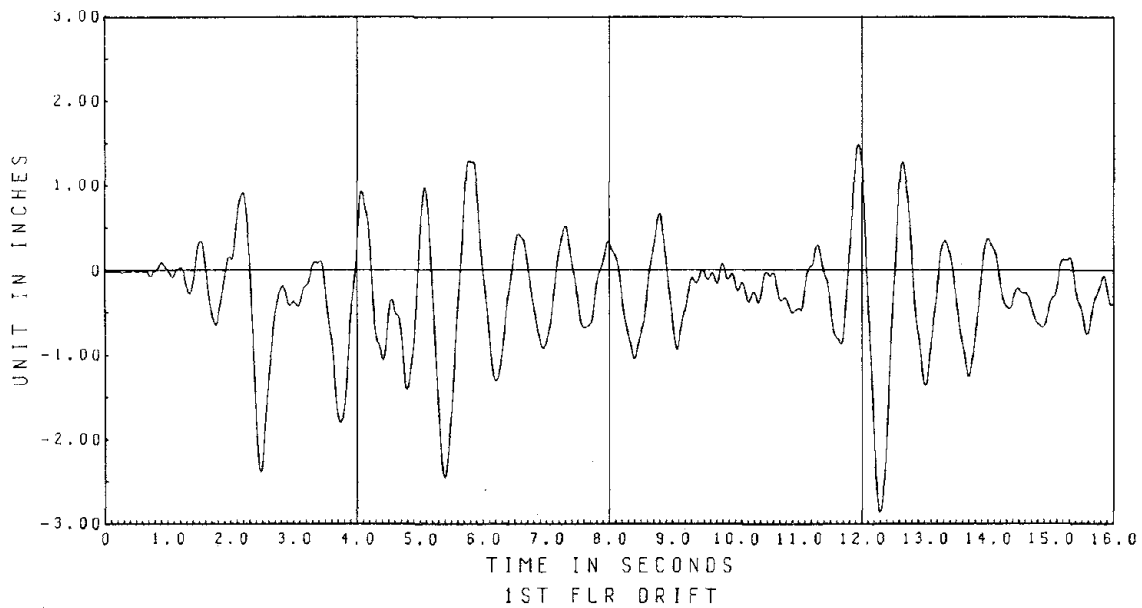
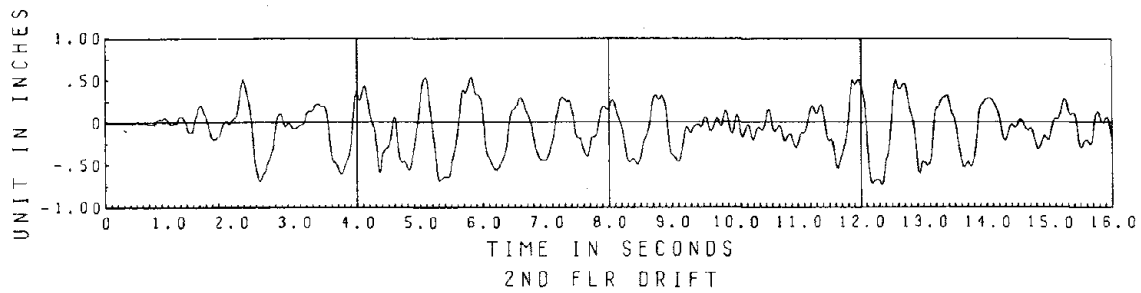
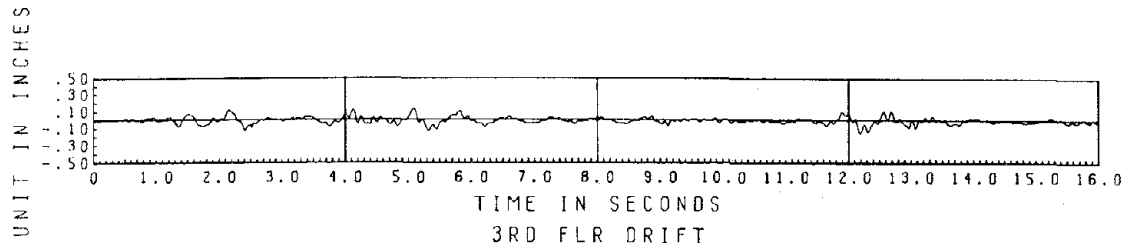


SOLID LINE=TABLE DISPL DASHED LINE=3RD FLOOR DISPL

EL CENTRO SPAN 950

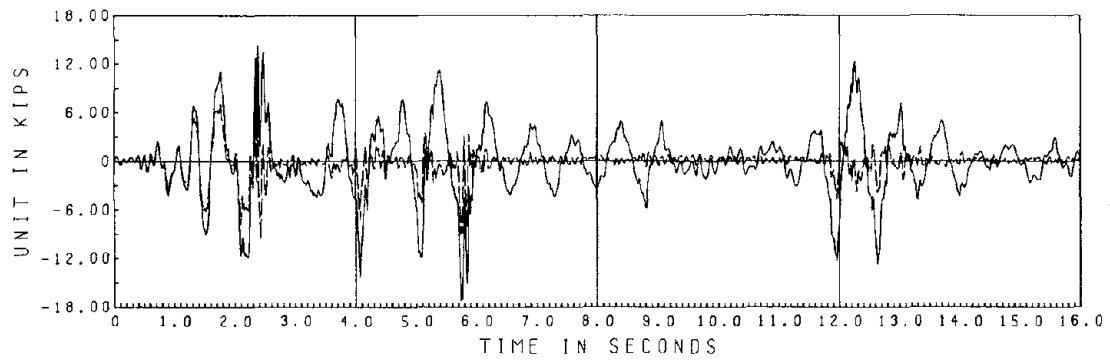
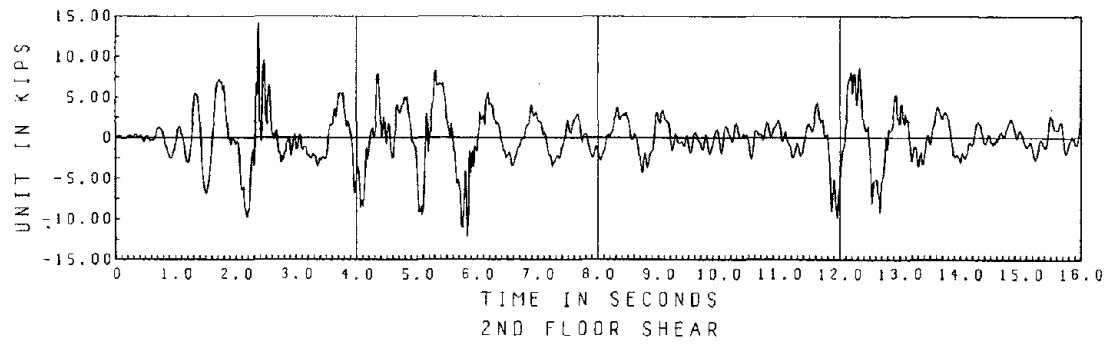
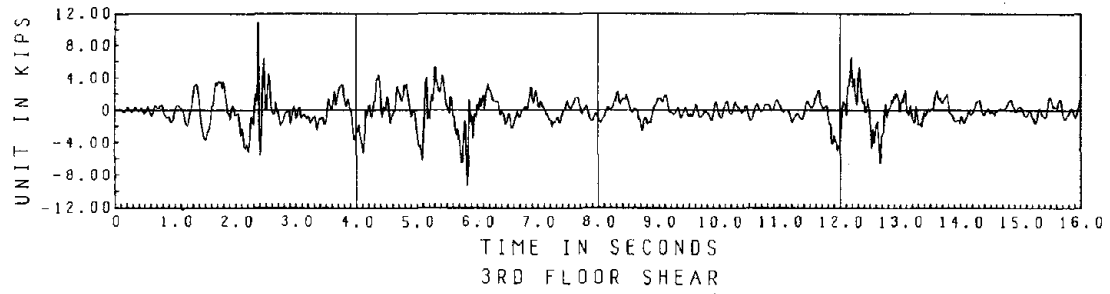
TEST RESULTS OF ROD BRACING

Fig. 5.1c.3 Table and Floor Displacements



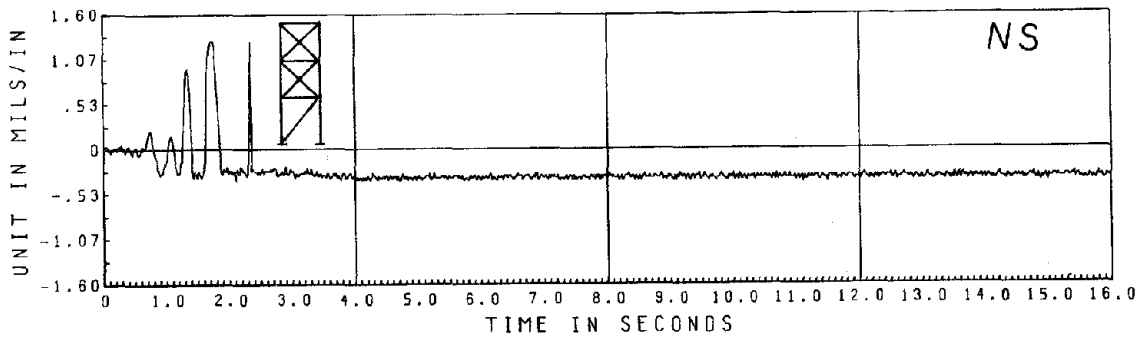
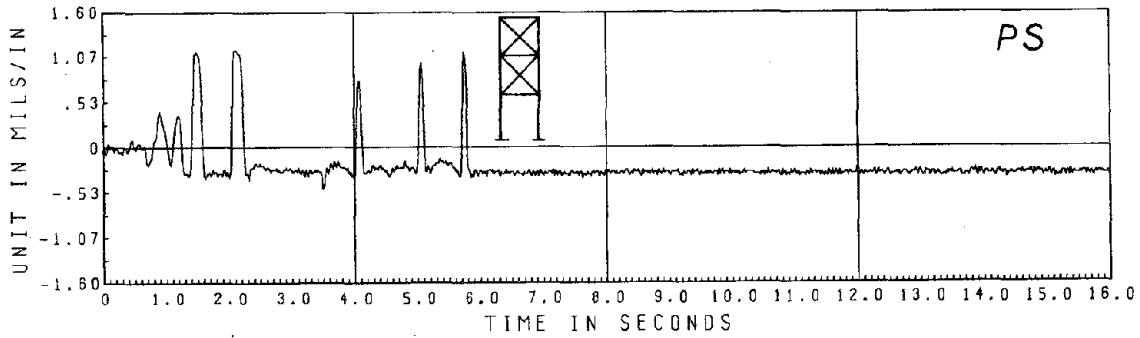
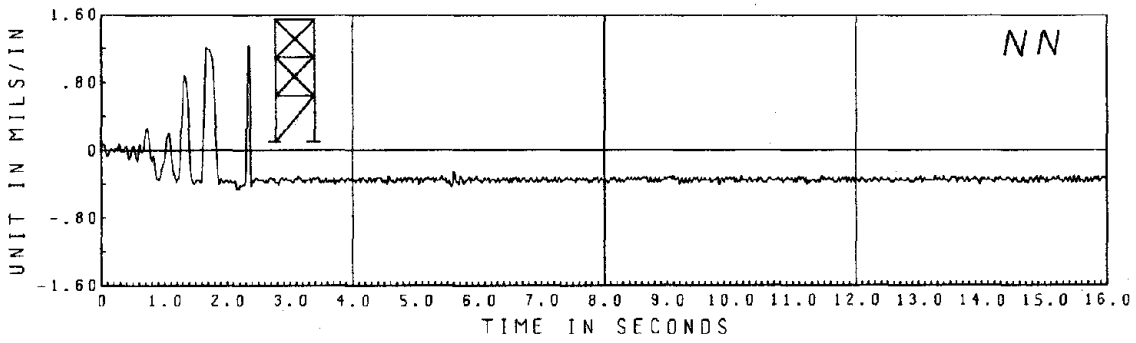
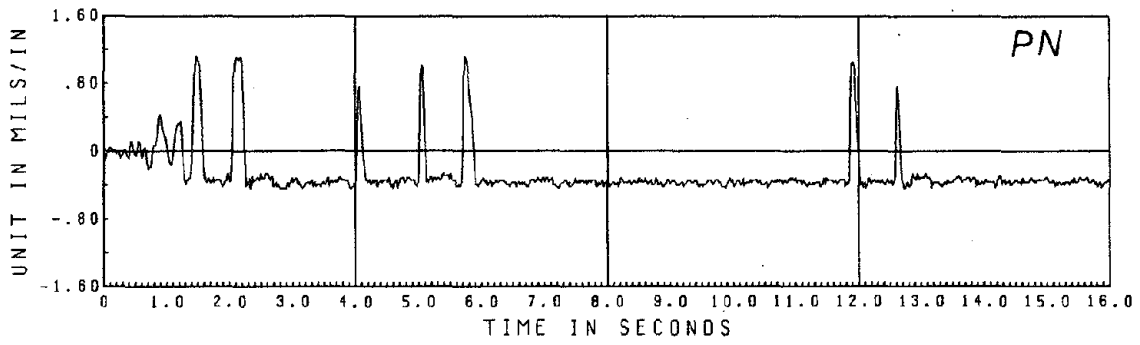
EL CENTRO SPAN 950
TEST RESULTS OF ROD BRACING

Fig. 5.1c.4 Floor Drifts



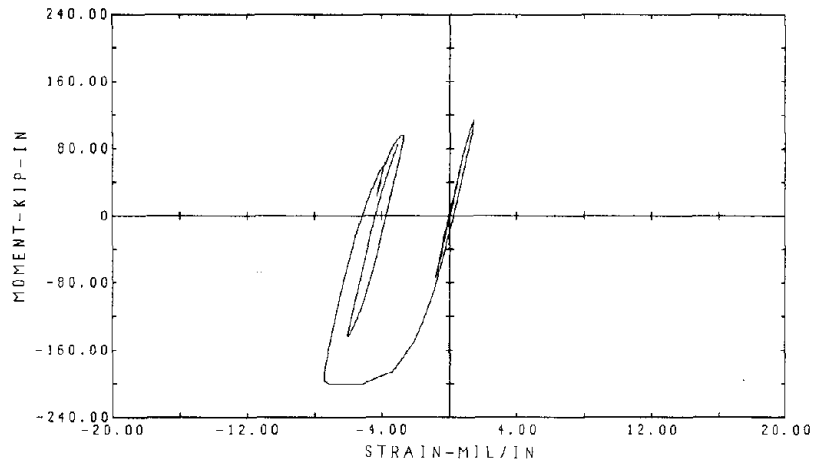
EL CENTRO SPAN 950
 TEST RESULTS OF ROD BRACING

Fig. 5.1c.5 Floor Shear Forces

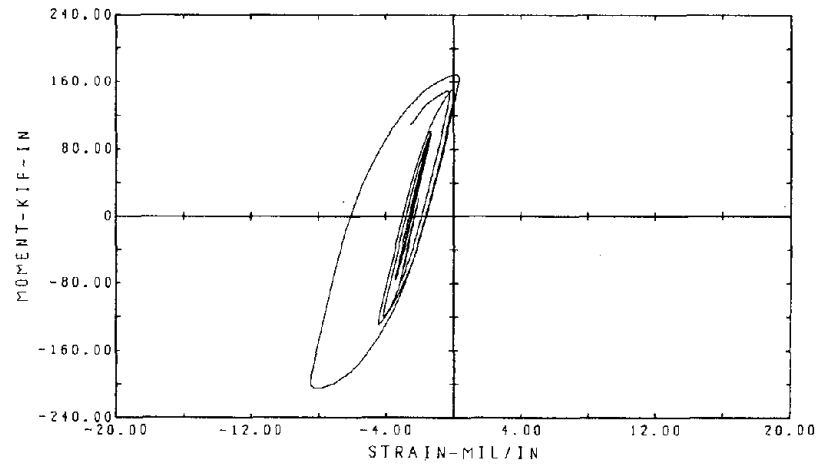


EL CENTRO SPAN 950
TEST RESULTS OF ROD BRACING

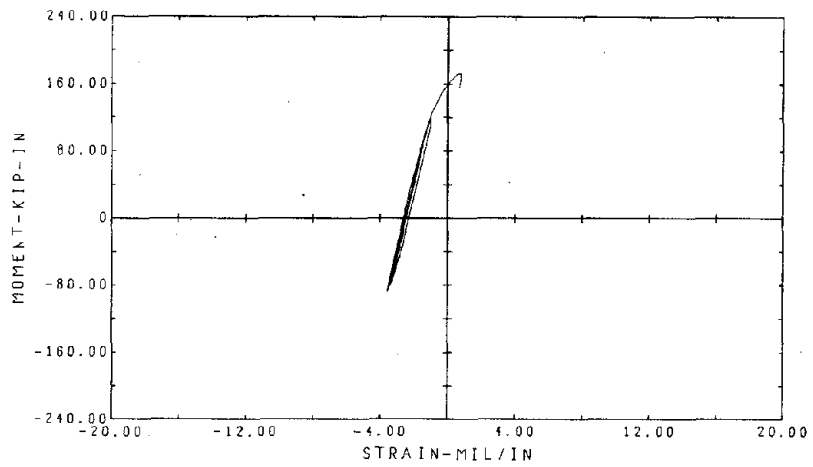
Fig. 5.1c.6 Strain Time-Histories of the 1st Floor Rod Braces



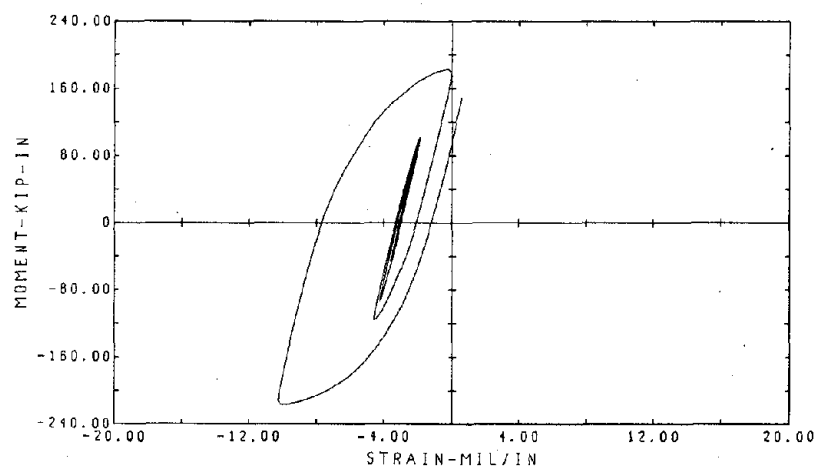
(0-4 SEC)



(4-8 SEC)

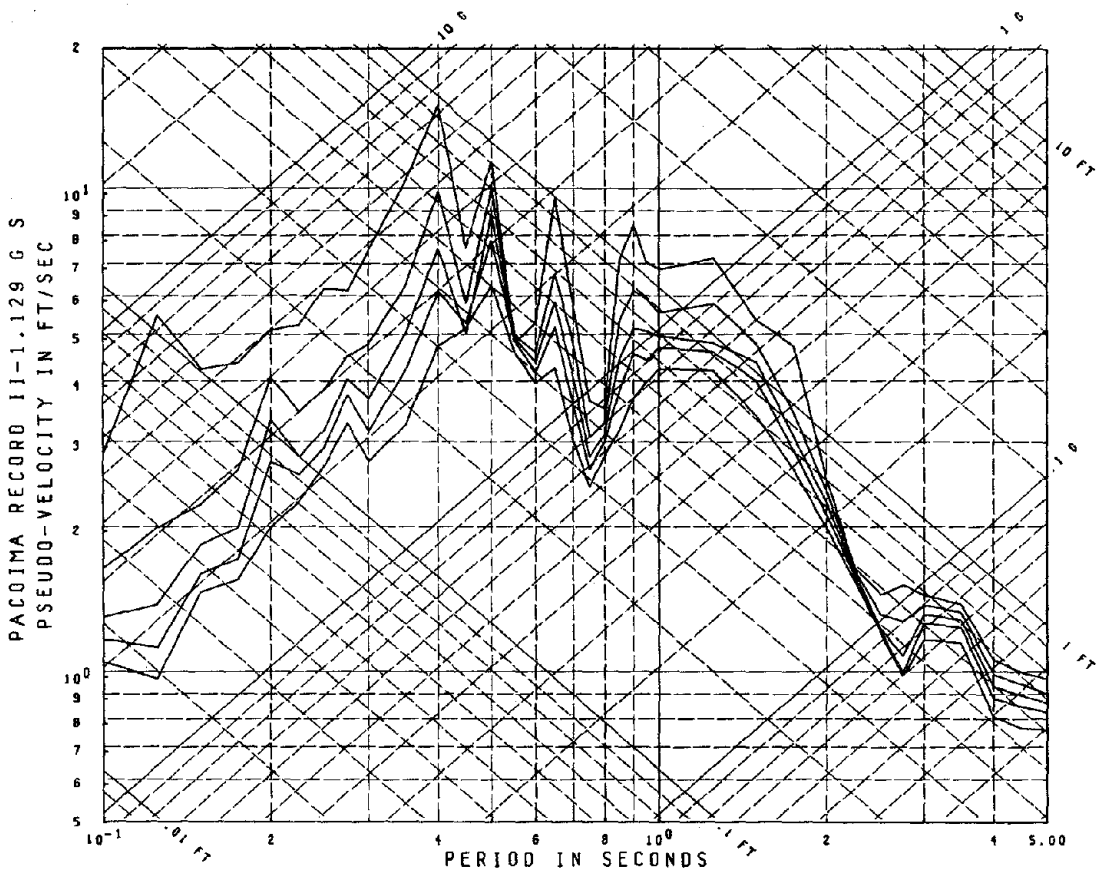
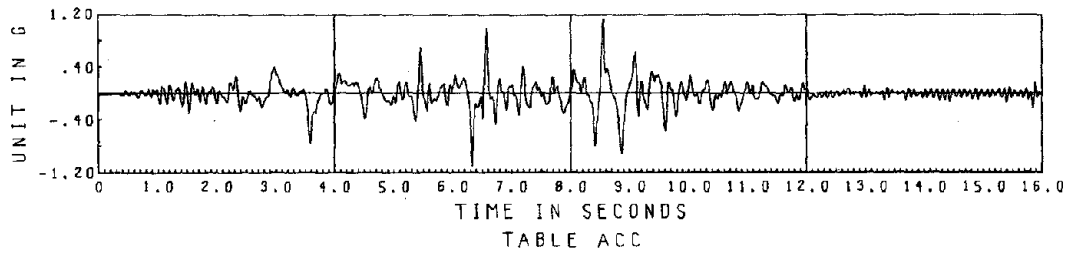
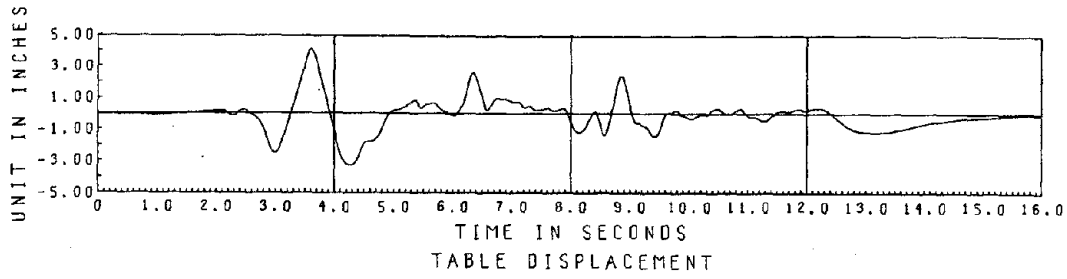


(8-12 SEC)



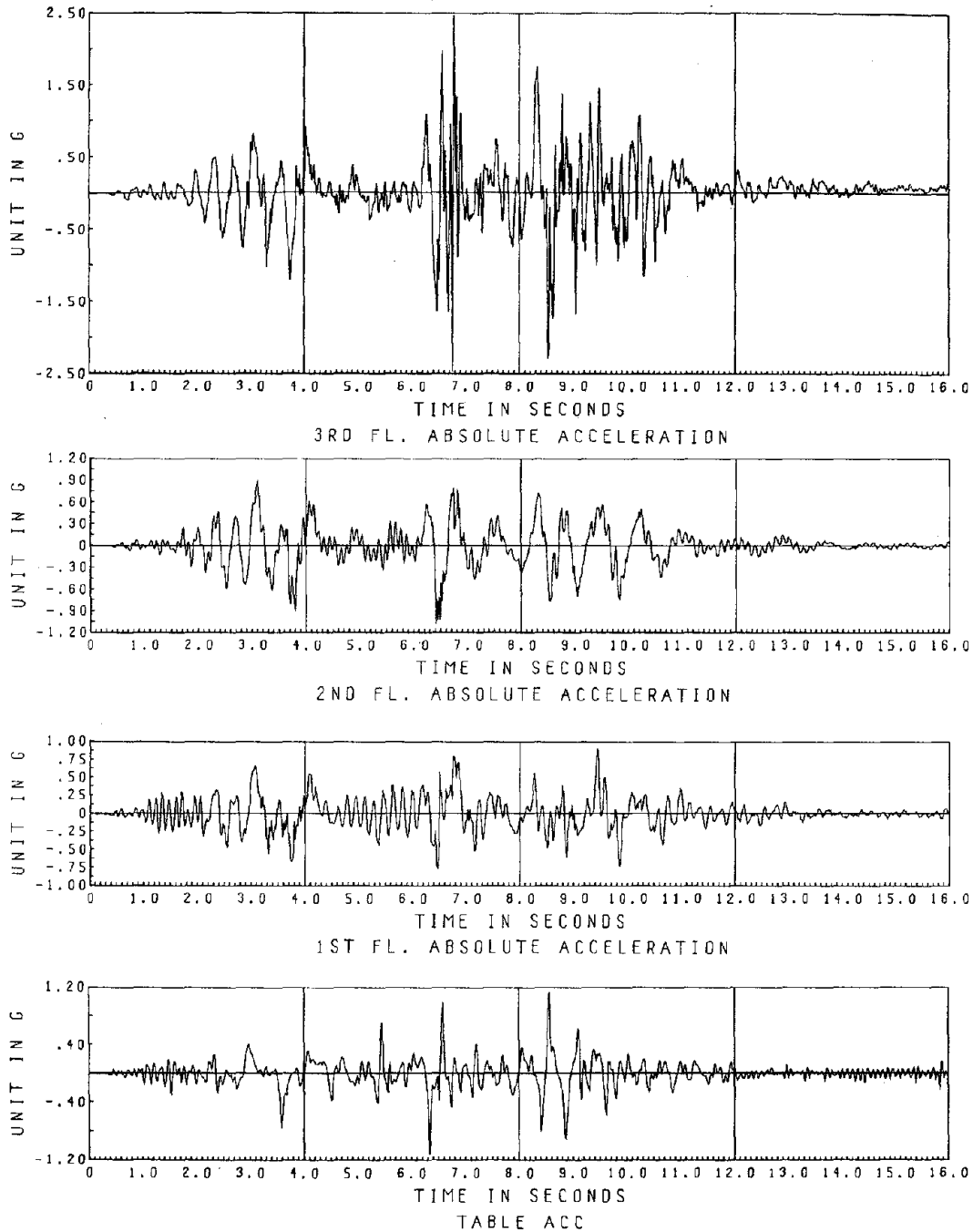
(12-16 SEC)

EL CENTRO SPAN 950 - TEST RESULTS OF ROD BRACING
 Fig. 5.1c.7 First Floor Column Moment-Strain Hysteresis Loops



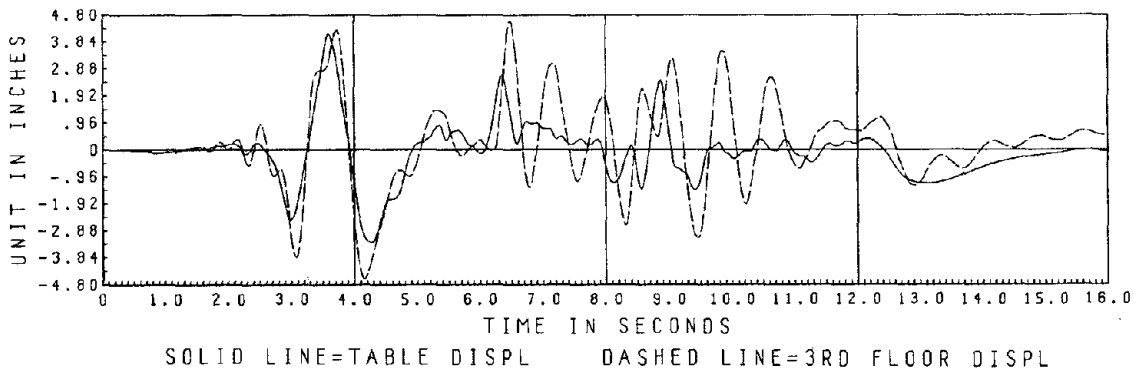
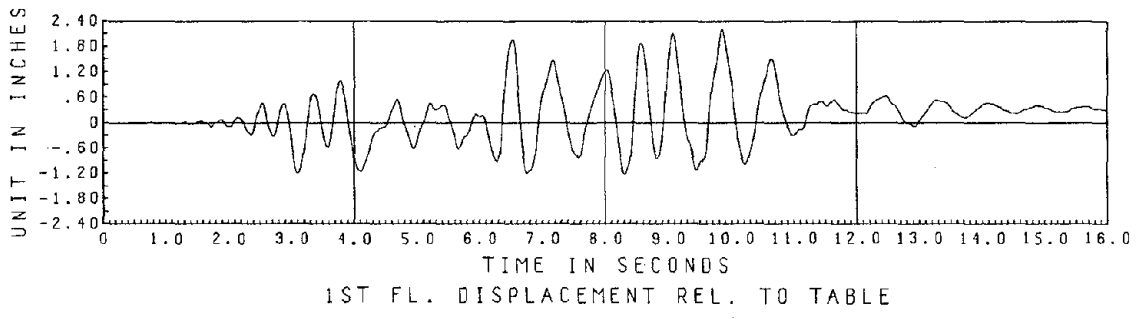
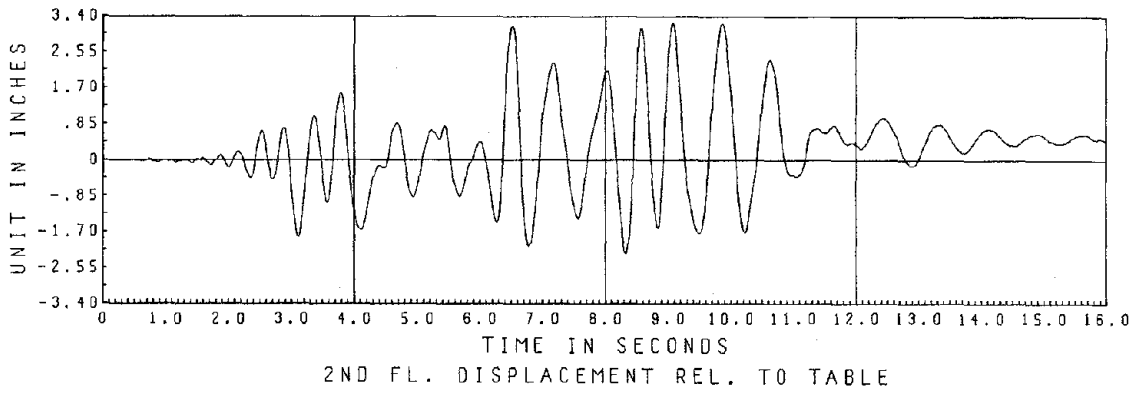
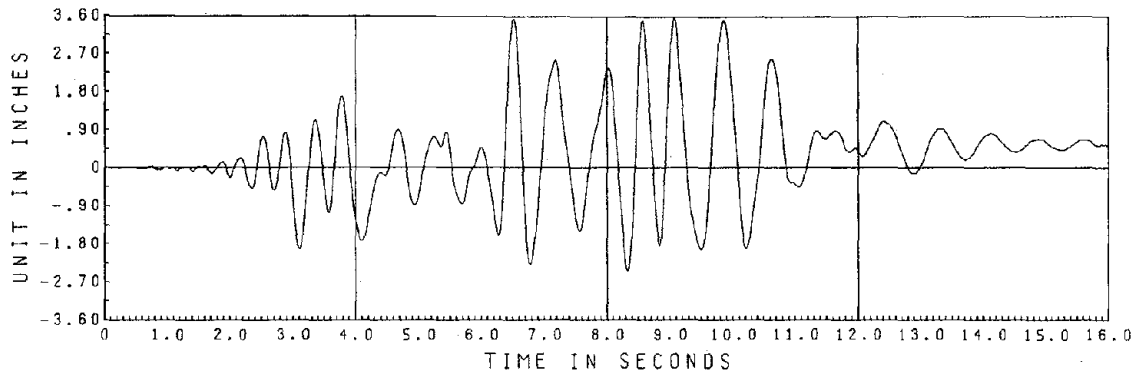
Damping = .00, .01, .02, .03, .05 Critical

Fig. 5.1d.1 Pacoima Record (1.129g) Horizontal Table Motion



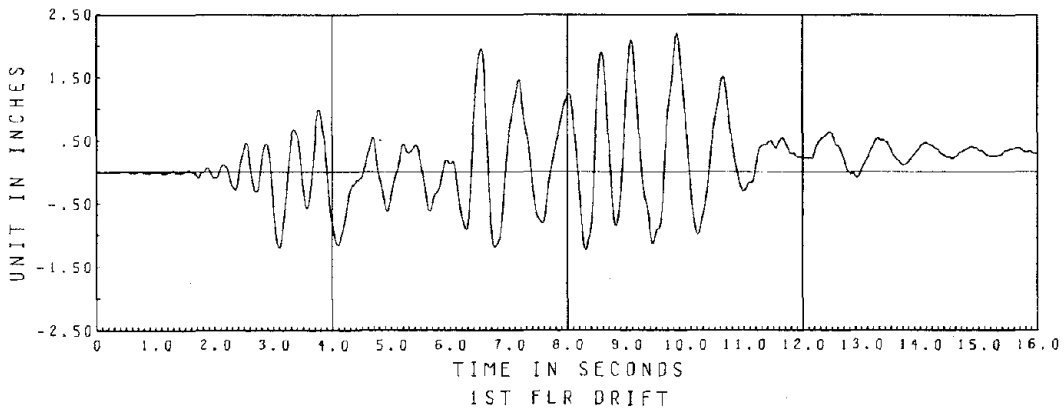
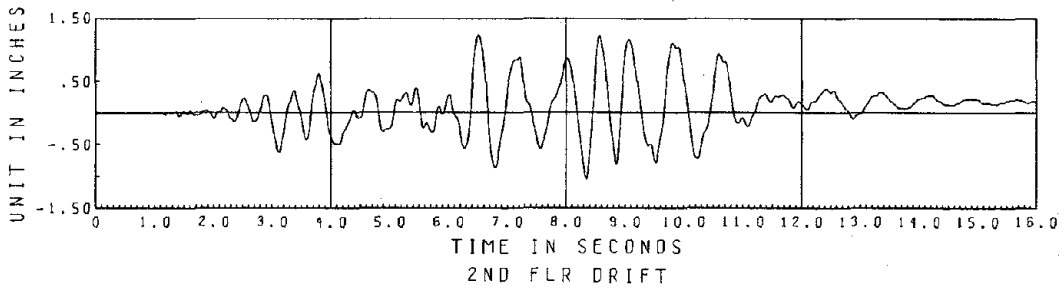
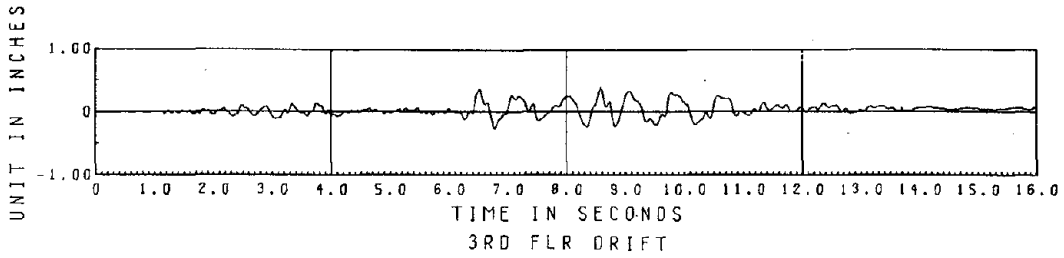
PACOIMA RECORD (1.129G)
 TEST RESULTS OF ROD BRACING

Fig. 5.1d.2 Table and Floor Accelerations



PACOIMA RECORD (1.129G)
TEST RESULTS OF ROD BRACING

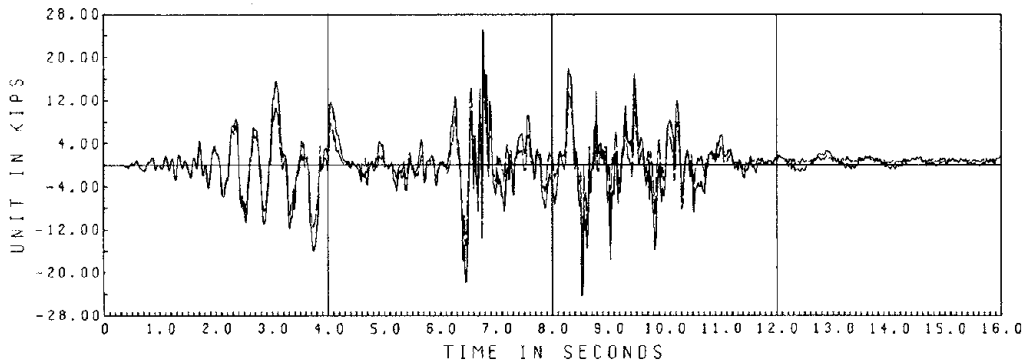
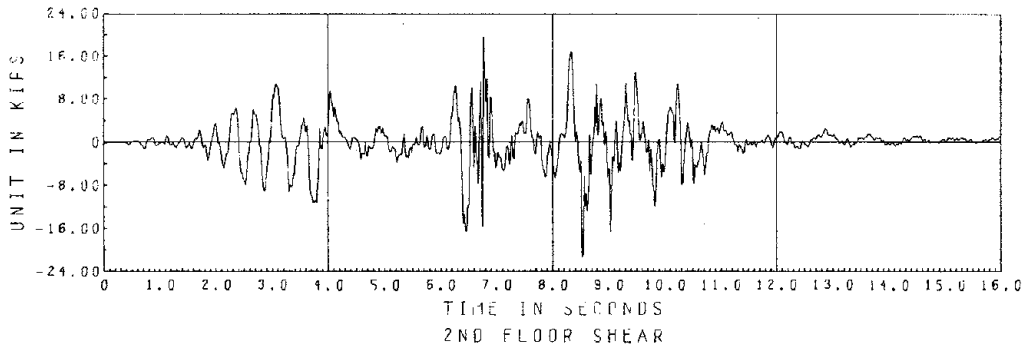
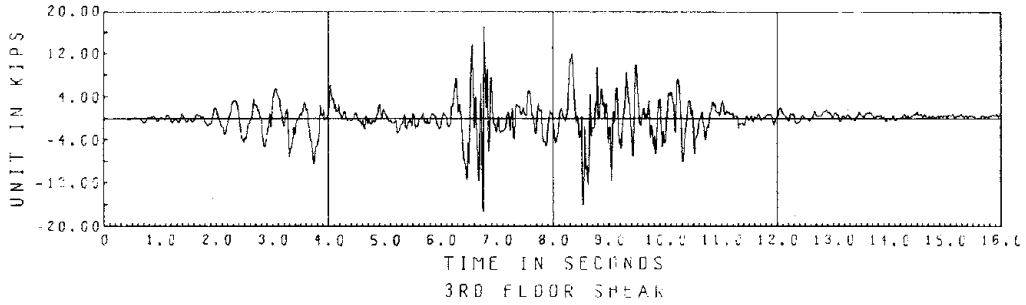
Fig. 5.1d.3 Table and Floor Displacements



PACOIMA RECORD (1.129G)

TEST RESULTS OF ROD BRACING

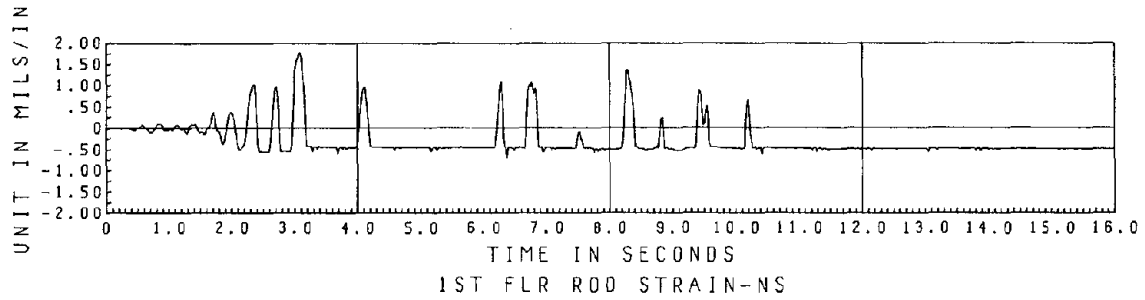
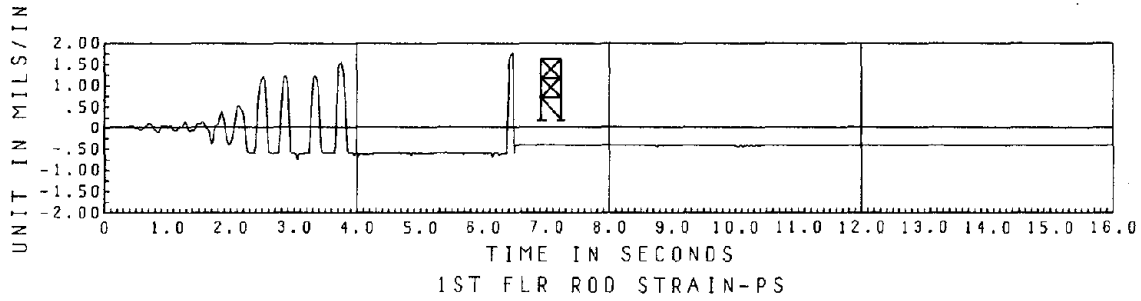
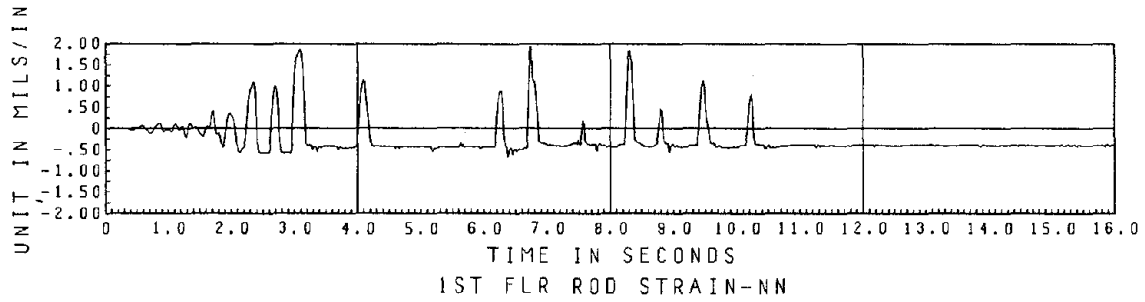
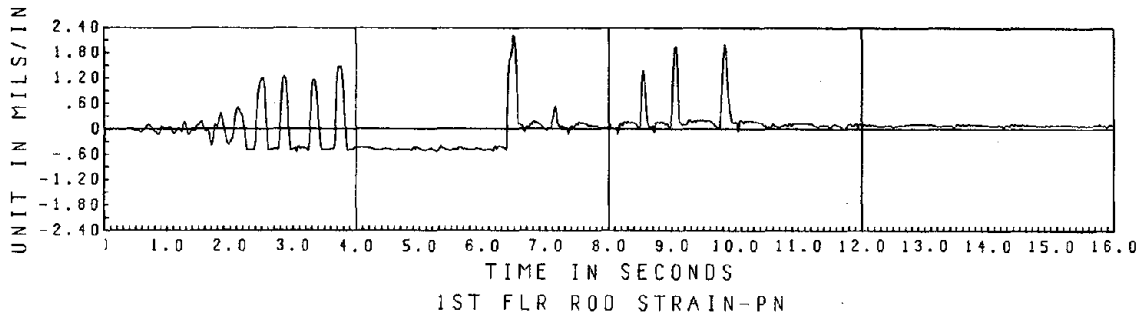
Fig. 5.1d.4 Floor Drifts



SOLID LINE=1ST FLOOR SHEAR DASHED LINE=1ST FLOOR BRACING DIAGONAL SHEAR

PACOIMA RECORD (1.129G)
 TEST RESULTS OF ROD BRACING

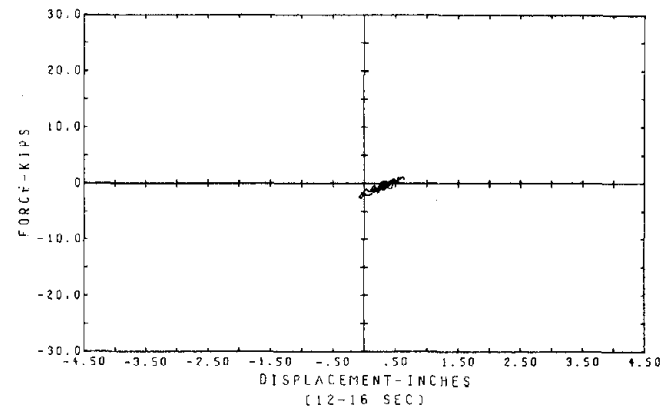
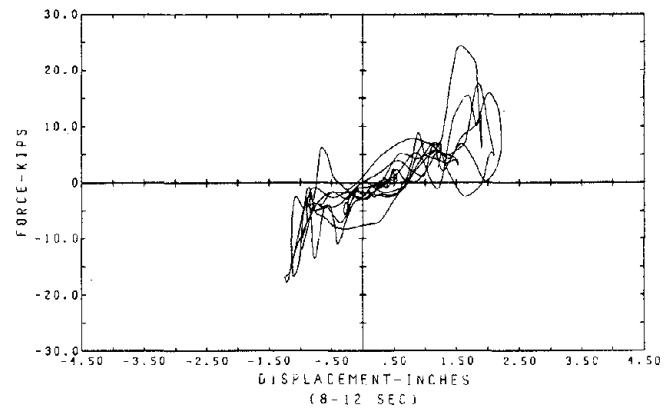
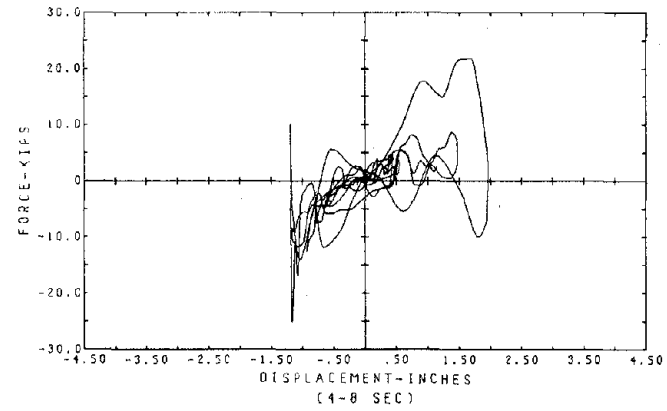
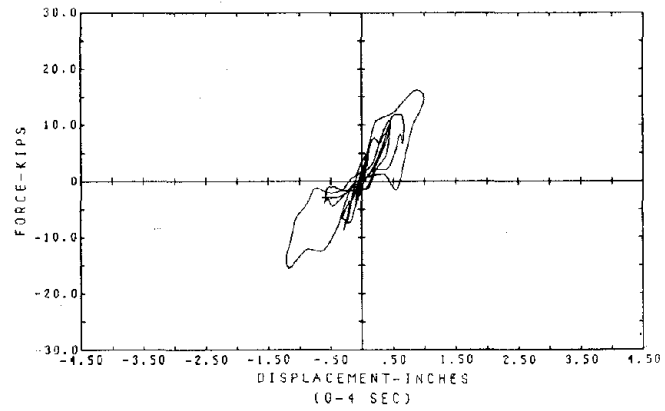
Fig. 5.1d.5 Floor Shear Forces



PACOIMA RECORD (1.129G)

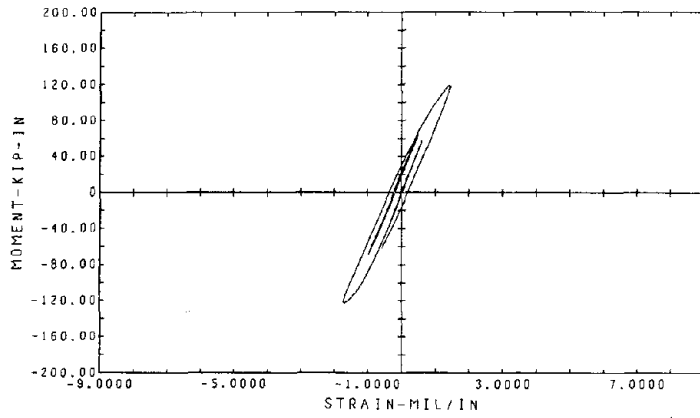
TEST RESULTS OF ROD BRACING

Fig. 5.1d.6 Strain Time-Histories of the 1st Floor Rod Braces

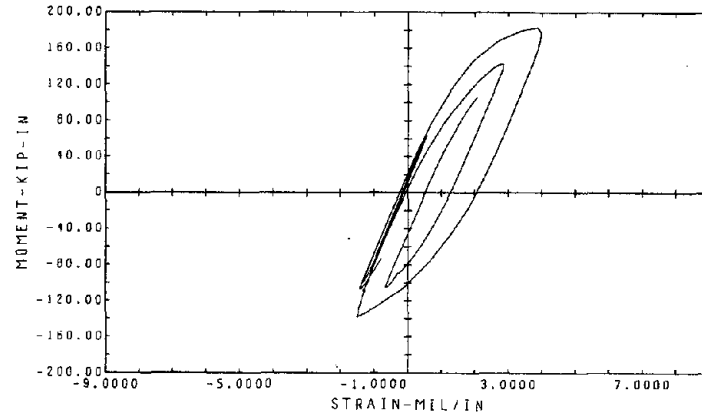


1ST FLR SHEAR VS. DISPLACEMENT
 REFERENCE FRAME N
 ROD DIAGONALLY BRACED STRUCTURE
 PACOIMA-1.129 G.S

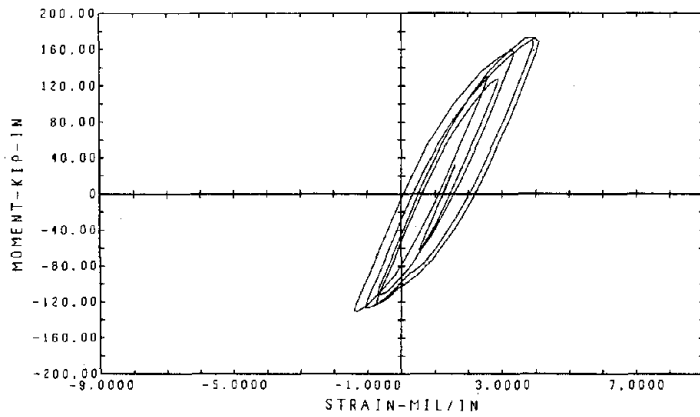
Fig. 5.1d.7 First Floor Shear-Displacement Hysteresis Loops



