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**VERTICAL MOTION
OF HIGHWAY BRIDGE
STRUCTURES DUE TO
AN EARTHQUAKE**

BY

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VERTICAL MOTION OF HIGHWAY BRIDGE STRUCTURES
DUE TO AN EARTHQUAKE

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ABSTRACT

A study of the dynamic response of continuous highway bridge superstructures due to pulsating support settlements induced by an earthquake is presented. A governing differential equation for the dynamic behavior of bridge superstructures under pulsating support settlements has been formulated and a Fourier series type solution method is presented. The results thus obtained have been successfully compared to those obtained by a dynamic three moment equation. Four separate computer programs have been developed to facilitate the analysis and comparisons. The concept of dynamic amplification factors as the ratios of dynamic responses to static responses has been developed and subsequent charts are plotted. Unusually high dynamic amplifications are observed, particularly in shearing forces and the reasons for such high values have been traced.

KEY WORDS

Earthquake; Highway bridge; Pulsating support settlement; Dynamic three moment equation; Dynamic amplification factor; Natural frequency; Structural analysis; Structural design.

PREFACE

The investigation with interpretation as described in this report was sponsored by the National Science Foundation, under Grant No. PFR78-22845 covering the period January 1, 1979 through December 31, 1980.

The general investigation reported herein is under the supervision and Technical responsibility of Professors Chai Hong Yoo and Jao-Shiun Kao. Professor Yoo acts as principal investigator.

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I. INTRODUCTION

A. Statement of the Problem

One of the most prevalent and destructive forms of dynamic disturbances is the earthquake. Contrary to the commonly held belief that the California Coast is the only area subject to strong seismic disturbances, the records of the United States Coast and Geodetic Survey show that these destructive forces have been felt all over the country [22]*. Moreover, these records suggest that where significant shocks have occurred, reoccurrences are very probable, and that in regions of little seismic activity, major shocks are not to be ruled out entirely.

Up to the end of the nineteenth century, little, if any, attention was given to the earthquake forces in structural design. With the high concentration of the population in the twentieth century cities on one side, and with the development of displacement meters and accelerometers capable of recording earthquake motions on the other, more studies about the earthquake emerged in this century. The majority of these studies, however, focus on the horizontal component of the ground motion and on design provisions that would guard against failure due to this motion [9,10,13]. The vertical component of these forces may have a considerable effect on structures with larger horizontal dimensions such as highway bridges.

A review of the literature has revealed that the American highway and railroad bridge and design codes (AASHTO and AREA) do not specify any provisions against the vertical effect of earthquakes on bridges as opposed to the Japanese code which requires a vertical seismic coefficient of 0.1 to be considered in bridge design [1,7,19]. Recent

*Numbers in square brackets refer to Reference numbers

research reports on highway bridge structures [5,21] from the University of California, Berkeley, are rigorous but do not include the effects of pulsating support settlement on the superstructure. Dynamic differential support settlements are very likely to prevail during an earthquake because of the difference in vertical excitation as well as different soil properties under each support. It is therefore the objective of this investigation to establish appropriate mathematical models which will yield realistic and practical seismic responses for a reasonable range of bridge superstructures due to vertical disturbances. Further, computer programs are written to carry out time - history dynamic analysis as well as to determine natural frequencies of bridge superstructures.

B. Scope of Present Investigation

The present investigation is a study of the dynamic response of highway bridge superstructures due to pulsating support settlements induced by earthquake excitation.

Chapter II presents the formulation of a governing differential equation for the dynamic behavior of bridge superstructures under pulsating support settlements. A Fourier series type solution method is discussed. A dynamic three moment equation is also discussed in Chapter II. Chapter III describes the general characteristic of four computer programs developed and the necessary input scheme and variables. Chapter IV discusses the computer programs further by using an actual example bridge analysis. The accuracy of the solution of the two methods is presented. The concept of dynamic amplification factors

is developed and subsequent charts are included. Chapter V presents certain conclusions and recommendations. Finally, sample input and output data and the listings of the computer programs are presented in the Appendices.

II. METHOD OF ANALYSIS

A. General Discussion

A highway bridge superstructure may be subjected to additional stresses from the vertical excitations of an earthquake, arising from:

- 1) The inertial force acting on the bridge as a result of the rigid body translation and
- 2) The effect of the dynamic differential settlement of one or more of the piers. The stresses due to the inertial force can be readily obtained by assuming that the bridge is subjected to an additional static-equivalent uniformly distributed load of a magnitude of ma , where m is mass per unit length of bridge superstructure and a is acceleration of the rigid body translation. More rigorous analyses using the finite element method are found in Ref. [5,21].