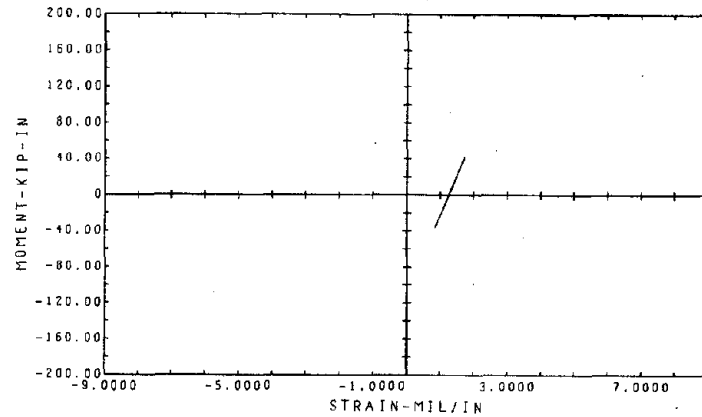
(0-4 SEC)



(4-8 SEC)



(8-12 SEC)

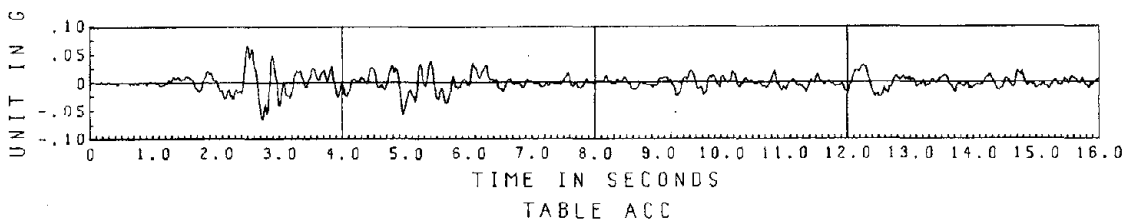
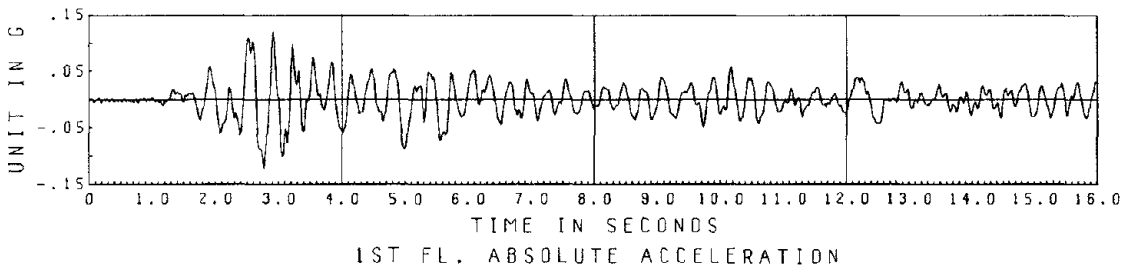
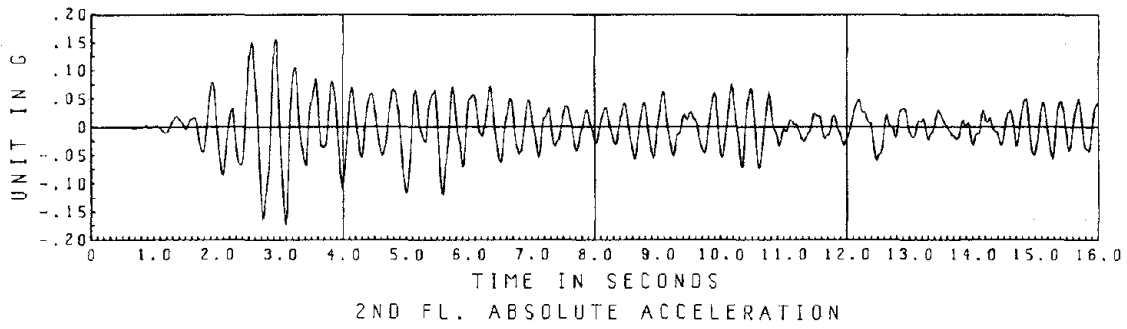
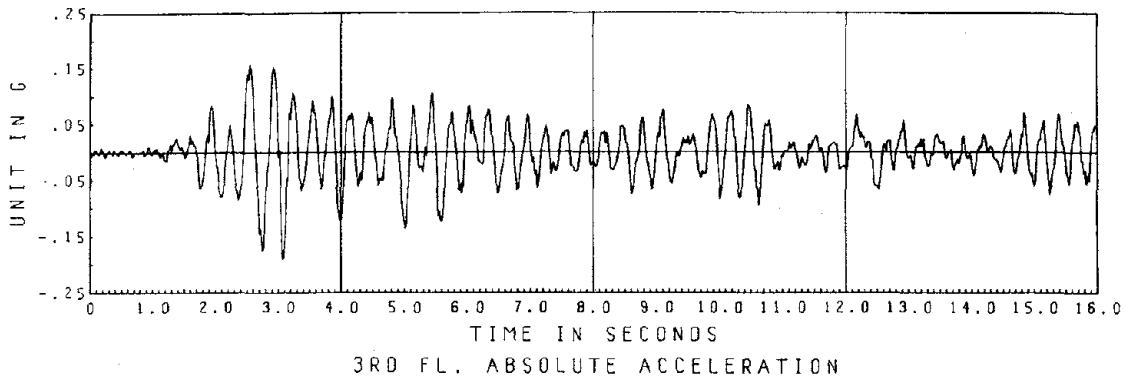


(12-16 SEC)

ROD DIAGONALLY BRACED STRUCTURE

1ST FLR. COL BOTTOM END, GAGES AT 1 IN. ABOVE STIFF. PLATE
REFERENCE COLUMN LINE NA

Fig. 5.1d.8 First Floor Column Moment-Strain Hysteresis Loops



EL CENTRO SPAN 100

TEST RESULTS OF PIPE BRACING

Fig. 5.2a.1 Table and Floor Accelerations

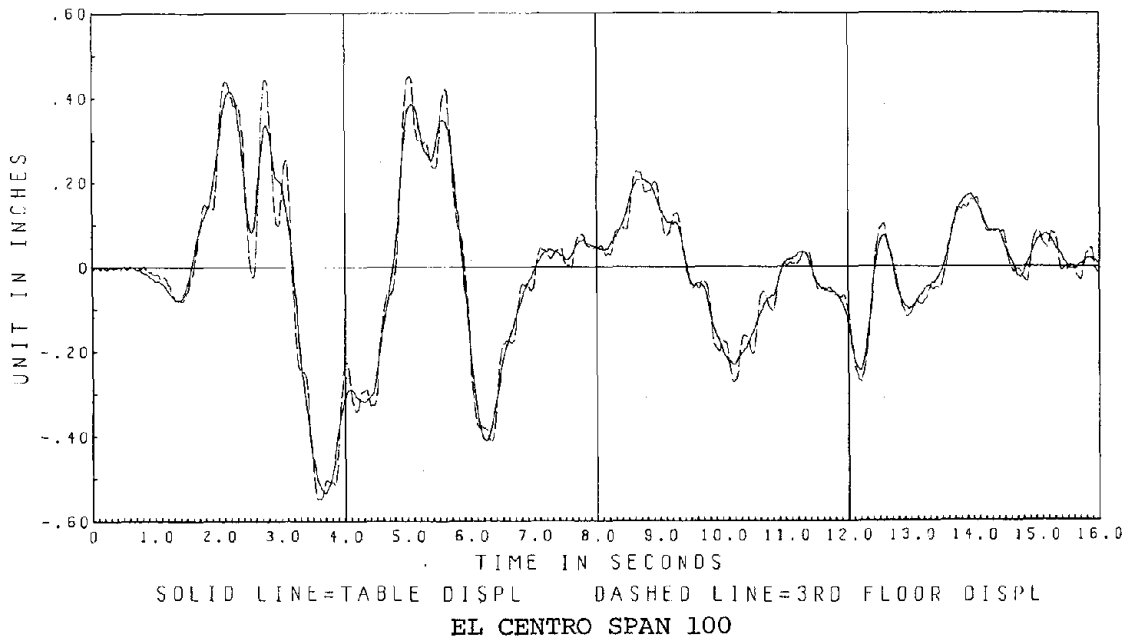
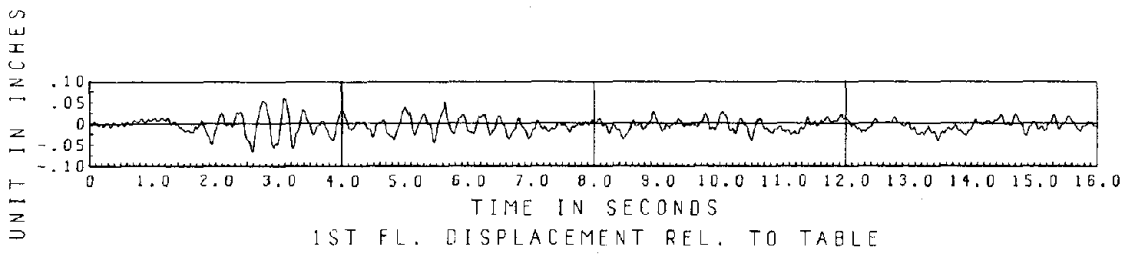
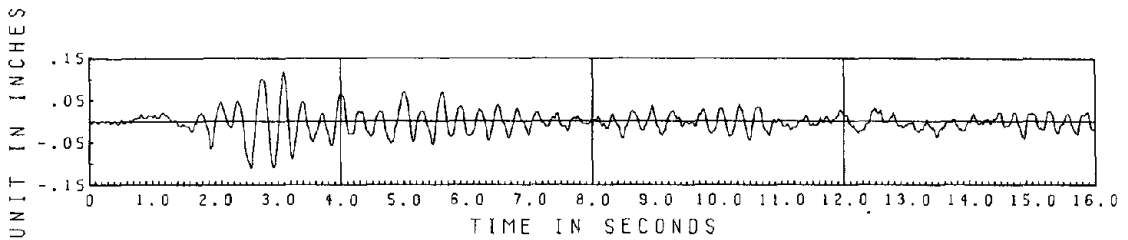
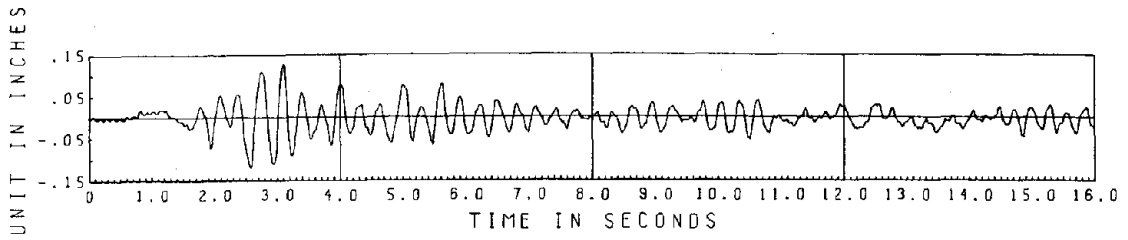
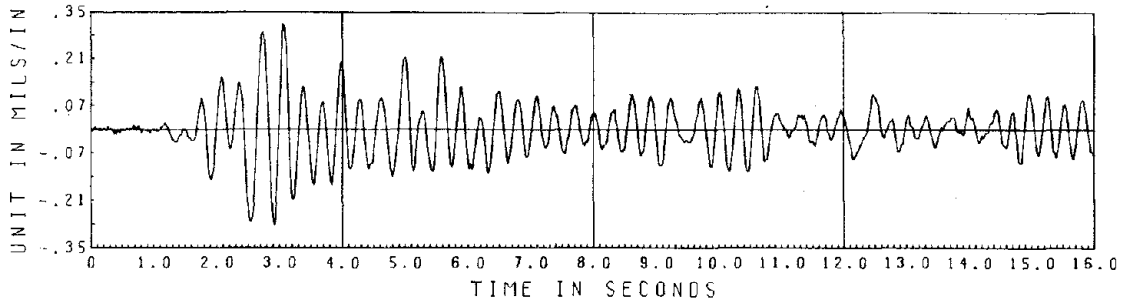
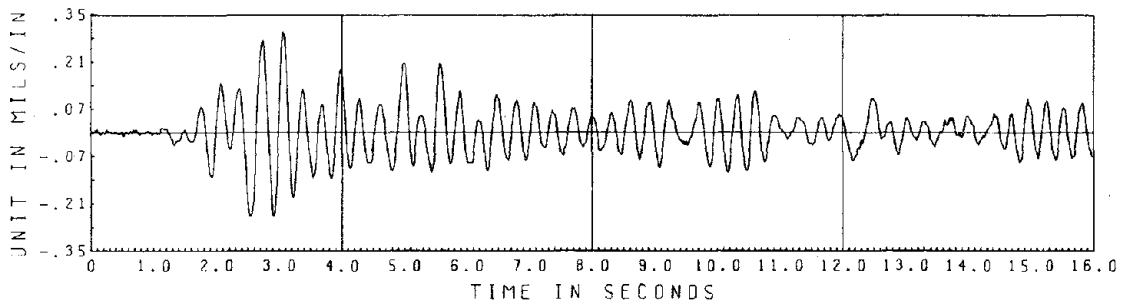


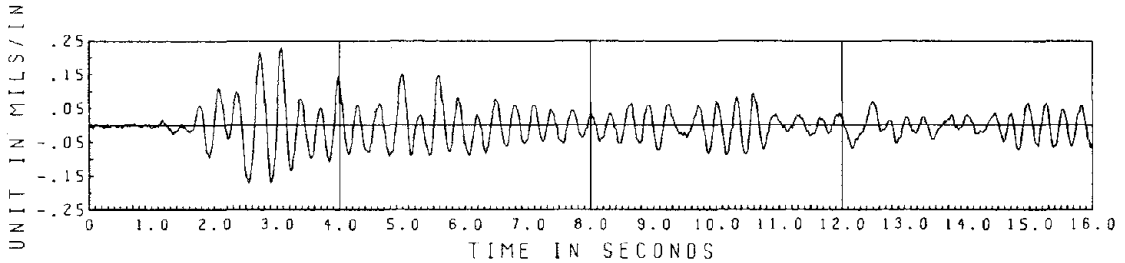
Fig. 5.2a.2 Table and Floor Displacements



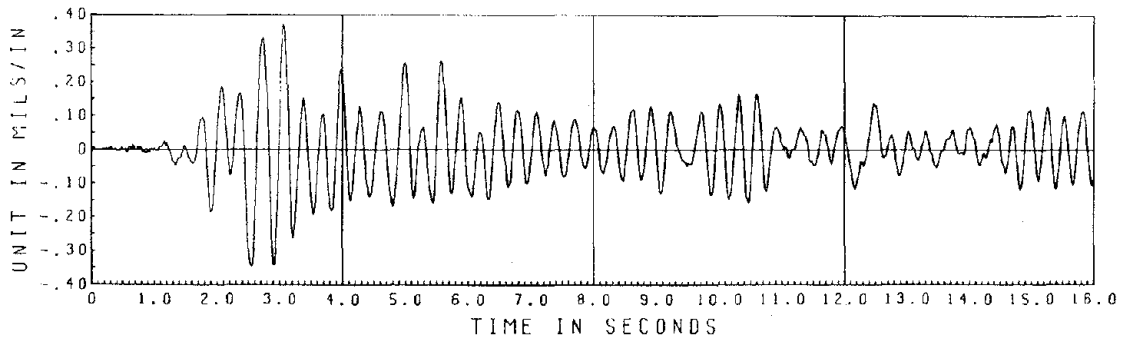
1ST FLR PIPE X-BR. STRAIN-PTN



1ST FLR PIPE X-BR. STRAIN-PBN



1ST FLR PIPE X-BR. STRAIN-PON

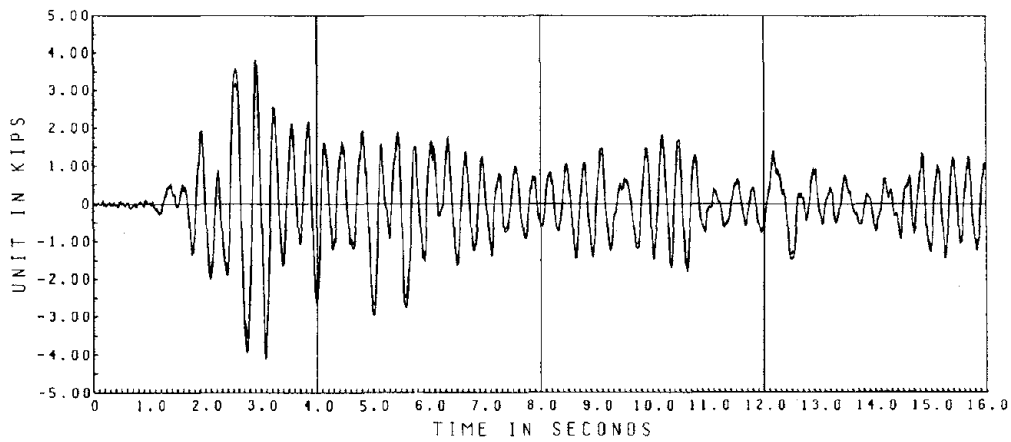
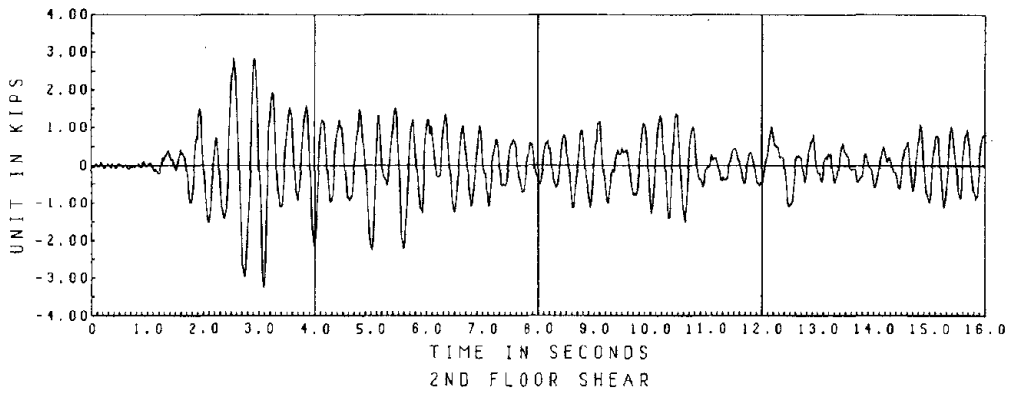
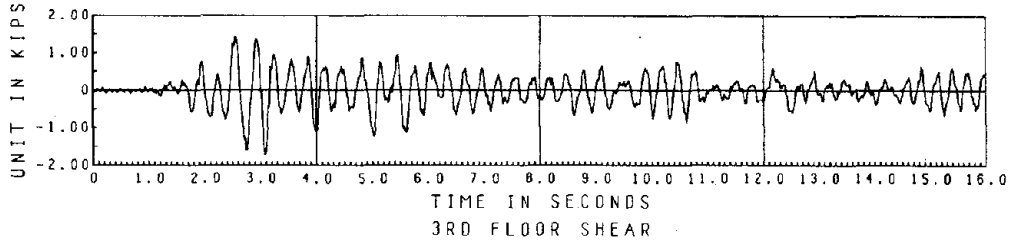


1ST FLR PIPE X-BR. STRAIN-PIN

EL CENTRO SPAN 100

TEST RESULTS OF PIPE BRACING

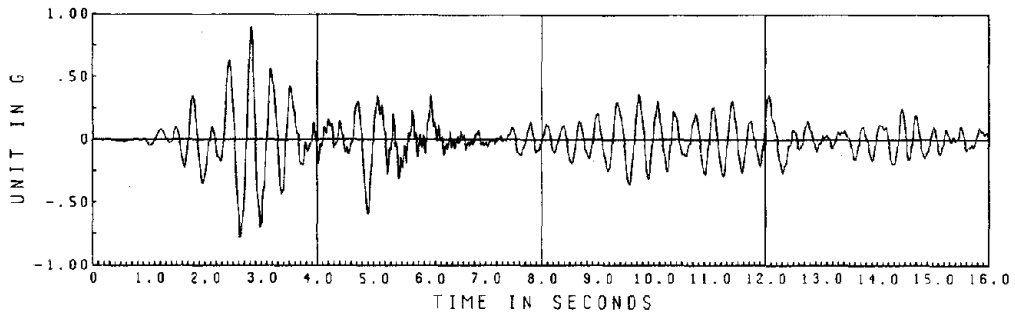
Fig. 5.2a.3 Strain Time-Histories of the 1st Floor Pipe Braces



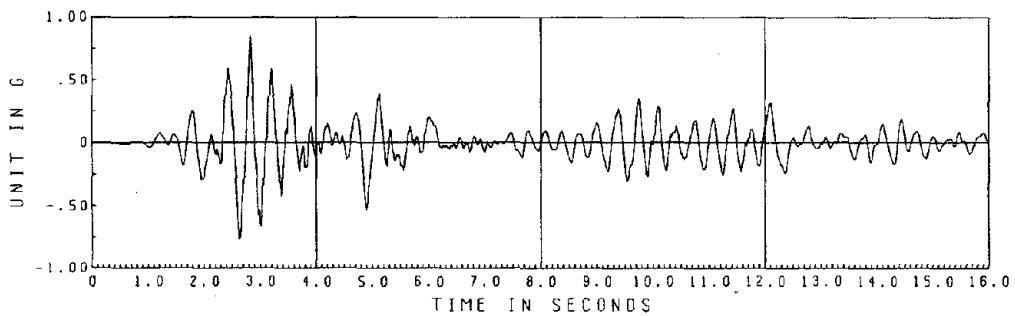
SOLID LINE=1ST FLOOR SHEAR DASHED LINE=1ST FLOOR BRACING DIAGONAL SHEAR

EL CENTRO SPAN 100
TEST RESULTS OF PIPE BRACING

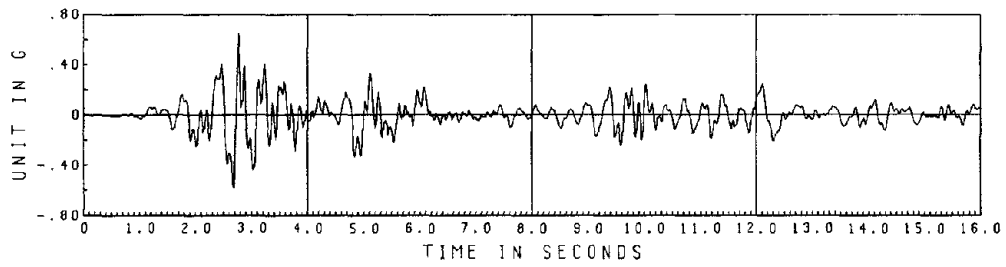
Fig. 5.2a.4 Floor Shear Forces



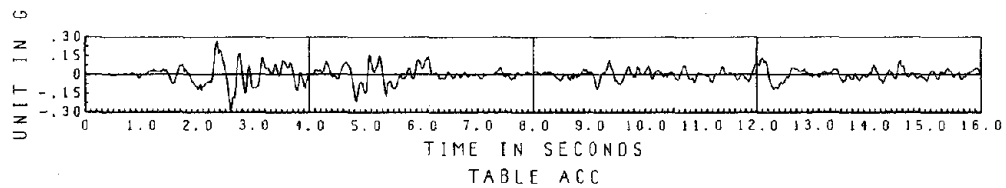
3RD FL. ABSOLUTE ACCELERATION



2ND FL. ABSOLUTE ACCELERATION

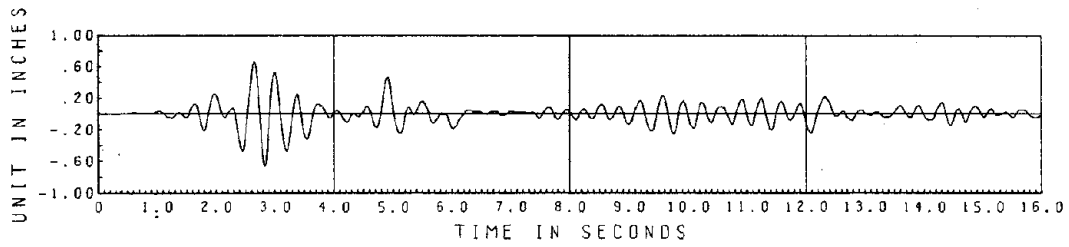


1ST FL. ABSOLUTE ACCELERATION

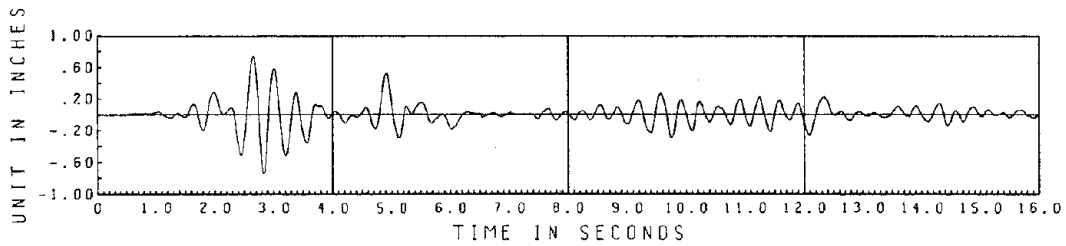


EL CENTRO SPAN 400
TEST RESULTS OF PIPE BRACING

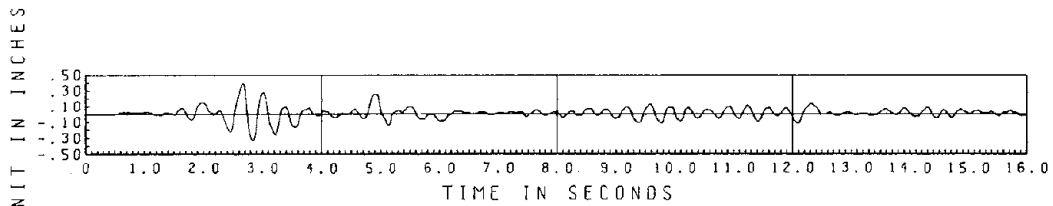
Fig. 5.2b.1 Table and Floor Accelerations



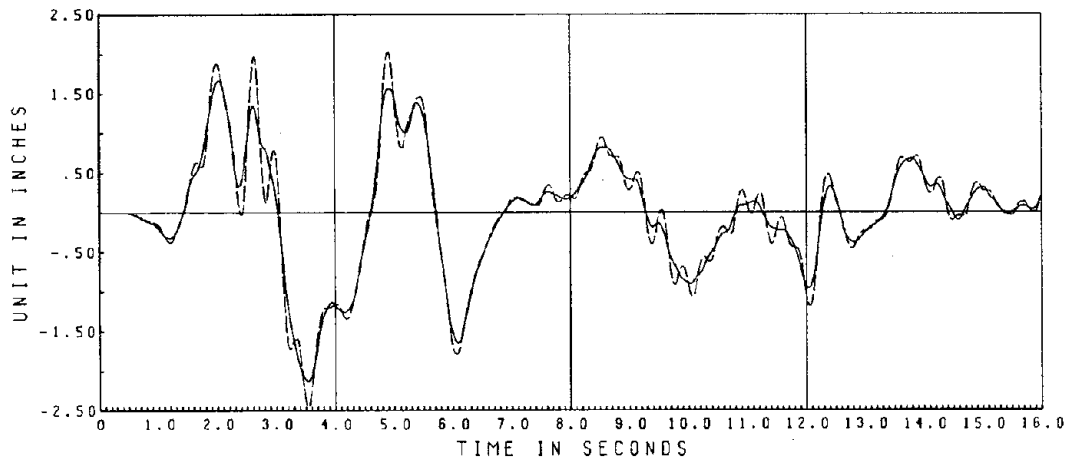
3RD FL. DISPLACEMENT REL. TO TABLE



2ND FL. DISPLACEMENT REL. TO TABLE



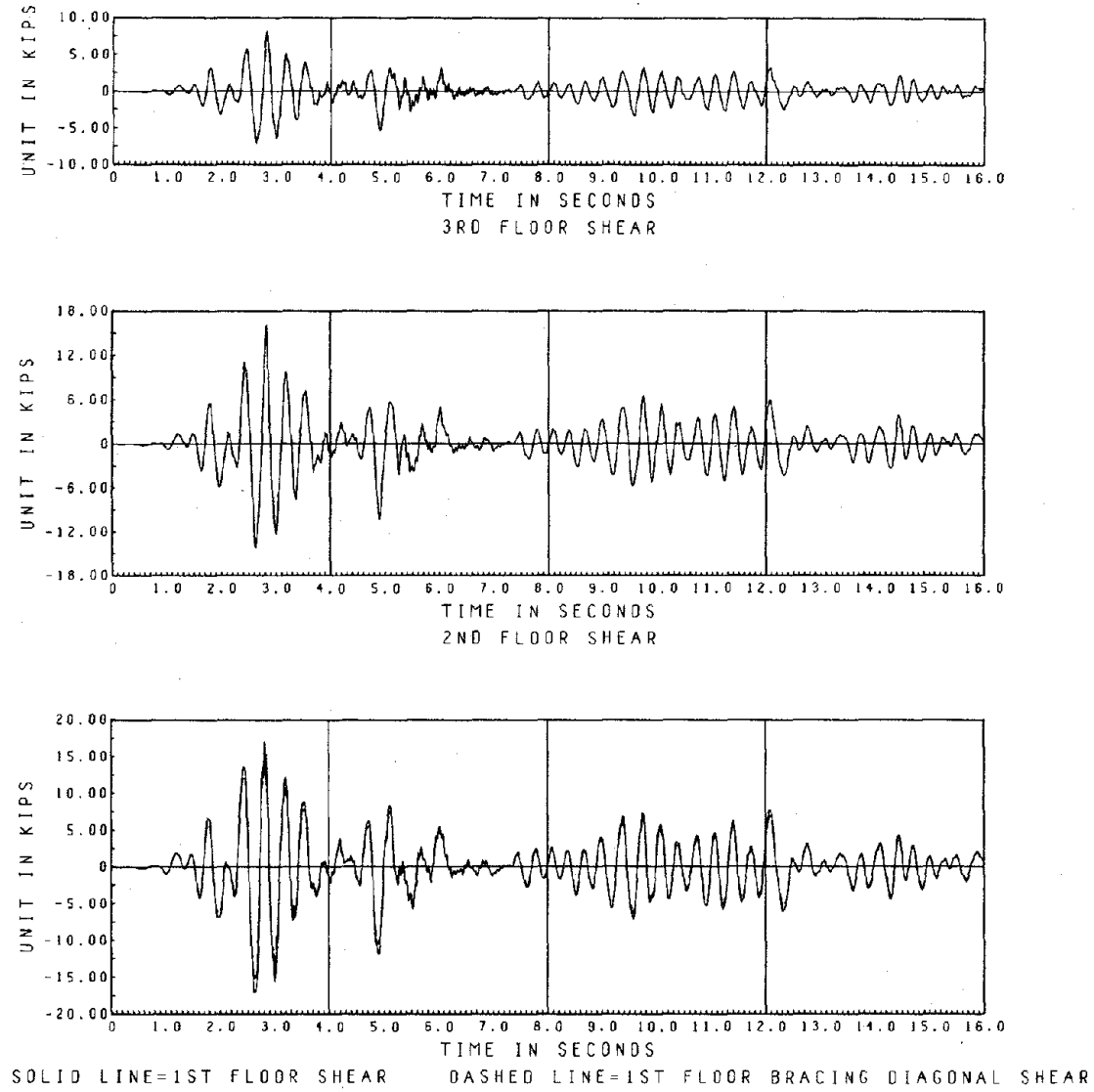
1ST FL. DISPLACEMENT REL. TO TABLE



SOLID LINE=TABLE DISPL DASHED LINE=3RD FLOOR DISPL

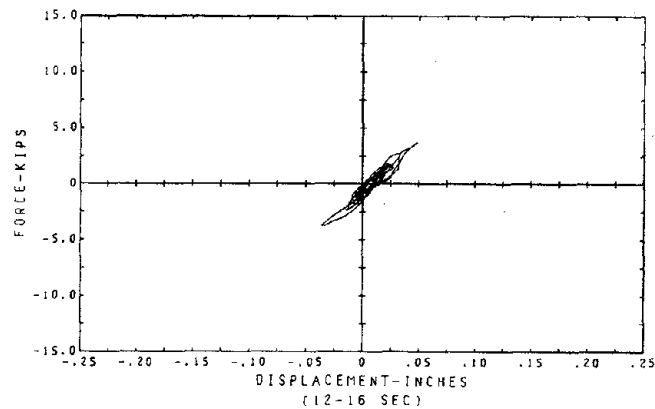
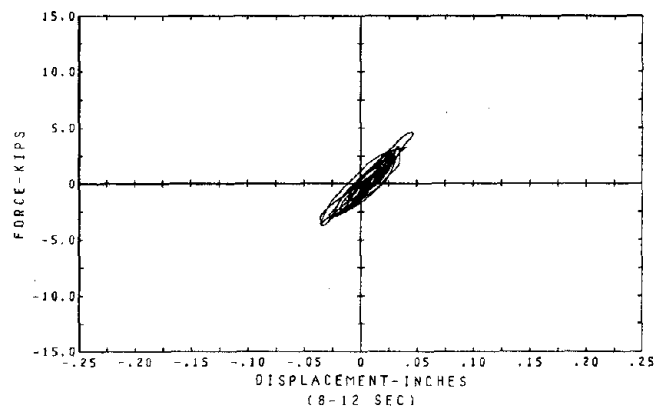
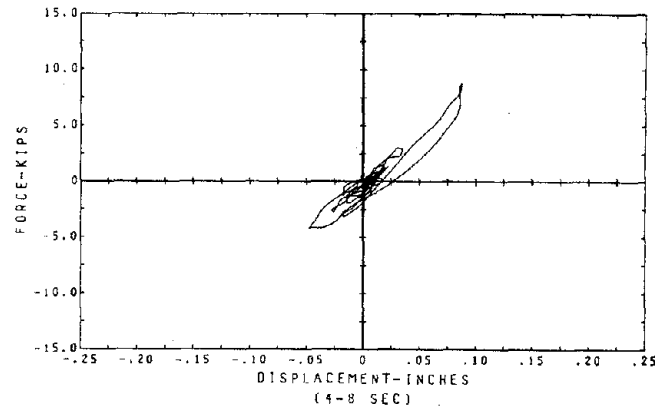
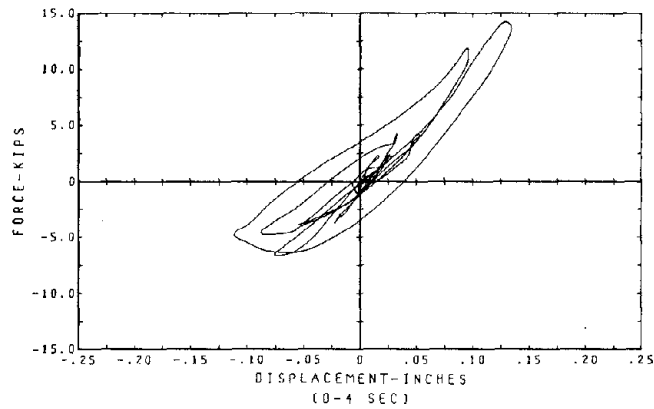
EL CENTRO SPAN 400
TEST RESULTS OF PIPE BRACING

Fig. 5.2b.2 Table and Floor Displacements



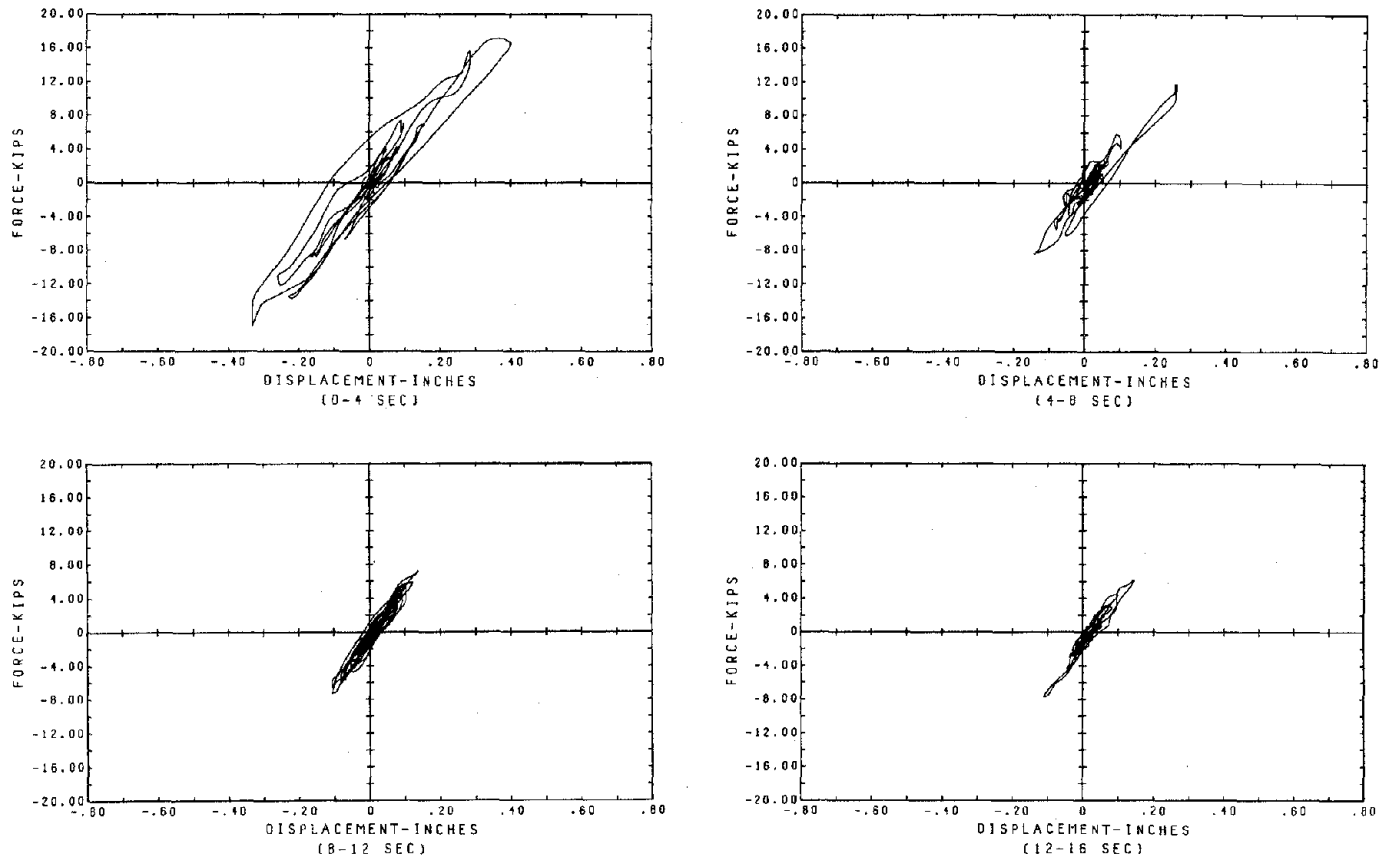
EL CENTRO SPAN 400
TEST RESULTS OF PIPE BRACING

Fig. 5.2b.3 Floor Shear Forces



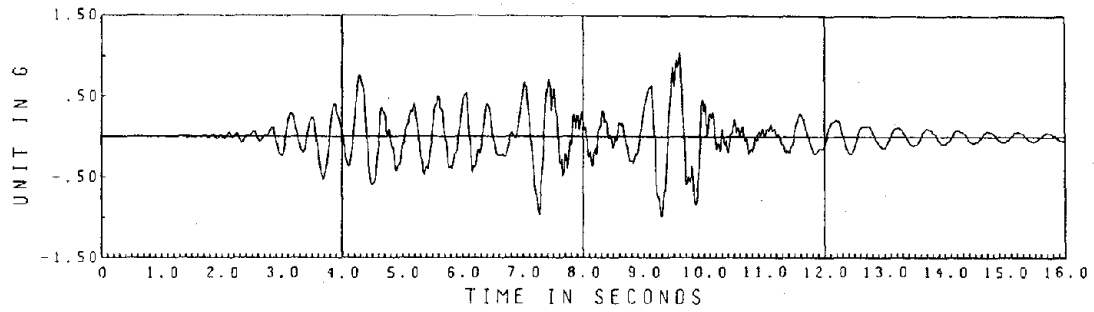
1ST FL. PIPE FORCE VS. DISPLACEMENT
 REFERENCE FRAME N
 PIPE DIAGONALLY BRACED STRUCTURE
 EL CENTRO SPAN 100

Fig. 5.2b.4 First Floor Pipe Force-Displacement Hysteresis Loops

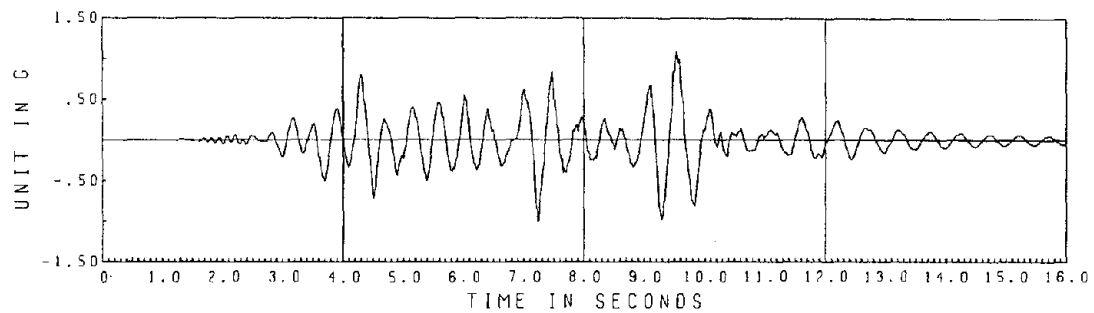


1ST FLR SHEAR VS. DISPLACEMENT
 REFERENCE FRAME N
 PIPE DIAGONALLY BRACED STRUCTURE
 EL CENTRO SPAN 400

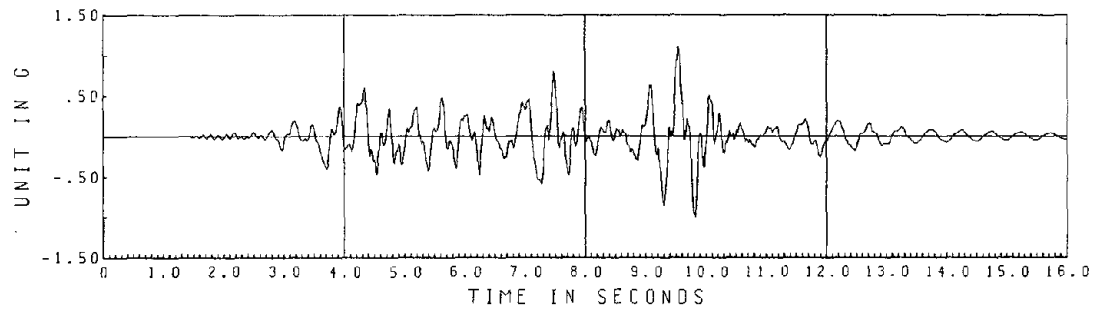
Fig. 5.2b.5 First Floor Shear-Displacement Hysteresis Loops



3RD FL. ABSOLUTE ACCELERATION



2ND FL. ABSOLUTE ACCELERATION



1ST FL. ABSOLUTE ACCELERATION

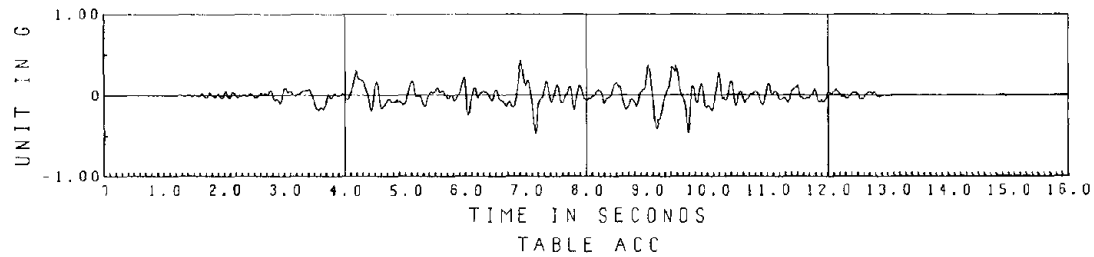
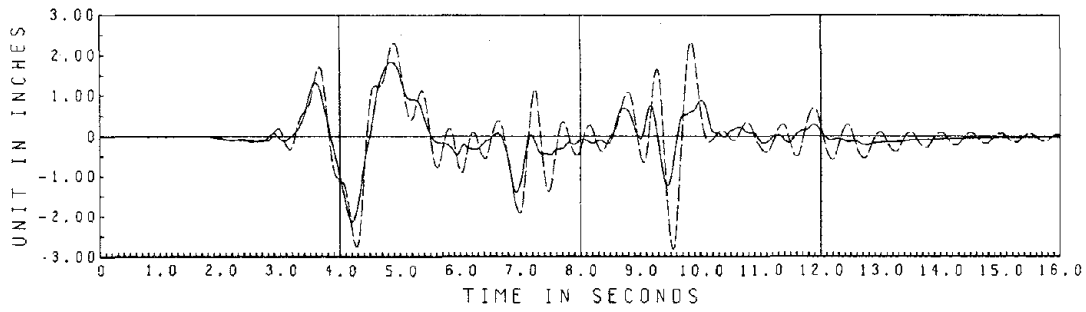
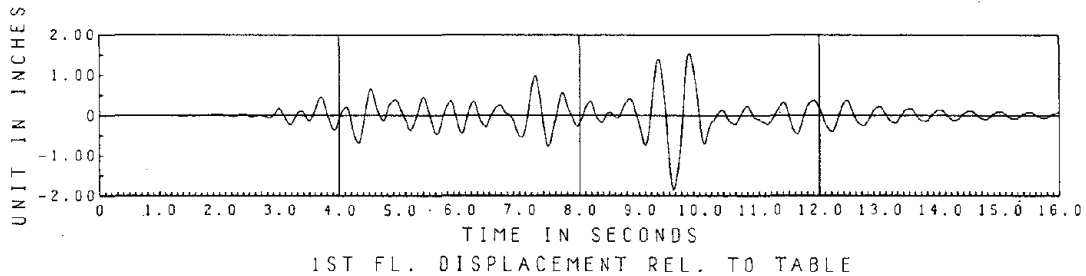
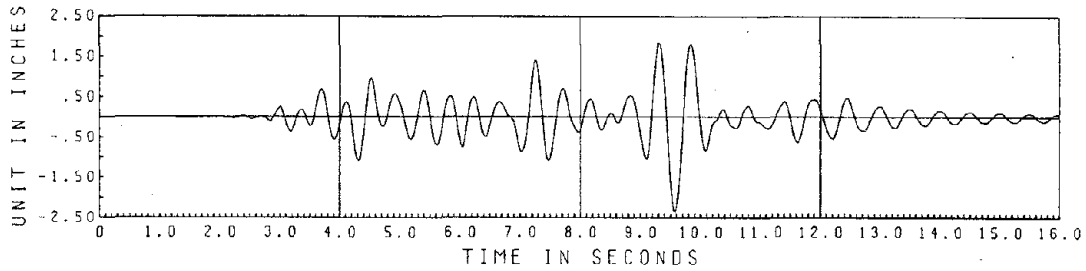
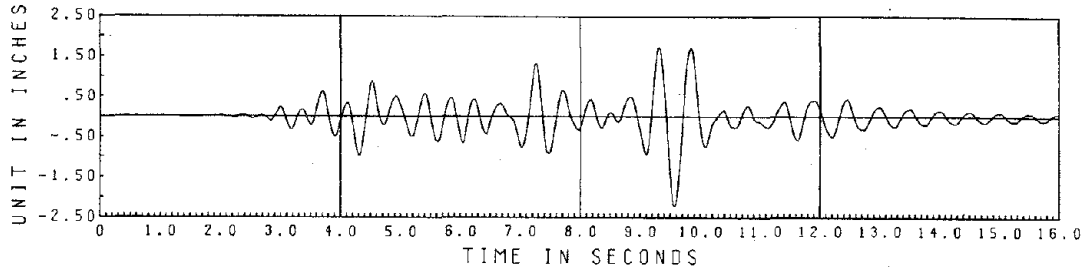


TABLE ACC

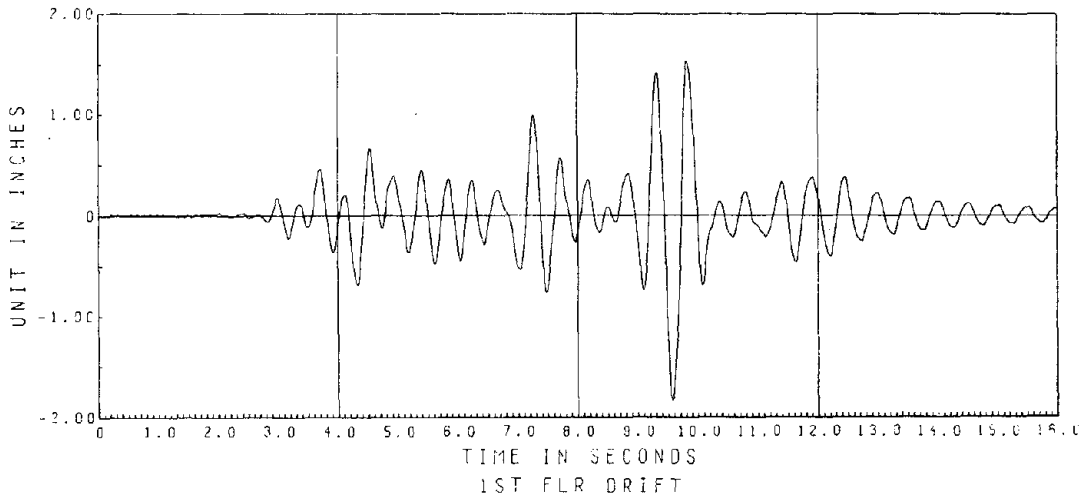
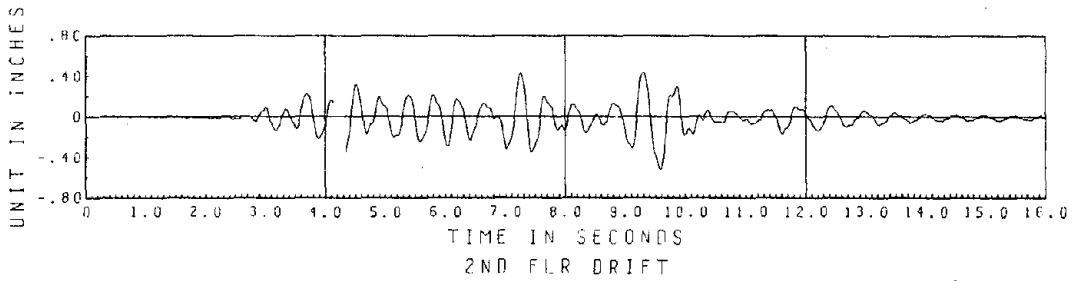
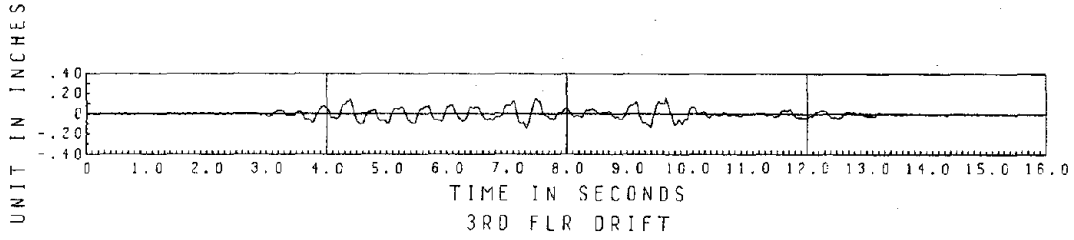
PACOIMA SPAN 400
PIPE DIAGONALLY BRACED FRAME

Fig. 5.2c.1 Table and Floor Accelerations



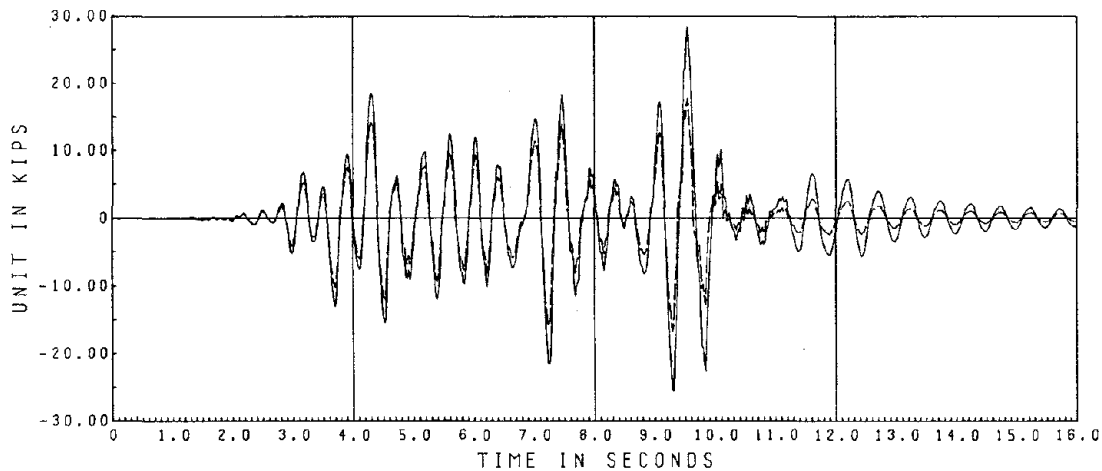
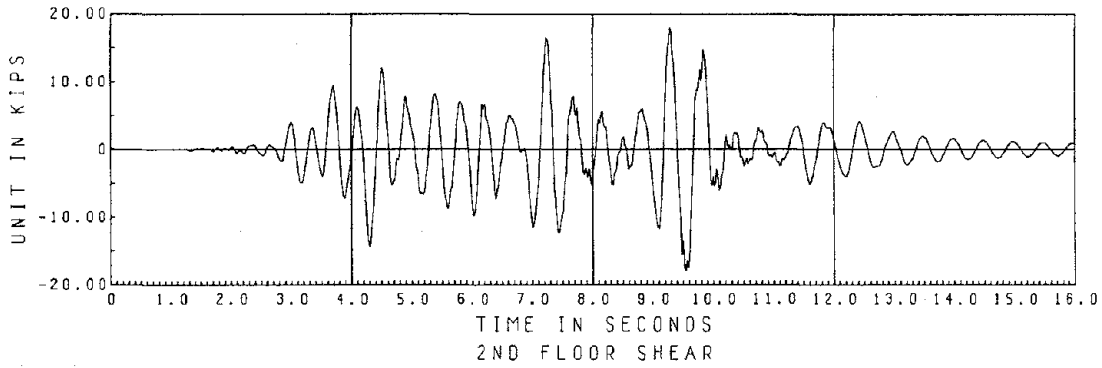
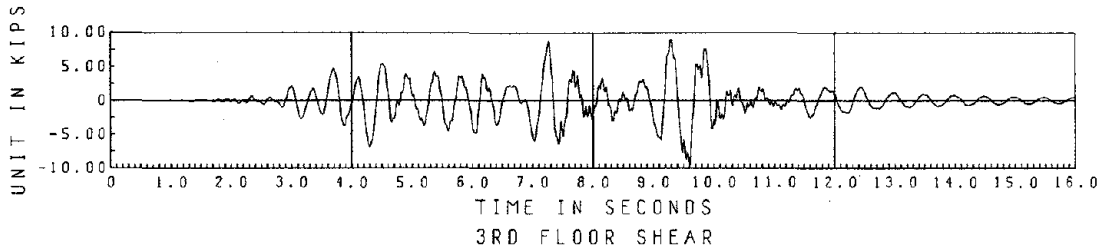
PACOIMA SPAN 400
TEST RESULTS OF PIPE BRACING

Fig. 5.2c.2 Table and Floor Displacements



PACOIMA SPAN 400
 PIPE DIAGONALLY BRACED FRAME

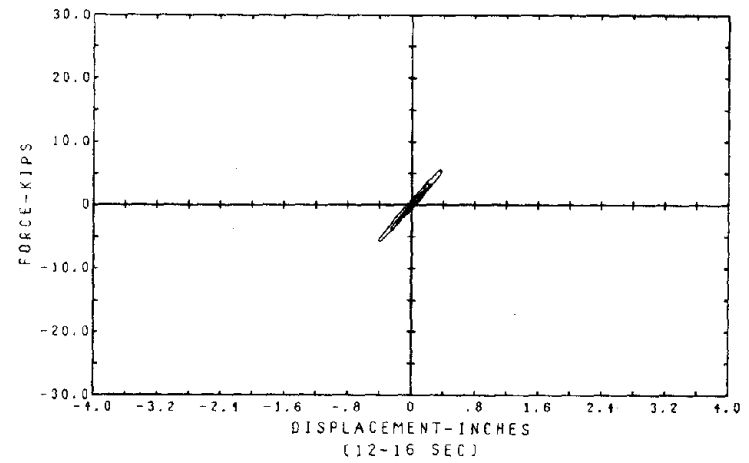
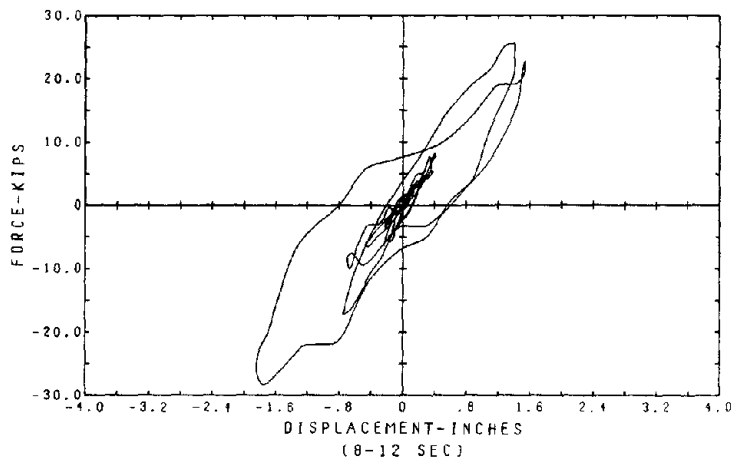
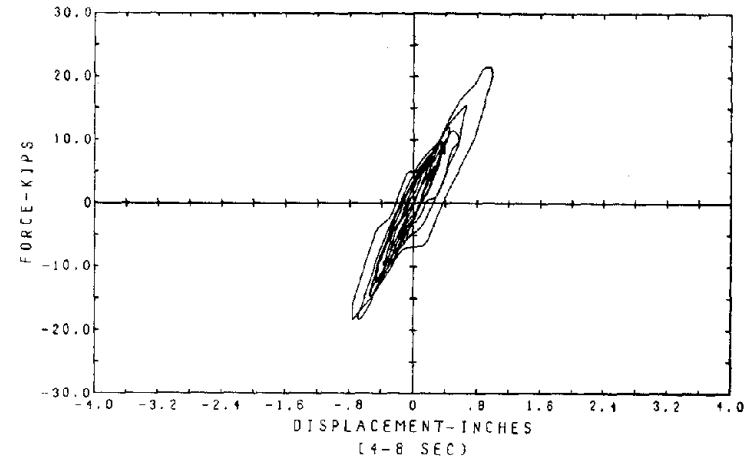
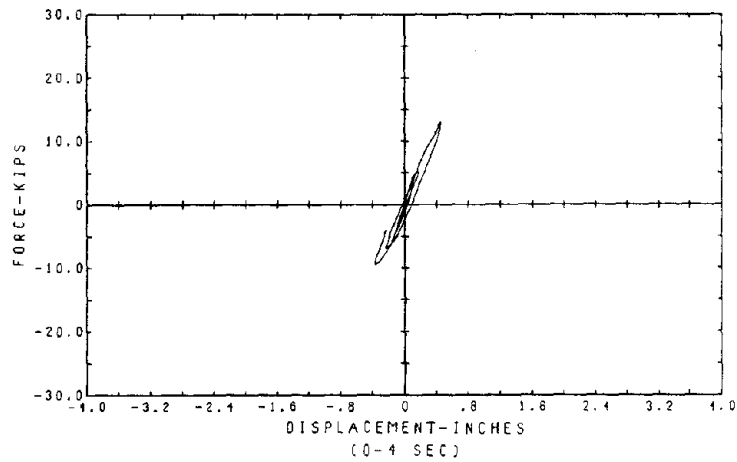
Fig. 5.2c.3 Floor Drifts



PACOIMA SPAN 400

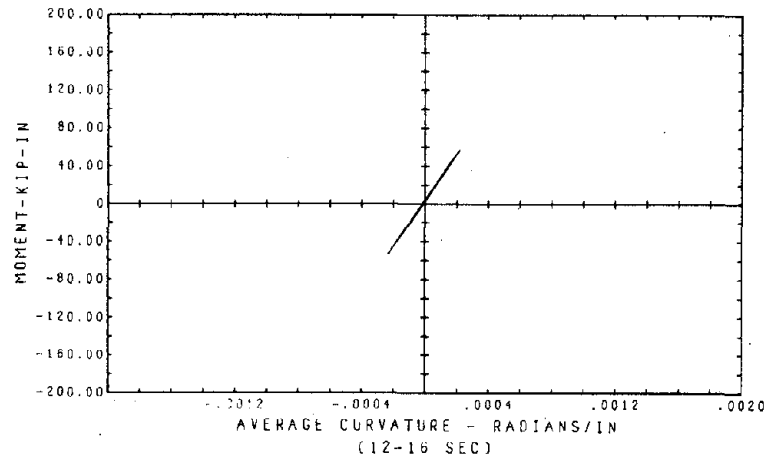
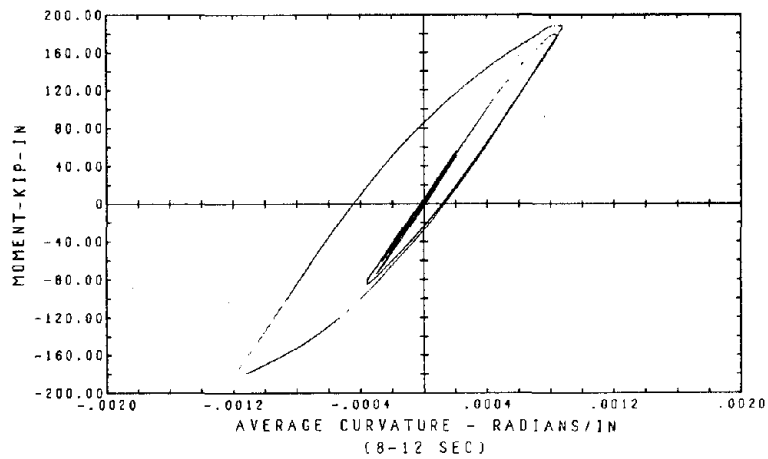
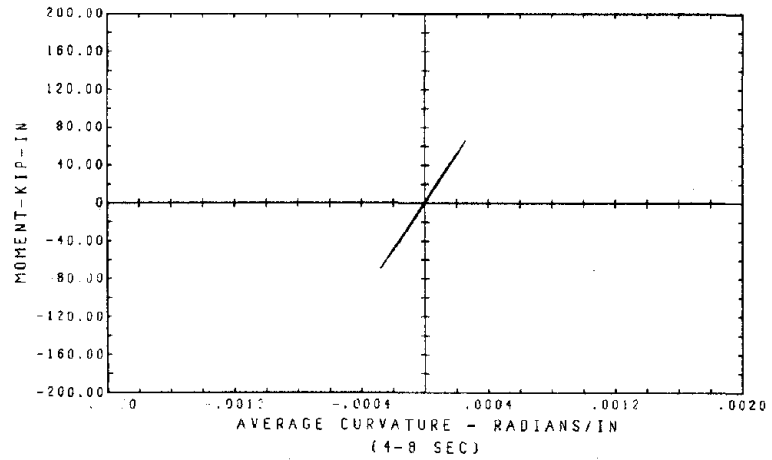
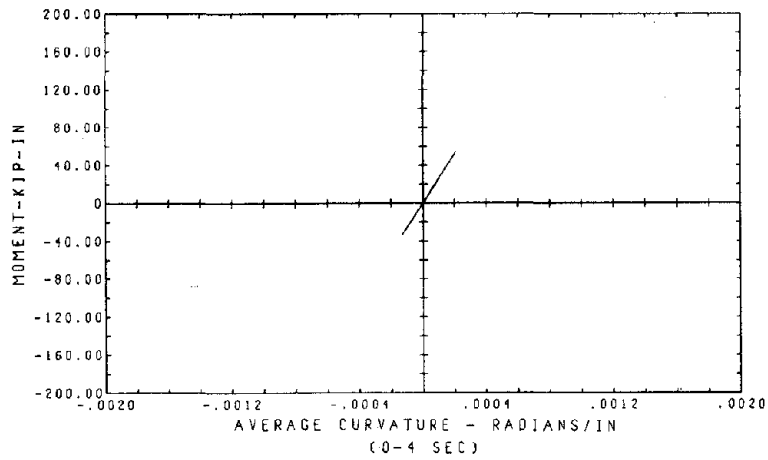
PIPE DIAGONALLY BRACED FRAME

Fig. 5.2c.4 Floor Shear Forces



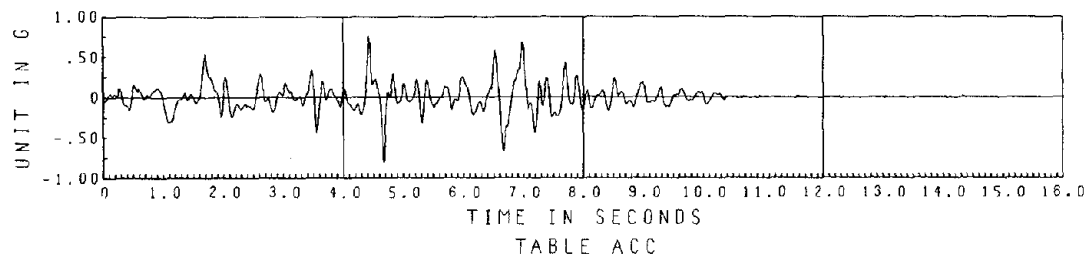
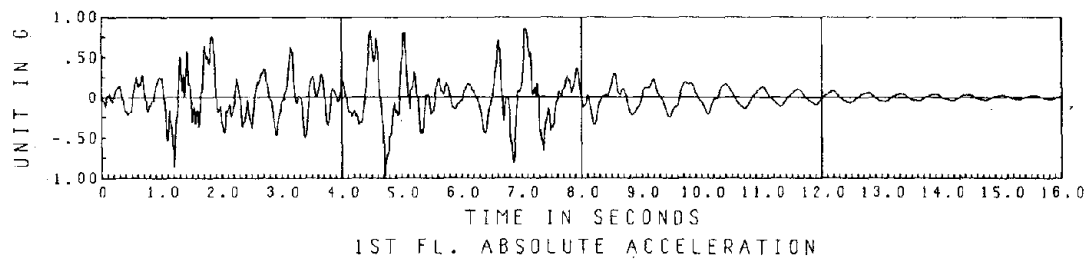
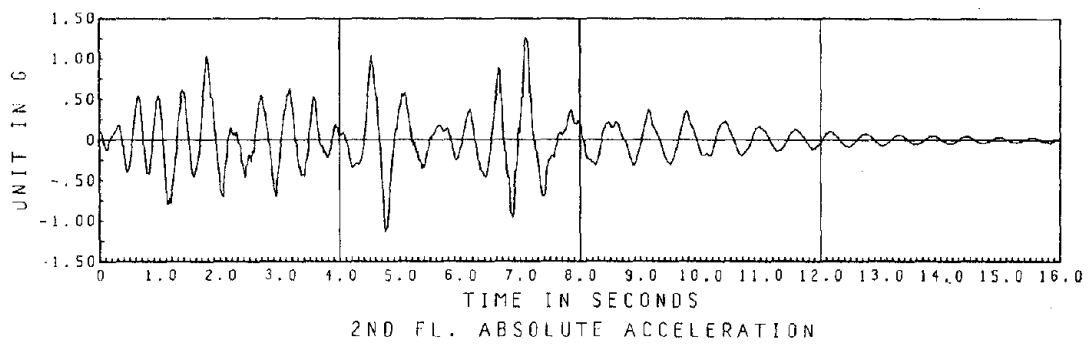
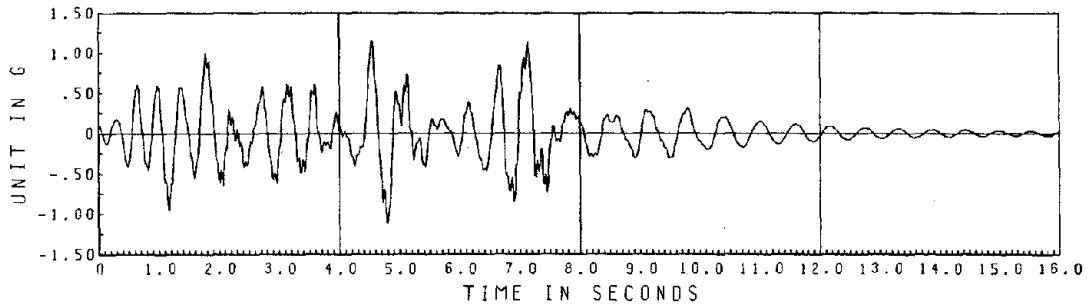
1ST FLR SHEAR VS. DISPLACEMENT
PIPE DIAGONALLY BRACED STRUCTURE
PACDTMA SPAN 400

Fig. 5.2c.5 First Floor Shear-Displacement Hysteresis Loops



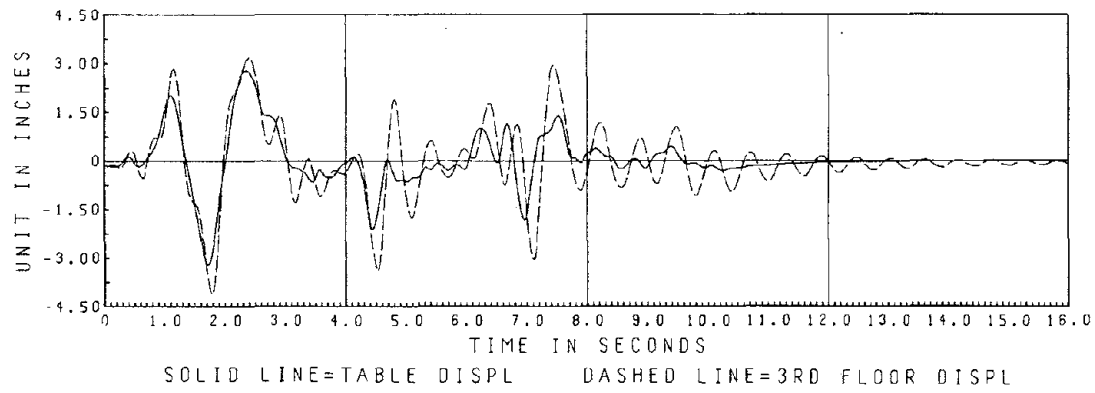
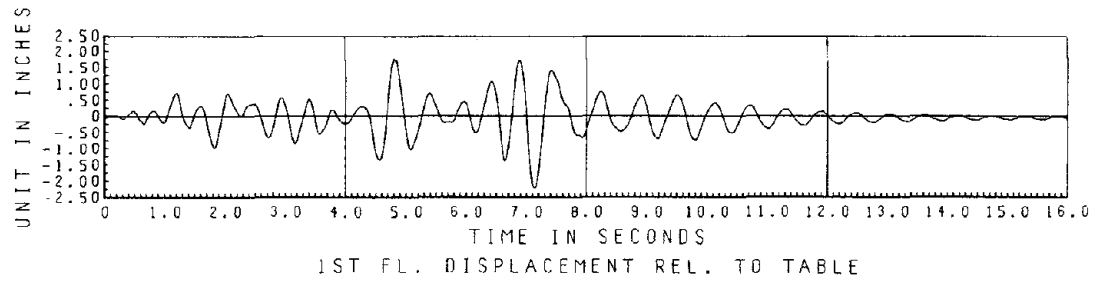
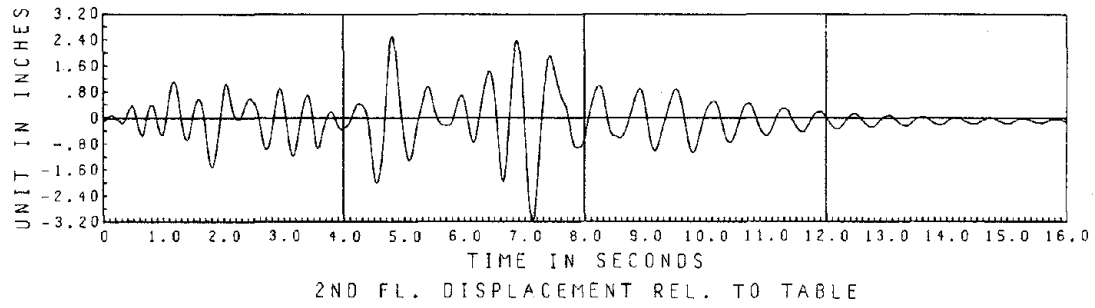
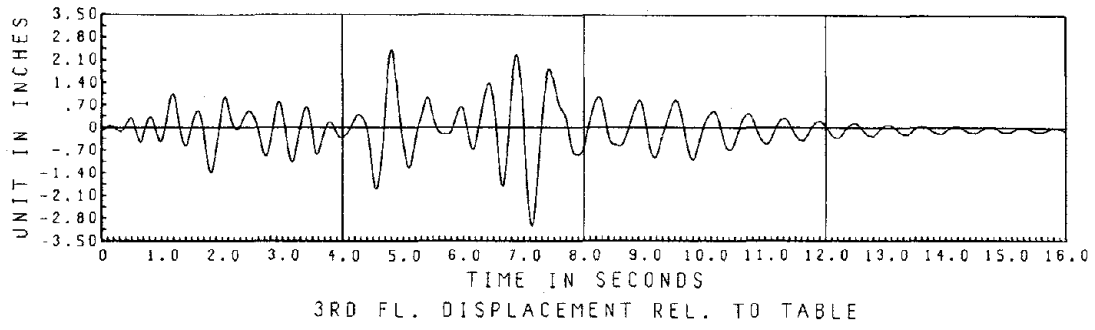
1ST FLR COL BOTTOM END , MOMENT VS. CURVATURE
 REFERENCE COLUMN LINE NA
 PIPE DIAGONALLY BRACED STRUCTURE
 PACOIMA SPAN 400

Fig. 5.2c.6 First Floor Column Moment-Curvature Hysteresis Loops



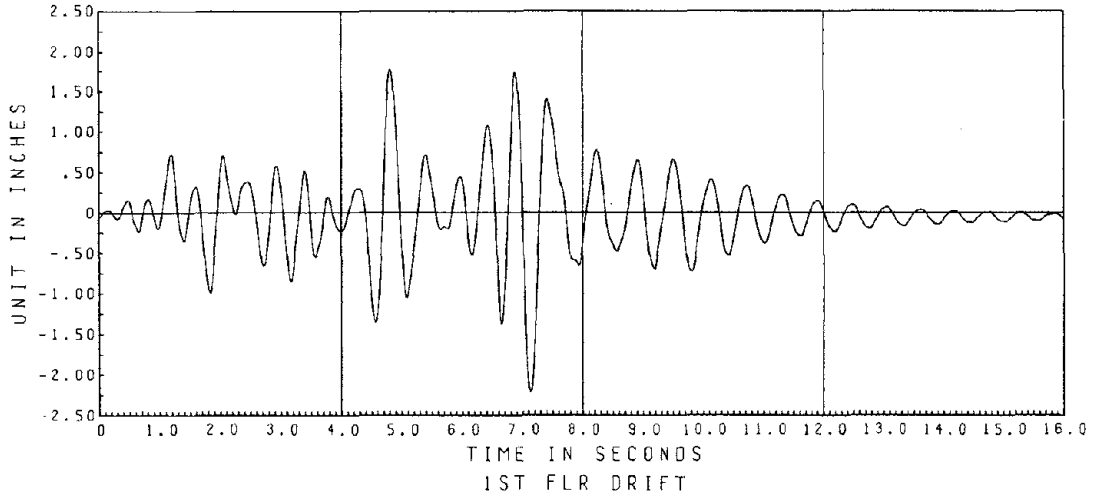
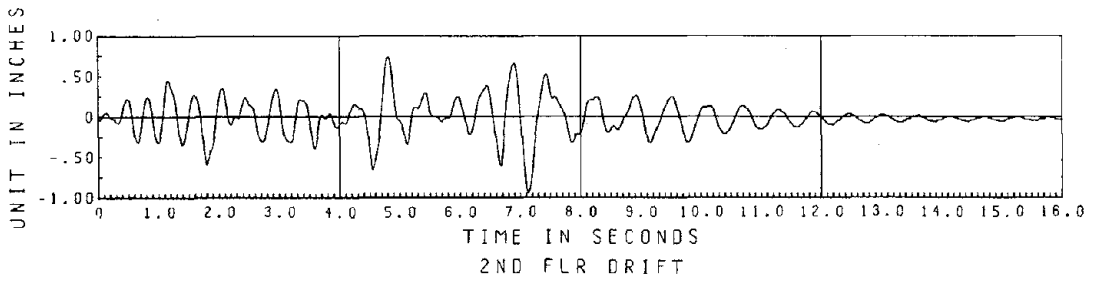
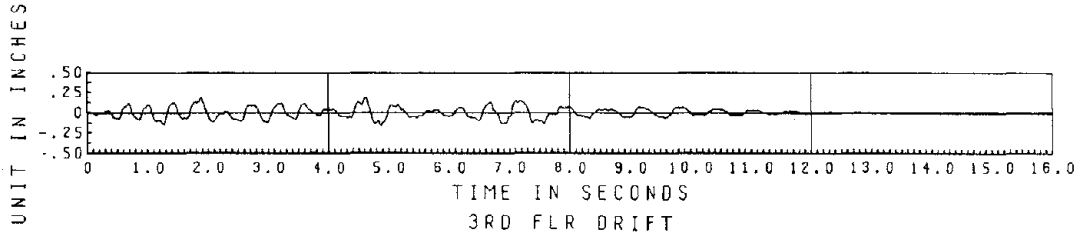
PACOIMA SPAN 600
TEST RESULTS OF PIPE BRACING

Fig. 5.2d.1 Table and Floor Accelerations



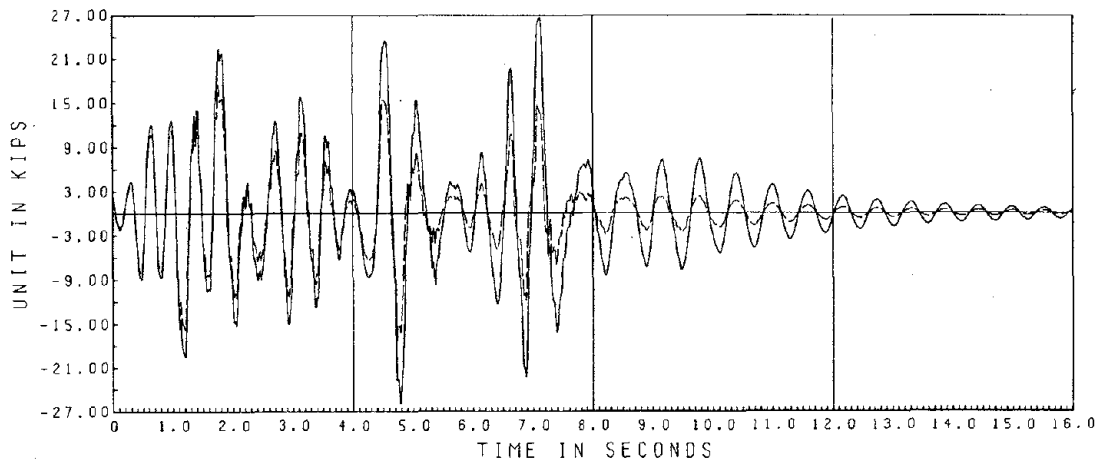
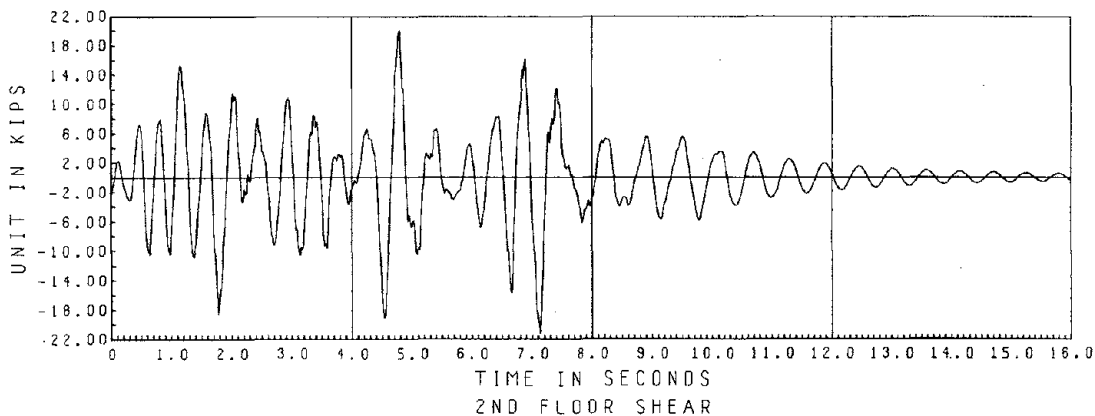
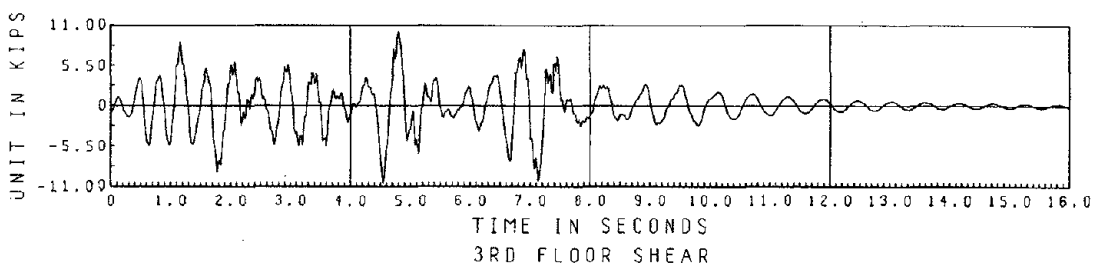
PACOIMA SPAN 600
TEST RESULTS OF PIPE BRACING

Fig. 5.2d.2 Table and Floor Displacements



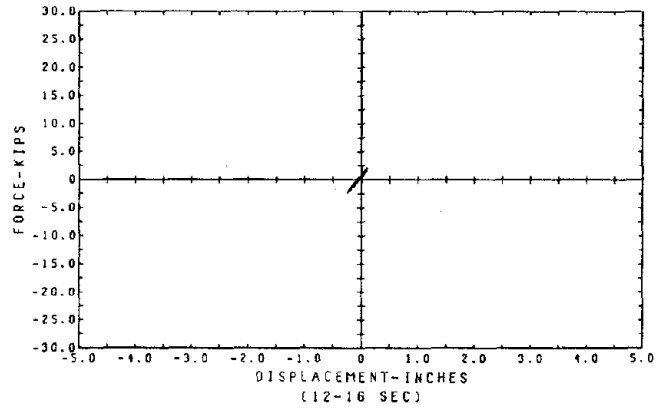
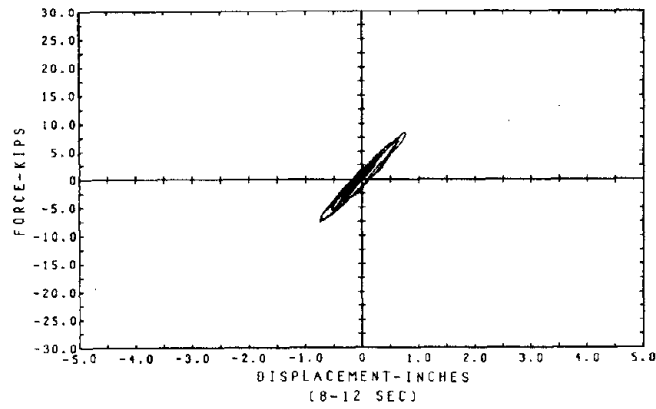
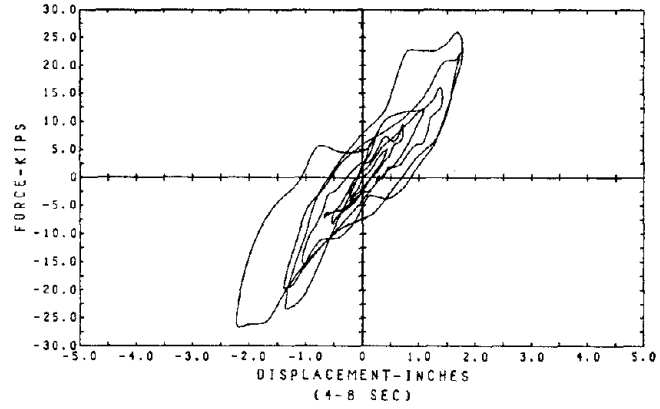
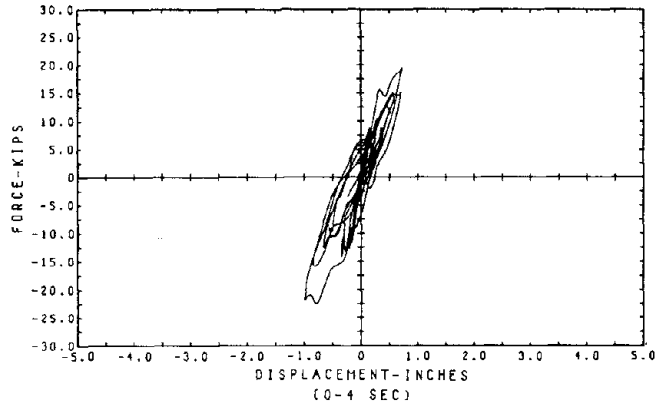
PACOIMA SPAN 600
TEST RESULTS OF PIPE BRACING

Fig. 5.2d.3 Floor Drifts



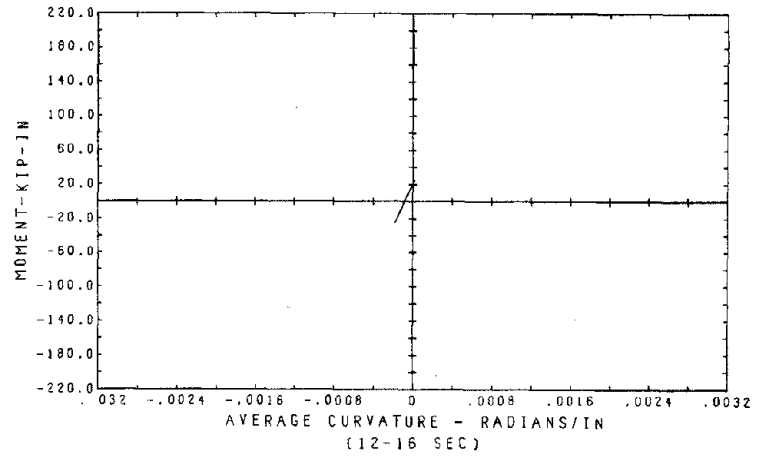
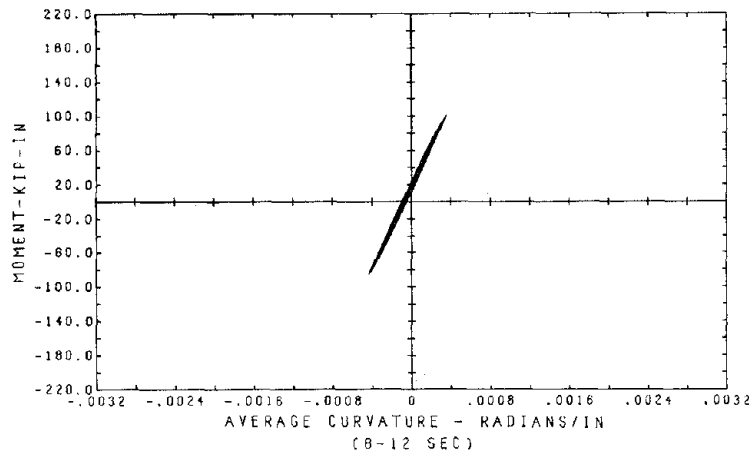
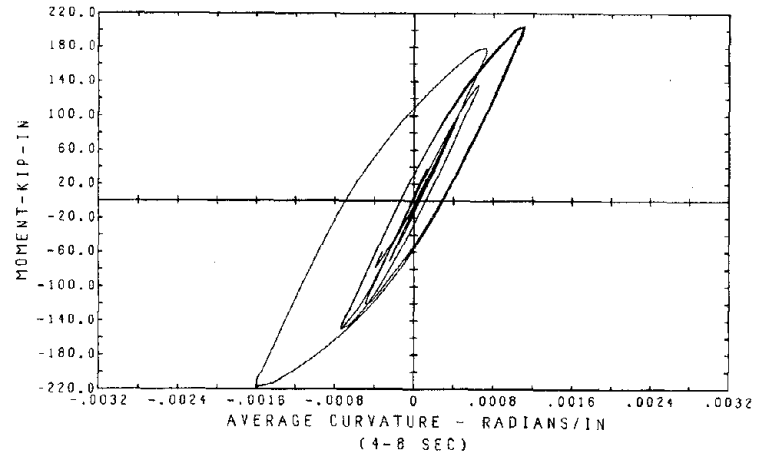
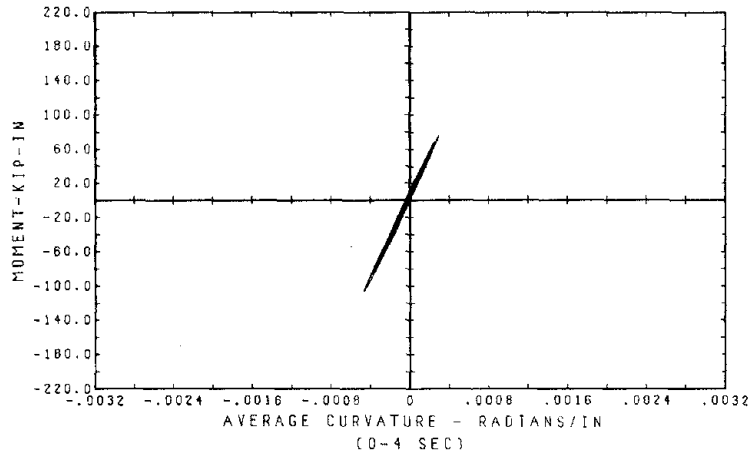
PACDIMA SPAN 600
TEST RESULTS OF PIPE BRACING

Fig. 5.2d.4 Floor Shear Forces



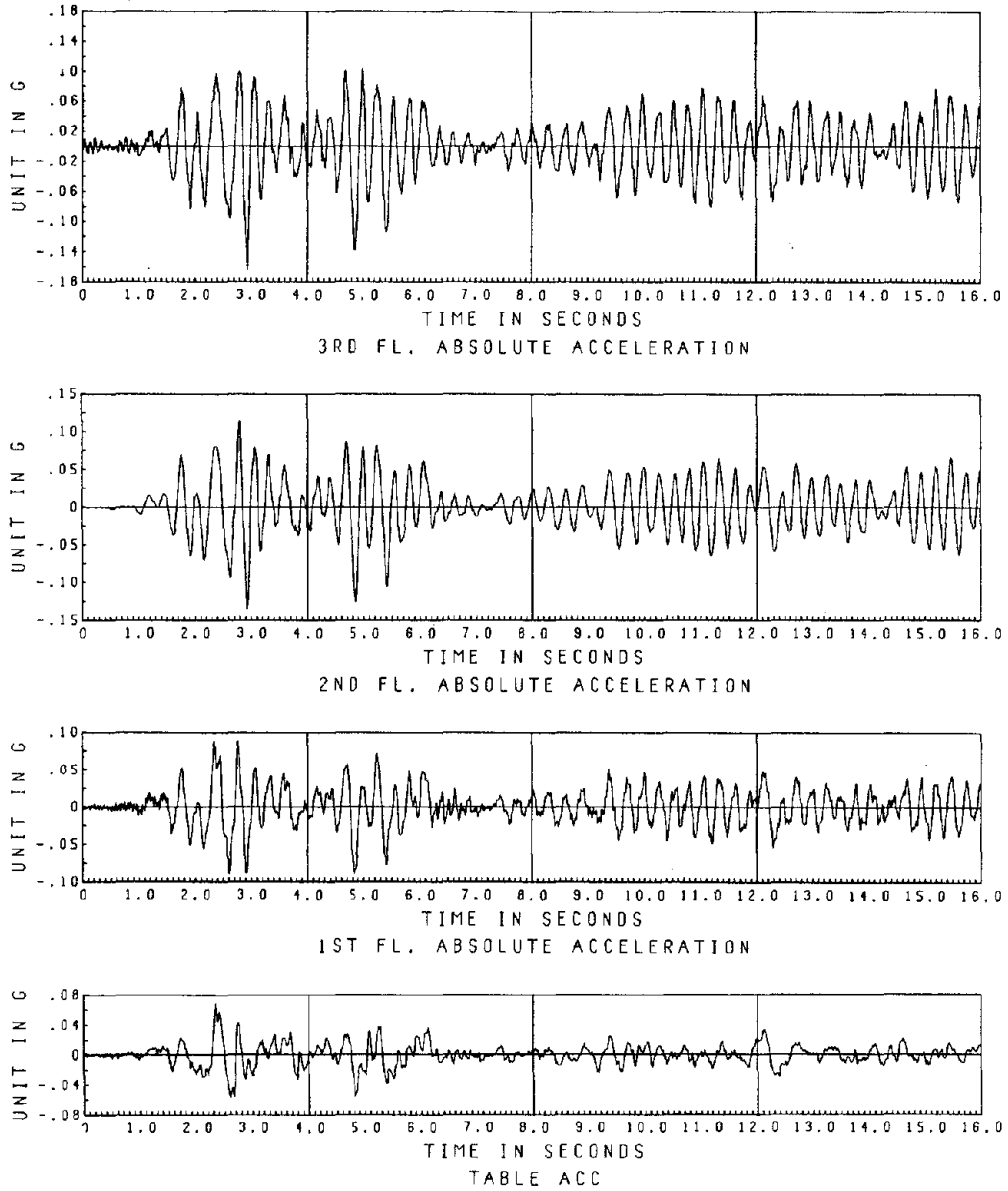
1ST FLR SHEAR VS. DISPLACEMENT
 REFERENCE FRAME N
 PIPE DIAGONALLY BRACED STRUCTURE
 PACOIMA SPAN 600

Fig. 5.2d.5 First Floor Shear-Displacement Hysteresis Loops



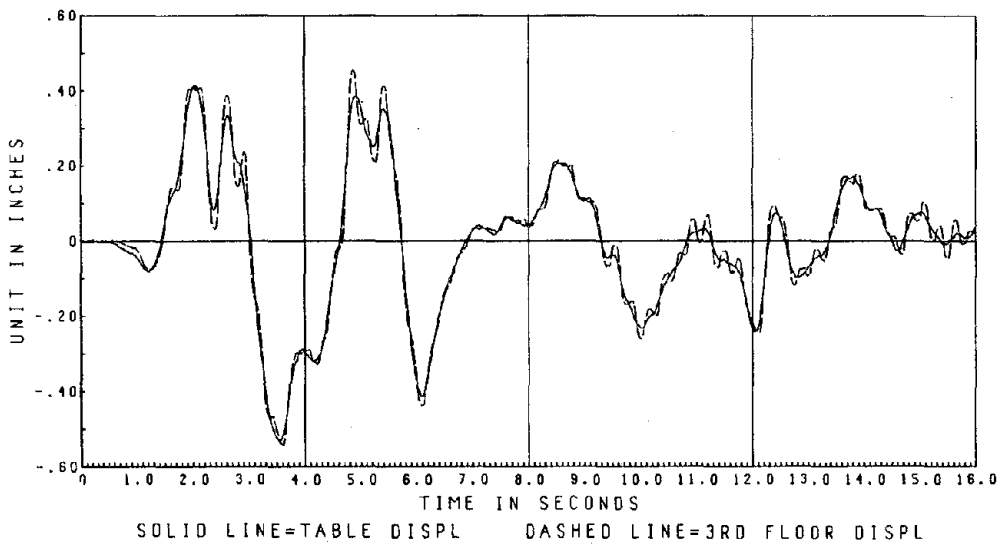
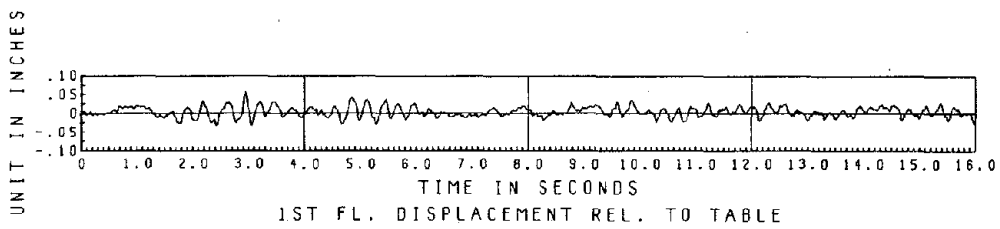
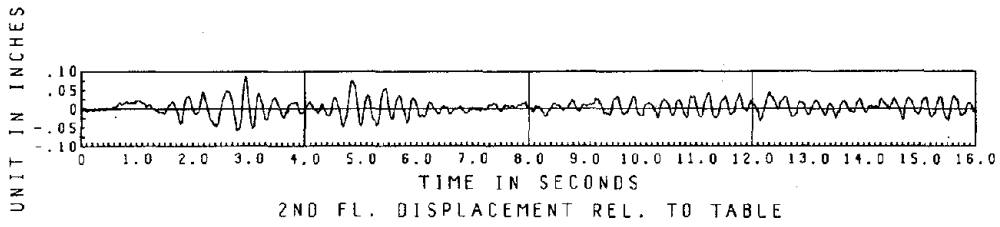
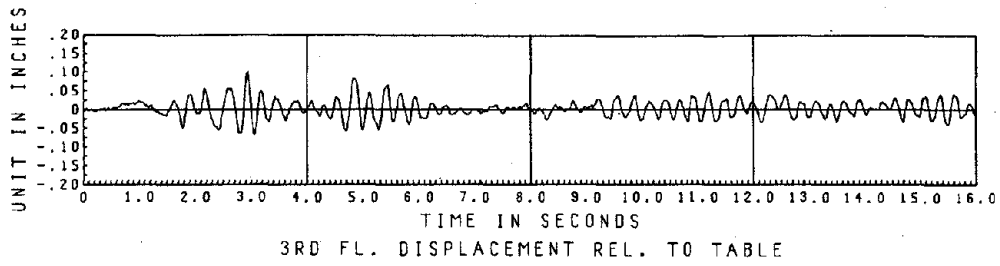
1ST FLR COL BOTTON END , MOMENT VS. CURVATURE
 REFERENCE COLUMN LINE NA
 PIPE DIAGONALLY BRACED STRUCTURE
 PACOIMA SPAN 600