As discussed earlier, the additional stresses due to the dynamic differential settlements require further analysis. Presented in this chapter is a mathematical formulation to determine the additional stresses and deflections resulting from these settlements.

B. Fourier Series Formulation

A multispan highway bridge superstructure is modelled by a wide continuous beam with both ends pinned and intermediate supports. The beam is considered elastic and prismatic and the effects of axial and shear deformations, rotary inertia, and damping are not included in the formulation simply to make the model reasonably uncomplicated. It is of interest to determine the maximum deflection, moment, shear, velocity

and acceleration at any time for any point along the span of a continuous beam subjected to dynamic differential support settlements. The steps taken to determine the maximum values of these functions are as follows:

- 1) The intermediate supports are removed, leaving an ordinary simple beam for which the differential equation, the vibration frequencies and normalized shape functions are known.
- 2) In lieu of the removed intermediate supports, constraints equivalent to the presence of these supports are imposed on the beam. The constraints are first transformed to concentrated loads on the simple beam, and then substituted by a generalized Fourier series in terms of normalized shape functions of the simple beam.
- 3) The deflection of the simple beam at any point along the span is also represented as a generalized Fourier series in terms of normalized shape functions of the beam.
- 4) Using the constraints at the intermediate supports as the necessary boundary conditions, the deflection function of the continuous beam is determined.
- 5) By successive differentiation of the deflection function with respect to the longitudinal coordinate along the span or with respect to time, moment and shear or velocity and acceleration functions are determined for the continuous beam.
- 6) Numerical values and subsequently the maximum values of these functions are evaluated by means of a computer program developed.

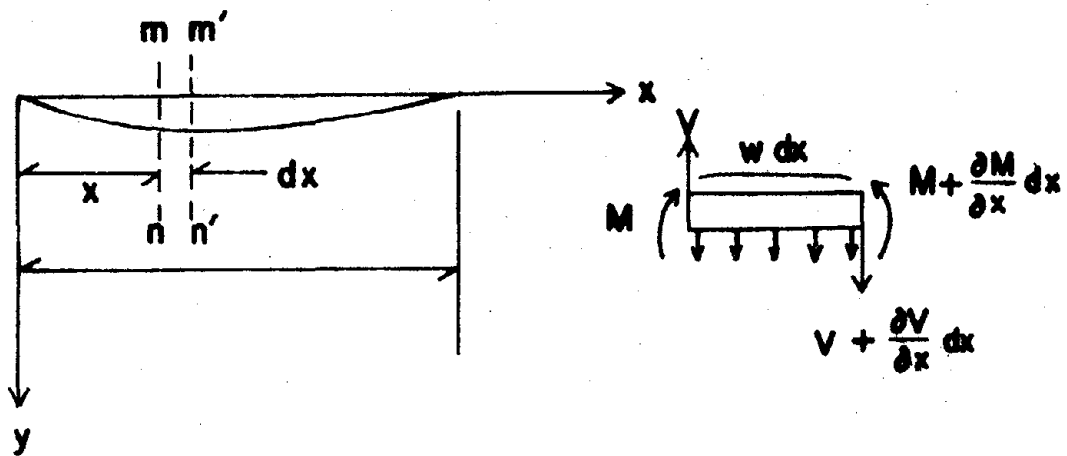
A similar procedure was used by Rogers [17] to investigate the dynamic behavior of a continuous beam subjected to a series of pulsating loads. Saibel and D'Appolonia [18] studied similar problems based on the energy of the system and Lagrangian constraints.

The differential equation governing the vibration behavior of a simple beam subjected to lateral forces as shown in Fig. 1 is given as

$$EI \frac{\partial^4 y}{\partial x^4} + m \frac{\partial^2 y}{\partial t^2} = w(x,t) \quad (1)$$

in which E is modulus of elasticity, I is moment of inertia of the beam cross section, m is mass per unit length and $w(x,t)$ is distributed lateral load.

Fig. 1 - Coordinates and Force Elements



The natural frequencies of the simple beam are given as:

$$P_n = n^2 \pi^2 \sqrt{\frac{EI}{mL^4}} \quad (2)$$

in which P_n is the natural circular frequency of the simple beam vibrating in n th mode, and L is the span length of the simple beam. The normalized shape functions of the simple beams are readily obtainable and can be expressed as [4,14,17].

$$\bar{X}_n(x) = \sqrt{\frac{2}{mL}} \sin \frac{n\pi x}{L} \quad (3)$$

in which \bar{X}_n is the normalized shape function when the simple beam is vibrating in the n th mode. The deflection of the simple beam as a function of longitudinal coordinate and time is expressed as [4,17].

$$y(x,t) = \sum_n \sqrt{\frac{2}{mL}} \sin \frac{n\pi x}{L} q_n(t) \quad (4)$$

in which $q_n(t)$ is a function of time only.