Fig. 5.2d.6 First Floor Column Moment-Curvature Hysteresis Loops



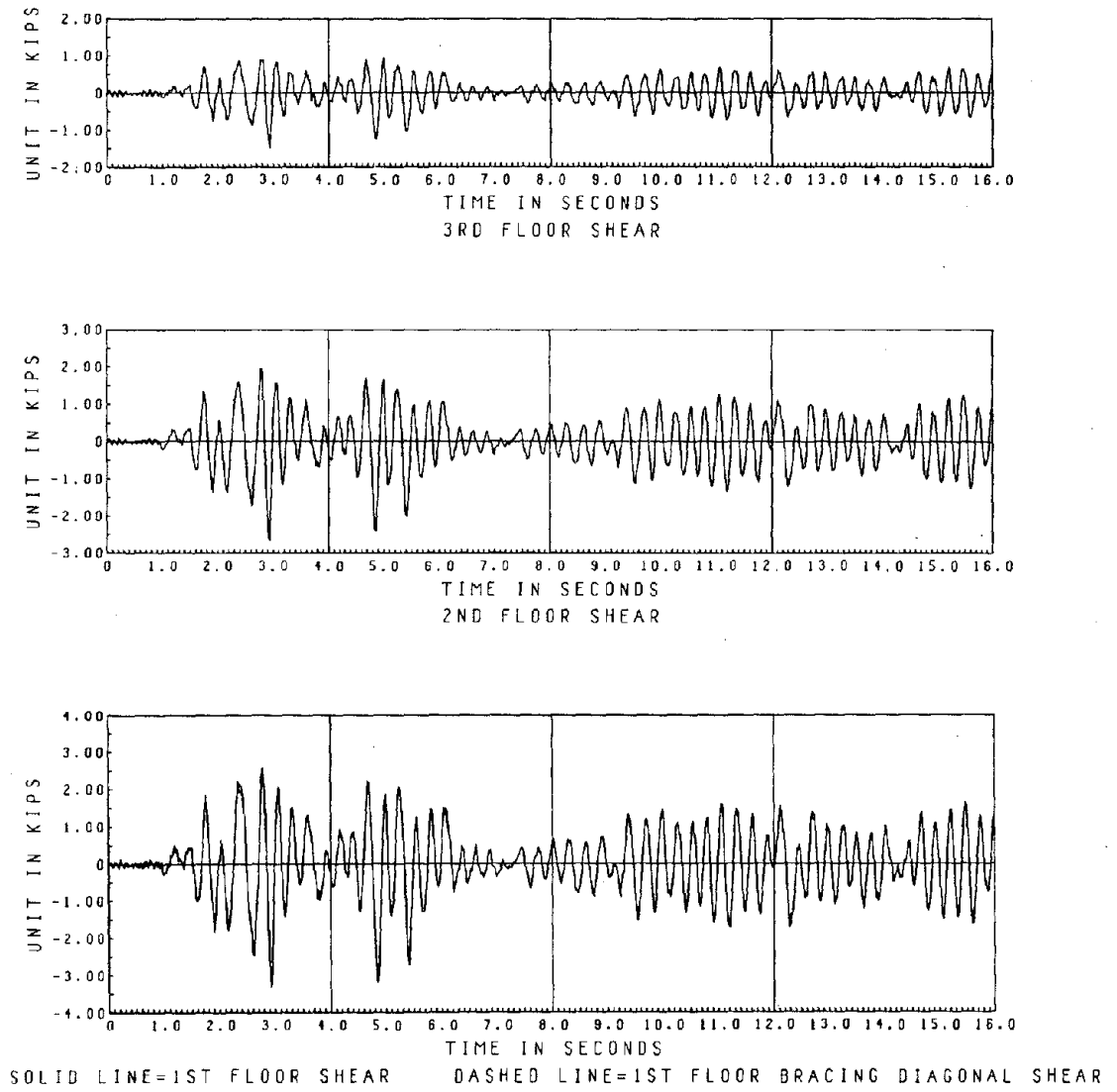
EL CENTRO SPAN 100
 TEST RESULTS OF DOUBLE ANGLE BRACING

Fig. 5.3a.1 Table and Floor Accelerations



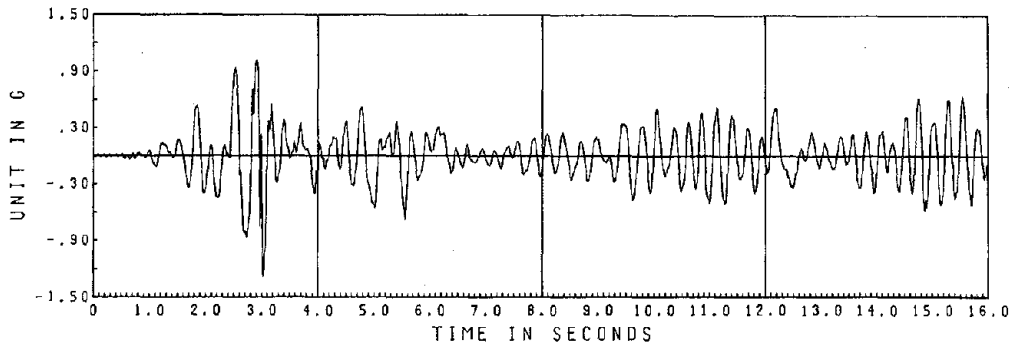
EL CENTRO SPAN 100
TEST RESULTS OF DOUBLE ANGLE BRACING

Fig. 5.3a.2 Table and Floor Displacements

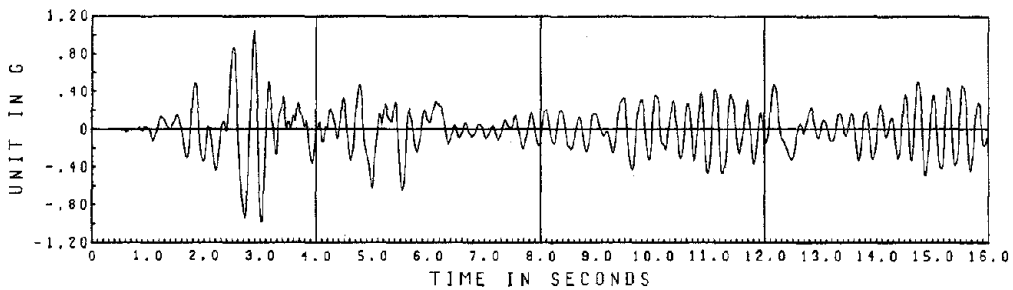


EL CENTRO SPAN 100
TEST RESULTS OF DOUBLE ANGLE BRACING

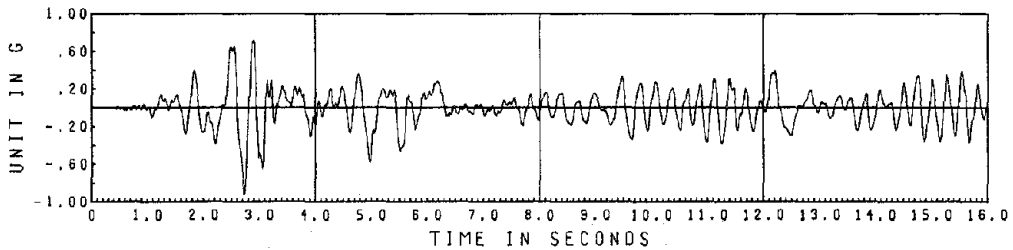
Fig. 5.3a.3 Floor Shear Forces



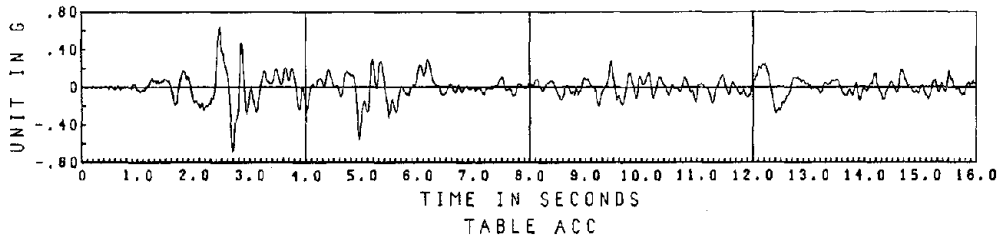
3RD FL. ABSOLUTE ACCELERATION



2ND FL. ABSOLUTE ACCELERATION

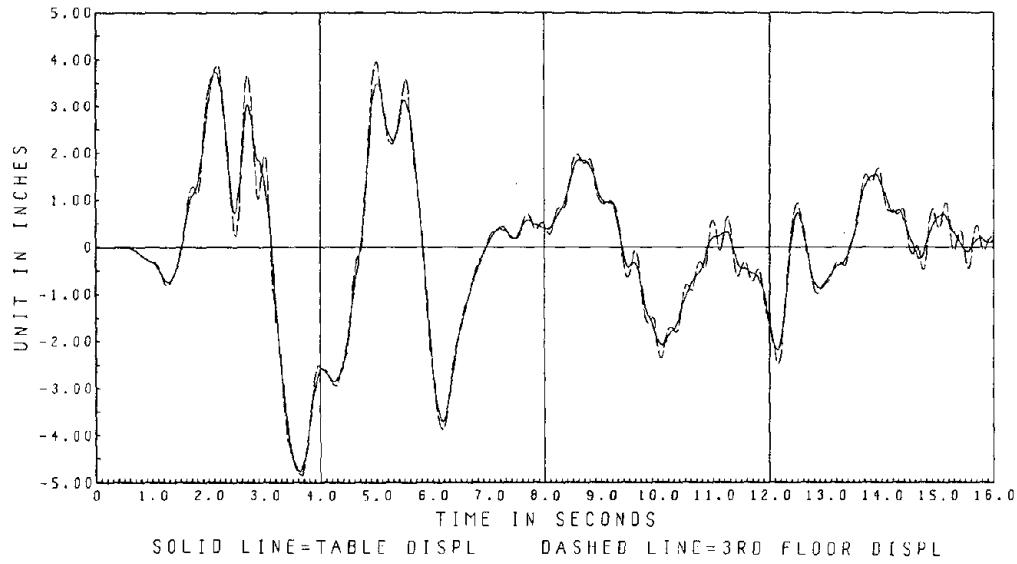
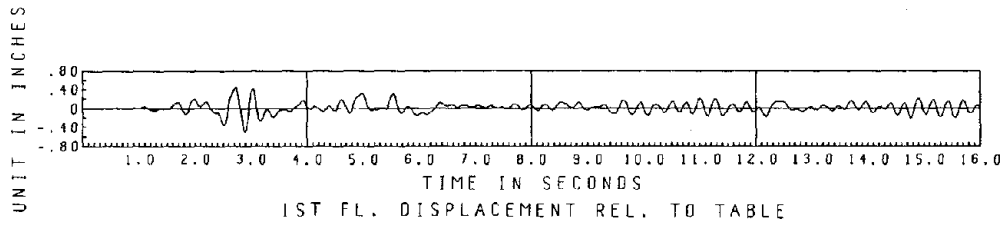
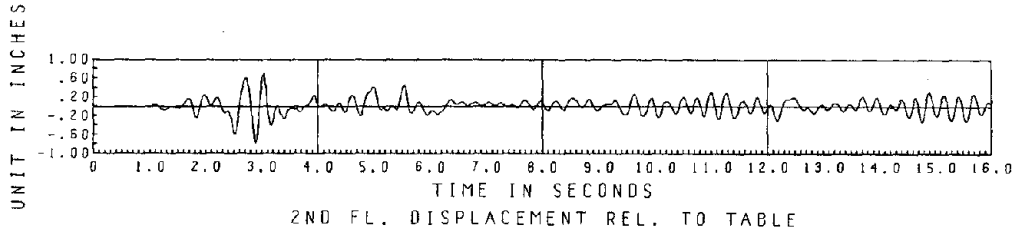
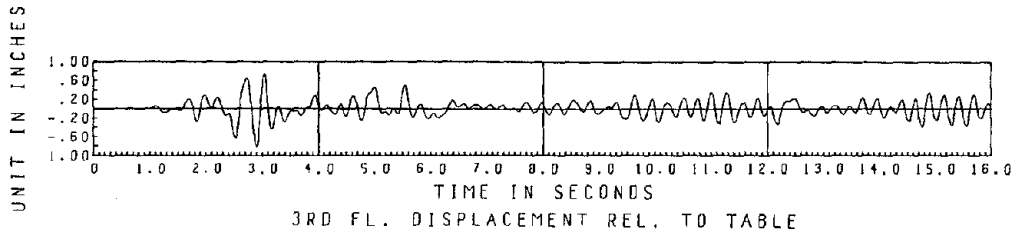


1ST FL. ABSOLUTE ACCELERATION



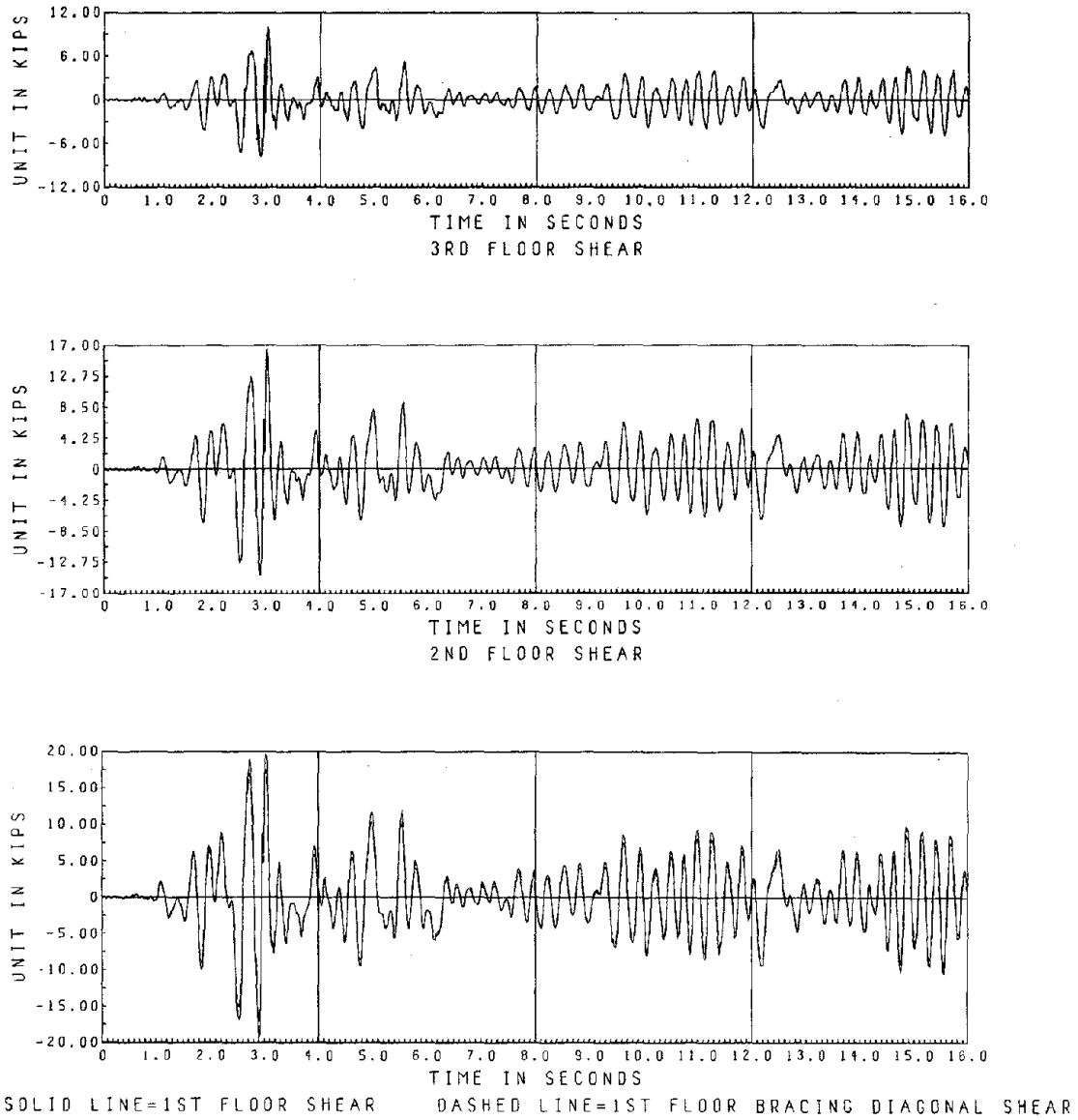
EL CENTRO SPAN 900
TEST RESULTS OF DOUBLE ANGLE BRACING

Fig. 5.3b.1 Table and Floor Accelerations



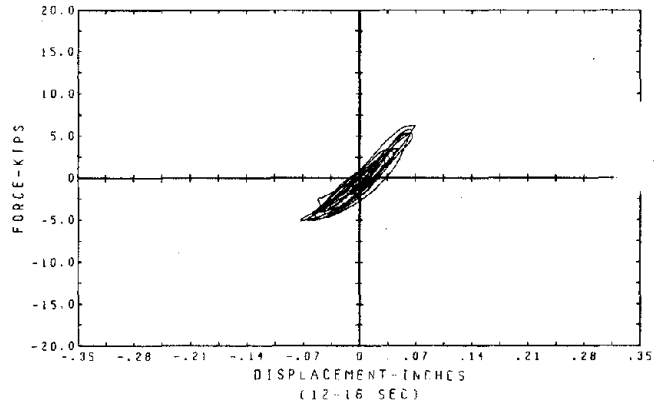
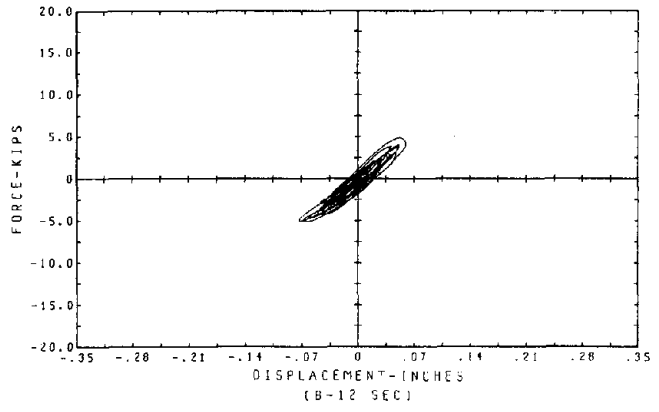
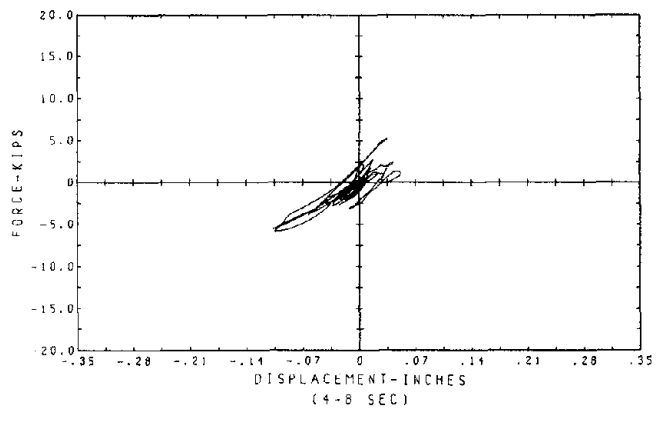
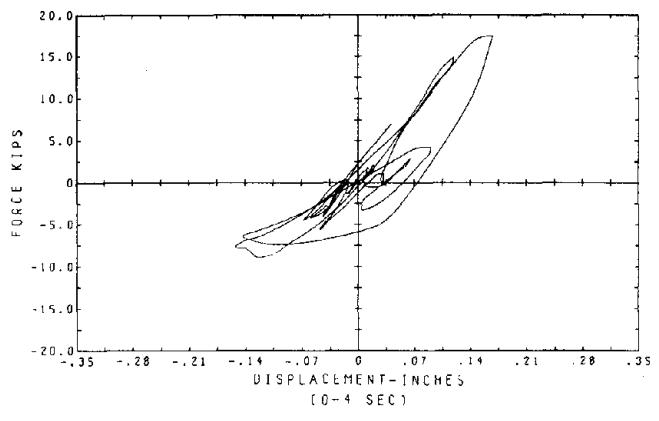
EL CENTRO SPAN 900
TEST RESULTS OF DOUBLE ANGLE BRACING

Fig. 5.3b.2 Table and Floor Displacements



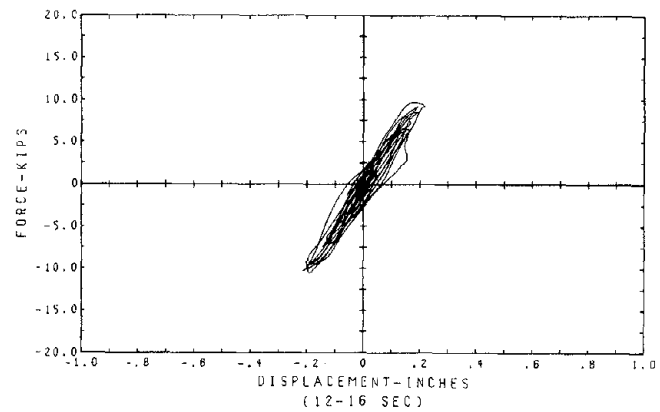
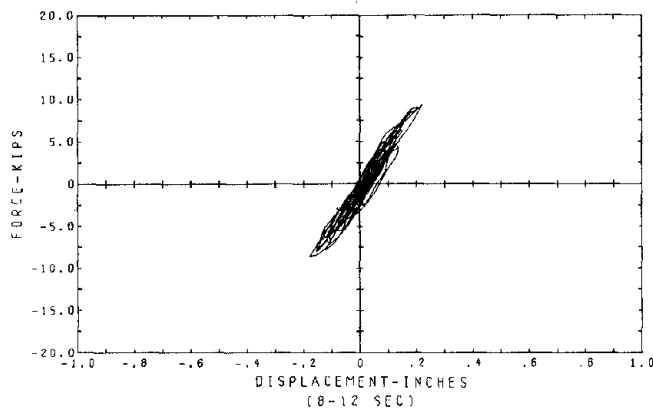
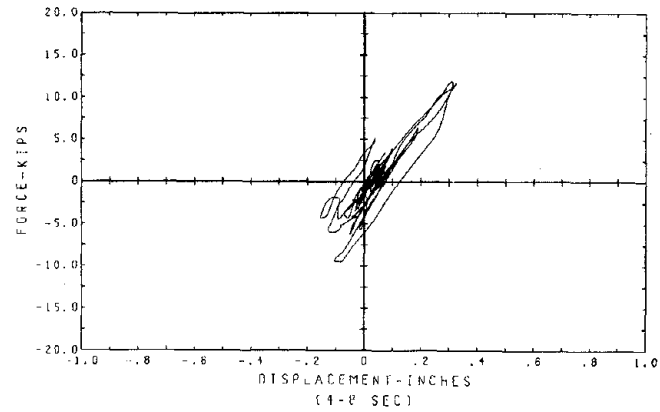
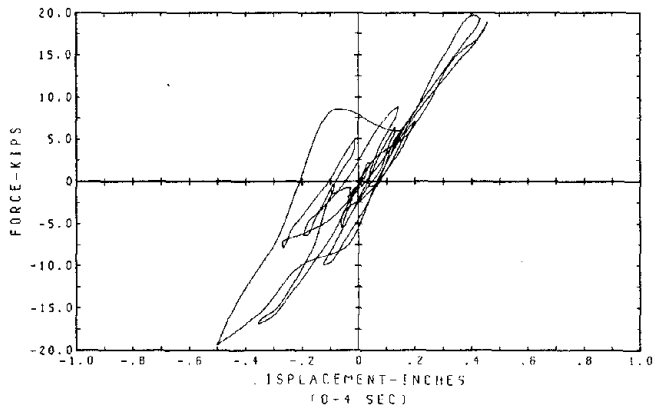
EL CENTRO SPAN 900
TEST RESULTS OF DOUBLE ANGLE BRACING

Fig. 5.3b.3 Floor Shear Forces



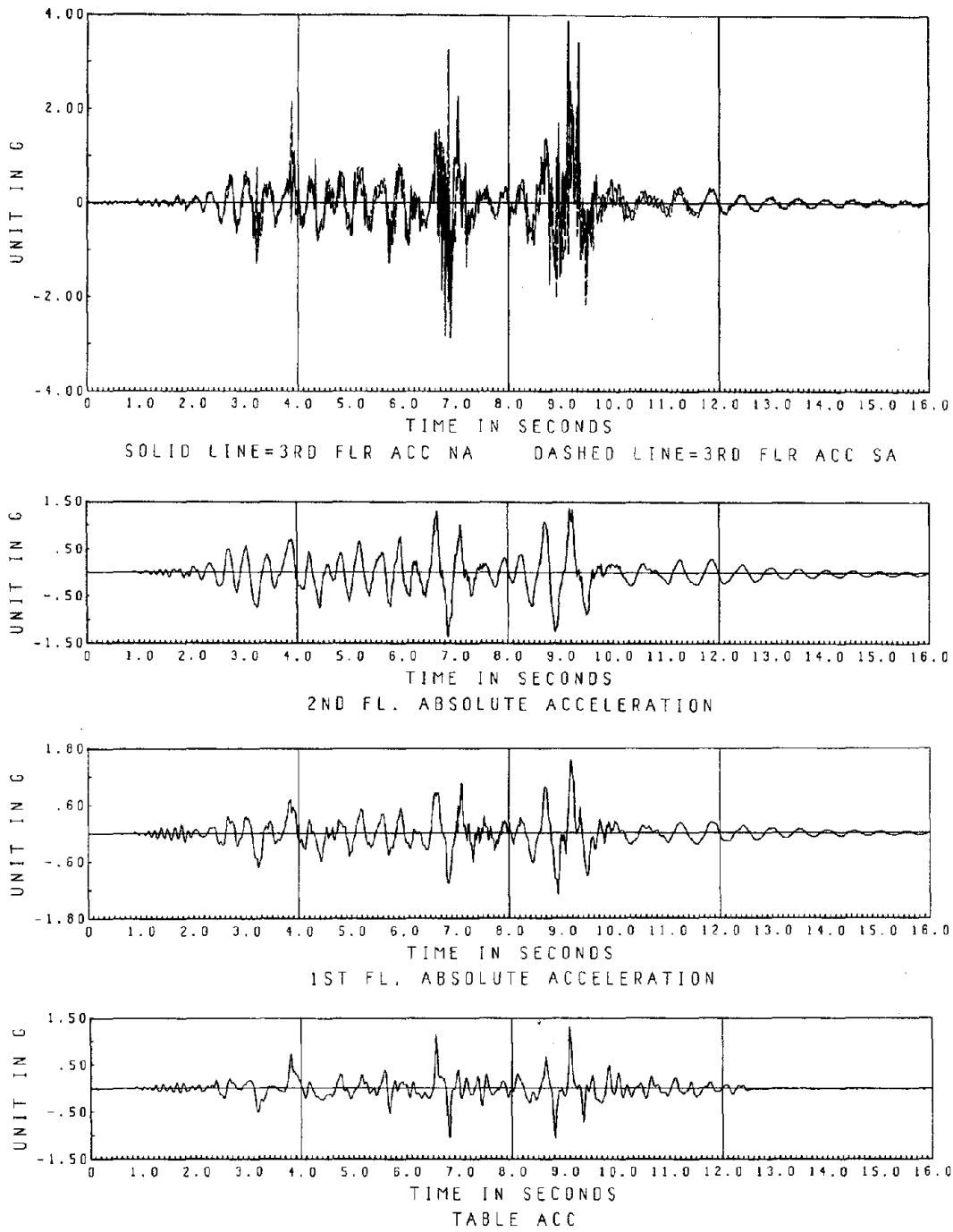
1ST FLR DOUBLE ANGLE FORCE VS. DISPLACEMENT
 REFERENCE FRAMC N
 DOUBLE ANGLE DIAGONALLY BRACED STRUCTURE
 EL CENTRO SPAN 900

Fig. 5.3b.4 First Floor Double Angle Force-Displacement Hysteresis Loops



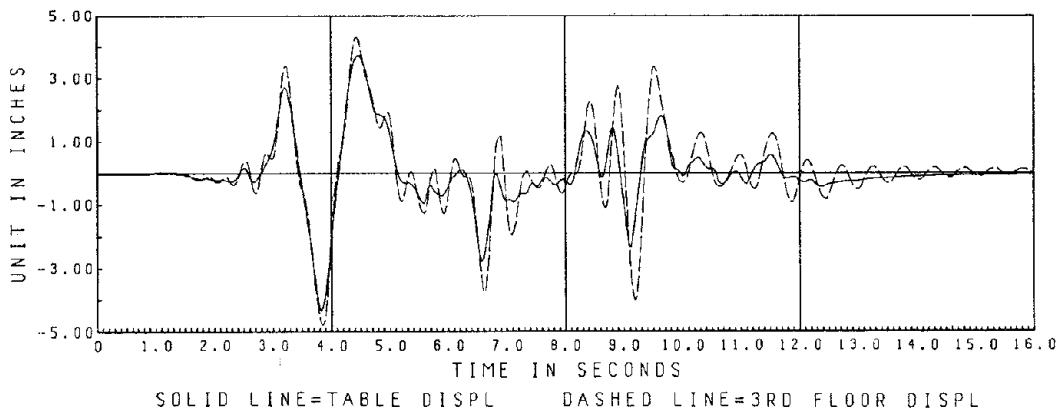
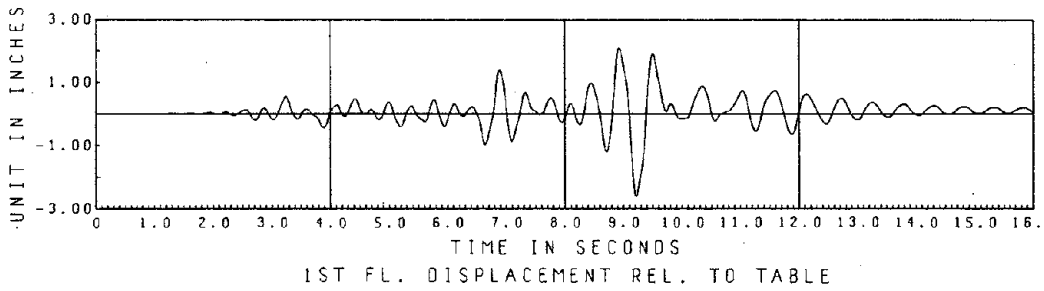
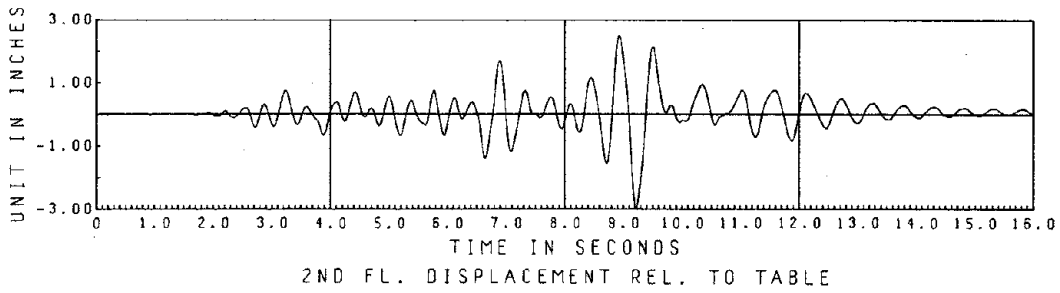
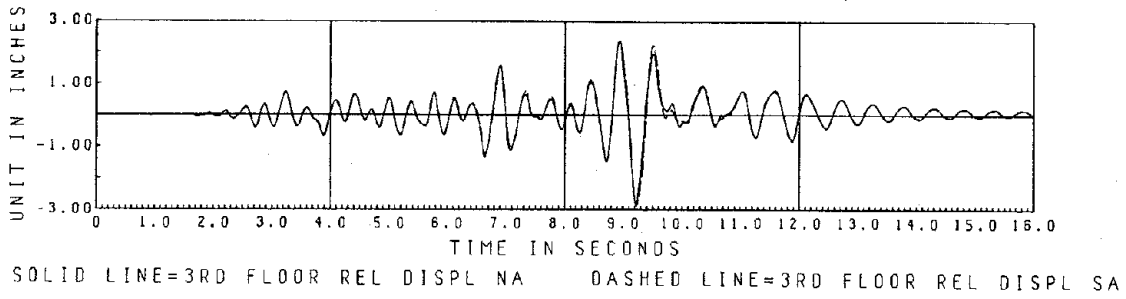
1ST FLR SHEAR VS. DISPLACEMENT
 REFERENCE FRAME N
 DOUBLE ANGLE DIAGONALLY BRACED STRUCTURE
 EL CENTRIC SPAN 900

Fig. 5.3b.5 First Floor Shear-Displacement Hysteresis Loops



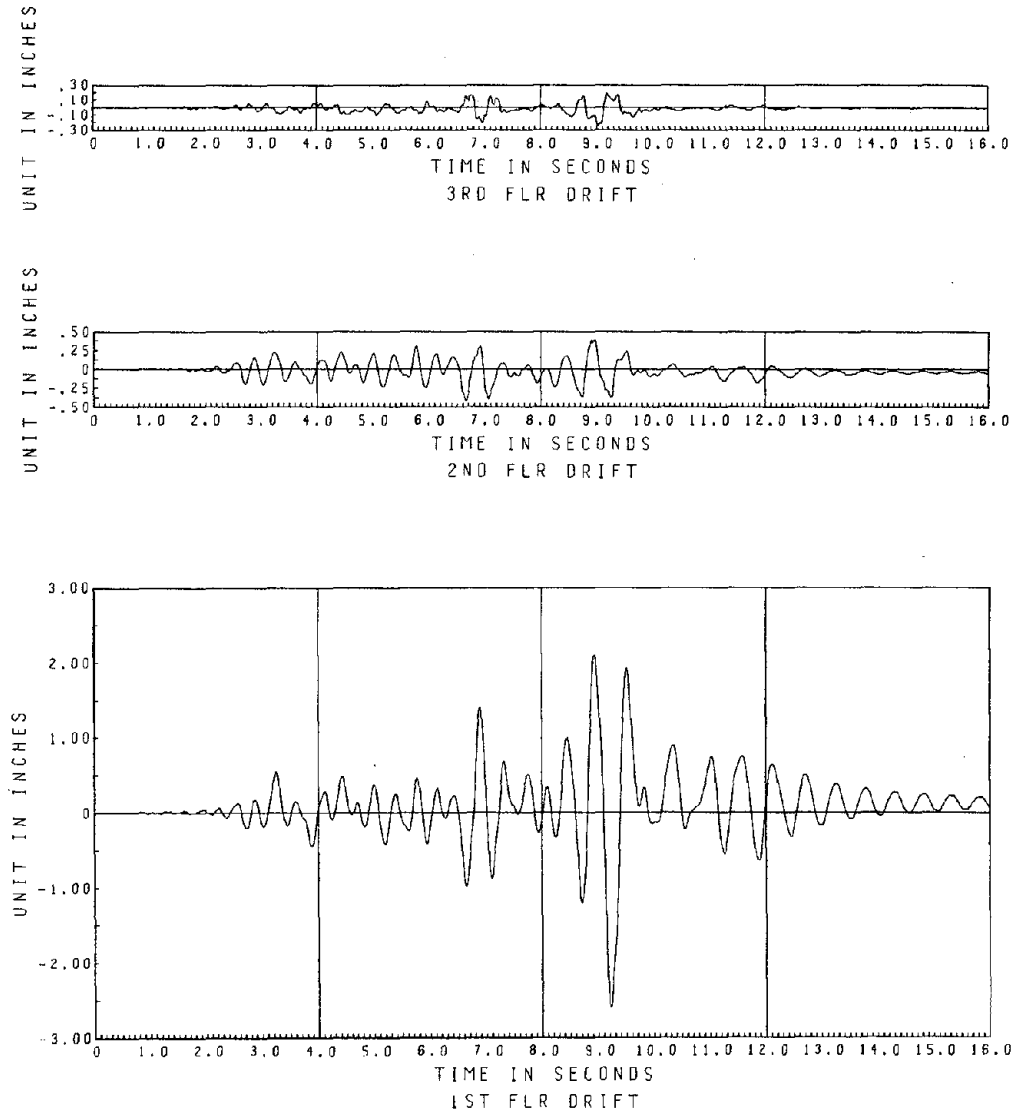
PACOIMA RECORD-1.314 G'S
TEST RESULTS OF DOUBLE ANGLE BRACING

Fig. 5.3c.1 Table and Floor Accelerations



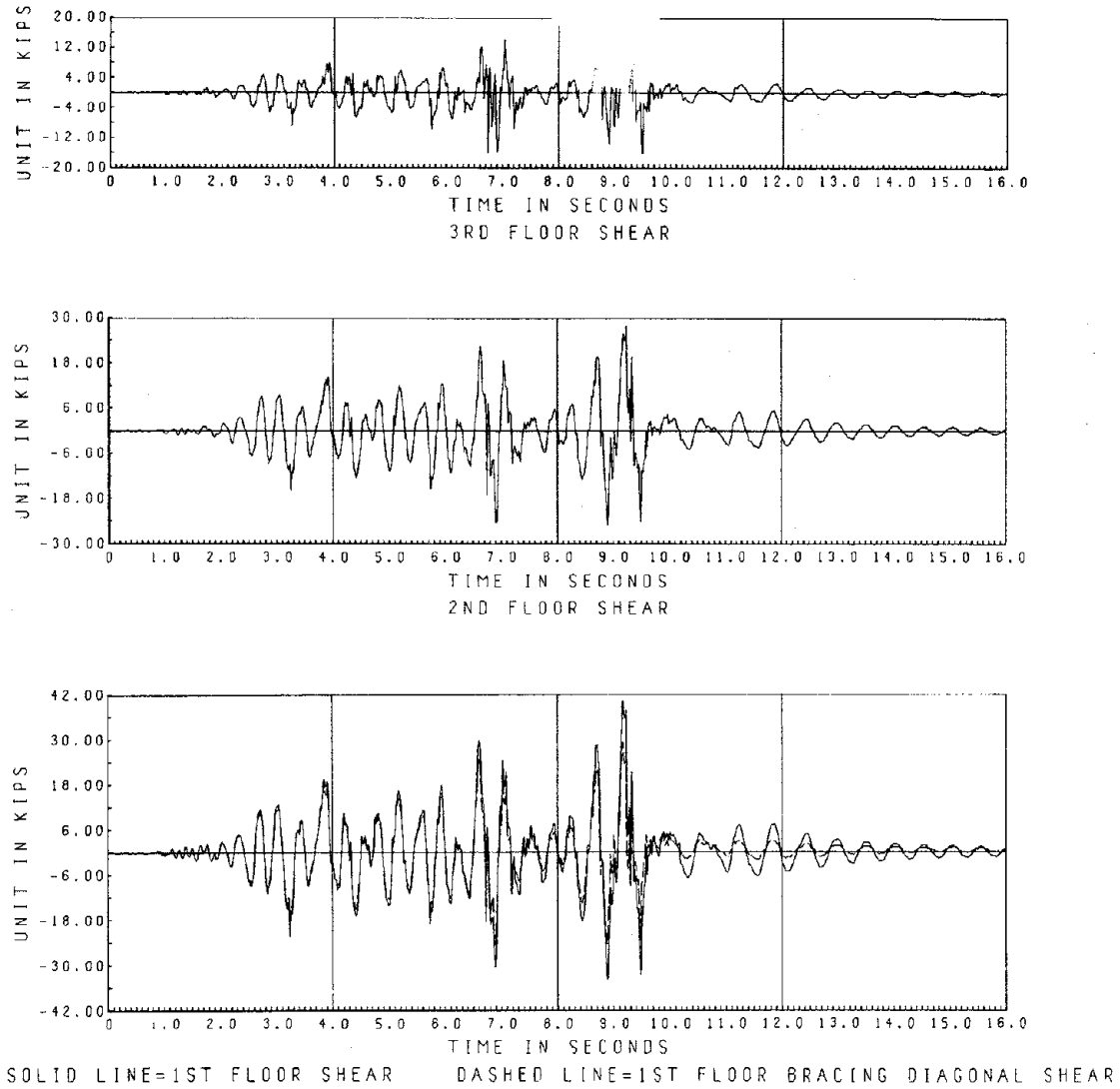
PACDIMA RECORD 1.314 G'S
 TEST RESULTS OF DOUBLE ANGLE BRACING

Fig. 5.3c.2 Table and Floor Displacements



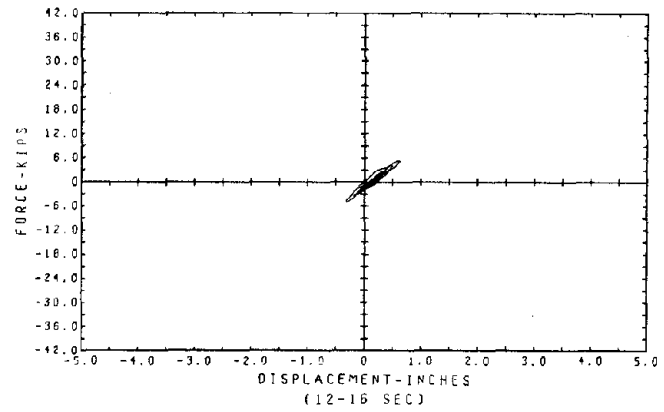
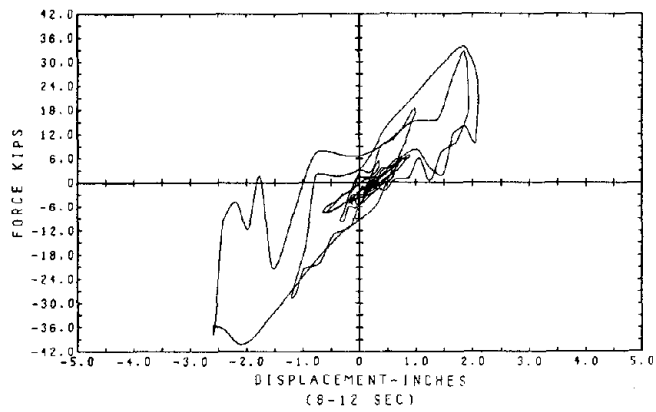
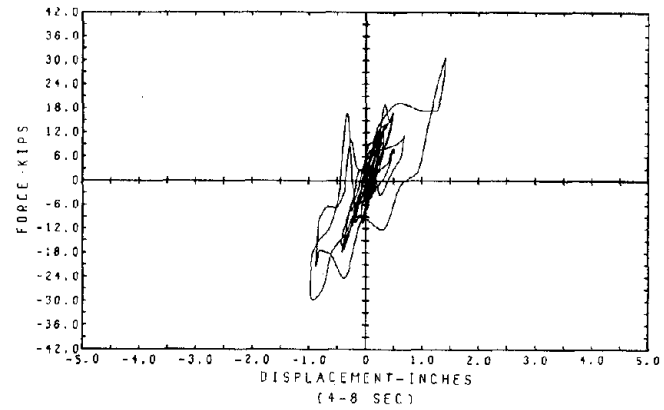
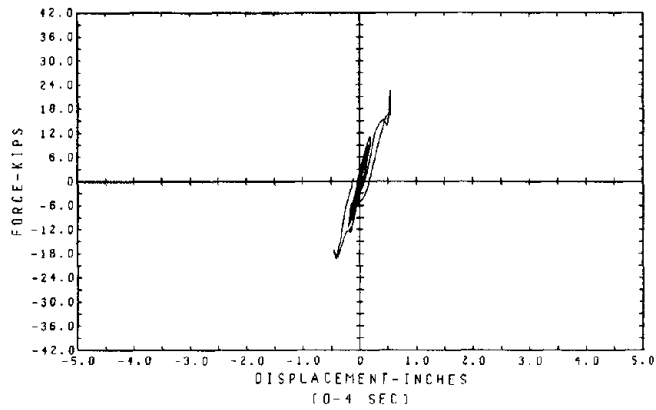
PACDIMA RECORD-1.314 G'S
TEST RESULTS OF DOUBLE ANGLE BRACING

Fig. 5.3c.3 Floor Drifts



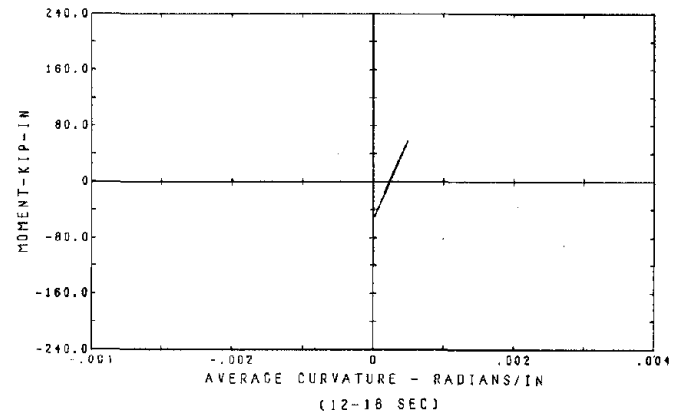
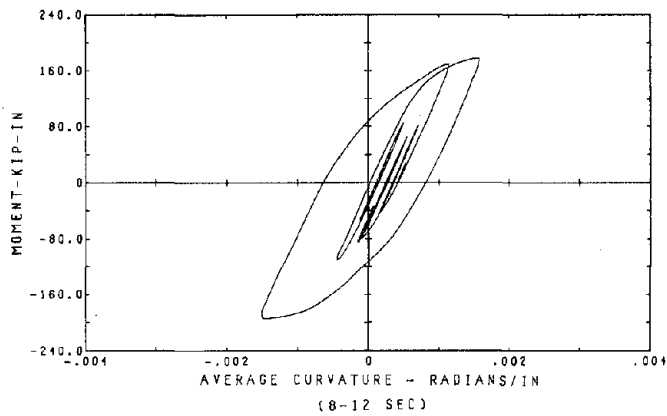
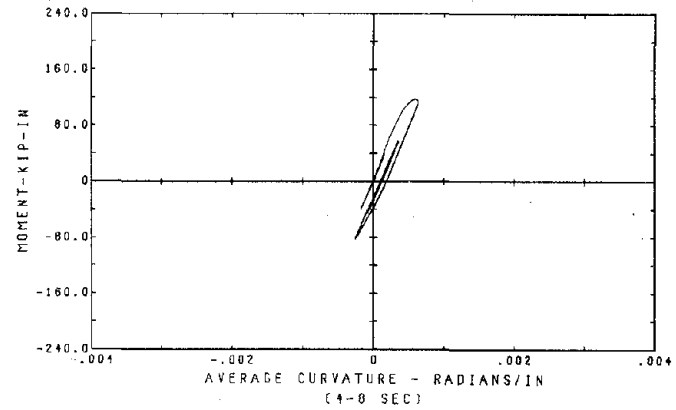
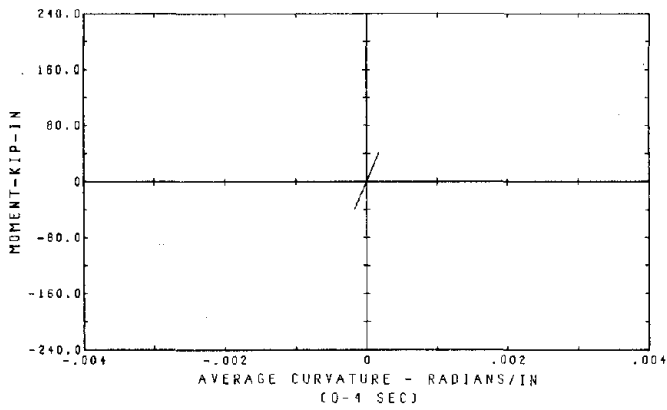
PACOIMA RECORD-1.314 G'S
TEST RESULTS OF DOUBLE ANGLE BRACING

Fig. 5.3c.4 Floor Shear Forces



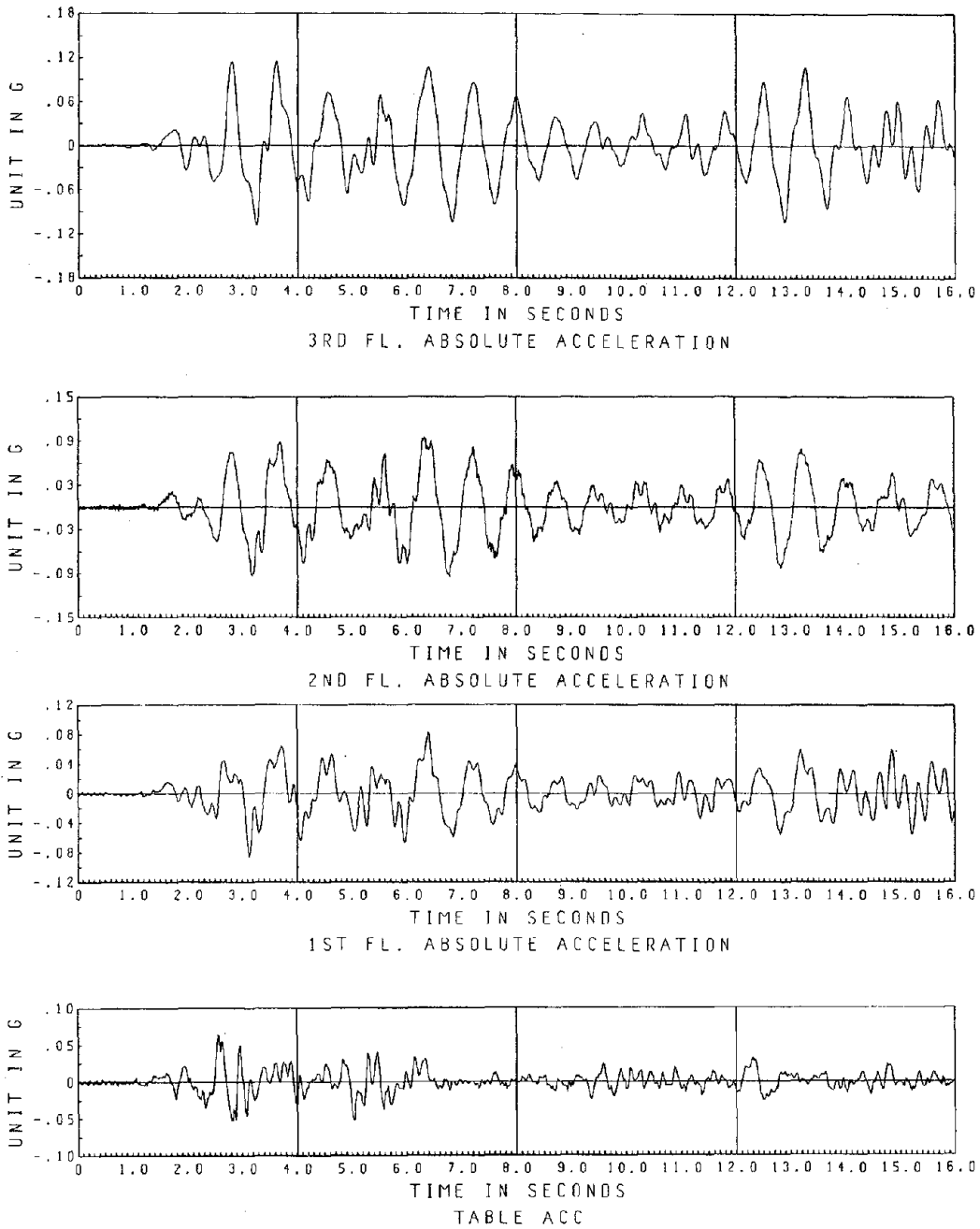
1ST FLR SHEAR VS. DISPLACEMENT
 REFERENCE FRAME N
 DOUBLE ANGLE DIAGONALLY BRACED STRUCTURE
 PACOIMA RECORD-1.314 G'S

Fig. 5.3c.5 First Floor Shear-Displacement Hysteresis Loops



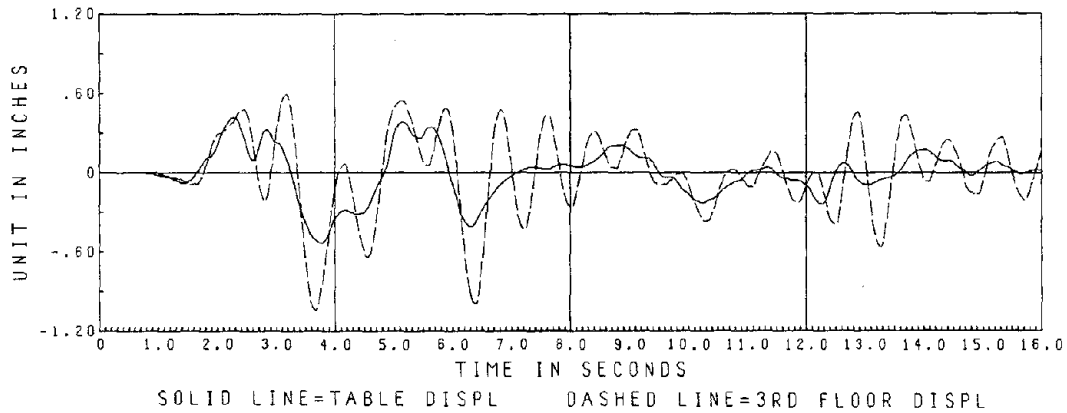
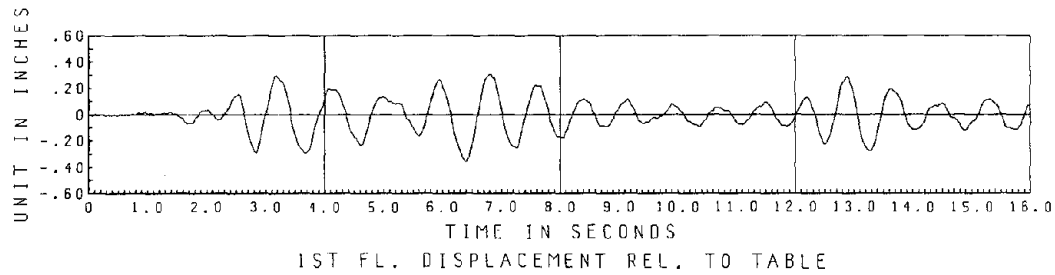
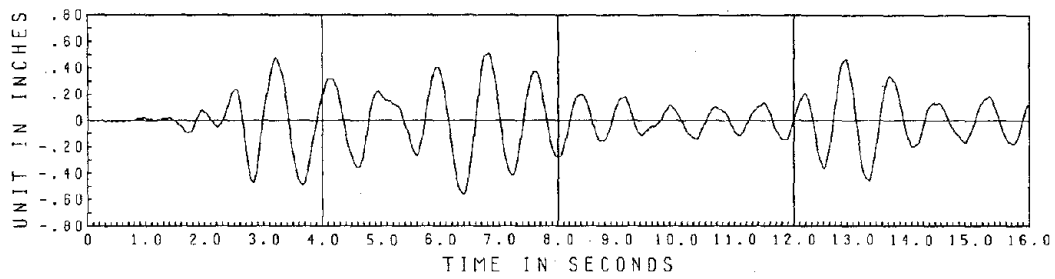
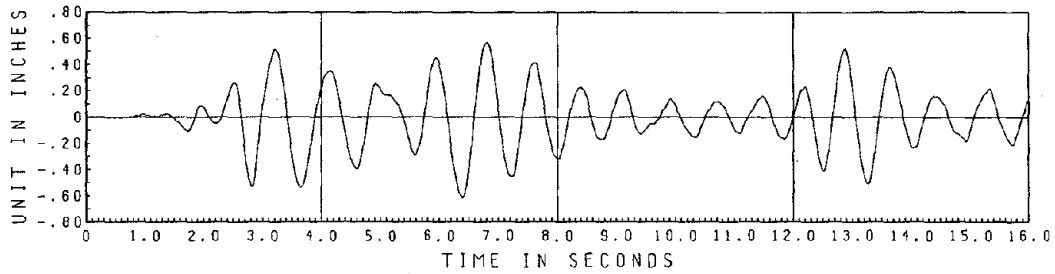
1ST FLR COL BOTTOM END , MOMENT VS. CURVATURE
 REFERENCE COLUMN LINE NA
 DOUBLE ANGLE DIAGONALLY BRACED STRUCTURE
 PACOIMA RECORD-1.314 G'S

Fig. 5.3c.6 First Floor Column Moment-Curvature Hysteresis Loops



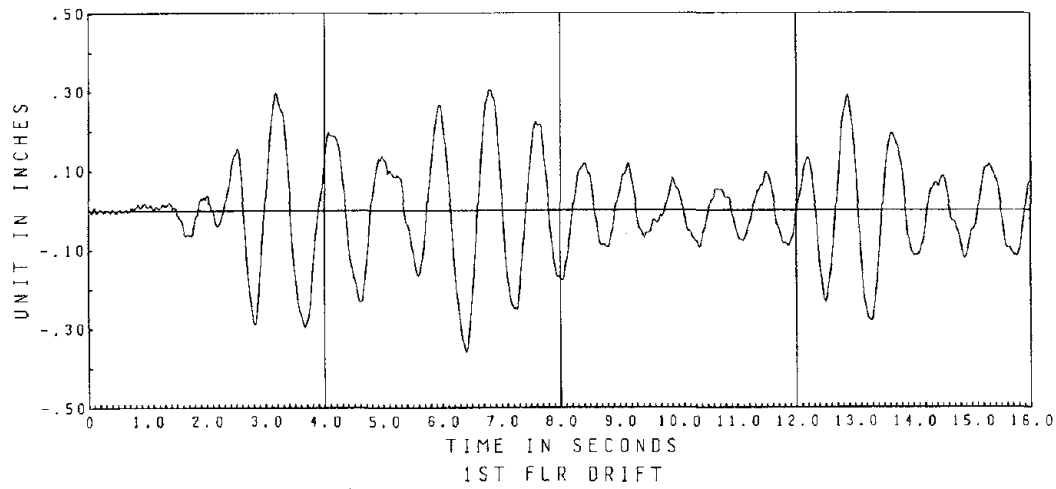
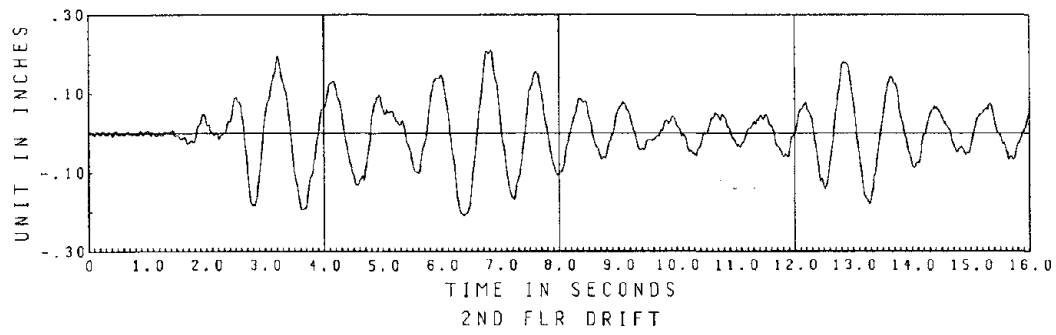
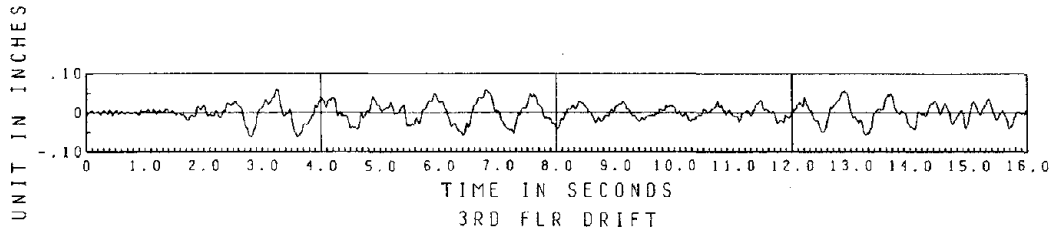
EL CENTRO SPAN 100
 TEST RESULTS OF THE UNBRACED STEEL FRAME

Fig. 5.4a.1 Table and Floor Accelerations



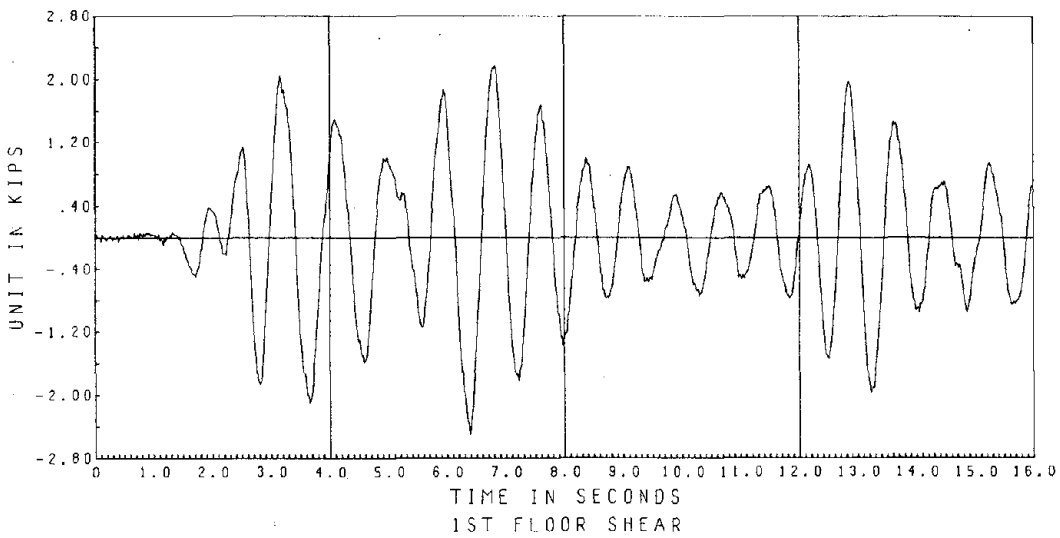
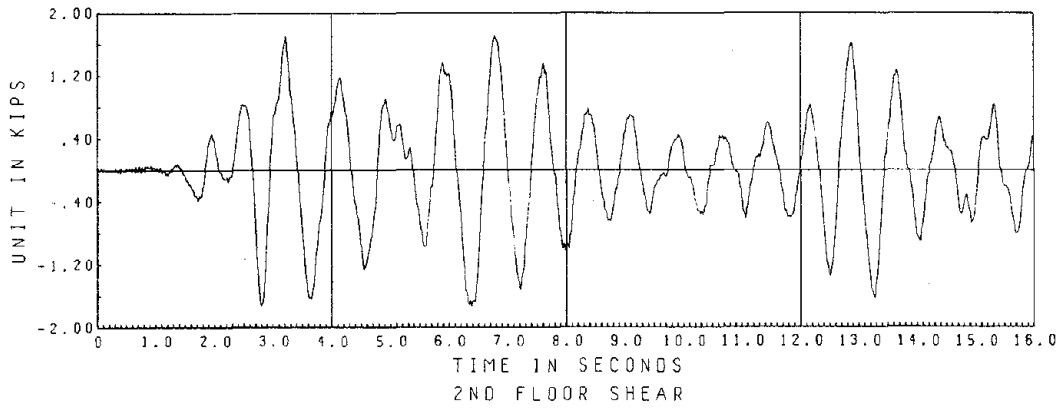
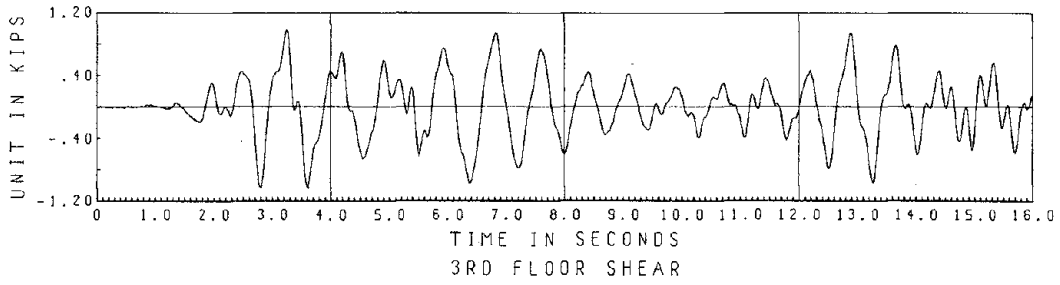
EL CENTRO SPAN 100
TEST RESULTS OF THE UNBRACED STEEL FRAME

Fig. 5.4a.2 Table and Floor Displacements



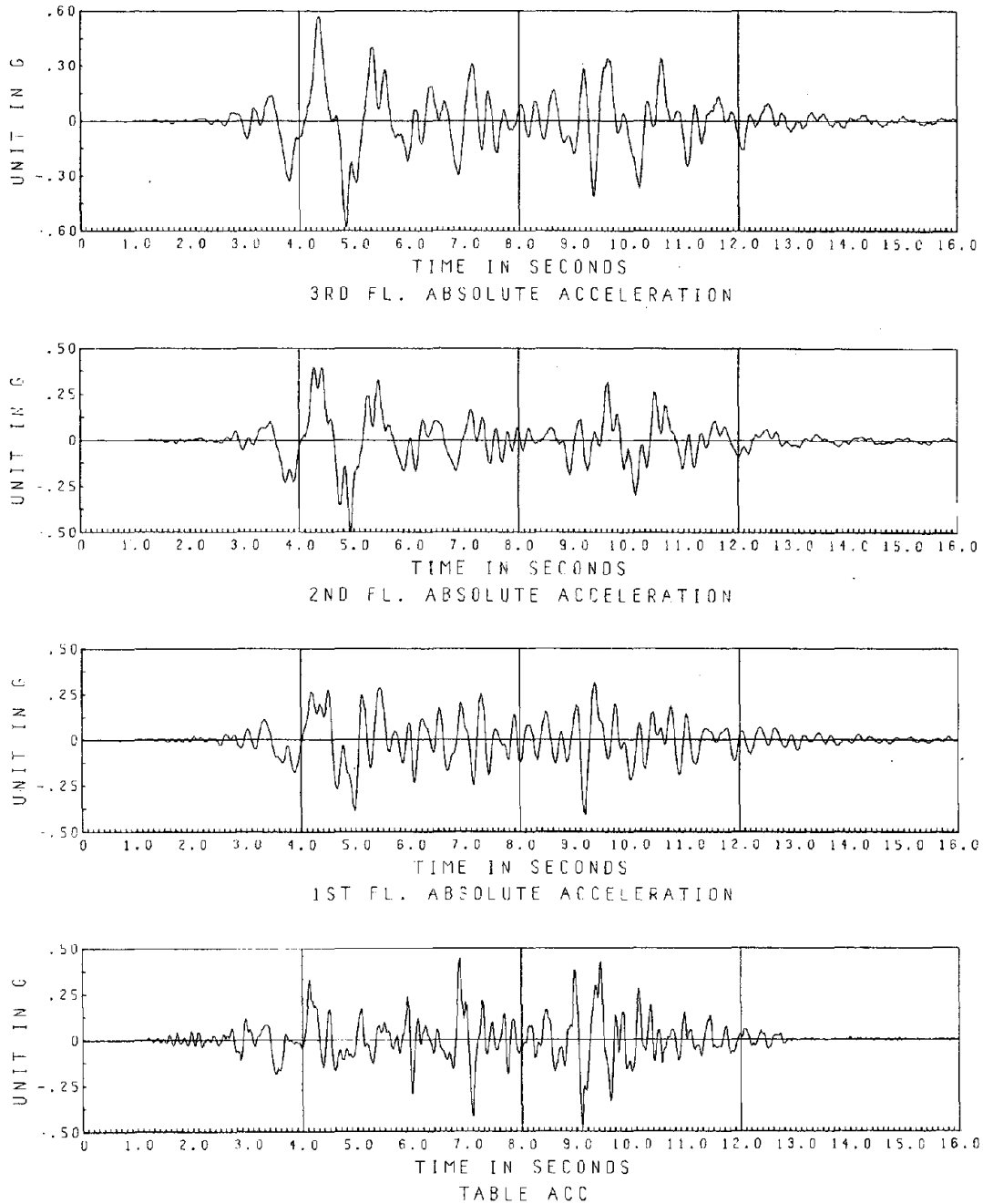
EL CENTRO SPAN 100
 TEST RESULTS OF THE UNBRACED STEEL FRAME

Fig. 5.4a.3 Floor Drifts



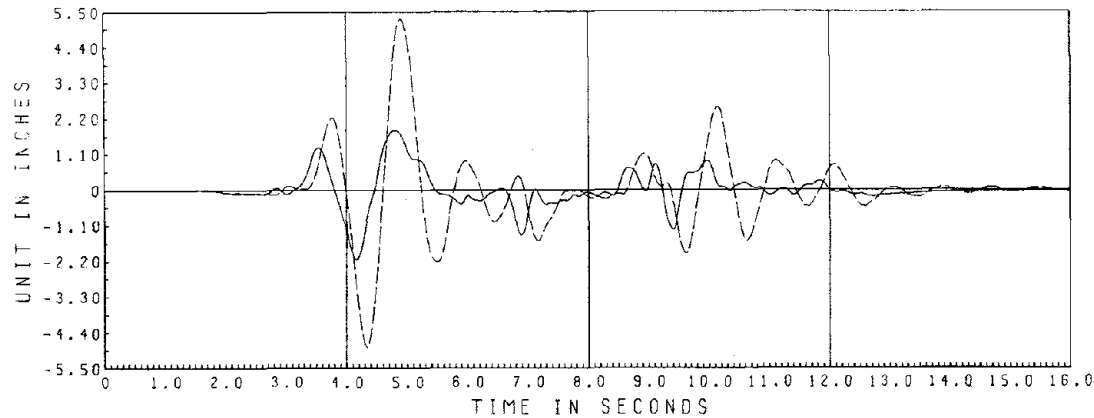
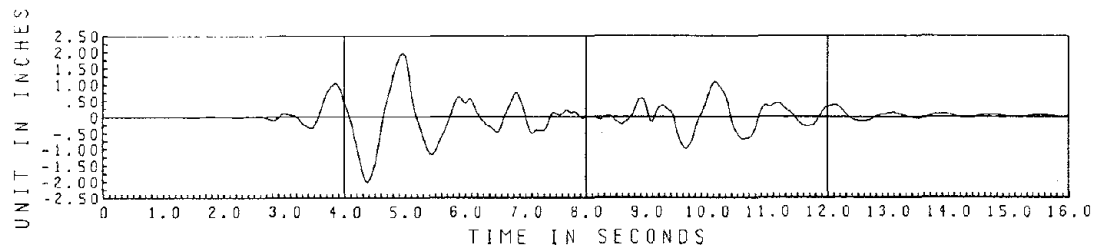
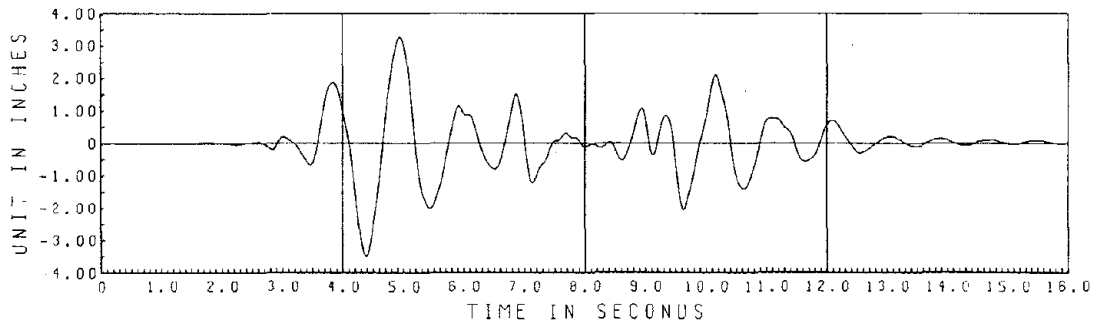
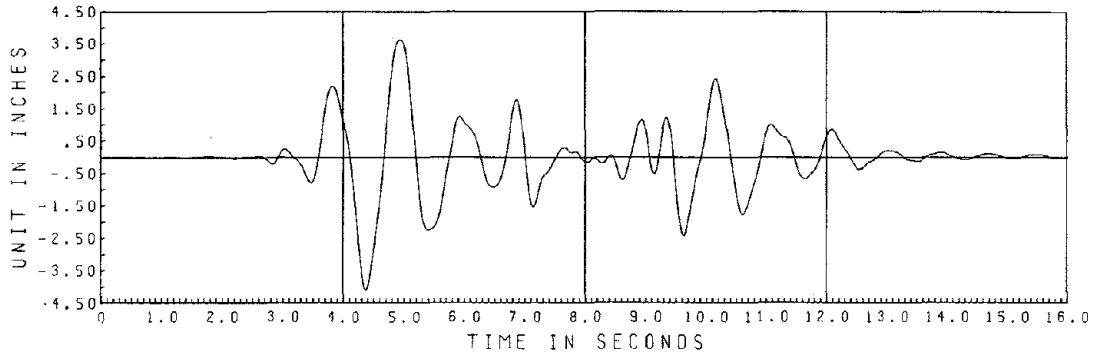
EL CENTRO SPAN 100
 TEST RESULTS OF THE UNBRACED STEEL FRAME

Fig. 5.4a.4 Floor Shear Forces



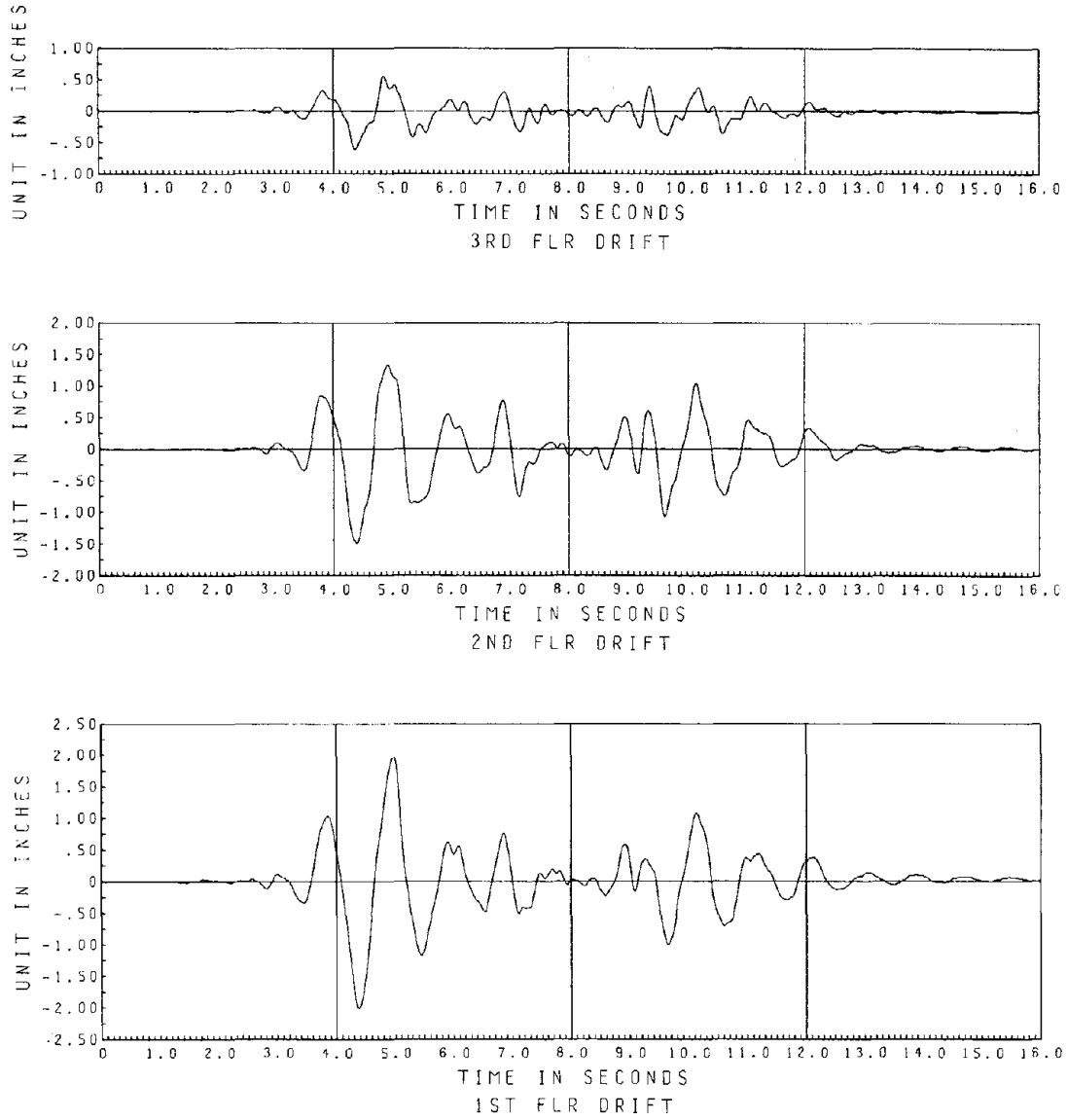
PACQIMA SPAN 400
TEST RESULTS OF THE UNBRACED STEEL FRAME

Fig. 5.4b.1 Table and Floor Accelerations



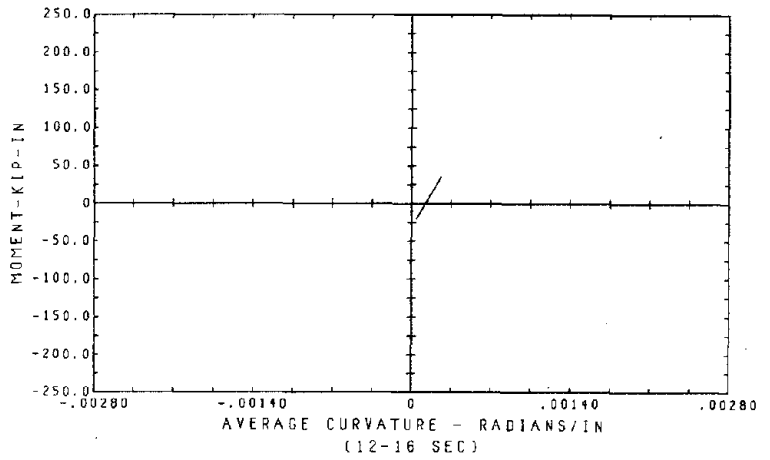
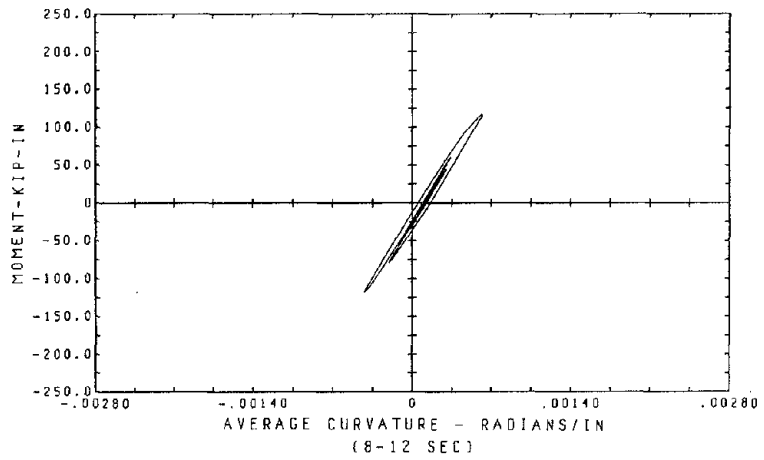
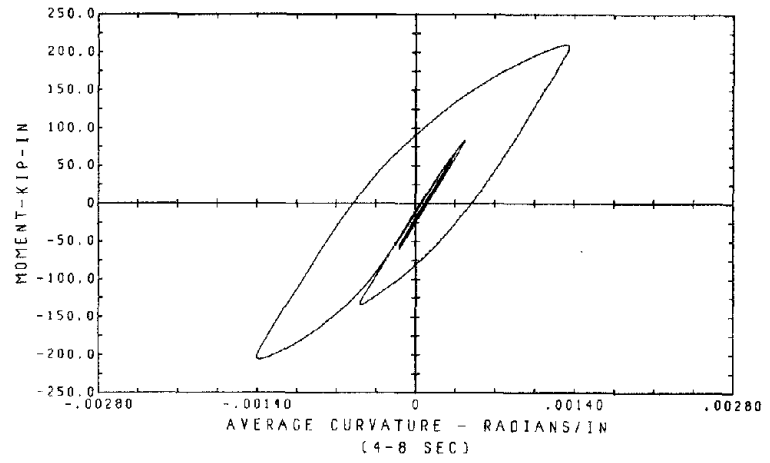
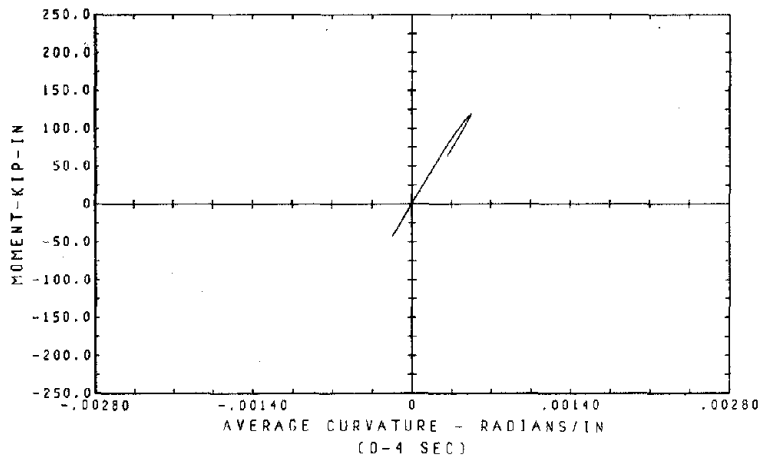
PACOIMA SPAN 400
TEST RESULTS OF THE UNBRACED STEEL FRAME

Fig. 5.4b.2 Table and Floor Displacements



PACDIMA SPAN 400
 TEST RESULTS OF THE UNBRACED STEEL FRAME

Fig. 5.4b.3 Floor Drifts



1ST FLR COL BOTTOM END, MOMENT VS. CURVATURE
 REFERENCE COLUMN LINE NA
 UNBRACED STRUCTURE
 PACOIMA SPAN 400

Fig. 5.4b.4 First Floor Column Moment-Curvature Hysteresis Loops

6. ANALYTICAL STUDY

One of the principal purposes of this investigation was to obtain actual response results for braced frame structures which would serve to demonstrate the effectiveness of an existing nonlinear structural program, in the analysis of diagonal bracing systems. The program under consideration is DRAIN-2D⁽⁸⁾, and of particular interest was the adequacy of the tension-compression bracing element included in that program. This program was selected because it is suitable for the inelastic dynamic analysis of plane frame structures, and allows for the addition of new element routines with no modification to the basic program. Accordingly, the post-buckling truss elements of Reference 7 and Reference 10 have been developed for use in this program. Other attractive features of the program are as follows: it includes semi-rigid connection elements; various yield interaction surfaces can be assumed for beam-column elements in defining plastic hinge mechanisms at the member ends; additional nodes can be specified along a member, so that spreading of plastic hinges can be studied; and, more than one member can be connected between two nodes so that a curvilinear load-deformation behavior can be approximated by the basic bilinear yielding mechanism.

The general purpose computer program DRAIN-2D, discussed fully by Kannan and Powell⁽⁸⁾, is for dynamic analysis of inelastic plane structures under identical in-phase motions of all support points. The analysis procedure makes use of the direct stiffness method with the nodal displacements as unknowns. The structure mass is assumed to be lumped at nodes each possessing up to three degrees of freedom. The earthquake excitations can be specified simultaneously in both horizontal and vertical directions by means of their acceleration time histories. Static

loads producing elastic structural response may be applied prior to the dynamic loading. The dynamic response is calculated by step-by-step integration of the equations of motion expressed in incremental form, assuming the acceleration to be constant within each step. The tangent stiffness of the structure is employed for each step assuming linear structural behavior during the time step. Unbalanced loads resulting from error due to the assumed linearity within the step are corrected in the subsequent time step. Note that greater accuracy can be obtained by selecting fairly short time steps to avoid large overshoots at instants of significant stiffness changes. Damping capabilities include optional combinations of mass-dependent, original stiffness-dependent, and tangent stiffness-dependent viscous damping.

6.1 Rod Bracing System

The two-dimensional frame model depicted in Fig. 6.1a.1, Model 1, was the first mathematical model developed for the rod bracing system. It was assumed that the column bases were fixed rigidly to the shaking table. The structure was discretized as nine beam-columns and twelve truss members interconnected by fourteen rigid points. The mass of the structural members and of the added concrete blocks was lumped at the nodes along the columns, and was associated with motions in the X-direction only. This model has five degrees of freedom per floor level; vertical displacement and rotation of each joint, and horizontal displacement of each floor. Nominal section properties and clear span dimensions were used in modelling all members; joint panel zones were assumed rigid. Bilinear flexural behavior was assumed for beams and columns; an axial force $P-\Delta$ effect was also considered in the first floor columns. Bracing properties are described separately in the discussion of the different mathematical models used for correlation with the experimental results.

6.1a Correlation with El Centro Span 100 - Model 1

In modelling the half-inch rod braces for the El Centro span 100 test, they were treated as composite axial force members consisting of two parts in series; solid rod, and turnbuckle. Based on static tests, the turnbuckles were assumed to behave elastically, while the rods were treated as bilinear yielding elements with very low compressive capacity (see Fig. 6.1a.2(a)). This is a standard DRAIN-2D element intended to model tensile yielding and elastic compression buckling. Static pretension loads applied to these members during the experiment were considered as initial static loads prior to the dynamic analysis. The

program DRAIN-2D does not consider the clear length of bracing members. Therefore, the actual flexibility of the braces was determined taking into account the rigidity of the end connections. The damping coefficient proportional to the initial stiffness was set at $\beta_0 = 0.00157$ leading to first mode damping of 2 percent of critical. The floor displacements calculated for this model are shown in Fig. 6.1a.3 with the measured floor displacements. The quality of this correlation is regarded as excellent considering that no system identification study was performed to determine the member properties.

6.1b Correlation with El Centro Span 1000 - Model 2

The previous model is not suitable for analysis of response to the El Centro motion with a peak acceleration of 0.775 g. Because of the pitching motion of the table, there would be significant interaction between the shaking table and the structure. The interaction of the shaking table in the dynamic response analysis was accounted for by providing vertical spring supports under the table to simulate the oil column flexibility of the hydraulic actuators.

Model 2, developed for this situation, is shown in Fig. 6.1b.1. The structure was discretized as ten beam-columns and twenty truss members interconnected by twenty-two rigid points. In modelling the half-inch rod braces for this model, they were treated as composite axial force members consisting of three parts in series; solid rod, threaded portion, and turnbuckle. The addition of the threaded portion was to localize the initial yielding of braces to these members as it actually occurred during the dynamic tests. Turnbuckles were assumed to behave elastically as before while both parts of the rod were considered as bilinear yielding elements with a very low compressive capacity (see Fig. 6.1a.2(a)). All

other assumptions for Model 1 are also valid in this case.

The stiffness-dependent damping coefficient was set at $\beta = 0.006$ for this analysis, leading to first mode damping of 5 percent critical. The area and length of the vertical springs under the shaking table were set arbitrarily at 10 square in. and 100 in., respectively. Thus, only the Young's modulus of the spring needed to be specified during data correlation of the model. Correlations between analytical results obtained with this model and the experimental results are presented in Fig. 6.1b.2 and Fig. 6.1b.3 for both global and local quantities. This correlation is excellent considering that tension yielding led to an impacting type of response in the resulting slack rod system. (Note that the Young's modulus of the vertical spring was set at 1800 ksi in this analysis.)

6.1c Correlation with the Pacoima Earthquake - Model 3

To model the rod bracing behavior in the Pacoima Dam test with a peak acceleration of 1.129 g, it was necessary to revise the element properties to account for rupture. In this case, the two resisting frames were represented as a single frame, but the rod braces from the two frames were treated as independent parallel members of appropriate uniform section. The different yield levels for threaded and solid sections were simulated by parallel members having different bilinear properties, resulting in a tri-linear mechanism. Also, the standard DRAIN-2D element was modified so that the stiffness became zero when the rupture load was reached (see Fig. 6.1a.2(b)); simultaneously the element force was transferred to the end nodes as an unbalanced load. The definition of the stiffness-dependent damping matrix for this model was the same as for Model 2, and the shaking table-structure interaction was also included; the modulus of the vertical spring was set at 1500 ksi. Correlation

between results obtained with this model and the experimental results described in Chapter 5.1d are shown for both global and local response quantities in Figs. 6.1c.1 and 6.1c.2, respectively. The model successfully predicts the rupture of the first and second floor braces and the time at which they occurred, but it is evident that the correlation is not as good as was obtained with Model 2 for the El Centro test. However, the analytical estimates of story displacements and column moments and shears are considered adequate; the high frequency "noise" in the experimental shear values is due to damage to the accelerometer attachments. Also, there is good reason to believe that the main source of discrepancy in the analysis is associated with inadequacy of the mathematical model for the bracing members in representing the impacting type response that developed in the slack-rod system during the more intense tests.

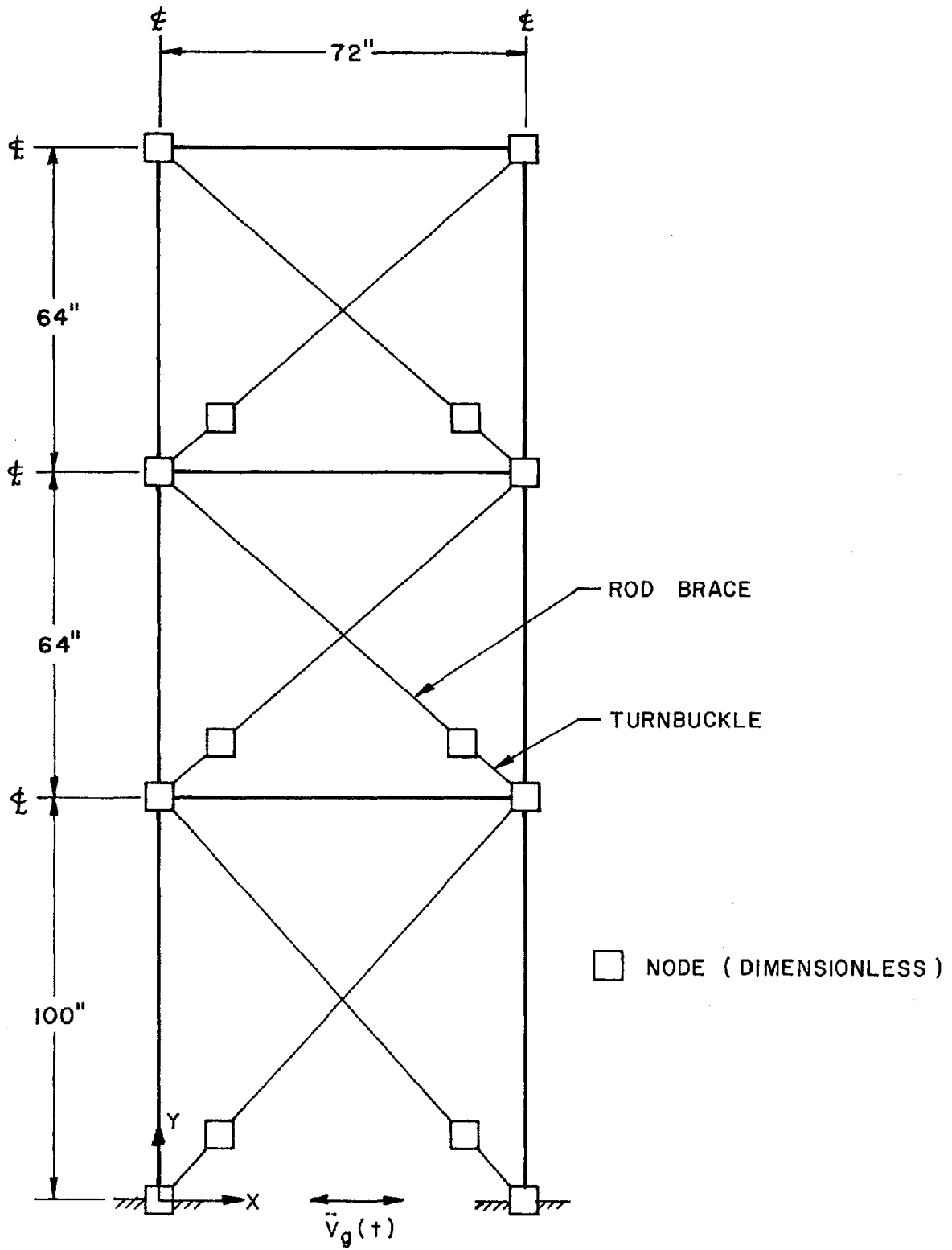
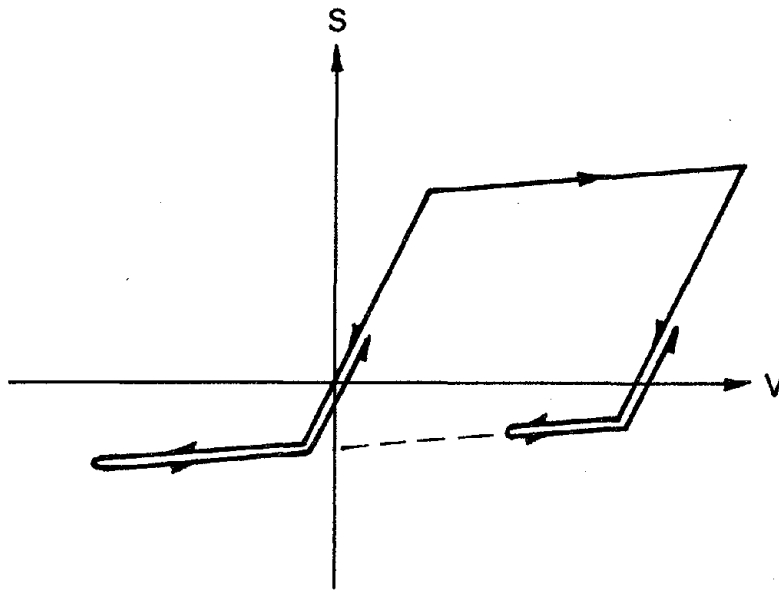
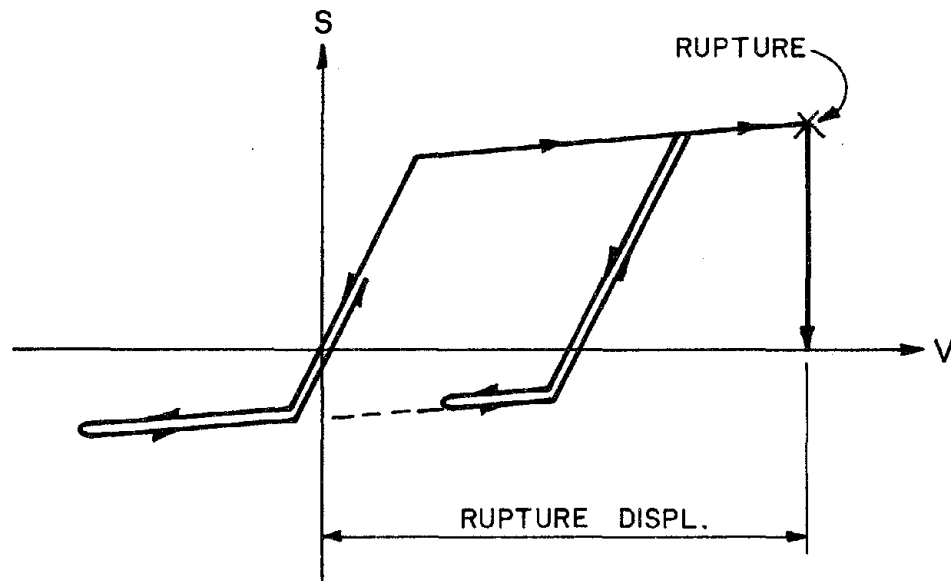


Fig. 6.1a.1 Mathematical Model 1 With Rod Diagonal Braces

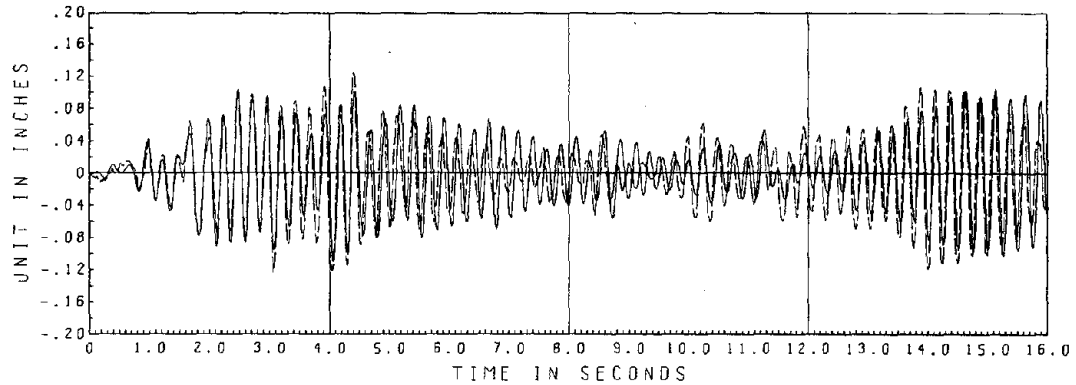


(a) Yield in Tension, Buckling in Compression

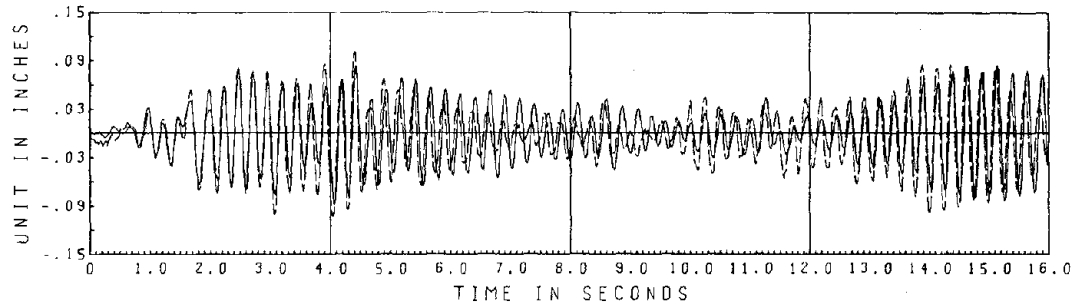


(b) Yield and Rupture in Tension, Buckling in Compression

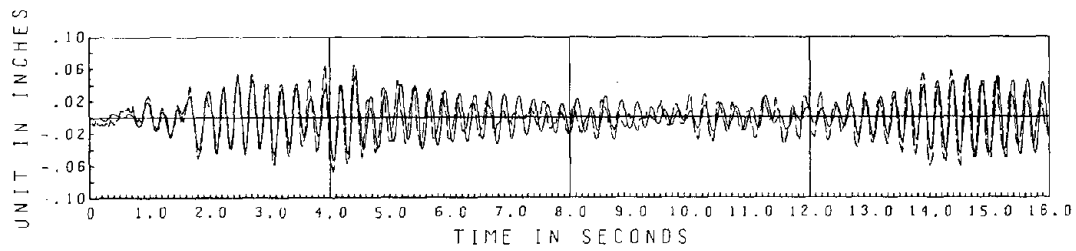
Fig. 6.1a.2 Force-Displacement for Rod Bracing Members



3RD FL. DISPLACEMENT REL. TO TABLE



2ND FL. DISPLACEMENT REL. TO TABLE



1ST FL. DISPLACEMENT REL. TO TABLE

EL CENTRO SPAN 100
 ROD DIAGONALLY BRACED FRAME
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.1a.3 Correlation of the Floor Displacements - Model 1