The constraints imposed by pulsating supports due to vertical earthquake excitations are idealized as the harmonic motions in this investigations.

$$y(x_i,t) = \Delta_i \sin (w_f t) \quad (5)$$

in which x_i is the position of support i along the spar, Δ_i the amplitude of the i th support settlement, and w_f is the circular forcing frequency (i.e. the estimated earthquake vertical circular frequency at the bridge location). The forces which excite the motion given by Eq. (5) coincide in phase with the motion if damping is neglected. For a damped system, the forces are out of phase with the motion. As can be seen in

Eq. (4), the deflection or motion of the simple beam is given by a summation of the deflection modes from one to infinity. These modes are not necessarily in phase with each other, and thus the phase angle between the forces that produces the motion and the motion itself is a function of the mode of vibration. This can be written as

$$F_{in} = R_i \sin (w_f t - \alpha_n) \quad (6)$$

in which α_n is the phase angle between the force and the nth mode of vibration, R_i is the undetermined amplitude of the equivalent force for the support i due to the dynamic differential settlement, and F_{in} is the force at support i for the nth mode of vibration at any time t. If damping is neglected, F_{in} becomes simply:

$$F_{in} = R_i \sin w_f t \quad (7)$$

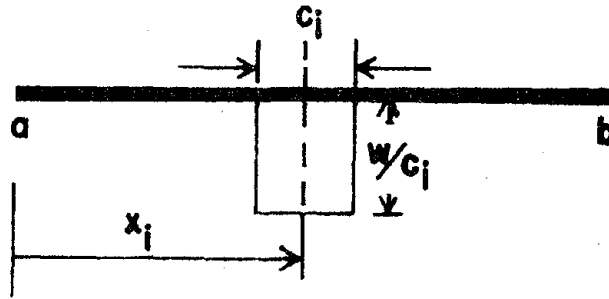
The Fourier series expression for F_{in} is given as

$$F_{in} = \sum_n A_{in} \bar{X}_n(x) \quad (8)$$

in which A_{in} is a constant for each model of vibration of the simple beam and is such that the series will correctly represent $F_{in}(t)$ along the span. Assuming that the concentrated load acts over an infinitely small distance c as shown in Fig. 2, and multiplying both sides of Eq. (8) by $m \bar{X}_k(x) dx$ and integrating over the distance range leads to:

$$\int_{x_i - \frac{c}{2}}^{x_i + \frac{c}{2}} m F_{in}(t) \bar{X}_k(x) dx = \int_{x_i - \frac{c}{2}}^{x_i + \frac{c}{2}} m \sum_n A_{in} \bar{X}_n(x) \bar{X}_k(x) dx \quad (9)$$

Fig. 2 - Fourier Series Expression of a Concentrated Load



For a beam with no energy dissipation at its supports and with a constant flexural rigidity along its span, the shape functions are orthogonal functions, Eq. (9) reduces to:

$$\int_{x_i - \frac{c}{2}}^{x_i + \frac{c}{2}} m F_{in}(t) \bar{X}_n(x) dx = A_{in} \quad (10)$$

and in the limiting case when c approaches zero, Eq. (10) is evaluated in Stieltjes' sense as

$$A_{in} = m F_{in}(A) \bar{X}_n(x_i) \quad (11)$$

For a beam with a multiple of intermediate supports, the equivalent distributed load becomes

$$w(x,t) = \sum_i \sum_n m F_{in} \bar{X}_n(x_i) \bar{X}_n(x) \quad (12)$$

The deflection of the continuous beam can also be expressed in terms of normalized shape functions as follows

$$y(x,t) = \sum_n \bar{X}_n(x) q_n(t) \quad (13)$$

in which $q_n(t)$ is a function of time only for the n th mode of vibration.

Eq. (12) is obtained by substitution of Eq. (3) into Eq. (4).