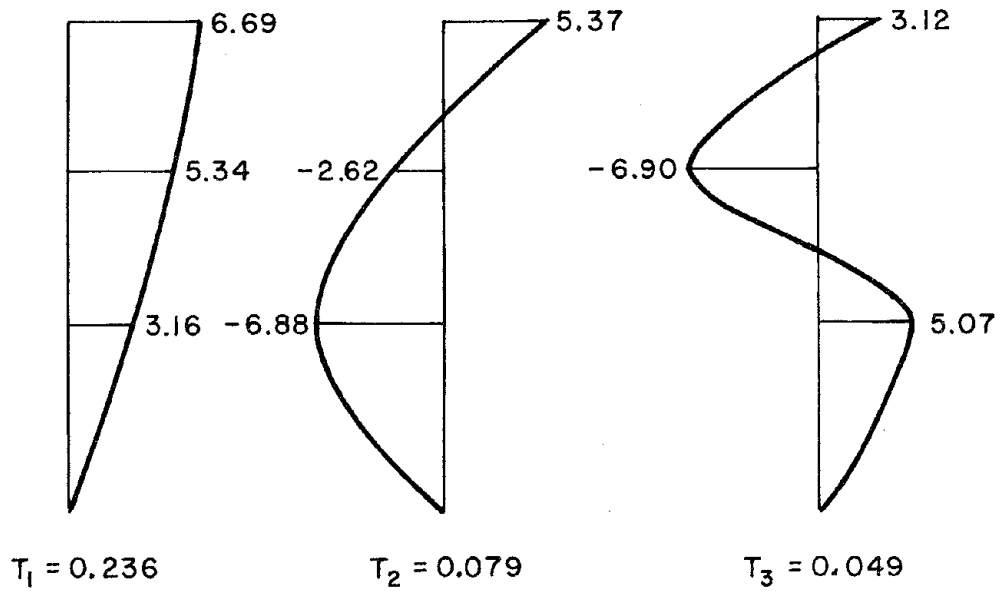
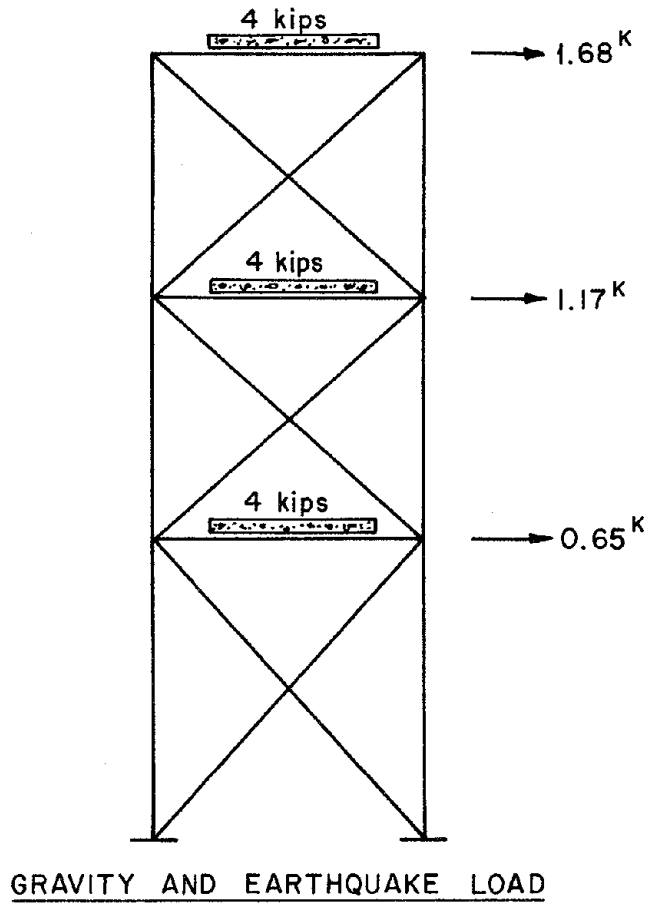


Fig. 6.1a.4 Calculated Periods and Mode Shapes
(Rod Braced Structure)

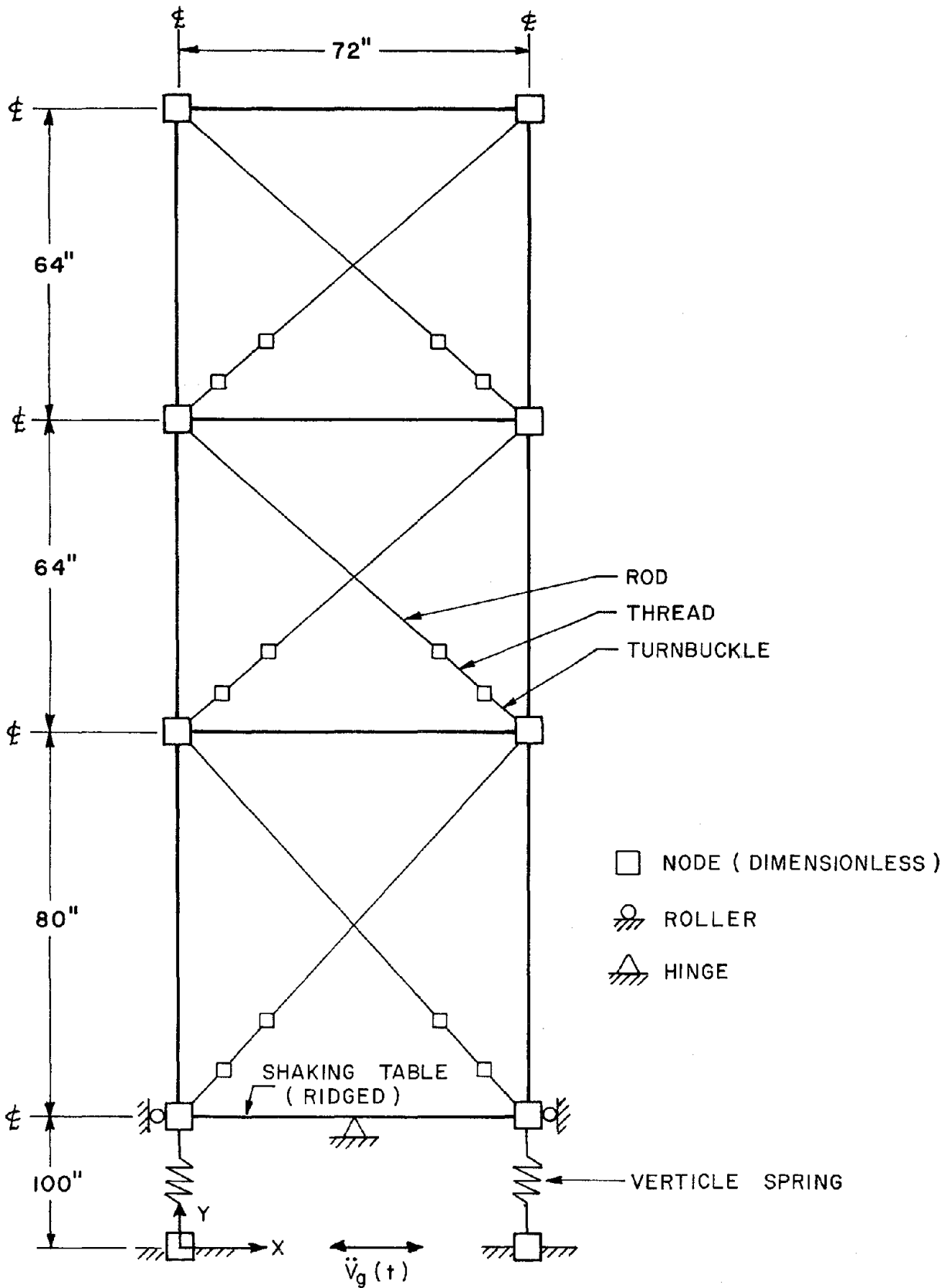
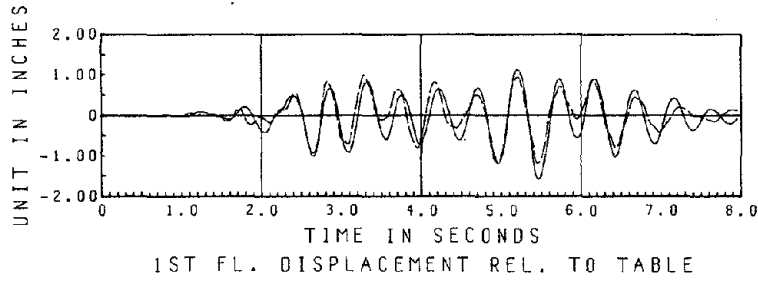
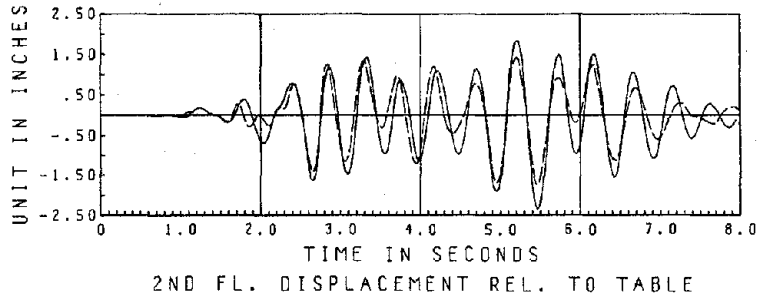
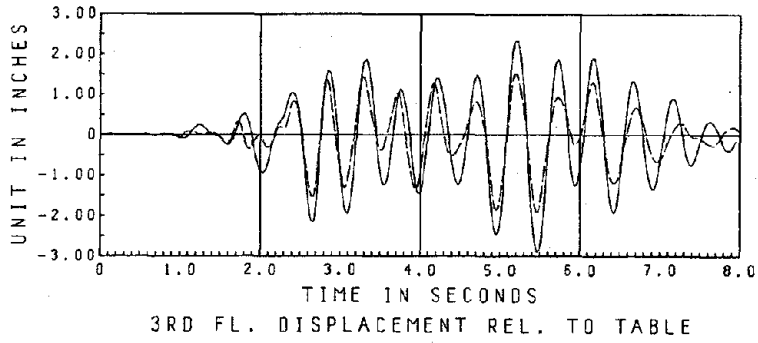
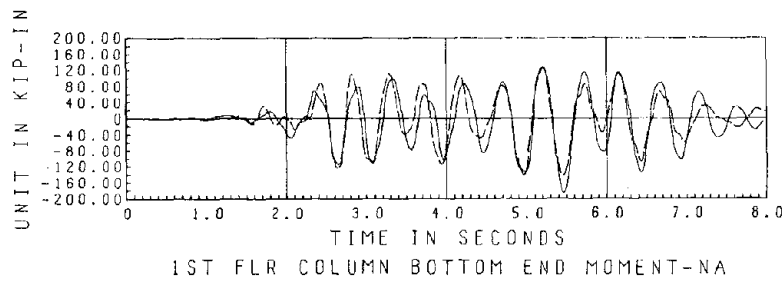
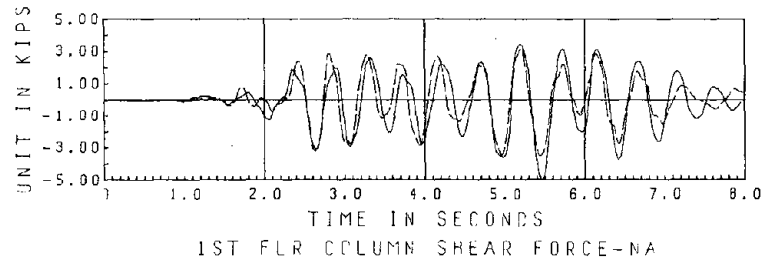
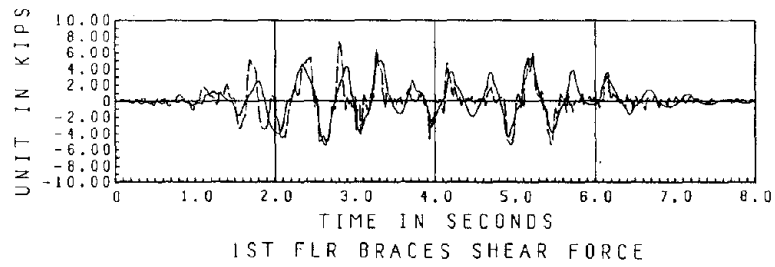
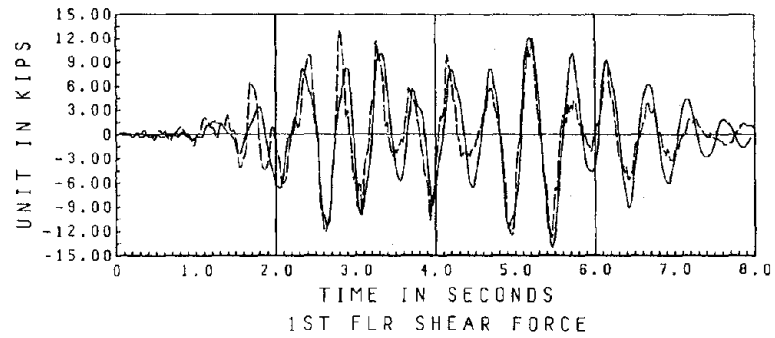


Fig. 6.lb.1 Mathematical Model 2
(shaking table-structure
interaction included)



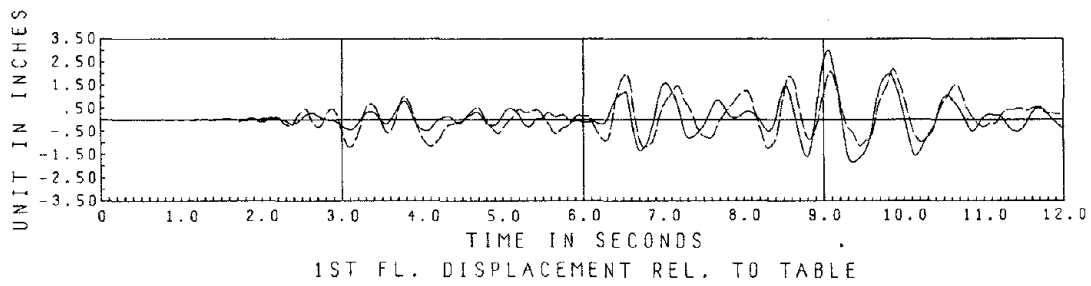
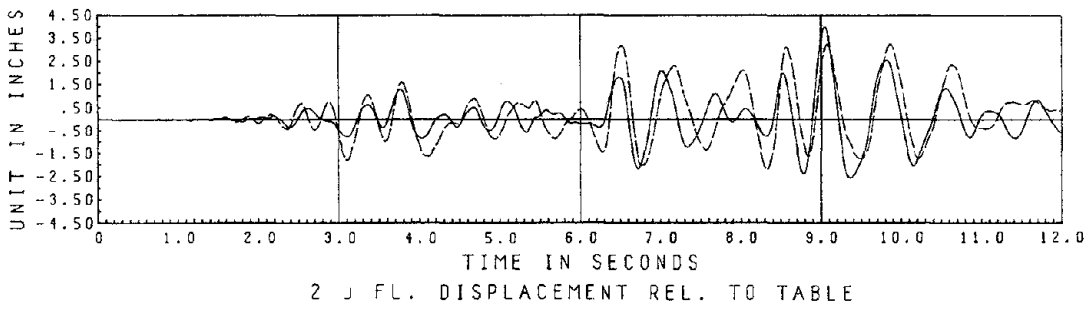
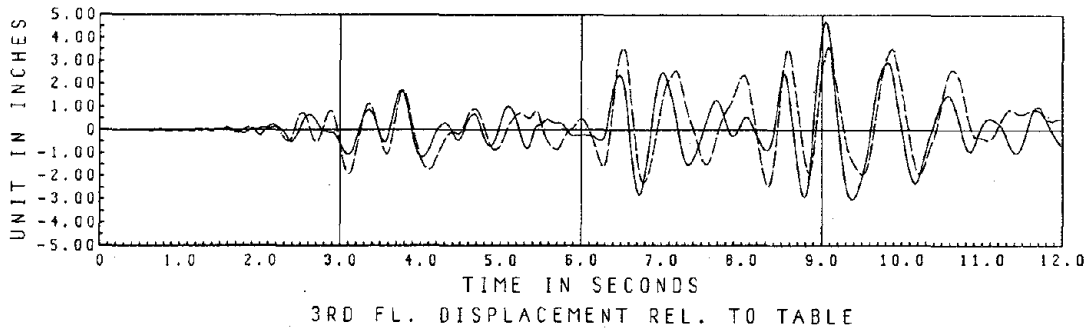
ROD DIAGONALLY BRACED FRAME
 EL CENTRO SPAN 1000
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.lb.2 Correlation of Floor Displacements - Model 2



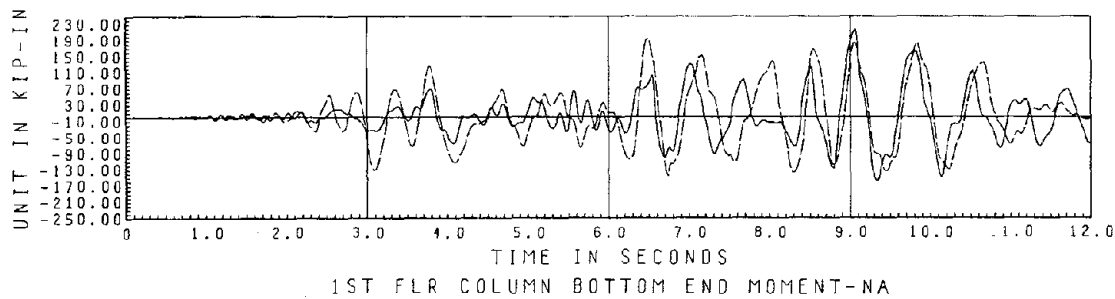
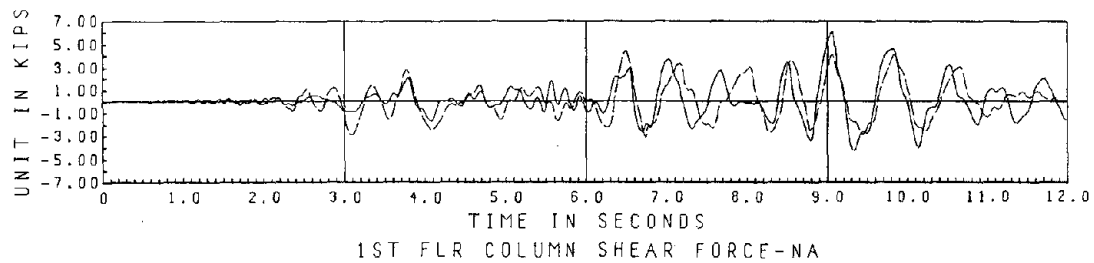
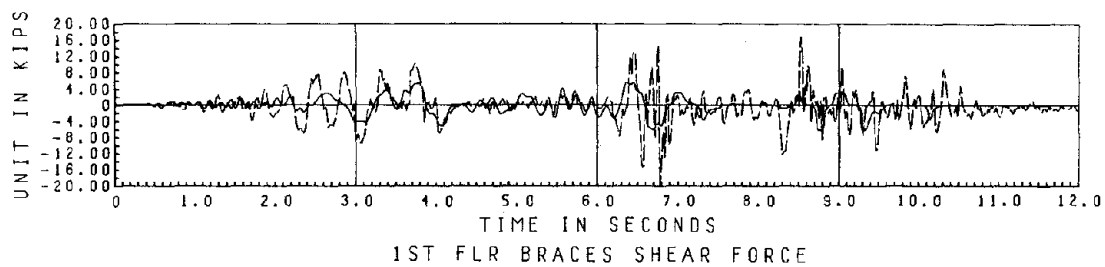
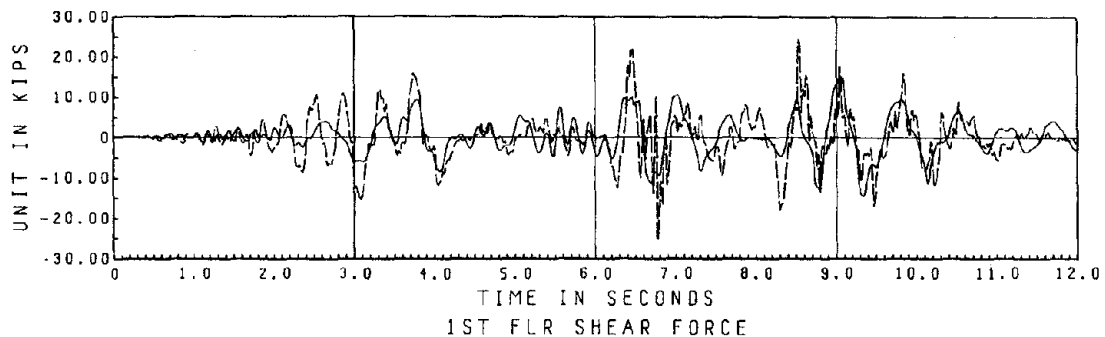
ROD DIAGONALLY BRACED FRAME
 EL CENTRO SPAN 1000
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.1b.3 Correlation of Global and Local Forces - Model 2



PACOIMA RECORD-1.129 G S , ROD DIAGONALLY BRACED FRAME
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.1c.1 Correlation of Floor Displacements - Model 3



PACOIMA RECORD-1.129 G S , ROD DIAGONALLY BRACED FRAME
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.lc.2 Correlation of Global and Local Forces - Model 3

6.2 Pipe Bracing System

The experimentally determined force-displacement relationship of the pipe bracing members, obtained in the El Centro test, is given in Fig. 5.2b.4. These hysteresis plots indicate that pipe braces with intermediate slenderness ratios have a significant compression strength and can dissipate energy in their post-buckling region. The model used in the analysis of rod braces does not reflect the actual behavior of pipe braces after they have buckled, thus it cannot be used in the inelastic analysis of these elements. Similarly, a brace mechanism which yields in compression, as assumed by the DRAIN-2D truss element, greatly overestimates the ability of a brace to dissipate energy and is not suitable for the pipe braces.

An accurate brace mechanism should include the post-buckling displacement behavior of pipe braces; such an element was developed originally by Nilforoushan⁽⁶⁾ and then modified by Roeder and Popov⁽⁷⁾. The brace model of Reference 7 is a linear approximation of the true behavior of bracing members as shown in Fig. 6.2.1, and has been adapted for use in the program DRAIN-2D. In this model, nine linear zones which are defined by the strain history and other critical parameters, are used in the approximation. The critical parameters are input values and they are specified by experimental results, by theoretical derivation or by other acceptable means. Although this general model can be used in the modeling of pipe braces, it was felt that a simplified version is more useful for practical purposes. Therefore, using this general model, a model with a smaller number of linear zones, shown in Fig. 6.2.2, was developed. The results of analytical studies based on this model are presented in the following subsections.

6.2a Correlation with El Centro Span 400 - Model 4

The mathematical model used to represent the pipe braces (Fig. 6.2.2) includes the hysteresis effects of both tension yielding and compression buckling. The critical input parameters were determined from experimental results. The model of the structure with pipe braces, depicted in Fig. 6.2.3, is similar to the previous models except for the braces, but with the addition of dynamic axial force $P-\Delta$ effects which were included in the second floor columns. This addition was made because of the larger axial forces induced in these columns by the bracing members.

In this analysis, the damping coefficient proportional to the initial stiffness was set at $\beta_0 = 0.007$ to obtain the desired first mode damping ratio of 6 percent. Also, a stiffness of 400 kips/in. was selected for the vertical shaking table spring supports to obtain a proper frequency match. Correlation between the analytical results obtained with this model and experimental data of the El Centro span 400 test is shown in Figs. 6.2.4 and 6.2.5. Agreement between analysis and experiment is considered to be good, especially considering that significant buckling occurred in the braces, as shown in Fig. 5.2b.4. Thus the modified bracing member hysteresis mechanism proved to be fairly good.

6.2b Correlation with Pacoima Span 400

To predict response to the Pacoima test with span 400, the previous model (Model 4) was employed with minor adjustments. The main change was associated with the buckling capacity of the first floor pipe braces. Because the braces had suffered buckling and yielding in earlier tests, it was determined that the residual buckling strength of the damaged first floor braces was only 3 kips.

In this analysis, the damping coefficient proportional to the initial

stiffness was set at $\beta_0 = 0.004$ corresponding to a first mode damping ratio of 3 percent and the stiffness of the shaking table compliance springs was estimated to be about 300 kips/in. Correlation between the results computed with this model and the experimental data of the Pacoima test are shown in Figs. 6.2.6 and 6.2.7. Agreement between analysis and experiment is considered excellent, particularly considering that the first floor braces buckled repeatedly and consequently induced a significant pinching effect, as shown in Fig. 5.2c.5. In addition, significant yielding occurred in the first floor columns. The greatest discrepancy in the analytical results is in the shear forces; this deviation resulted from assuming the bracing stiffness to vanish after tensile yielding occurred.

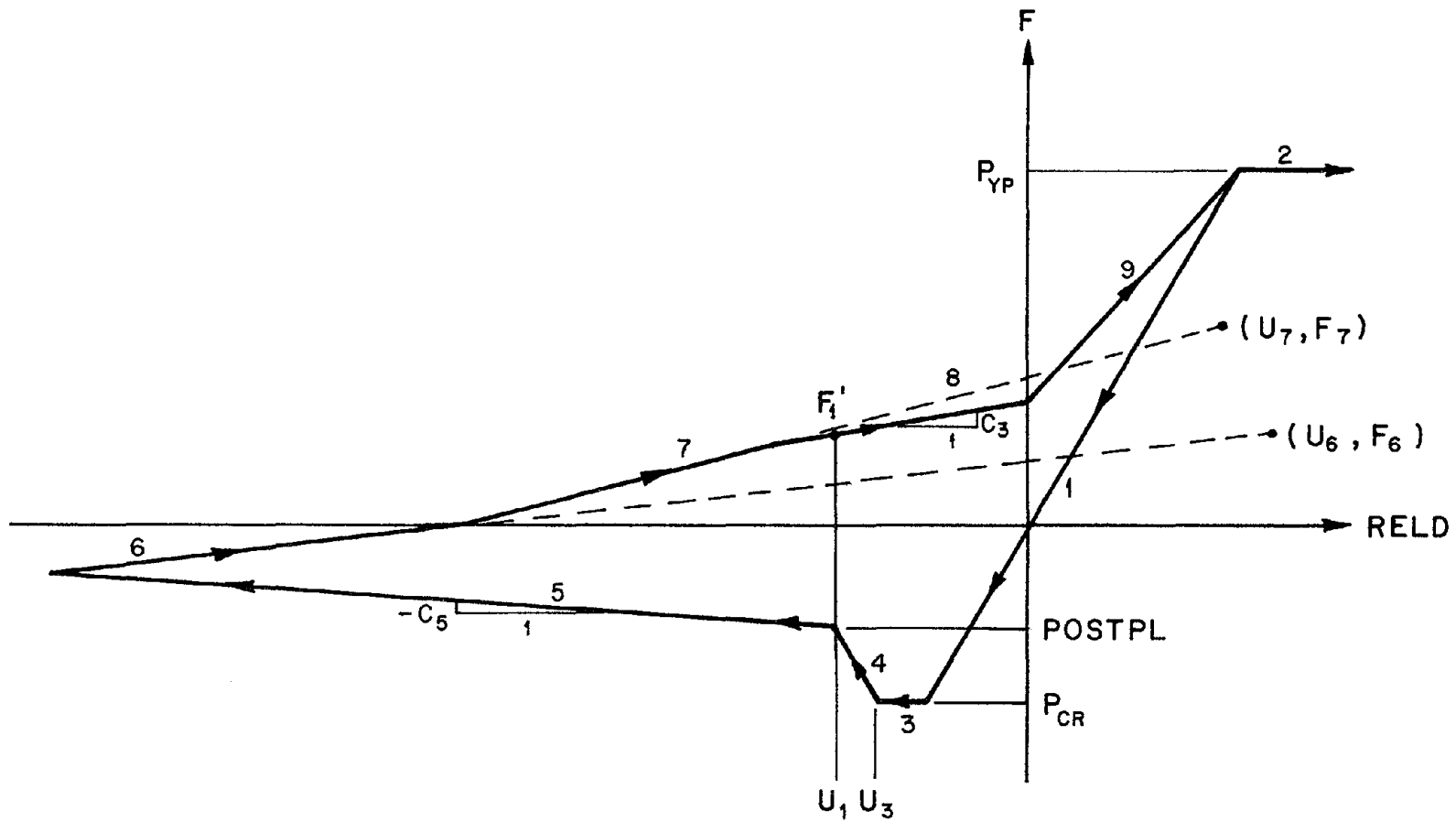


Fig. 6.2.1 Force-Displacement of Post-Buckling Brace Element
(after Roeder and Popov)

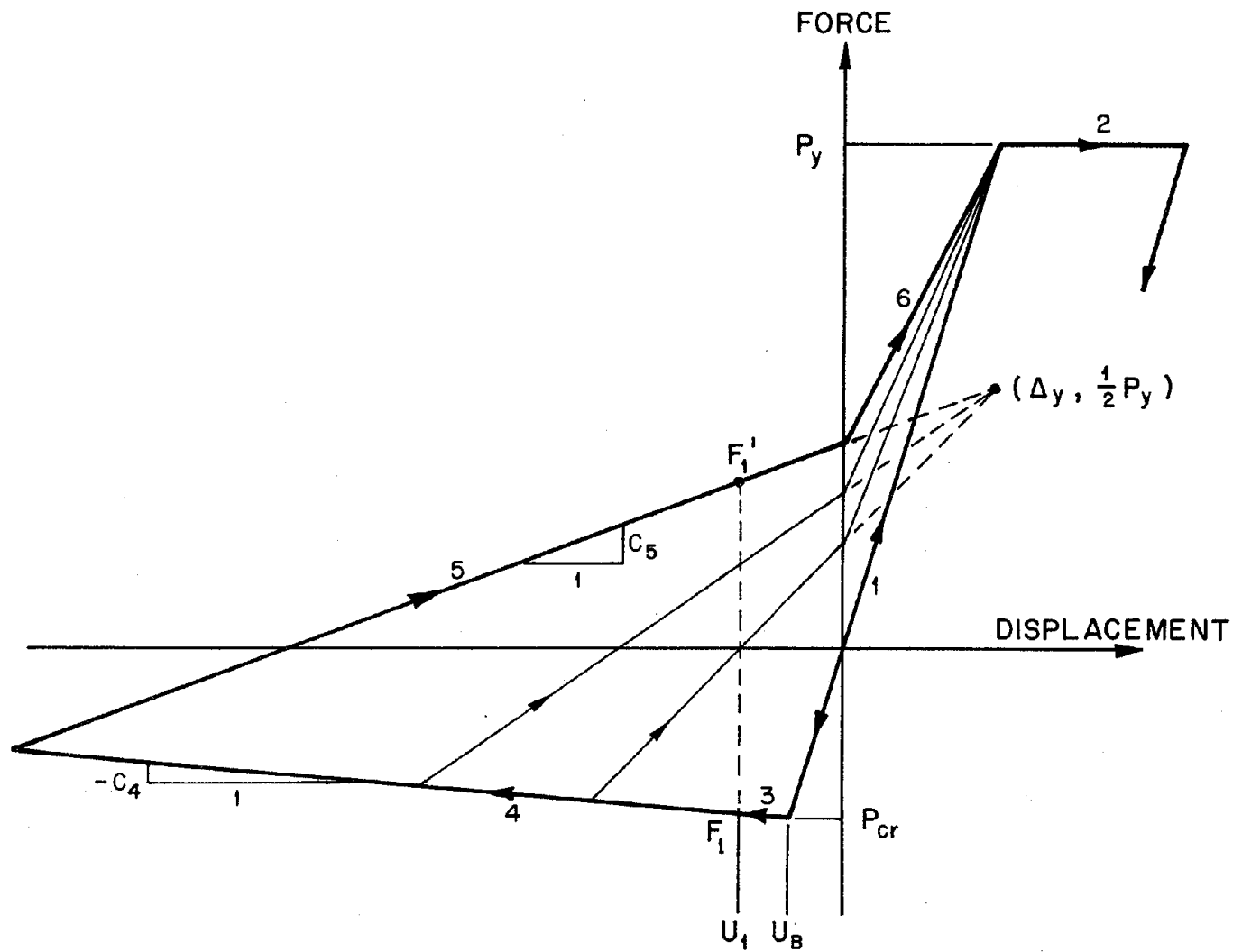


Fig. 6.2.2 Proposed Force-Displacement Model for Pipe Bracing Members

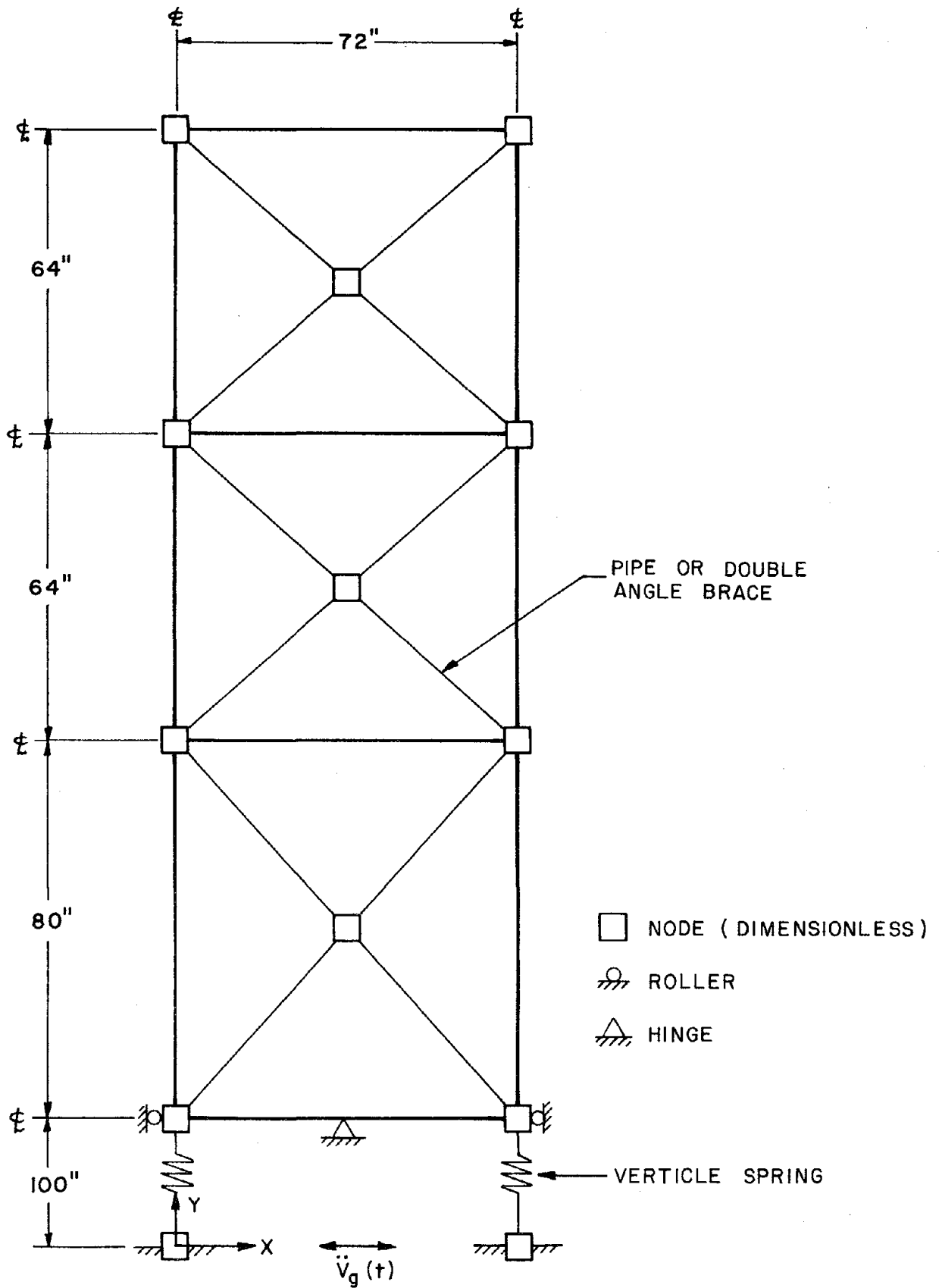
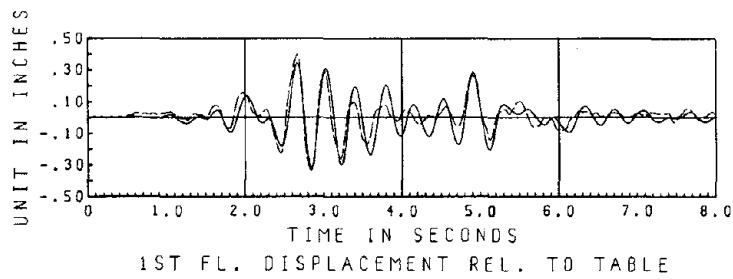
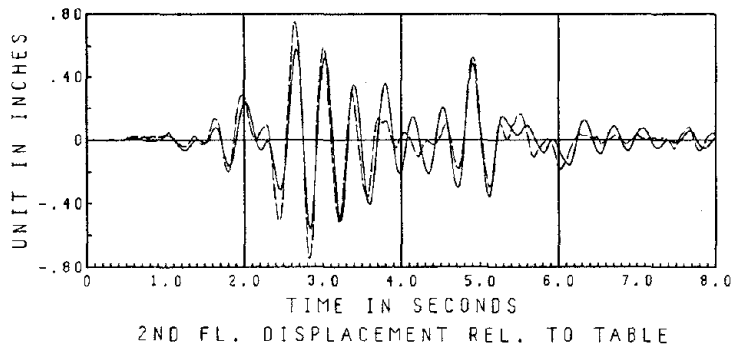
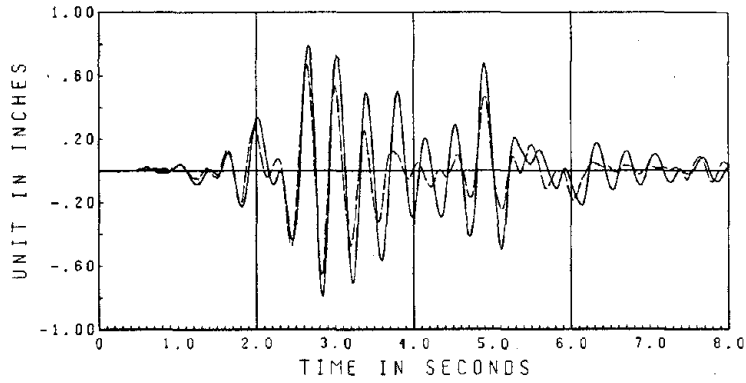
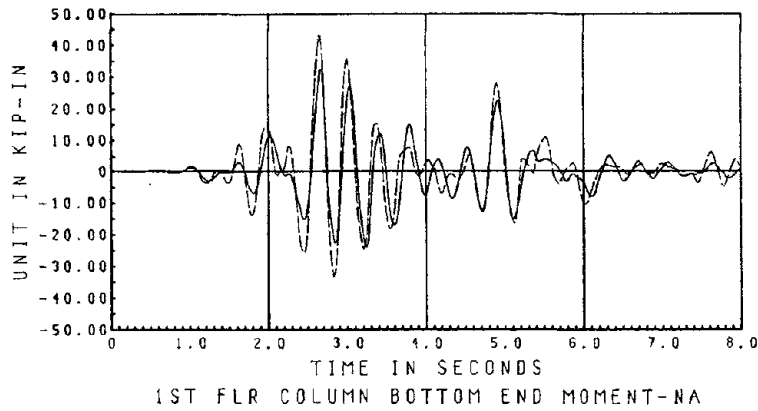
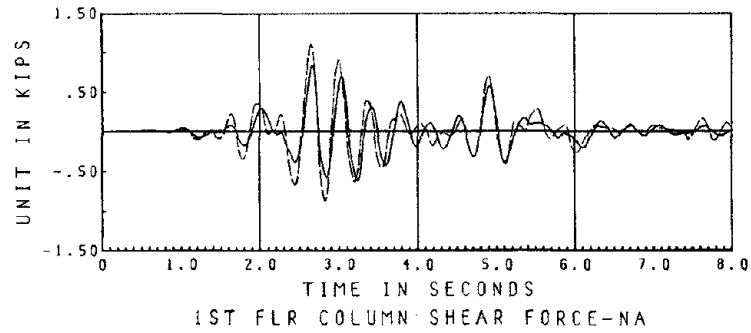


Fig. 6.2.3 Mathematical Model for Pipe and Double Angle Braced Structures



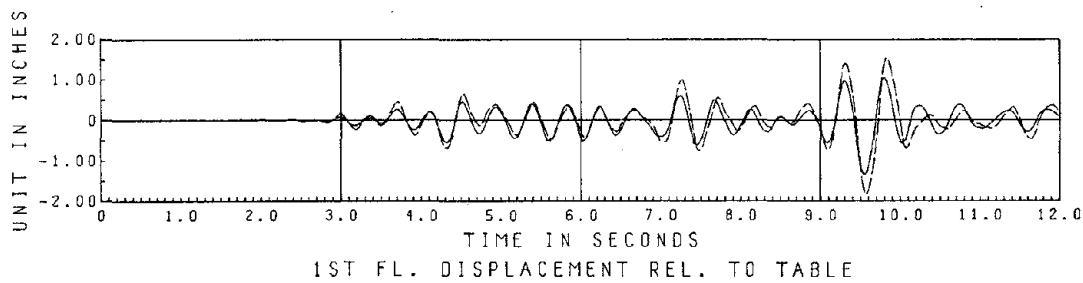
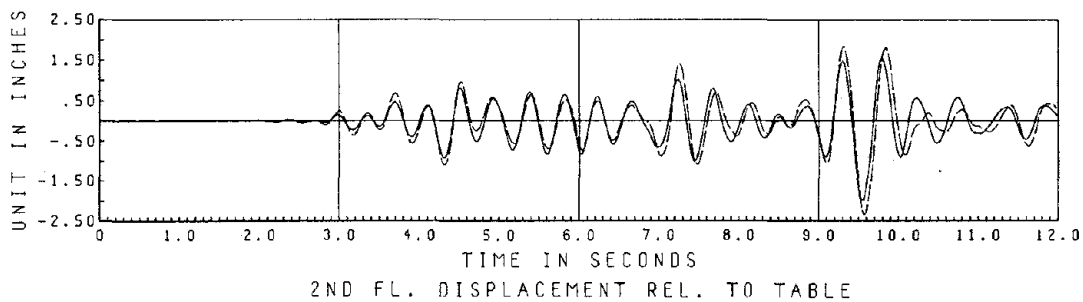
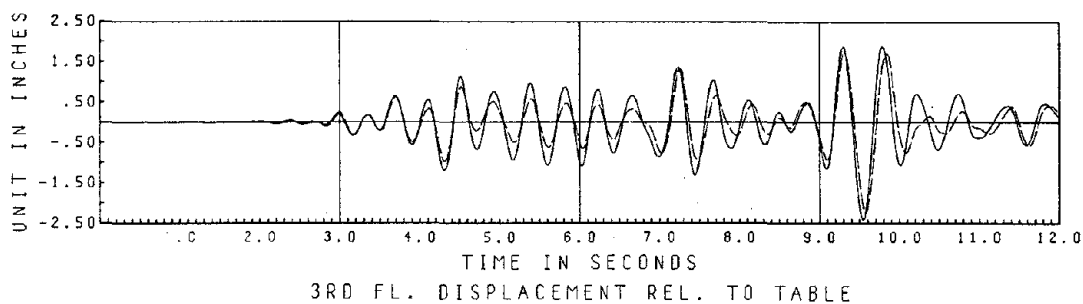
EC SPAN 400 , PIPE DIAGONALLY BRACED FRAME
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.2.4 Correlation of Floor Displacements



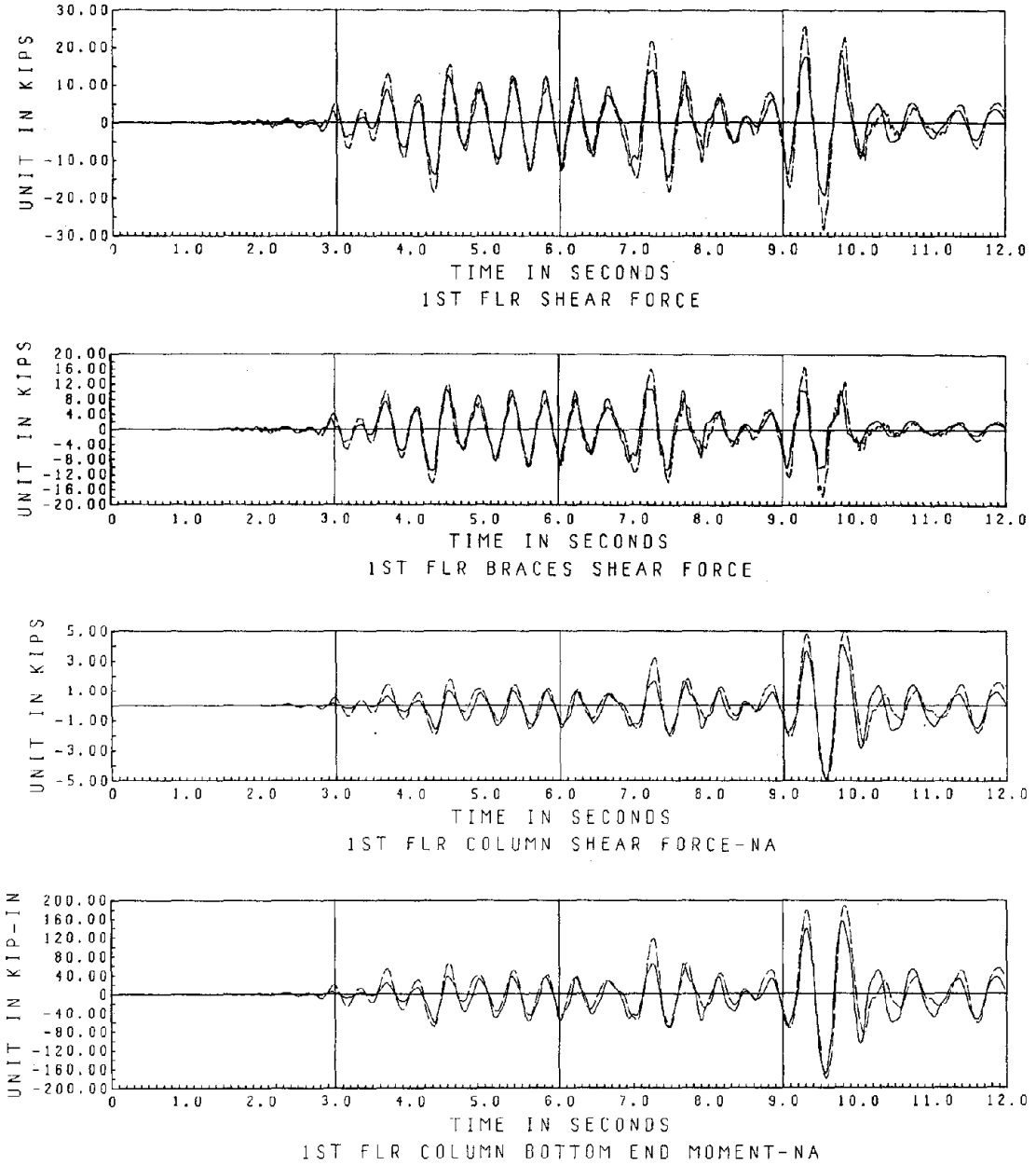
EC SPAN 400 , PIPE DIAGONALLY BRACED FRAME
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.2.5 Correlation of Column Forces



PAC 400 , PIPE DIAGONALLY BRACED FRAME
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.2.6 Correlation of Floor Displacements



PAC 400 , PIPE DIAGONALLY BRACED FRAME
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.2.7 Correlation of Global and Local Forces

6.3 Double Angle Bracing System

The mathematical model developed for the structure with double angle braces was similar to the pipe brace model (Model 4) except for the hysteresis mechanism of the bracing members. A preliminary analysis indicated that the hysteresis model used in the pipe bracing system was not suitable for the double angle braces, and buckling element of Reference 10 shown in Fig. 6.3.1 was selected. This hysteresis model is described fully by Jain and Goel⁽¹⁰⁾. It includes two significant characteristics of a brace; residual elongation, and reduction in compressive strength with number of cycles. These parameters are especially important in braces with small slenderness ratios. The input parameters of this model are fewer and were directly determined from the experimental results of the El Centro span 900 test described in Section 5.3b.

In this analysis, the damping coefficient proportional to the initial stiffness was set at $\beta_0 = 0.0044$ to obtain the desired first mode damping ratio of 5 percent. Also, the stiffness of the shaking table spring support was set at 600 kips/in. to account for the shaking table-structure interaction. Correlation between the results calculated with this model and measured results of the El Centro span 900 test described earlier are shown in Figs. 6.3.2 and 6.3.3. Both global and local quantities of the analytical model are in good general agreement with the experimental values, and the peak values have been predicted fairly well. This correlation is considered excellent, especially considering that the first floor angles had been distorted slightly during earlier tests; therefore, the selected buckling element is regarded as satisfactory.

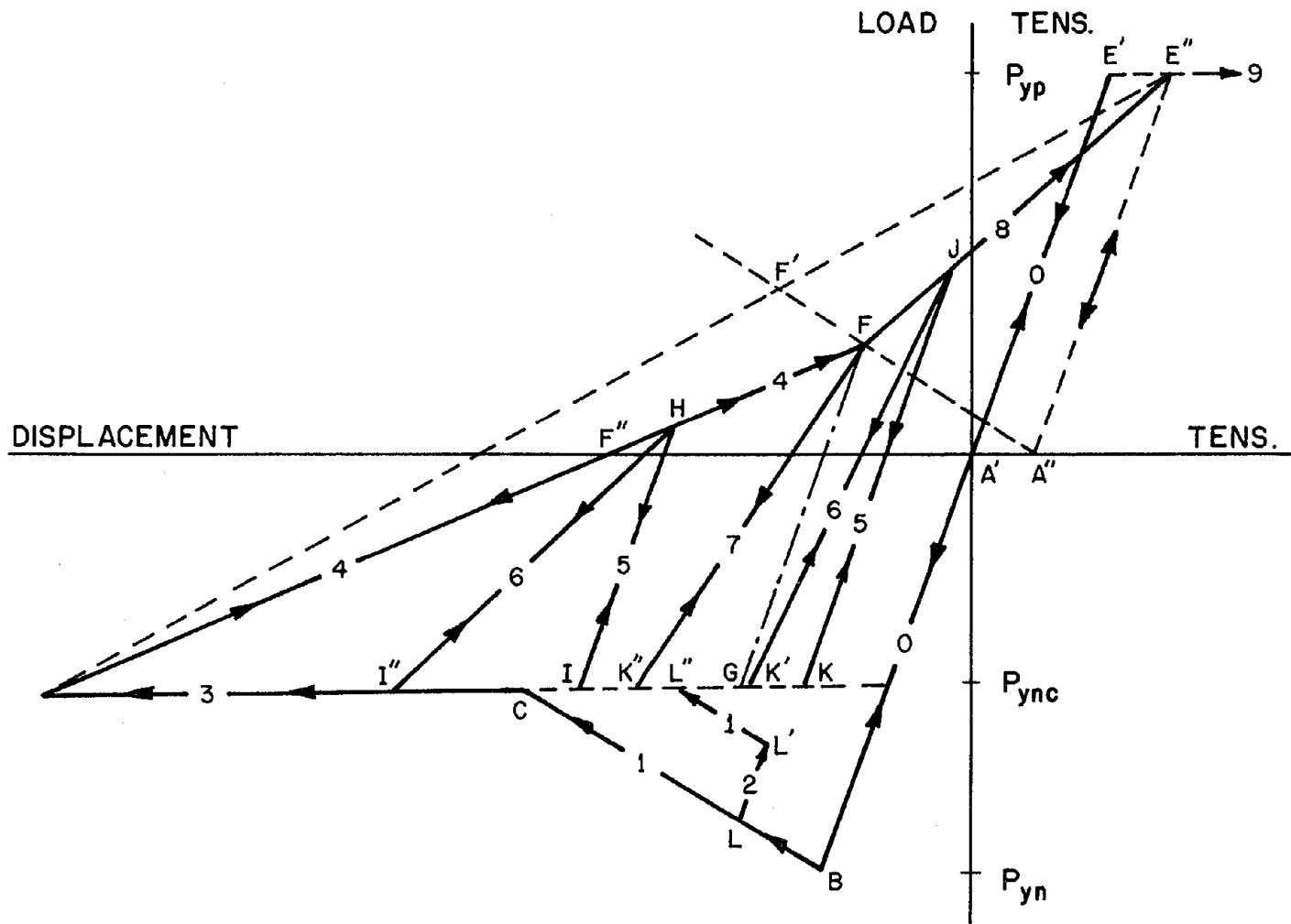
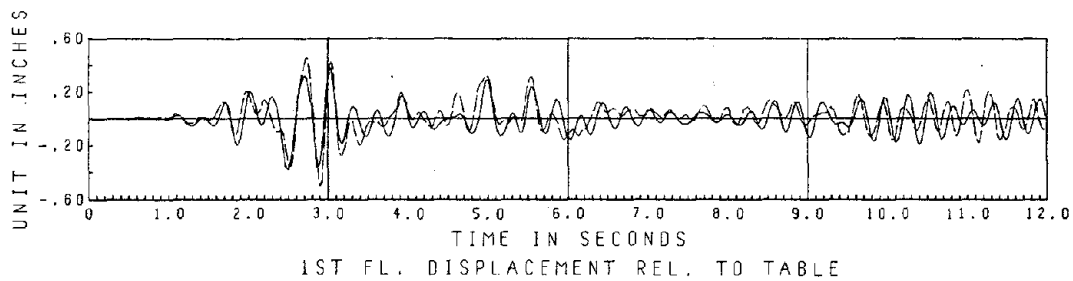
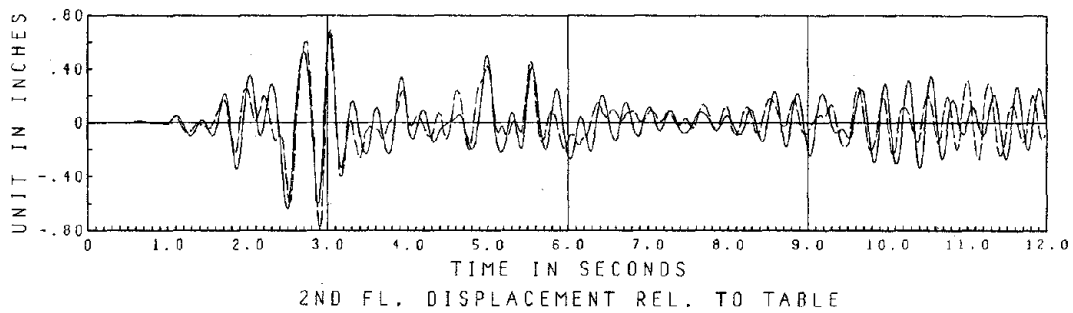
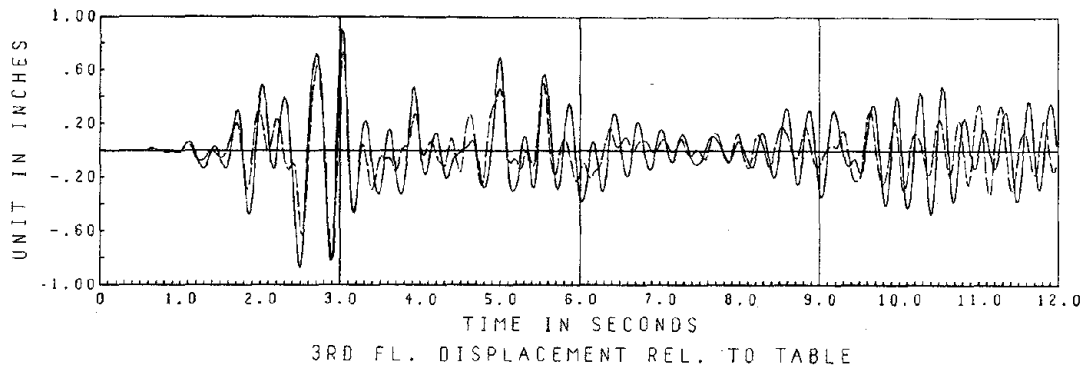
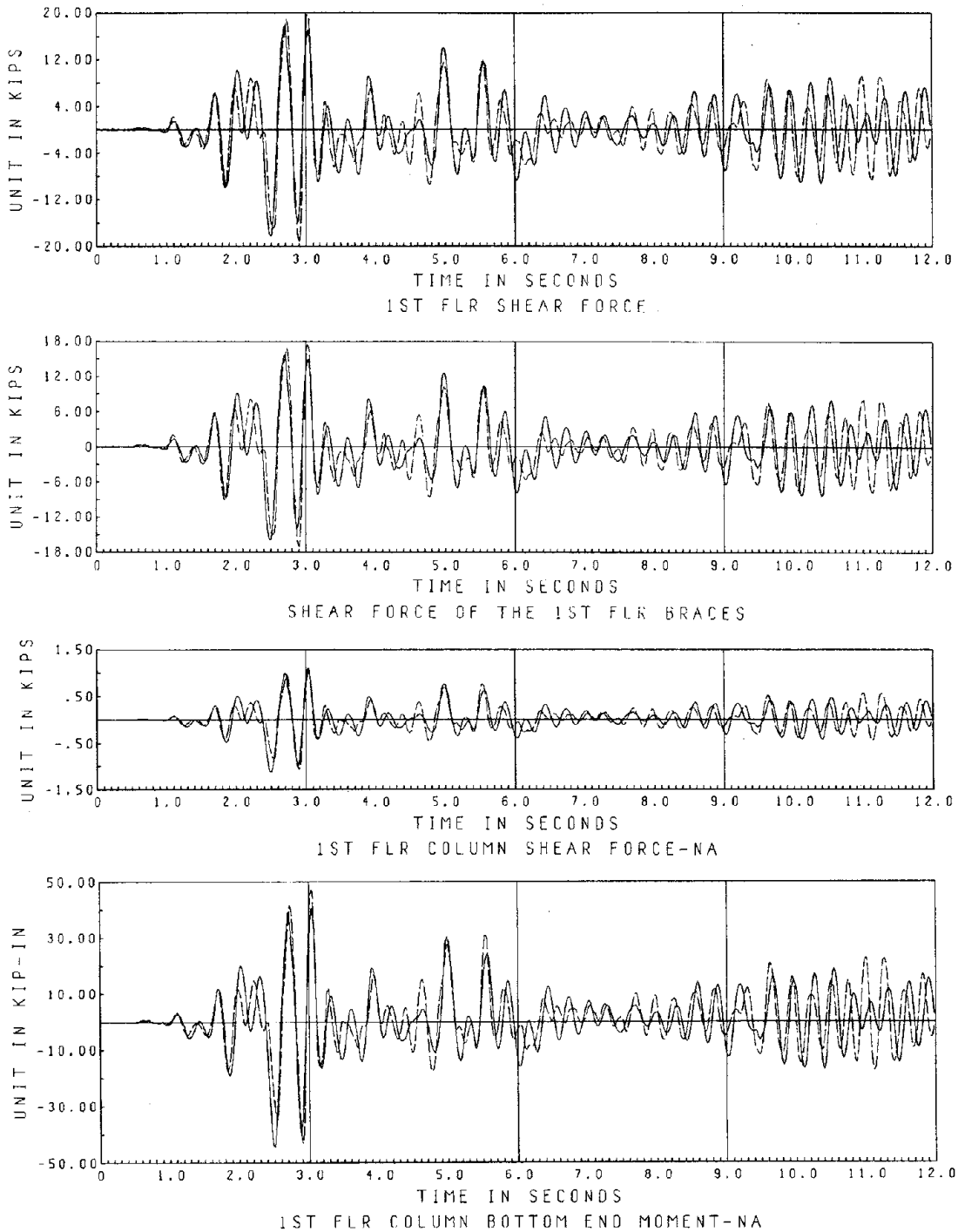


Fig. 6.3.1 Axial Hysteresis Behavior of Double Angle Braces
(after Jain and Goel)



EL CENTRO SPAN 900 , ANGLE DIAGONALLY BRACED FRAME
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.3.2 Correlation of Floor Displacements



EL CENTRO SPAN 900 , ANGLE DIAGONALLY BRACED FRAME
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.3.3 Correlation of Global and Local Forces

6.4 Unbraced Frame

The mathematical model developed for the unbraced moment-resistant frame is shown in Fig. 6.4.1. Note that eight semi-rigid connection elements have been introduced at the beam-to-column connections, and at the column ends in this model. These elements were used to model the type of joint connection shown in Fig. 3.2.3, to account for the significant angle change which occurred between the connected beams and columns. Each semi-rigid connection is connected to two nodes, and is influenced only by the relative rotational displacement between the nodes. The rotational stiffnesses of these connections were determined by frequency analysis of the mathematical model, treating them as the unknown parameters. To perform this frequency analysis, an explicit stiffness matrix formulation corresponding to the mathematical model was derived. All matrix operations were performed by the Symbolic Matrix Interpretive System (SMIS), a computer program described in Reference 17. In this process, the stiffness properties of all structural components except those of the semi-rigid connections were kept unchanged. Then, by varying the stiffness of the connections, a trial and error procedure was used until a close match was obtained between the frequency of the analytical model and that of the actual structure. The resulting rotational stiffness of the connections was then utilized in the dynamic response analysis of the unbraced frame model.

6.4a Correlation with El Centro Span 400

The model depicted in Fig. 6.4.1 was used to predict the results of the El Centro span 400 test. In this analysis, an initial stiffness-dependent damping coefficient of $\beta_0 = 0.018$ was selected to provide the desired

first mode damping ratio of 6 percent. The stiffness properties of the various members used in the frequency analysis were 400 kips/in. for the vertical spring supports, 15,900 kip-in./rad for the beam-to-column connection joint, and 30,000 kip-in./rad for the connection joint of the column ends. Bilinear hysteresis behavior was assumed for all members except for the vertical supports which were treated as elastic axial members. Figure 6.4.2 shows that the predicted floor displacements obtained with this model appear to correlate adequately with the observed floor displacements of the El Centro test. These results are considered to be fairly good; more accurate matching is only possible using a mathematically optimized system identification method, which is beyond the scope of this study.

6.4b Correlation with Pacoima Span 400

The response of the unbraced structure to the Pacoima span 400 earthquake, as described in Section 4.5b, was nonlinear and induced significant column yielding. These results were used to examine the applicability of the model discussed above for predicting the response of severely nonlinear cases. In this analysis, the same initial stiffness-dependent damping coefficient of $\beta_0 = 0.018$ provided the first mode damping ratio of 6 percent. The stiffness values of the various structural components were kept the same except that of the vertical spring supports. This stiffness was reduced to 300 kips/in. to represent the more intense response behavior. Figures 6.4.3 and 6.4.4 show the correlation between the calculated and the measured results. Both global and local quantities are in excellent agreement. Note that the column moments were predicted quite accurately despite the assumed bilinear behavior instead of a more realistic curvilinear

mechanism. Thus, based on the assumptions made, this correlation is regarded as excellent and the model is adequate.

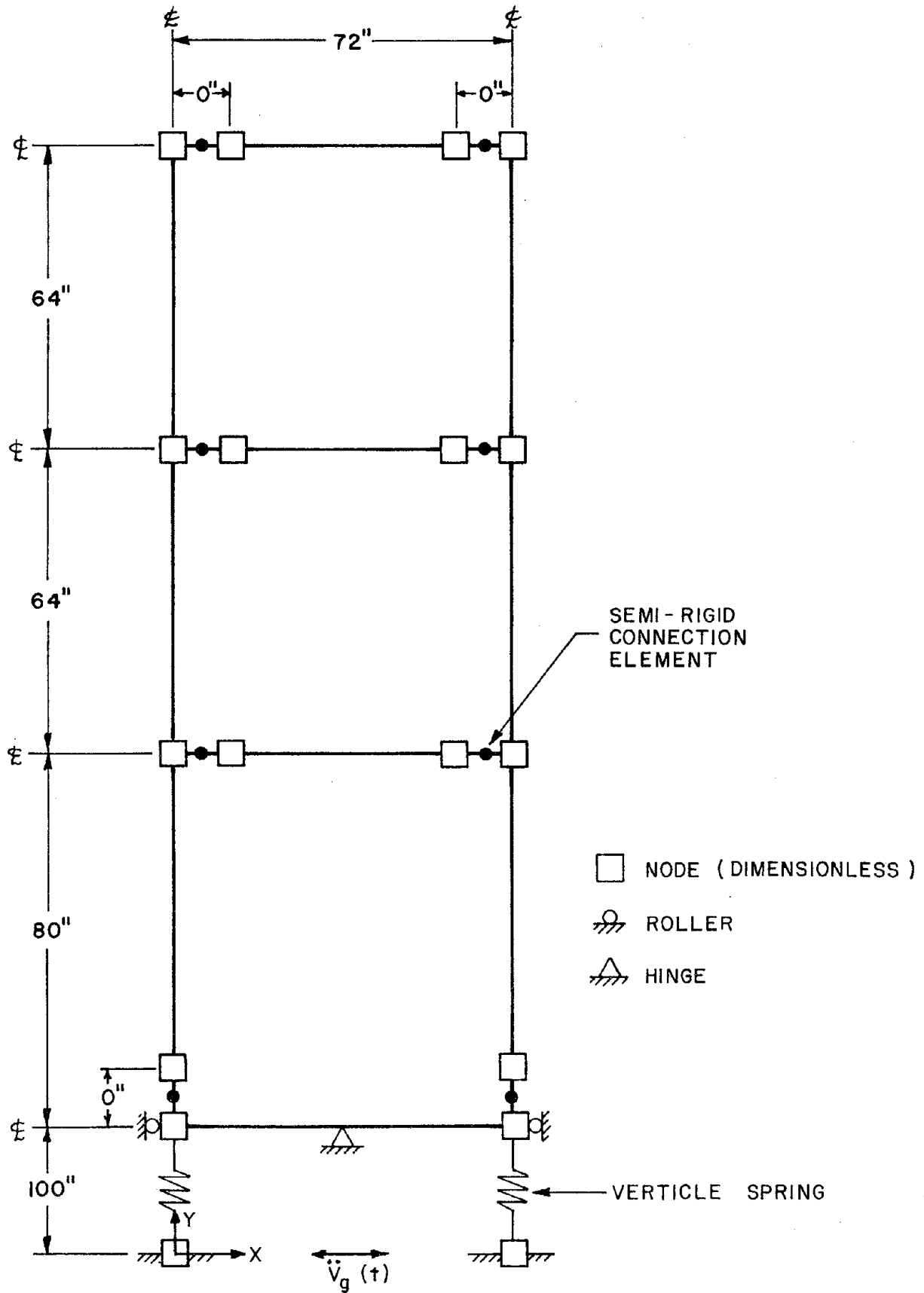
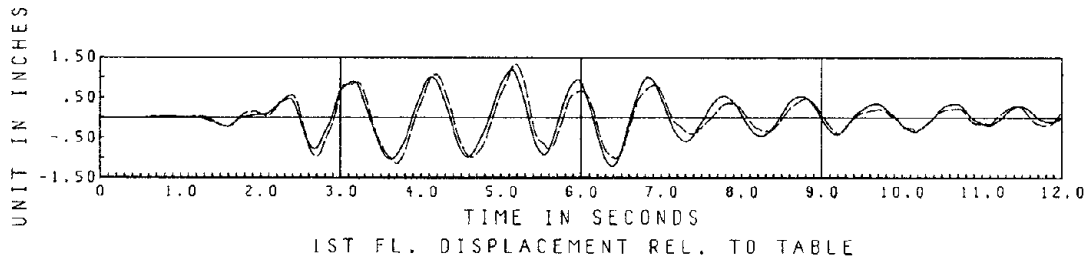
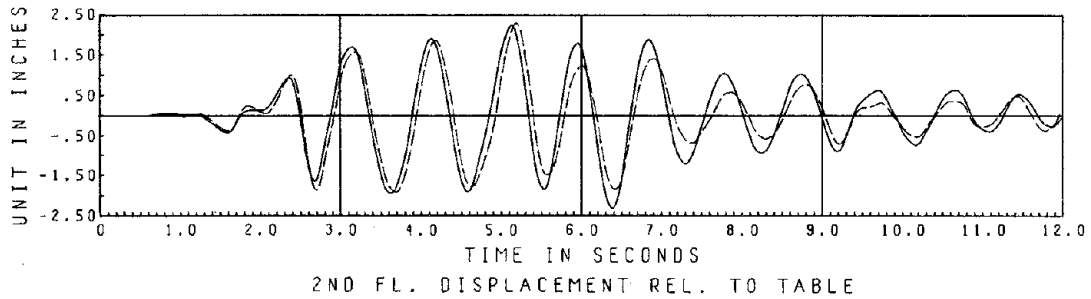
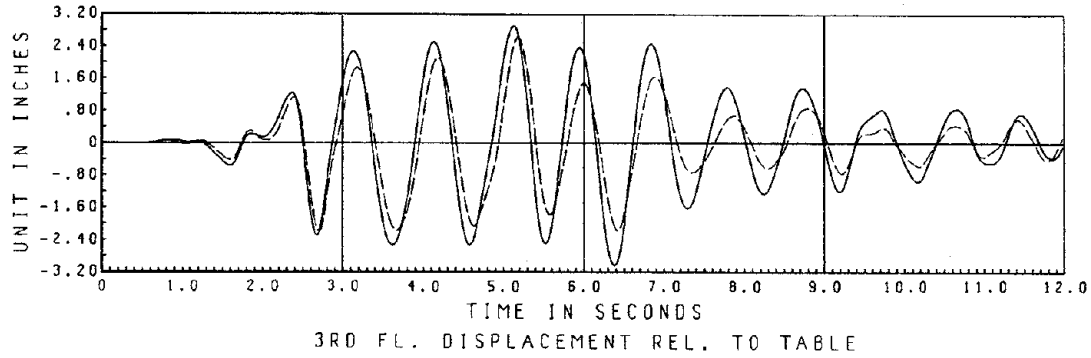
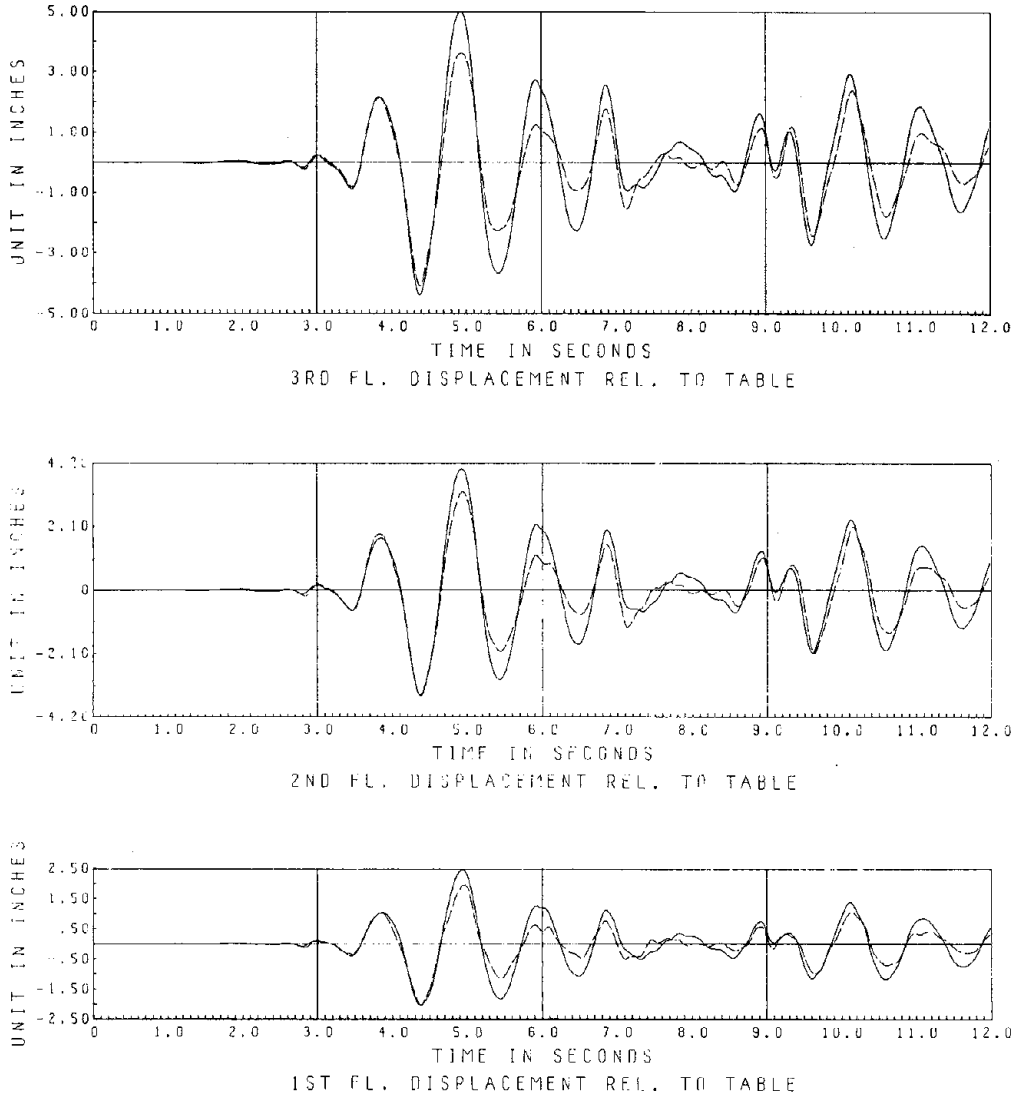


Fig. 6.4.1 Mathematical Model for Unbraced Frame



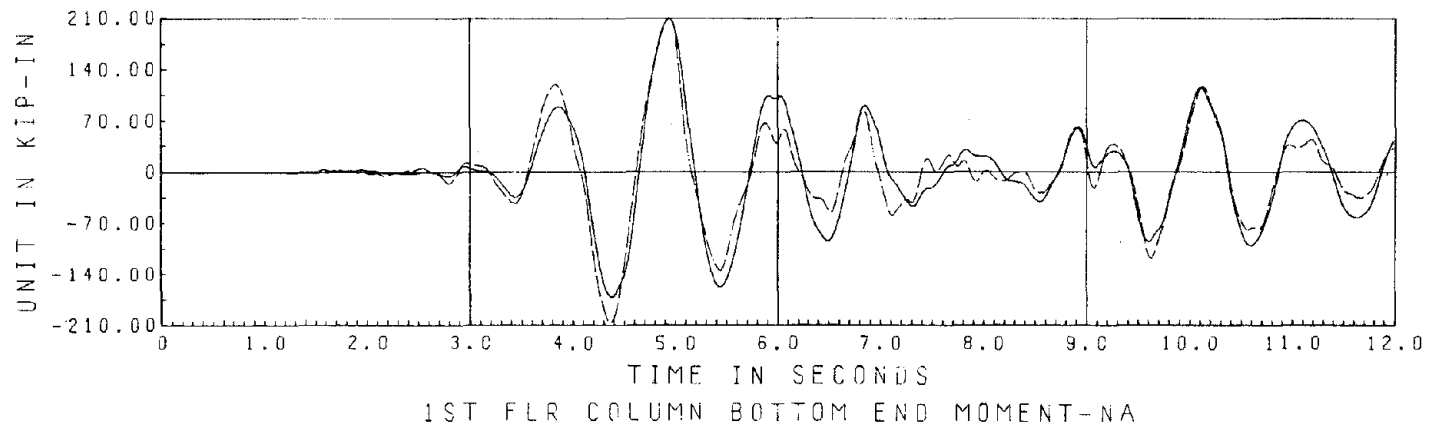
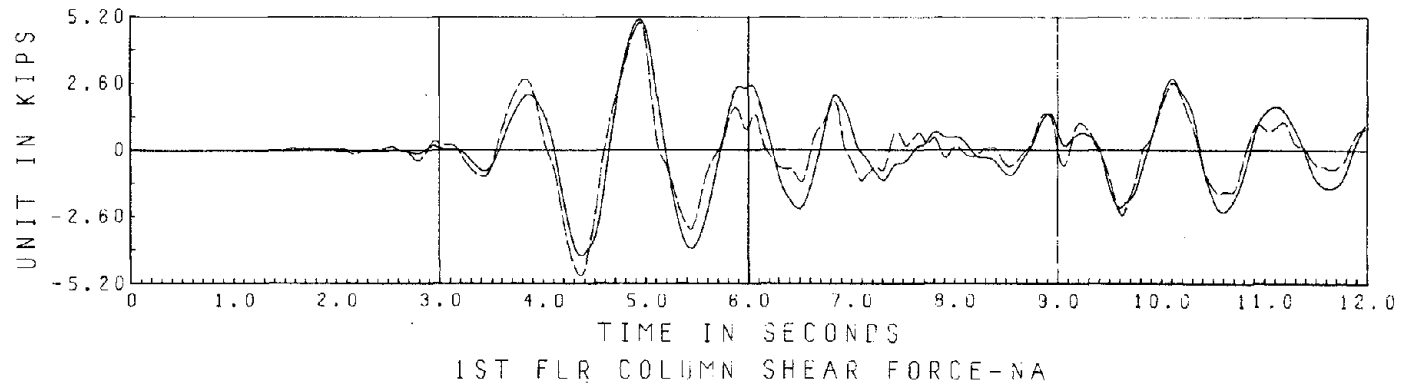
EL CENTRO SPAN 400 , UNBRACED FRAME
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.4.2 Correlation of Floor Displacements



PACDIMA SPAN 400 , UNBRACED FRAME
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.4.3 Correlation of Floor Displacements



PACDIMA SPAN 400 , UNBRACED FRAME
 CALCULATED RESULT IN SOLID LINE, MEASURED RESULT IN DASH LINE

Fig. 6.4.4 Correlation of Column Forces

7. COMPARISON OF DIFFERENT BRACING SYSTEMS

The experimental results presented in Chapter 5 cannot be directly used in a comparison of the seismic efficiency of the bracing systems considered in this study, because each structure was tested under different load conditions and/or was subjected to the earthquake motions in a different sequence and with different intensities. However, the mathematical models developed in Chapter 6 satisfactorily predicted the experimental response of each structure. Thus, it was decided to apply the same analytical procedures to predict the seismic response of unbraced frame, rod, pipe, and double angle braced structures under identical dead loads and identical earthquake motions. In these analyses, all four structures were loaded with 17 kips dead load per floor and were subjected to motions patterned after the El Centro earthquake record. These analytical results were then used to compare the seismic behavior and efficiency of each structure.

The maximum calculated floor drifts and lateral forces were selected for purposes of comparison, and they are displayed versus the peak acceleration of the input motions in Figs. 7.1 to 7.6. In general, the largest floor drifts and largest floor shear forces were associated with the unbraced (the softest) and the double angle braced (the stiffest) structures, respectively. The first floor drifts of the rod braced structure were the largest among the bracing systems and became relatively close to those of the unbraced structure as the slacking mechanism developed in rod members during moderate earthquake motions (about 0.3 g peak acceleration). Rupture of the rod braces started during motions with a peak acceleration of about 0.5 g (see Fig. 7.1). At the second floor level, the drifts of the rod bracing system were also relatively large due to the same type of

behavior, but the slacking and rupture occurred at the second floor during higher input intensities (see Fig. 7.2). Yielding of the third floor rods was minor and no rupture occurred at this level, so drifts were relatively smaller than those of the unbraced frame.

The floor drifts of the pipe and double angle bracing systems were the smallest despite their frequent buckling and/or yielding during the moderate and strong input motions. In fact, the pipe and particularly the double angle braces were very effective in limiting the floor drifts even at an input peak acceleration of 0.8 g, and their strength loss was not significant at this level.

Although the pipe and double angle braces were efficient in reducing the floor drifts, as shown in Figs. 7.4 to 7.6, they provided larger floor shear forces. However, this behavior is tolerable considering that the major portion of these lateral forces were resisted by the bracing members, and the plastic deformations in the main structural components such as the columns were correspondingly reduced. The lowest floor shear forces were associated with the unbraced frame, but they were resisted only by the columns. In this case the shear resisting capacity of the first floor was estimated to be about 11 kips per frame, with plastic hinges formed at the column ends during a peak acceleration of 0.775 g. At this same input intensity, plastic hinges were not formed at the column ends of the pipe and double angle braced structures. Thus, even though the lateral forces of the unbraced frame were not as large as those of the braced structures, the columns suffered significant plastic deformations and the resulting lateral drifts were larger than what can be considered tolerable. Damage of the non-structural components such as the partitions as well as of the major structural members

would be greatest in this type of construction.

The shear forces of the rod braced structure were about the same magnitude as those of the pipe and double angle braced structures for low intensity motions (up to 0.2 g) even though the rod bracing provided a softer system. This was due to the fact that the participation of the columns in the shear resistance was more significant in a system with lighter braces having low compressive capacity. As was mentioned earlier, the efficiency of the rod braces diminished as their slacking behavior developed during moderate earthquake motions, and they ruptured in response to stronger excitations. Accordingly, the resisting force capacity of the rod braced structure decreased and was limited to about 14 kips per frame even with complete participation of the columns, this behavior resembled the response of the unbraced structure.

To provide a more useful comparison and to evaluate the efficiency of the bracing systems, the floor drifts of the different systems were also compared with the drift limits allowed by the Uniform Building Code⁽¹³⁾ and the Applied Technology Council (ATC) regulations⁽¹⁸⁾. The UBC requires that the lateral deflections or drifts of a story relative to its adjacent stories shall not exceed 0.005 times the story height, and the allowable drifts according to the seismic design regulations provided by the ATC shall not exceed 0.015 times the story height. The limits for both codes are shown in Figs. 7.1 to 7.3.

According to the UBC requirements, the unbraced structure should not be subjected to motions with peak accelerations greater than 0.07 g. This limits the applicability of this type of structure to very low

intensity earthquake regions where only elastic response is expected. The newly developed ATC seismic design regulation, which considers the modern concept of ductility, modifies the UBC demand and allows the unbraced structure to be subjected to earthquakes with peak accelerations up to 0.2 g (see Fig. 7.1).

The rod bracing system, according to the UBC is limited to earthquakes with peak accelerations less than 0.12 g corresponding to the elastic behavior of the rod braces. On the other hand, under ATC regulations, rod bracing is beneficial up to a peak acceleration of 0.3 g. The pipe and double angle bracing systems satisfy the UBC requirements with great efficiency up to peak accelerations of 0.28 g and 0.34 g, respectively. At moderate to relatively strong motions they buckle and yield frequently, but their drifts do not exceed the ATC drift limits up to a peak acceleration of 0.68 g in the pipe system and 0.8 g in the double angle system. According to these results, the pipe brace and particularly the angle brace are very efficient and effective in controlling lateral displacements of the steel frame structure. But their efficiencies diminish during very strong earthquake motions as the braces undergo severe yielding and buckling. For instance, the maximum first floor drift of the double angle braced structure was 2.22 in. in the Paçoima test described in Section 5.3c.

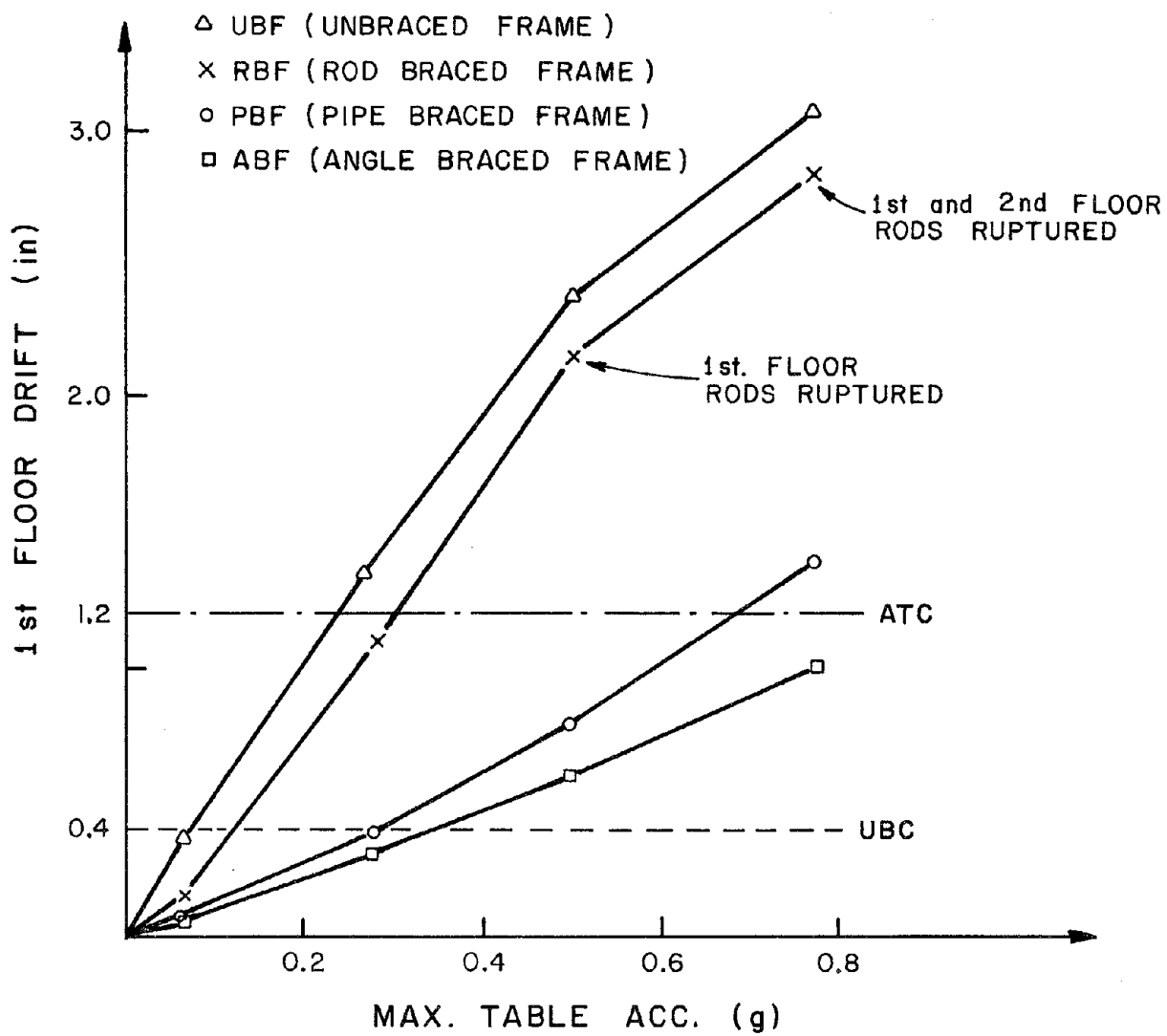


Fig. 7.1 Maximum 1st Floor Drift vs. Peak Table Acc.

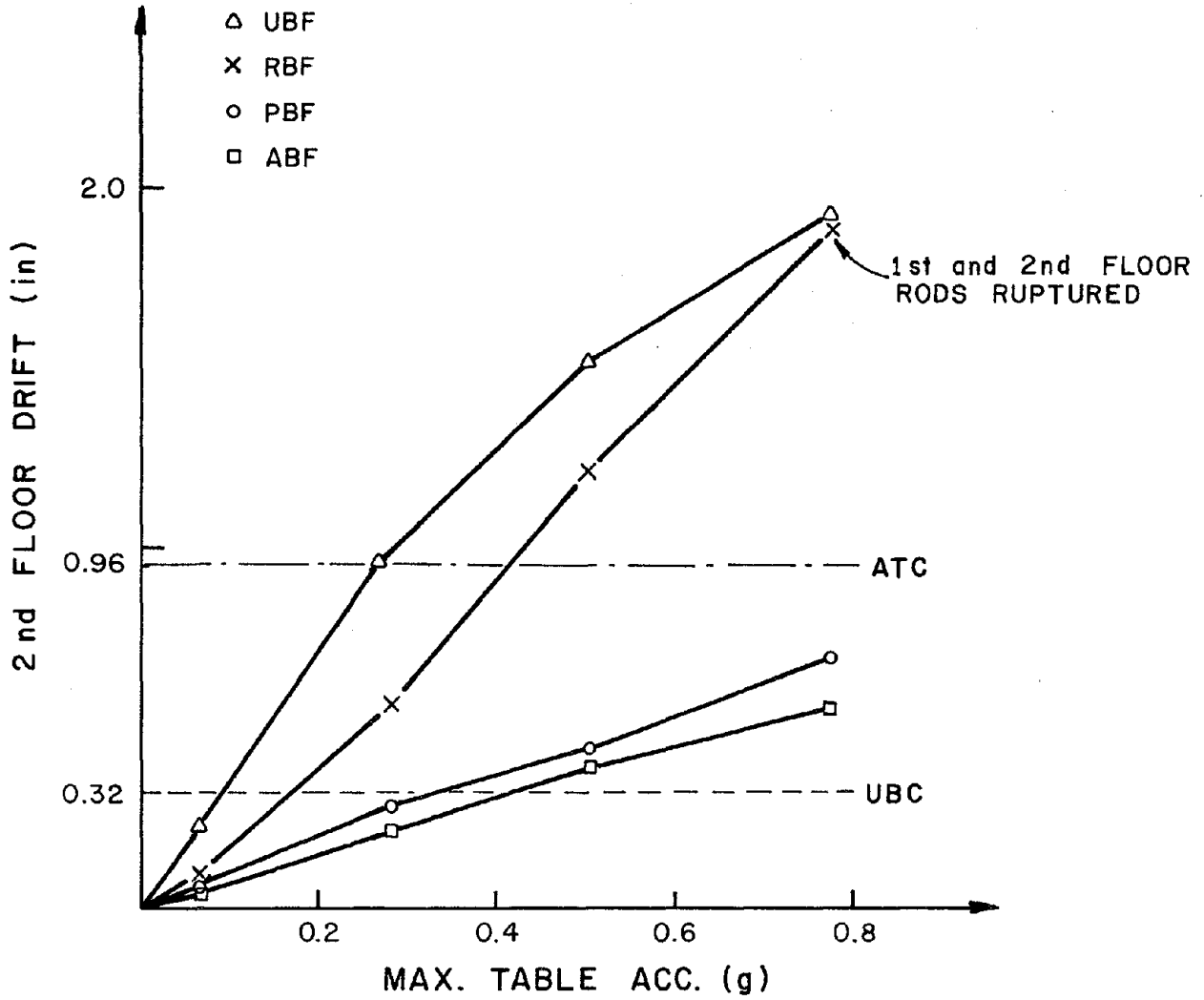


Fig. 7.2 Maximum 2nd Floor Drift vs. Peak Table Acceleration

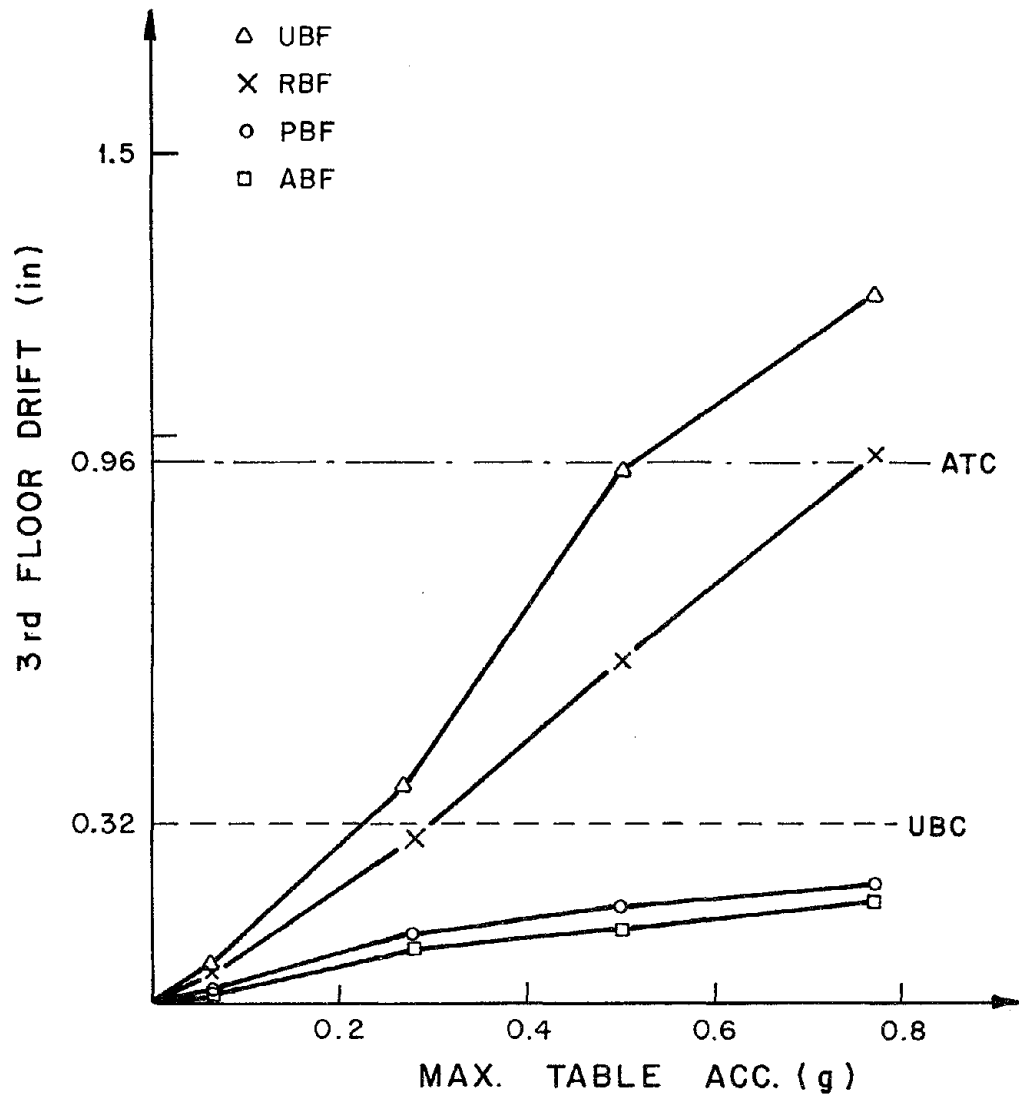


Fig. 7.3 Maximum 3rd Drift vs. Peak Table Acceleration

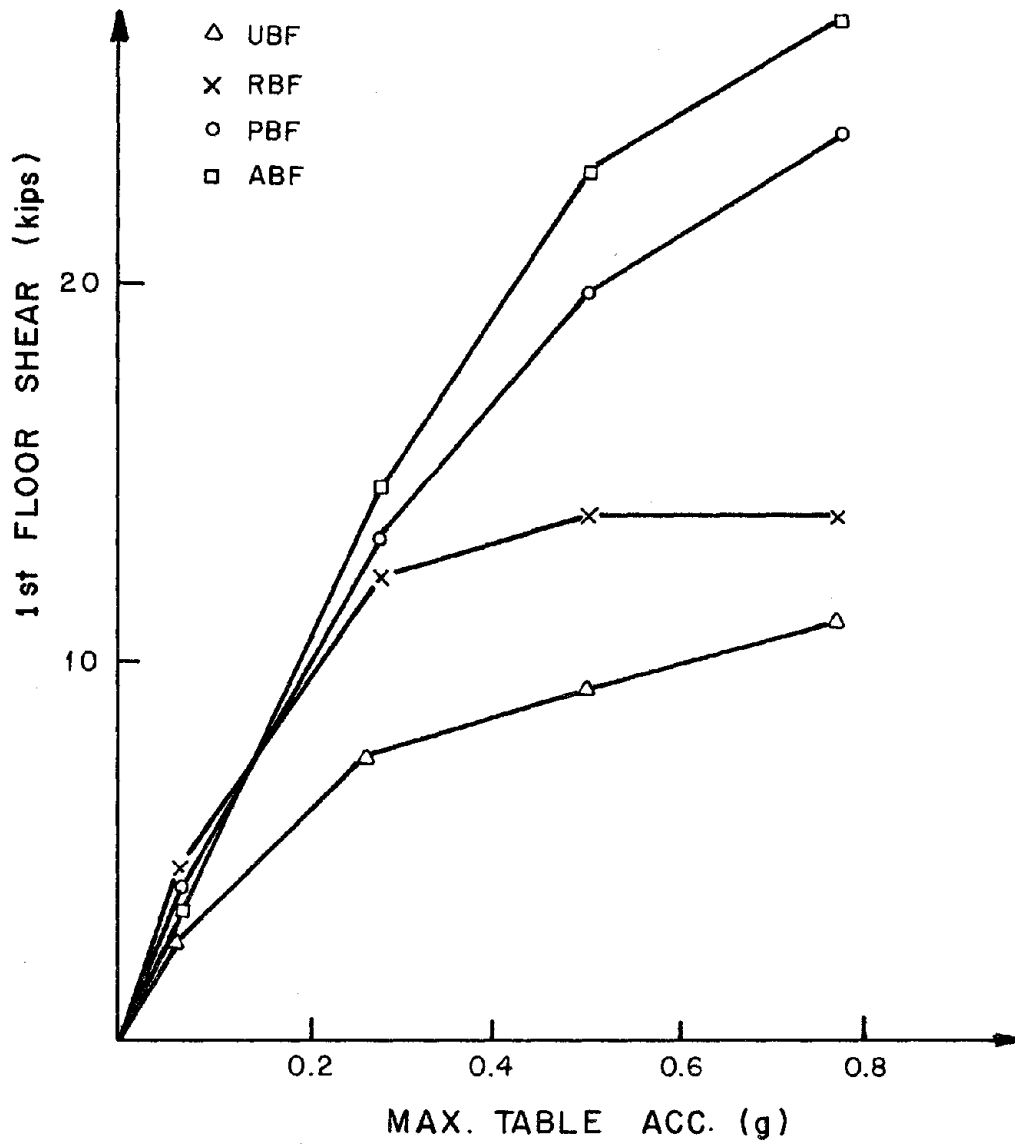


Fig. 7.4 Maximum 1st Floor Shear vs. Peak Table Acceleration

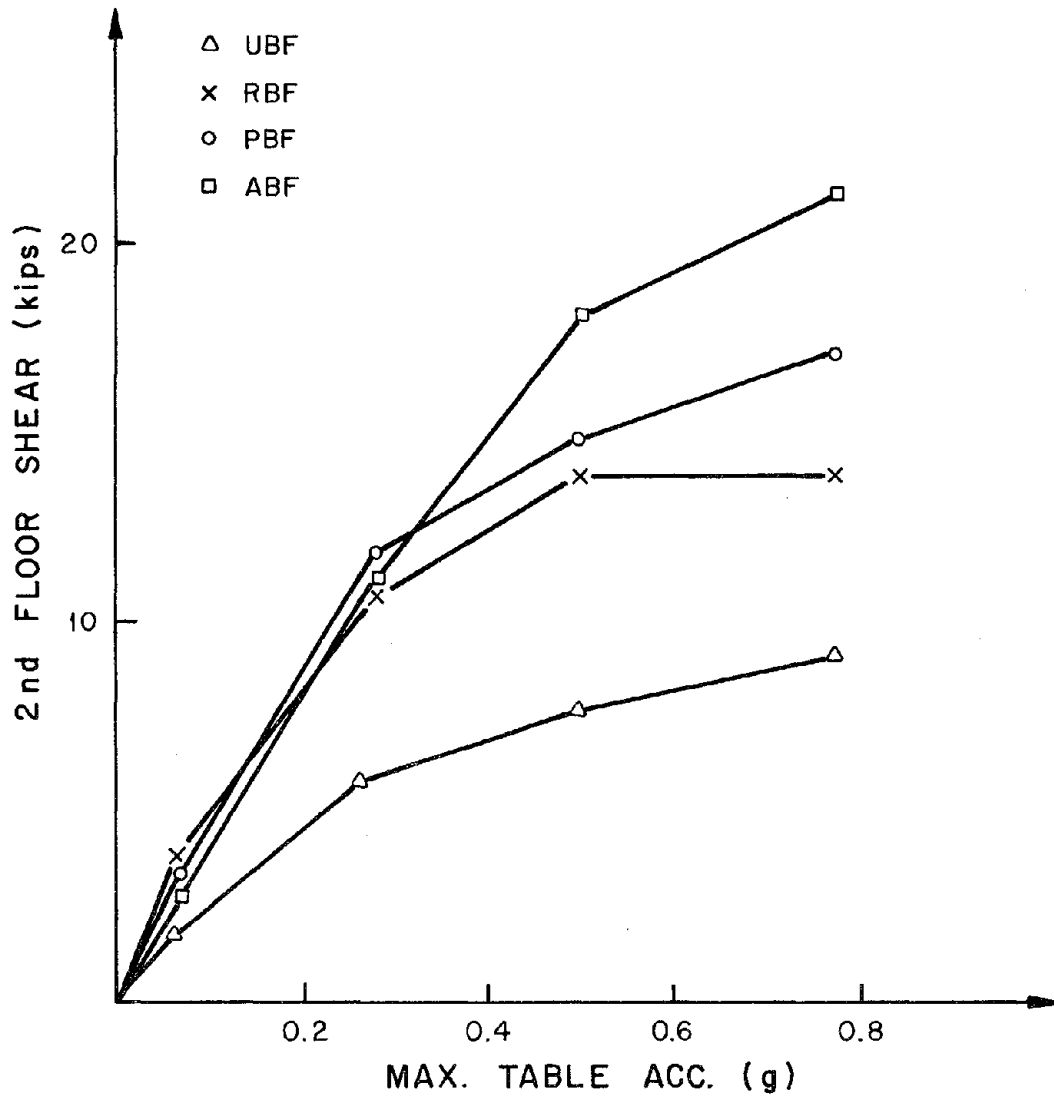


Fig. 7.5 Maximum 2nd Floor Shear vs. Peak Table Acceleration

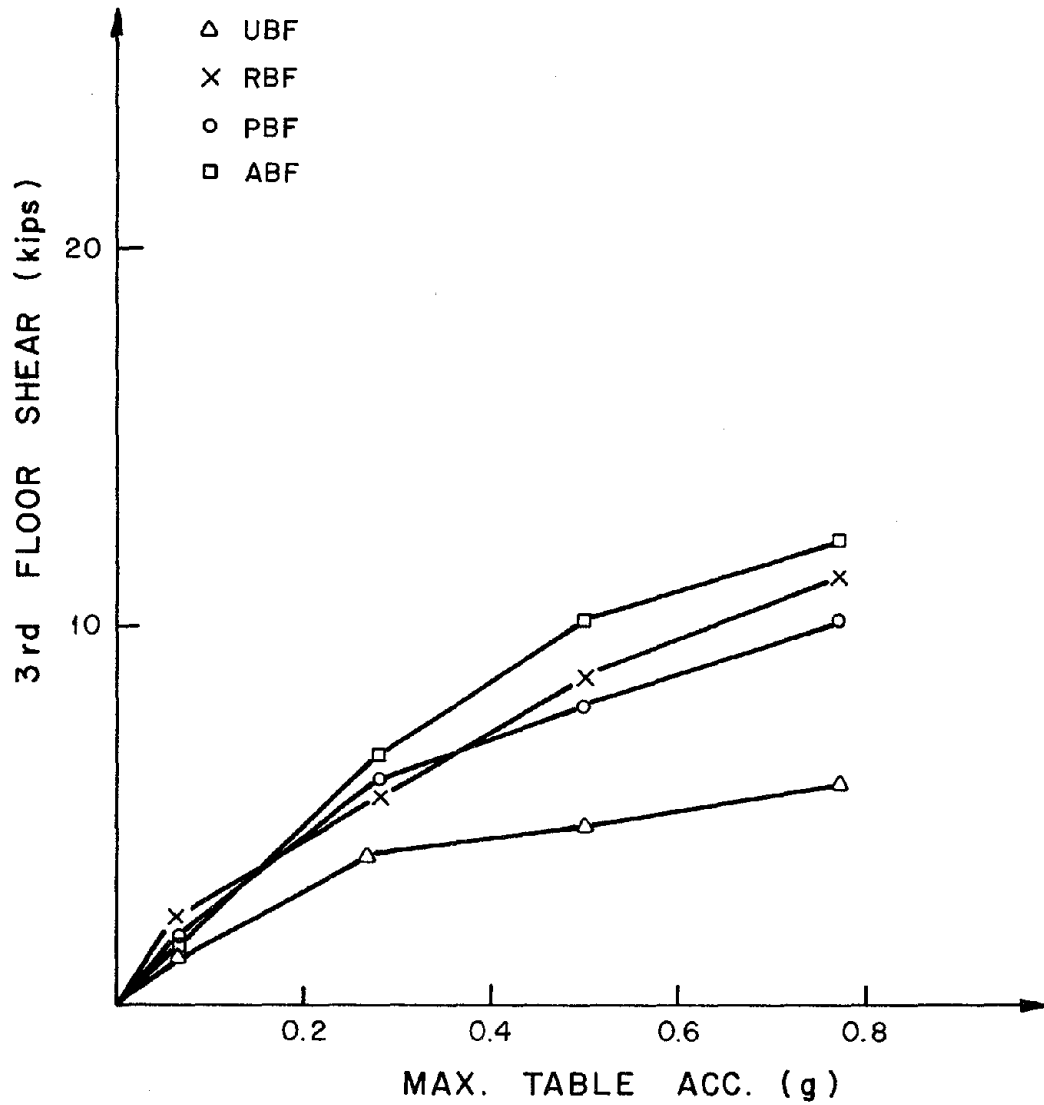


Fig. 7.6 Maximum 3rd Floor Shear vs. Peak Table Acceleration

8. CONCLUSIONS

This investigation demonstrated the seismic response behavior of steel frames with and without wind bracing and the feasibility of predicting the response by existing nonlinear frame analysis programs. The most important feature of the response behavior with rod braces is that pre-tension is lost during moderate earthquakes, leading to an impacting type response in the resulting slack rod system. As a result, the efficiency of rod braces diminishes and the story drifts of the structure become relatively large. For larger earthquakes, the rods may be expected to break during the successive impacts; but it is significant that the structure without braces may still survive the earthquake motions, if it is stable under the action of the static gravity loads.

The structures with pipe and double angle braces behave better because the pipes and double angles retain some significant compressive capacity after buckling. Thus, there is no slack response mechanism with associated impacting, and significant energy is absorbed by the braces in post-buckling displacement cycles. The pipe braces and the double angle braces each are quite effective in limiting story drifts during moderate and moderate-to-strong earthquakes, respectively. Their efficiencies reduce only for rather strong earthquakes as a "pinching" effect develops in the force-displacement hysteresis loops because of accumulated tensile yield deformations.

The lateral stiffness of the unbraced structure is very small so its story drifts are very large even for moderate earthquakes. The structure is not expected to collapse under strong earthquakes because the $P-\Delta$ effect for such a low rise structure is not large enough.

But, damage of the non-structural members such as partitions and also residual distortion of the columns may be very significant.

The results of this study show that supplemental diagonal wind bracing has a beneficial effect on the earthquake resistance of the steel frame structure because it tends to limit the story drifts and thus to reduce damage of both structural and non-structural components. However, each bracing type has its own limitations and specific recommendations can be made as follows:

1. Tension type wind braces with large slenderness ratios (greater than 200) and correspondingly low compressive capacity are only beneficial for wind loading and very low intensity earthquakes.

2. Compression type wind braces with intermediate slenderness ratios (less than 200) and compressive strength not less than 50 percent of the tension strength provide a bracing system which can also resist moderate earthquakes with great efficiency.

3. Compression type wind braces with slenderness ratios less than 100 and compressive strength not less than 50 percent of tension strength are very effective and beneficial for moderate-to-strong earthquakes.

Analyses made with the DRAIN-2D program showed generally good agreement with the experimental results. Rod bracing models simulated both tension yielding and elastic buckling behavior satisfactorily; the tension rod rupture mechanism was modelled least satisfactorily, but such ruptures also were found to be quite random in the experiments.

Hysteresis models of pipe and double angle bracing members included both tension yielding and post-buckling deflection behavior; the hysteresis model of the double angle braces also considered residual elongation

and reduction in compressive capacity with number of cycles. Both the pipe and the double angle mathematical models proved to be satisfactory.

The analytical response prediction of the unbraced frame showed that angle changes in the joints between connected beams and column are significant in this type of construction, and should be included in the analysis.

This study indicated that diagonal cross bracing systems, such as pipe and double angle braces, are very efficient for moderate earthquakes and their energy dissipation will be significant if their compressive capacity is not less than 50 percent of their tension capacity. However, the energy dissipation characteristics of these diagonals may be less satisfactory during major earthquakes due to the pinching effect in the bracing hysteresis loops.

A braced frame suitable for resisting major ground motion may be achieved by using a split K-bracing system whose diagonal members have significant eccentricities. In such a system, the benefits of bracing elements to minimize drift (with consequent damage control and increased stability) are combined with the ductility of the moment-resistant frame. In addition, the split K-bracing system can be adapted to accommodate architectural requirements such as windows, doors, or utilities in the walls. The principles of the eccentric K-bracing system discussed here have not been verified by dynamic tests; but, it is believed that dynamic testing of a steel frame with split K-bracing members can provide significant information regarding the applicability of the concept.

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APPENDIX A

List of Data Channels

Channel No.	Name	Description
0	T/R ACC-1	Command Horizontal Acceleration Signal
1	T/R ACC-2	Command Vertical Acceleration Signal
2	CMD H DISP	Command Horizontal Displacement Signal
3	CMD V DISP	Command Vertical Displacement Signal
4	AV H T DISP	Average Horizontal Table Displacement
5	AV V T DISP	Average Vertical Table Displacement
6	AV H T ACC	Average Horizontal Table Acceleration
7	AV V T ACC	Average Vertical Table Acceleration
8	PITCH	Angular Acceleration in Pitching Mode
9	ROLL	Angular Acceleration in Rolling Mode
10	TWIST	Angular Acceleration in Twisting Mode
11	FORCE H1	Force in Horizontal Actuator
12	FORCE H2	Force in Horizontal Actuator
13	FORCE H3	Force in Horizontal Actuator
14	ACC H1	Horizontal Table Acceleration at Actuator H1
15	ACC H2	Horizontal Table Acceleration at Actuator H2
16	ACC V1	Vertical Table Acceleration at Actuator V1
17	ACC V2	Vertical Table Acceleration at Actuator V2
18	ACC V3	Vertical Table Acceleration at Actuator V3
19	ACC V4	Vertical Table Acceleration at Actuator V4
20	FORCE V1	Force in Vertical Actuator V1
21	FORCE V2	Force in Vertical Actuator V2

Table A-1 Data channel listing for diagonal rod bracing

Channel No.	Name	Description
22	FORCE V3	Force in Vertical Actuator V3
23	FORCE V4	Force in Vertical Actuator V4
24	DISP V1	Vertical Table Displacement at Actuator V1
25	DISP V2	Vertical Table Displacement at Actuator V2
26	DISP V3	Vertical Table Displacement at Actuator V3
27	DISP V4	Vertical Table Displacement at Actuator V4
28	DISP H1	Horizontal Table Displacement at Actuator H1
29	DISP H2	Horizontal Table Displacement at Actuator H2
30	DISP H3	Horizontal Table Displacement at Actuator H3
31	BLANK	
32	PS FORCE-1	Force in Passive Stabilizer
33	PS FORCE-2	Force in Passive Stabilizer
34	PS FORCE-3	Force in Passive Stabilizer
35	PS FORCE-4	Force in Passive Stabilizer
36	CLELG-NAOF	Column NA DCDT, Outside Face
37	CLELG-NAIF	Column NA DCDT, Inside Face
38	CLELG-NBOF	Column NB DCDT, Outside Face
39	CLELG-NBIF	Column NB DCDT, Inside Face
40	CLELG-SAOF	Column SA DCDT, Outside Face
41	CLELG-SAIF	Column SA DCDT, Inside Face
42	CLELG-SBOF	Column SB DCDT, Outside Face
43	CLELG-SBIF	Column SB DCDT, Inside Face

Channel No.	Name	Description
44	FLR DSP-NA3	3rd Floor Absolute Displacement at Column NA
45	FLR DSP-SA3	3rd Floor Absolute Displacement at Column SA
46	FLR DSP-NA2	2nd Floor Absolute Displacement
47	FLR DSP-NA1	1st Floor Absolute Displacement
48	ROD DFM-N1	1st Floor Rod Brace Displacement, Frame N
49	ROD DFM-S1	1st Floor Rod Brace Displacement, Frame S
50	BLANK	
51	BLANK	
52	FLR ACC-1	1st Floor Absolute Acceleration
53	FLR ACC-2	2nd Floor Absolute Acceleration
54	FLR ACC-3N	3rd Floor Absolute Acceleration, Frame N
55	FLR ACC-3S	3rd Floor Absolute Acceleration, Frame S
56	CLPYSTR-NAB	1st Floor Column NA Post-Yield Flexural Strain, Bottom End
57	CLPYSTR-NAT	1st Floor Column NA Post-Yield Flexural Strain, Top End
58	CLPYSTR-NBB	1st Floor Column NB Post-Yield Flexural Strain, Bottom End
59	CLPYSTR-NBT	1st Floor Column NB Post-Yield Flexural Strain, Top End
60	CLSTRG-NAB1	1st Floor Column NA Elastic Flexural Strain, Bottom End
61	CLSTRG-NAT1	1st Floor Column NA Elastic Flexural Strain, Top End
62	CLSTRG-NBB1	1st Floor Column NB Elastic Flexural Strain, Bottom End
63	CLSTRG-NBT1	1st Floor Column NB Elastic Flexural Strain, Top End
64	CLSTRG-SAB1	1st Floor Column SA Elastic Flexural Strain, Bottom End
65	CLSTRG-SAT1	1st Floor Column SA Elastic Flexural Strain, Top End
66	CLSTRG-SBB1	1st Floor Column SB Elastic Flexural Strain, Bottom End

Channel No.	Name	Description
67	CLSTRG-SBT1	1st Floor Column SB Elastic Flexural Strain, Top End
68	CLSTRG-NAB2	2nd Floor Column NA Elastic Flexural Strain, Bottom End
69	CLSTRG-NAT2	2nd Floor Column NA Elastic Flexural Strain, Top End
70	CLSTRG-NBB2	2nd Floor Column NB Elastic Flexural Strain, Bottom End
71	CLSTRG-NBT2	2nd Floor Column NB Elastic Flexural Strain, Top End
72	CLSTRG-SAB2	2nd Floor Column SA Elastic Flexural Strain, Bottom End
73	CLSTRG-SAT2	2nd Floor Column SA Elastic Flexural Strain, Top End
74	CLSTRG-SBB2	2nd Floor Column SB Elastic Flexural Strain, Bottom End
75	CLSTRG-SBT2	2nd Floor Column SB Elastic Flexural Strain, Top End
76	RPYSTR-PN1	1st Floor Rod Post-Yield Axial Strain, North Frame Positive Direction
77	RPYSTR-NN1	1st Floor Rod Post-Yield Axial Strain, North Frame Negative Direction
78	RPYSTR-PS1	1st Floor Rod Post-Yield Axial Strain, South Frame Positive Direction
79	RPYSTR-NS1	1st Floor Rod Post-Yield Axial Strain, South Frame Negative Direction
80	RPYSTR-PN2	2nd Floor Rod Post-Yield Axial Strain, North Frame Positive Direction
81	RPYSTR-NN2	2nd Floor Rod Post-Yield Axial Strain, North Frame Negative Direction
82	RPYSTR-PS2	2nd Floor Rod Post-Yield Axial Strain, South Frame Positive Direction
83	RPYSTR-NS2	2nd Floor Rod Post-Yield Axial Strain, South Frame Negative Direction
84	RPYSTR-PN3	3rd Floor Rod Post-Yield Axial Strain, North Frame Positive Direction
85	RPYSTR-NN3	3rd Floor Rod Post-Yield Axial Strain, North Frame Negative Direction
86	RPYSTR-PS3	3rd Floor Rod Post-Yield Axial Strain, South Frame Positive Direction
87	RPYSTR-NS3	3rd Floor Rod Post-Yield Axial Strain, South Frame Negative Direction
88	CLFRC-NAO	1st Floor Column NA Elastic Axial Strain, Outside Face
89	CLFRC-NAI	1st Floor Column NA Elastic Axial Strain, Inside Face

Channel No.	Name	Description
90	CLFRC-NBO	1st Floor Column NB Elastic Axial Strain, Outside Face
91	CLFRC-NBI	1st Floor Column NB Elastic Axial Strain, Inside Face
92	CLFRC-SAO	1st Floor Column SA Elastic Axial Strain, Outside Face
93	CLFRC-SAI	1st Floor Column SA Elastic Axial Strain, Inside Face
94	CLFRC-SBO	1st Floor Column SB Elastic Axial Strain, Outside Face
95	CLFRC-SBI	1st Floor Column SB Elastic Axial Strain, Inside Face

Channel No.	Name	Description
0	T/R ACC-1	Command Horizontal Acceleration Signal
1	T/R ACC-2	Command Vertical Acceleration Signal
2	CMD H DISP	Command Horizontal Displacement Signal
3	CMD V DISP	Command Vertical Displacement Signal
4	AV H T DISP	Average Horizontal Table Displacement
5	AV V T DISP	Average Vertical Table Displacement
6	AV H T ACC	Average Horizontal Table Acceleration
7	AV V T ACC	Average Vertical Table Acceleration
8	PITCH	Angular Acceleration in Pitching Mode
9	ROLL	Angular Acceleration in Rolling Mode
10	TWIST	Angular Acceleration in Twisting Mode
11	FORCE H1	Force in Horizontal Actuator H1
12	FORCE H2	Force in Horizontal Actuator H2
13	FORCE H3	Force in Horizontal Actuator H3
14	ACC H1	Horizontal Table Acceleration at Actuator H1
15	ACC H2	Horizontal Table Acceleration at Actuator H2
16	ACC V1	Vertical Table Acceleration at Actuator V1
17	ACC V2	Vertical Table Acceleration at Actuator V2
18	ACC V3	Vertical Table Acceleration at Actuator V3
19	ACC V4	Vertical Table Acceleration at Actuator V4
20	FORCE V1	Force in Vertical Actuator V1

Table A-2 Data channel listing for diagonal pipe bracing

Channel No.	Name	Description
21	FORCE V2	Force in Vertical Actuator V2
22	FORCE V3	Force in Vertical Actuator V3
23	FORCE V4	Force in Vertical Actuator V4
24	DISP V1	Vertical Table Displacement at Actuator V1
25	DISP V2	Vertical Table Displacement at Actuator V2
26	DISP V3	Vertical Table Displacement at Actuator V3
27	DISP V4	Vertical Table Displacement at Actuator V4
28	DISP H1	Horizontal Table Displacement at Actuator H1
29	DISP H2	Horizontal Table Displacement at Actuator H2
30	DISP H3	Horizontal Table Displacement at Actuator H3
31	BLANK	
32	PPYS-PTS	1st Floor Pipe Post-Yield Strain, Positive Direction, Top Face, Frame S
33	PPYS-PBS	1st Flr Pipe Post-Yield Strain, Positive Direction, Bottom Face, Frame S
34	PPYS-POS	1st Flr Pipe Post-Yield Strain, Positive Direction, Outside Face, Frame S
35	PPYS-PIS	1st Flr Pipe Post-Yield Strain, Positive Direction, Inside Face, Frame S
36	CLELG-NAOF	Column NA DCDT, Outside Face
37	CLELG-NAIF	Column NA DCDT, Inside Face
38	CLELG-NBOF	Column NB DCDT, Outside Face
39	CLELG-NBIF	Column NB DCDT, Inside Face
40	CLELG-SAOF	Column SA DCDT, Outside Face
41	CLELG-SAIF	Column SA DCDT, Inside Face
42	CLELG-SBOF	Column SB DCDT, Outside Face

Channel No.	Name	Description
43	CLELG-SBIF	Column SB DCDT, Inside Face
44	FLR DSP-MA1	1st Floor Absolute Displacement
45	FLR DSP-MA2	2nd Floor Absolute Displacement
46	FLR DSP-NA3	3rd Floor Absolute Displacement at Column NA
47	FLR DSP-SA3	3rd Floor Absolute Displacement at Column SA
48	FLR ACC-1	1st Floor Absolute Acceleration
49	FLR ACC-2	2nd Floor Absolute Acceleration
50	FLR ACC-N3	3rd Floor Absolute Acceleration, Frame N
51	FLR ACC-S3	3rd Floor Absolute Acceleration, Frame S
52	CPYS-NABO	1st Floor Column NA Post-Yield Flexural Strain, Bottom End, Outside Face
53	CPYS-NABI	1st Floor Column NA Post-Yield Flexural Strain, Bottom End, Inside Face
54	CPYS-NBBO	1st Floor Column NB Post-Yield Flexural Strain, Bottom End, Outside Face
55	CPYS-NBBI	1st Floor Column NB Post-Yield Flexural Strain, Bottom End, Inside Face
56	CESTR-NAB1	1st Floor Column NA Elastic Flexural Strain, Bottom End
57	CESTR-NAT1	1st Floor Column NA Elastic Flexural Strain, Top End
58	CESTR-NBB1	1st Floor Column NB Elastic Flexural Strain, Bottom End
59	CESTR-NBT1	1st Floor Column NB Elastic Flexural Strain, Top End
60	CESTR-SAB1	1st Floor Column SA Elastic Flexural Strain, Bottom End
61	CESTR-SAT1	1st Floor Column SA Elastic Flexural Strain, Top End
62	CESTR-SBB1	1st Floor Column SB Elastic Flexural Strain, Bottom End
63	CESTR-SBT1	1st Floor Column SB Elastic Flexural Strain, Top End
64	CESTR-NAB2	2nd Floor Column NA Elastic Flexural Strain, Bottom End
65	CESTR-NAT2	2nd Floor Column NA Elastic Flexural Strain, Top End

Channel No.	Name	Description
66	CESTR-NBB2	2nd Floor Column NB Elastic Flexural Strain, Bottom End
67	CESTR-NBT2	2nd Floor Column NB Elastic Flexural Strain, Top End
68	CESTR-SAB2	2nd Floor Column SA Elastic Flexural Strain, Bottom End
69	CESTR-SAT2	2nd Floor Column SA Elastic Flexural Strain, Top End
70	CESTR-SBB2	2nd Floor Column SB Elastic Flexural Strain, Bottom End
71	CESTR-SBT2	2nd Floor Column SB Elastic Flexural Strain, Top End
72	BESTR-LN1	1st Floor Beam Elastic Flexural Strain, Left End, Frame N
73	BESTR-RN1	1st Floor Beam Elastic Flexural Strain, Right End, Frame N
74	PPYS-NN1	1st Floor Pipe Post-Yield Axial Strain, Negative Direction, Frame N
75	PPYS-PN1	1st Floor Pipe Post-Yield Axial Strain, Positive Direction, Frame N
76	PPYS-NS1	1st Floor Pipe Post-Yield Axial Strain, Negative Direction, Frame S
77	PPYS-PS1	1st Floor Pipe Post-Yield Axial Strain, Positive Direction, Frame S
78	PPYS-PN2	2nd Floor Pipe Post-Yield Axial Strain, Positive Direction, Frame N
79	PPYS-NN2	2nd Floor Pipe Post-Yield Axial Strain, Negative Direction, Frame N
80	PPYS-PS2	2nd Floor Pipe Post-Yield Axial Strain, Positive Direction, Frame S
81	PPYS-NS2	2nd Floor Pipe Post-Yield Axial Strain, Negative Direction, Frame S
82	PPYS-PN3	3rd Floor Pipe Post-Yield Axial Strain, Positive Direction, Frame N
83	PPYS-NN3	3rd Floor Pipe Post-Yield Axial Strain, Negative Direction, Frame N
84	PPYS-PS3	3rd Floor Pipe Post-Yield Axial Strain, Positive Direction, Frame S
85	PPYS-NS3	3rd Floor Pipe Post-Yield Axial Strain, Negative Direction, Frame S
86	PPYS-NTS	1st Flr Pipe Post-Yield Strain, Negative Direction, Top Face, Frame S
87	PPYS-NBS	1st Flr Pipe Post-Yield Strain, Negative Direction, Bottom Face, Frame S
88	PPYS-NOS	1st Flr Pipe Post-Yield Strain, Negative Direction, Outside Face, Frame S

Channel No.	Name	Description
89	PPYS-NIS	1st Flr Pipe Post-Yield Strain, Negative Direction, Inside Face, Frame S
90	PPYS-NTN	1st Flr Pipe Post-Yield Strain, Negative Direction, Top Face, Frame N
91	PPYS-NBN	1st Flr Pipe Post-Yield Strain, Negative Direction, Bottom Face, Frame N
92	PPYS-NON	1st Flr Pipe Post-Yield Strain, Negative Direction, Outside Face, Frame N
93	PPYS-NIN	1st Flr Pipe Post-Yield Strain, Negative Direction, Inside Face, Frame N
94	CLFRC-NA	1st Floor Column NA Elastic Axial Strain
95	CLFRC-NB	1st Floor Column NB Elastic Axial Strain
96	CLFRC-SA	1st Floor Column SA Elastic Axial Strain
97	CLFRC-SB	1st Floor Column SB Elastic Axial Strain
98	CPYS-NAT	1st Floor Column NA Post-Yield Flexural Strain, Top End
99	CPYS-NBT	1st Floor Column NB Post-Yield Flexural Strain, Top End
100	PPYS-PTN	1st Flr Pipe Post-Yield Strain, Positive Direction, Top Face, Frame N
101	PPYS-PBN	1st Flr Pipe Post-Yield Strain, Positive Direction, Bottom Face, Frame N
102	PPYS-PON	1st Flr Pipe Post-Yield Strain, Positive Direction, Outside Face, Frame N
103	PPYS-PIN	1st Flr Pipe Post-Yield Strain, Positive Direction, Inside Face, Frame N

Channel No.	Name	Description
0	T/R ACC-1	Command Horizontal Table Acceleration Signal
1	T/R ACC-2	Command Vertical Table Acceleration Signal
2	CMD H DISP	Command Horizontal Table Displacement Signal
3	CMD V DISP	Command Vertical Table Displacement Signal
4	AV H T DISP	Average Horizontal Table Displacement
5	AV V T DISP	Average Vertical Table Displacement
6	AV H T ACC	Average Horizontal Table Acceleration
7	AV V T ACC	Average Vertical Table Acceleration
8	PITCH	Angular Table Acceleration in Pitching Mode
9	ROLL	Angular Table Acceleration in Rolling Mode
10	TWIST	Angular Table Acceleration in Twisting Mode
11	FORCE H1	Force in Horizontal Actuator H1
12	FORCE H2	Force in Horizontal Actuator H2
13	FORCE H3	Force in Horizontal Actuator H3
14	ACC H1	Horizontal Table Acceleration at Actuator H1
15	ACC H2	Horizontal Table Acceleration at Actuator H2
16	ACC V1	Vertical Table Acceleration at Actuator V1
17	ACC V2	Vertical Table Acceleration at Actuator V2
18	ACC V3	Vertical Table Acceleration at Actuator V3
19	ACC V4	Vertical Table Acceleration at Actuator V4
20	FORCE V1	Force in Vertical Actuator V1
21	FORCE V2	Force in Vertical Actuator V2

Table A-3 Data channel listing for diagonal double angle bracing

Channel No.	Name	Description
22	FORCE V3	Force in Vertical Actuator V3
23	FORCE V4	Force in Vertical Actuator V4
24	DISP V1	Vertical Displacement at Actuator V1
25	DISP V2	Vertical Displacement at Actuator V2
26	DISP V3	Vertical Displacement at Actuator V3
27	DISP V4	Vertical Displacement at Actuator V4
28	DISP H1	Horizontal Table Displacement at Actuator H1
29	DISP H2	Horizontal Table Displacement at Actuator H2
30	DISP H3	Horizontal Table Displacement at Actuator H3
31	BLANK	
32	PS FORCE-1	Force in Passive Stabilizer 1
33	PS FORCE-2	Force in Passive Stabilizer 2
34	PS FORCE-3	Force in Passive Stabilizer 3
35	PS FORCE-4	Force in Passive Stabilizer 4
36	CLELG-NAOF	1st Floor Column NA DCDT, Outside Face
37	CLELG-NAIF	1st Floor Column NA DCDT, Inside Face
38	CLELG-NBOF	1st Floor Column NB DCDT, Outside Face
39	CLELG-NBIF	1st Floor Column NB DCDT, Inside Face
40	CLELG-SAOF	1st Floor Column SA DCDT, Outside Face
41	CLELG-SAIF	1st Floor Column SA DCDT, Inside Face
42	CLELG-SBOF	1st Floor Column SB DCDT, Outside Face
43	CLELG-SBIF	1st Floor Column SB DCDT, Inside Face

Channel No.	Name	Description
44	FLR DSP-MA1	1st Floor Absolute Displacement
45	FLR DSP-MA2	2nd Floor Absolute Displacement
46	FLR DSP-NA3	3rd Floor Absolute Displacement at Column NA
47	FLR DSP-SA3	3rd Floor Absolute Displacement at Column SA
48	FLR ACC-1	1st Floor Absolute Acceleration
49	FLR ACC-2	2nd Floor Absolute Acceleration
50	FLR ACC-N3	3rd Floor Absolute Acceleration, Frame N
51	FLR ACC-S3	3rd Floor Absolute Acceleration, Frame S
52	CPYS-NABO	1st Floor Column NA Post-Yield Flexural Strain, Bottom End, Outside Face
53	CPYS-NABI	1st Floor Column NA Post-Yield Flexural Strain, Bottom End, Inside Face
54	CPYS-NBBO	1st Floor Column NB Post-Yield Flexural Strain, Bottom End, Outside Face
55	CPYS-NBBI	1st Floor Column NB Post-Yield Flexural Strain, Bottom End, Inside Face
56	CESTR-NAB1	1st Floor Column NA Elastic Flexural Strain, Bottom End
57	CESTR-NAT1	1st Floor Column NA Elastic Flexural Strain, Top End
58	CESTR-NBB1	1st Floor Column NB Elastic Flexural Strain, Bottom End
59	CESTR-NBT1	1st Floor Column NB Elastic Flexural Strain, Top End
60	CESTR-SAB1	1st Floor Column SA Elastic Flexural Strain, Bottom End
61	CESTR-SAT1	1st Floor Column SA Elastic Flexural Strain, Top End
62	CESTR-SBB1	1st Floor Column SB Elastic Flexural Strain, Bottom End
63	CESTR-SBT1	1st Floor Column SB Elastic Flexural Strain, Top End
64	CESTR-NAB2	2nd Floor Column NA Elastic Flexural Strain, Bottom End
65	CESTR-NAT2	2nd Floor Column NA Elastic Flexural Strain, Top End
66	CESTR-NBB2	2nd Floor Column NB Elastic Flexural Strain, Bottom End

Channel No.	Name	Description
67	CESTR-NBT2	2nd Floor Column NB Elastic Flexural Strain, Top End
68	CESTR-SAB2	2nd Floor Column SA Elastic Flexural Strain, Bottom End
69	CESTR-SAT2	2nd Floor Column SA Elastic Flexural Strain, Top End
70	CESTR-SBB2	2nd Floor Column SB Elastic Flexural Strain, Bottom End
71	CESTR-SBT2	2nd Floor Column SB Elastic Flexural Strain, Top End
72	BESTR-LN1	1st Floor Beam Elastic Flexural Strain, Left End, Frame N
73	BESTR-RN1	1st Floor Beam Elastic Flexural Strain, Right End, Frame N
74	DAPYS-PN1	1st Flr Angle Post-Yield Axial Strain, Positive Direction, Frame N
75	DAPYS-NN1	1st Flr Angle Post-Yield Axial Strain, Negative Direction, Frame N
76	DAPYS-PS1	1st Flr Angle Post-Yield Axial Strain, Positive Direction, Frame S
77	DAPYS-NS1	1st Flr Angle Post-Yield Axial Strain, Negative Direction, Frame S
78	DAPYS-PN2	2nd Flr Angle Post-Yield Axial Strain, Positive Direction, Frame N
79	DAPYS-NN2	2nd Flr Angle Post-Yield Axial Strain, Negative Direction, Frame N
80	DAPYS-PS2	2nd Flr Angle Post-Yield Axial Strain, Positive Direction, Frame S
81	DAPYS-NS2	2nd Flr Angle Post-Yield Axial Strain, Negative Direction, Frame S
82	DAPYS-PN3	3rd Flr Angle Post-Yield Axial Strain, Positive Direction, Frame N
83	DAPYS-NN3	3rd Flr Angle Post-Yield Axial Strain, Negative Direction, Frame N
84	DAPYS-PS3	3rd Flr Angle Post-Yield Axial Strain, Positive Direction, Frame S
85	DAPYS-NS3	3rd Flr Angle Post-Yield Axial Strain, Negative Direction, Frame S
86	DAPYS-PBIF	1st Flr Angle Post-Yield Strain, Positive Direction, Bottom Inside Face
87	DAPYS-PBOF	1st Flr Angle Post-Yield Strain, Positive Direction, Bottom Outside Face
88	DAPYS-PTIF	1st Flr Angle Post-Yield Strain, Positive Direction, Top Inside Face
89	DAPYS-PTOF	1st Flr Angle Post-Yield Strain, Positive Direction, Top Outside Face

Channel No.	Name	Description
90	DAPYS-NBOF	1st Flr Angle Post-Yield Strain, Negative Direction, Bottom Outside Face
91	DAPYS-NBIF	1st Flr Angle Post-Yield Strain, Negative Direction, Bottom Inside Face
92	DAPYS-NTOF	1st Flr Angle Post-Yield Strain, Negative Direction, Top Outside Face
93	DAPYS-NTIF	1st Flr Angle Post-Yield Strain, Negative Direction, Top Inside Face
94	CLFRC-NA	1st Floor Column NA Elastic Axial Strain
95	CLFRC-NB	1st Floor Column NB Elastic Axial Strain
96	CLFRC-SA	1st Floor Column SA Elastic Axial Strain
97	CLFRC-SB	1st Floor Column SB Elastic Axial Strain
98	CPYS-NAT	1st Floor Column NA Post-Yield Flexural Strain, Top End
99	CPYS-NBT	1st Floor Column NB Post-Yield Flexural Strain, Top End
100	DAPYS-PMBO	1st Flr Angle P.Y. Strain, Positive Direction, Mid-Section Bottom Outside
101	DAPYS-PMBI	1st Flr Angle P.Y. Strain, Positive Direction, Mid-Section Bottom Inside
102	DAPYS-PMTO	1st Flr Angle P.Y. Strain, Positive Direction, Mid-Section Top Outside Face
103	DAPYS-PMTI	1st Flr Angle P.Y. Strain, Positive Direction, Mid-Section Top Inside Face

Channel No.	Name	Description
0	T/R ACC-1	Command Horizontal Table Acceleration Signal
1	T/R ACC-2	Command Vertical Table Acceleration Signal
2	CMD H DISP	Command Horizontal Table Displacement Signal
3	CMD V DISP	Command Vertical Table Displacement Signal
4	AV H T DISP	Average Horizontal Table Displacement
5	AV V T DISP	Average Vertical Table Displacement
6	AV H T ACC	Average Horizontal Table Acceleration
7	AV V T ACC	Average Vertical Table Acceleration
8	PITCH	Angular Table Acceleration in Pitching Mode
9	ROLL	Angular Table Acceleration in Rolling Mode
10	TWIST	Angular Table Acceleration in Twisting Mode
11	FORCE H1	Force in Horizontal Actuator H1
12	FORCE H2	Force in Horizontal Actuator H2
13	FORCE H3	Force in Horizontal Actuator H3
14	ACC H1	Horizontal Table Acceleration at Actuator H1
15	ACC H2	Horizontal Table Acceleration at Actuator H2
16	ACC V1	Vertical Table Acceleration at Actuator V1
17	ACC V2	Vertical Table Acceleration at Actuator V2
18	ACC V3	Vertical Table Acceleration at Actuator V3
19	ACC V4	Vertical Table Acceleration at Actuator V4
20	FORCE V1	Force in Vertical Actuator V1

Table A-4 Dta channel listing for diagonal double angle bracing- Phase II

Channel No.	Name	Description
21	FORCE V2	Force in Vertical Actuator V2
22	FORCE V3	Force in Vertical Actuator V3
23	FORCE V4	Force in Vertical Actuator V4
24	DISP V1	Vertical Table Displacement at Actuator V1
25	DISP V2	Vertical Table Displacement at Actuator V2
26	DISP V3	Vertical Table Displacement at Actuator V3
27	DISP V4	Vertical Table Displacement at Actuator V4
28	DISP H1	Horizontal Table Displacement at Actuator H1
29	DISP H2	Horizontal Table Displacement at Actuator H2
30	DISP H3	Horizontal Table Displacement at Actuator H3
31	BLANK	
32	DAPYS-PBOS	1st Flr Angle P.Y.Strain, Pos. Direction, Bottom Outside Face, Frame S
33	DAPYS-PBIS	1st Flr Angle P.Y.Strain, Pos. Direction, Bottom Inside Face, Frame S
34	DAPYS-PTOS	1st Flr Angle P.Y.Strain, Pos. Direction, Top Outside Face, Frame S
35	DAPYS-PTIS	1st Flr Angle P.Y. Strain, Pos. Direction, Top Inside Face, Frame S
36	CLELG-NAOF	1st Floor Column NA DCDT, Outside Face
37	CLELG-NAIF	1st Floor Column NA DCDT, Inside Face
38	CLELG-NBOF	1st Floor Column NB DCDT, Outside Face
39	CLELG-NBIF	1st Floor Column NB DCDT, Inside Face
40	CLELG-SAOF	1st Floor Column SA DCDT, Outside Face
41	CLELG-SAIF	1st Floor Column SA DCDT, Inside Face
42	CLELG-SBOF	1st Floor Column SB DCDT, Outside Face

Channel No.	Name	Description
43	CLELG-SBIF	1st Floor Column SB DCDT, Inside Face
44	FLR DSP-MA1	1st Floor Absolute Displacement
45	FLR DSP-MA2	2nd Floor Absolute Displacement
46	FLR DSP-NA3	3rd Floor Absolute Displacement at Column NA
47	FLR DSP-SA3	3rd Floor Absolute Displacement at Column SA
48	FLR ACC-1	1st Floor Absolute Acceleration
49	FLR ACC-2	2nd Floor Absolute Acceleration
50	FLR ACC-N3	3rd Floor Absolute Acceleration, Frame N
51	FLR ACC-S3	3rd Floor Absolute Acceleration, Frame S
52	CPYS-NABO	1st Floor Column NA P.Y. Flexural Strain, Bottom End, Outside Face
53	CPYS-NABI	1st Floor Column NA P.Y. Flexural Strain, Bottom End, Inside Face
54	CPYS-NBBO	1st Floor Column NB P.Y. Flexural Strain, Bottom End, Outside Face
55	CPYS-NBBI	1st Floor Column NB P.Y. Flexural Strain, Bottom End, Inside Face
56	CESTR-NAB1	1st Floor Column NA Elastic Flexural Strain, Bottom End
57	CESTR-NAT1	1st Floor Column NA Elastic Flexural Strain, Top End
58	CESTR-NBB1	1st Floor Column NB Elastic Flexural Strain, Bottom End
59	CESTR-NBT1	1st Floor Column NB Elastic Flexural Strain, Top End
60	CESTR-SAB1	1st Floor Column SA Elastic Flexural Strain, Bottom End
61	CESTR-SAT1	1st Floor Column SA Elastic Flexural Strain, Top End
62	CESTR-SBB1	1st Floor Column SB Elastic Flexural Strain, Bottom End
63	CESTR-SBT1	1st Floor Column SB Elastic Flexural Strain, Top End
64	CESTR-NAB2	2nd Floor Column NA Elastic Flexural Strain, Bottom End

Channel No.	Name	Description
65	CESTR-NAT2	2nd Floor Column NA Elastic Flexural Strain, Top End
66	CESTR-NBB2	2nd Floor Column NB Elastic Flexural Strain, Bottom End
67	CESTR-NBT2	2nd Floor Column NB Elastic Flexural Strain, Top End
68	CESTR-SAB2	2nd Floor Column SA Elastic Flexural Strain, Bottom End
69	CESTR-SAT2	2nd Floor Column SA Elastic Flexural Strain, Top End
70	CESTR-SBB2	2nd Floor Column SB Elastic Flexural Strain, Bottom End
71	CESTR-SBT2	2nd Floor Column SB Elastic Flexural Strain, Top End
72	BESTR-LN1	1st Floor Beam Elastic Flexural Strain, Left End, Frame N
73	BESTR-RN1	1st Floor Beam Elastic Flexural Strain, Right End, Frame N
74	DAPYS-NN1	1st Flr Angle P.Y. Axial Strain, Negative Direction, Frame N
75	DAPYS-PN1	1st Flr Angle P.Y. Axial Strain, Positive Direction, Frame N
76	DAPYS-NS1	1st Flr Angle P.Y. Axial Strain, Negative Direction, Frame S
77	DAPYS-PS1	1st Flr Angle P.Y. Axial Strain, Positive Direction, Frame S
78	DAPYS-PN2	2nd Flr Angle P.Y. Axial Strain, Positive Direction, Frame N
79	DAPYS-NN2	2nd Flr Angle P.Y. Axial Strain, Negative Direction, Frame N
80	DAPYS-PS2	2nd Flr Angle P.Y. Axial Strain, Positive Direction, Frame S
81	DAPYS-NS2	2nd Flr Angle P.Y. Axial Strain, Negative Direction, Frame S
82	DAPYS-PN3	3rd Flr Angle P.Y. Axial Strain, Positive Direction, Frame N
83	DAPYS-NN3	3rd Flr Angle P.Y. Axial Strain, Negative Direction, Frame N
84	DAPYS-PS3	3rd Flr Angle P.Y. Axial Strain, Positive Direction, Frame S
85	DAPYS-NS3	3rd Flr Angle P.Y. Axial Strain, Negative Direction, Frame S
86	DAPYS-NBOS	1st Flr Angle P.Y. Strain, Neg. Direction, Bottom Outside Face, Frame S

Channel No.	Name	Description
87	DAPYS-NBIS	1st Flr Angle P.Y. Strain, Neg. Direction, Bottom Inside Face, Frame S
83	DAPYS-NTOS	1st Flr Angle P.Y. Strain, Neg. Direction, Top Outside Face, Frame S
89	DAPYS-NTIS	1st Flr Angle P.Y. Strain, Neg. Direction, Top Inside Face, Frame S
90	DAPYS-NBON	1st Flr Angle P.Y. Strain, Neg. Direction, Bottom Outside Face, Frame N
91	DAPYS-NBIN	1st Flr Angle P.Y. Strain, Neg. Direction, Bottom Inside Face, Frame N
92	DAPYS-NTON	1st Flr Angle P.Y. Strain, Neg. Direction, Top Outside Face, Frame N
93	DAPYS-NTIN	1st Flr Angle P.Y. Strain, Neg. Direction, Top Inside Face, Frame N
94	CLFRC-NA	1st Floor Column NA Elastic Axial Strain
95	CLFRC-NB	1st Floor Column NB Elastic Axial Strain
96	CLFRC-SA	1st Floor Column SA Elastic Axial Strain
97	CLFRC-SB	1st Floor Column SB Elastic Axial Strain
98	CPYS-NAT	1st Floor Column NA Post-Yield Flexural Strain, Top End
99	CPYS-NBT	1st Floor Column NB Post-Yield Flexural Strain, Top End
100	DAPYS-PBON	1st Flr Angle P.Y. Strain, Pos. Direction, Bottom Outside Face, Frame N
101	DAPYS-PBIN	1st Flr Angle P.Y. Strain, Pos. Direction, Bottom Inside Face, Frame N
102	DAPYS-PTON	1st Flr Angle P.Y. Strain, Pos. Direction, Top Outside Face, Frame N
103	DAPYS-PTIN	1st Flr Angle P.Y. Strain, Pos. Direction, Top Inside Face, Frame N

Channel No.	Name	Description
0	T/R ACC-1	Command Horizontal Table Acceleration Signal
1	T/R ACC-2	Command Vertical Table Acceleration Signal
2	CMD H DISP	Command Horizontal Table Displacement Signal
3	CMD V DISP	Command Vertical Table Displacement Signal
4	AV H T DISP	Average Horizontal Table Displacement
5	AV V T DISP	Average Vertical Table Displacement
6	AV H T ACC	Average Horizontal Table Acceleration
7	AV V T ACC	Average Vertical Table Acceleration
8	PITCH	Angular Table Acceleration in Pitching Mode
9	ROLL	Angular Table Acceleration in Rolling Mode
10	TWIST	Angular Table Acceleration in Twisting Mode
11	FORCE H1	Force in Horizontal Actuator H1
12	FORCE H2	Force in Horizontal Actuator H2
13	FORCE H3	Force in Horizontal Actuator H3
14	ACC H1	Horizontal Table Acceleration at Actuator H1
15	ACC H2	Horizontal Table Acceleration at Actuator H2
16	ACC V1	Vertical Table Acceleration at Actuator V1
17	ACC V2	Vertical Table Acceleration at Actuator V2
18	ACC V3	Vertical Table Acceleration at Actuator V3
19	ACC V4	Vertical Table Acceleration at Actuator V4
20	FORCE V1	Force in Vertical Actuator V1

Table A-5 Data Channel Listing for Unbraced Structure

Channel No.	Name	Description
21	FORCE V2	Force in Vertical Actuator V2
22	FORCE V3	Force in Vertical Actuator V3
23	FORCE V4	Force in Vertical Actuator V4
24	DISP V1	Vertical Table Displacement at Actuator V1
25	DISP V2	Vertical Table Displacement at Actuator V2
26	DISP V3	Vertical Table Displacement at Actuator V3
27	DISP V4	Vertical Table Displacement at Actuator V4
28	DISP H1	Horizontal Table Displacement at Actuator H1
29	DISP H2	Horizontal Table Displacement at Actuator H2
30	DISP H3	Horizontal Table Displacement at Actuator H3
31	BLANK	
:	:	
:	:	
36	CLELG-NAOF	1st Floor Column NA DCDT, Outside Face
37	CLELG-NAIF	1st Floor Column NA DCDT, Inside Face
38	CLELG-NBOF	1st Floor Column NB DCDT, Outside Face
39	CLELG-NBIF	1st Floor Column NB DCDT, Inside Face
40	CLELG-SAOF	1st Floor Column SA DCDT, Outside Face
41	CLELG-SAIF	1st Floor Column SA DCDT, Inside Face
42	CLELG-SBOF	1st Floor Column SB DCDT, Outside Face
43	CLELG-SBIF	1st Floor Column SB DCDT, Inside Face
44	FLR DSP-MA1	1st Floor Absolute Displacement
45	FLR DSP-MA2	2nd Floor Absolute Displacement

Channel No.	Name	Description
46	FLR DSP-NA3	3rd Floor Absolute Displacement at Column NA
47	FLR DSP-SA3	3rd Floor Absolute Displacement at Column SA
48	FLR ACC-1	1st Floor Absolute Acceleration
49	FLR ACC-2	2nd Floor Absolute Acceleration
50	FLR ACC-N3	3rd Floor Absolute Acceleration, Frame N
51	FLR ACC-S3	3rd Floor Absolute Acceleration, Frame S
52	CPYS-NABO	1st Floor Column NA Post-Yield Flexural Strain, Bottom End, Outside Face
53	CPYS-NABI	1st Floor Column NA Post-Yield Flexural Strain, Bottom End, Inside Face
54	CPYS-NBBO	1st Floor Column NB Post-Yield Flexural Strain, Bottom End, Outside Face
55	CPYS-NBBI	1st Floor Column NB Post-Yield Flexural Strain, Bottom End, Inside Face
56	CESTR-NAB1	1st Floor Column NA Elastic Flexural Strain, Bottom End
57	CESTR-NAT1	1st Floor Column NA Elastic Flexural Strain, Top End
58	CESTR-NBB1	1st Floor Column NB Elastic Flexural Strain, Bottom End
59	CESTR-NBT1	1st Floor Column NB Elastic Flexural Strain, Top End
60	CESTR-SAB1	1st Floor Column SA Elastic Flexural Strain, Bottom End
61	CESTR-SAT1	1st Floor Column SA Elastic Flexural Strain, Top End
62	CESTR-SBB1	1st Floor Column SB Elastic Flexural Strain, Bottom End
63	CESTR-SBT1	1st Floor Column SB Elastic Flexural Strain, Top End
64	CESTR-NAB2	2nd Floor Column NA Elastic Flexural Strain, Bottom End
65	CESTR-NAT2	2nd Floor Column NA Elastic Flexural Strain, Top End
66	CESTR-NBB2	2nd Floor Column NB Elastic Flexural Strain, Bottom End
67	CESTR-NBT2	2nd Floor Column NB Elastic Flexural Strain, Top End
68	CESTR-SAB2	2nd Floor Column SA Elastic Flexural Strain, Bottom End

Channel No.	Name	Description
69	CESTR-SAT2	2nd Floor Column SA Elastic Flexural Strain, Top End
70	CESTR-SBB2	2nd Floor Column SB Elastic Flexural Strain, Bottom End
71	CESTR-SBT2	2nd Floor Column SB Elastic Flexural Strain, Top End
72	BESTR-LN1	1st Floor Beam Elastic Flexural Strain, Left End, Frame N
73	BESTR-RN1	1st Floor Beam Elastic Flexural Strain, Right End, Frame N
74	BLANK	
⋮	⋮	
94	CLFRC-NA	1st Floor Column NA Elastic Axial Strain
95	CLFRC-NB	1st Floor Column NB Elastic Axial Strain
96	CLFRC-SA	1st Floor Column SA Elastic Axial Strain
97	CLFRC-SB	1st Floor Column SB Elastic Axial Strain

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- EERC 68-1 Unassigned
- EERC 68-2 "Inelastic Behavior of Beam-to-Column Subassemblages Under Repeated Loading," by V.V. Bertero - 1968 (PB 184 888)A05
- EERC 68-3 "A Graphical Method for Solving the Wave Reflection-Refraction Problem," by H.D. McNiven and Y. Mengi - 1968 (PB 187 943)A03
- EERC 68-4 "Dynamic Properties of McKinley School Buildings," by D. Rea, J.G. Bouwkamp and R.W. Clough - 1968 (PB 187 902)A07
- EERC 68-5 "Characteristics of Rock Motions During Earthquakes," by H.B. Seed, I.M. Idriss and F.W. Kiefer - 1968 (PB 188 338)A03
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- EERC 69-7 "Rock Motion Accelerograms for High Magnitude Earthquakes," by H.B. Seed and I.M. Idriss - 1969 (PB 187 940)A02
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EERC-5

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