Substituting the expressing of $w(x,t)$ given in Eq. (11) into Eq. (1) yields:

$$EI \frac{\partial^4 y}{\partial x^4} + m \frac{\partial^2 y}{\partial t^2} = \sum_i \sum_n m F_{in} \bar{X}_n(x_i) \ddot{X}_n(x) \quad (14)$$

Substituting the partial derivative of Eq. (13) into Eq. (14) gives:

$$\begin{aligned} & \sum_n \frac{\partial^2}{\partial x^2} [EI \bar{X}_n''(x)] q_n(t) + \sum_n m \bar{X}_n(x) \ddot{q}_n(t) \\ &= \sum_i \sum_n m F_{in} \bar{X}_n(x_i) \ddot{X}_n(x) \end{aligned} \quad (15)$$

in which

$$\frac{\partial^2}{\partial x^2} [EI \bar{X}_n''(x)] = m P_n^2 \bar{X}_n(x) \quad (16)$$

Substituting Eq. (16) into Eq. (15) leads to

$$\ddot{q}_n(t) + P_n^2 q_n(t) = \sum_i R_i \bar{X}_n(x_i) \sin w_f t \quad (17)$$

Taking a solution to Eq. (17) to be

$$q_n(t) = D_{n1} \sin w_f t + D_{n2} \cos w_f t \quad (18)$$

Then

$$\dot{q}_n(t) = w_f D_{n1} \cos w_f t - w_f D_{n2} \sin w_f t \quad (19)$$

$$\ddot{q}_n(t) = -w_f^2 D_{n1} \sin w_f t - w_f^2 D_{n2} \cos w_f t \quad (20)$$

Substituting Eqs. (18) and (20) into Eq. (17) yields

$$\begin{aligned}
 & -w_f^2 (D_{n1} S \sin w_f t + D_{n2} \cos w_f t) + P_n^2 (P_{n1} S \sin w_f t + D_{n2} \cos w_f t) \\
 & = \sum_i R_i \bar{X}_n(x_i) S \sin w_f t
 \end{aligned} \tag{21}$$

or

$$\begin{aligned}
 & (P_n^2 - w_f^2) D_{n1} S \sin w_f t + (P_n^2 - w_f^2) D_{n2} \cos w_f t \\
 & = \sum_i R_i \bar{X}_n(x_i) S \sin w_f t
 \end{aligned} \tag{22}$$

Equating the coefficients of sine and cosine functions on both sides

gives

$$D_{n1} = \frac{\sum_i R_i \bar{X}_n(x_i)}{(P_n^2 - w_f^2)} \tag{23}$$

and

$$D_{n2} = 0 \tag{24}$$

Substituting Eqs. (18), (23) and (24) into Eq. (12) gives

$$y(x, t) = \sum_n \bar{X}_n(x) \frac{\sum_i R_i \bar{X}_n(x_i)}{P_n^2 - w_f^2} \sin w_f t \tag{25}$$

in which the magnitudes of the support force R_{is} can be determined from

the constraints given by Eq. (5). Hence,

$$\sum_n \bar{X}_n(x_j) \frac{\sum_i R_i \bar{X}_n(x_i)}{P_n^2 - w_f^2} \sin w_f t = \Delta_j \sin w_f t \tag{26}$$

or

$$\sum_n \bar{X}_n(x_j) \frac{\sum_i R_i \bar{X}_n(x_i)}{P_n^2 - w_f^2} = \Delta_j \tag{27}$$

Eq. (27) is essentially a system of linear equations that has as many unknowns (R_{iS}) as there are interior supports and can be solved readily.

It is of interest to note that Eq. (27) may also be used to evaluate the natural frequencies of the continuous beam. The system frequencies can be obtained by setting the constraints such that the deflection at each support is zero for every instant in time. That is

$$y(x_j, t) \equiv 0 \text{ for all } t \quad (28)$$

which implies that

$$\sum_n \bar{X}_n(x_j) \frac{\sum_i R_i X_n(x_i)}{P_n^2 - \omega_f^2} = 0 \quad (29)$$

Eq. (29) is another system of linear equations. For a non-trivial solution for R_{iS} , the determinant must vanish. Using numerical methods (combined determinant search with convergency by Newton-Raphson method), a computer program has been developed to compute the system frequencies.

C. Dynamic Three Moment Equation

The governing differential equation for the free lateral vibration of an elastic beam with constant cross section shown in Fig. 1 is

[4,6,12,14,15,20,23]

$$\frac{\partial^4 y}{\partial x^4} + \frac{m}{EI} \frac{\partial^2 y}{\partial t^2} = 0 \quad (30)$$

The solution of Eq. (30) for normal mode of vibration is given by

$$y = X \cdot q \quad (31)$$

in which

$$X = C_1 \sin \lambda x + C_2 \cos \lambda x + C_3 \sinh \lambda x + C_4 \cosh \lambda x \quad (32)$$

$$q = D_1 \sin pt + D_2 \cos pt \quad (33)$$

$$\lambda = \sqrt[4]{\frac{mp^2}{EI}} \quad (34)$$

X , the shape function, q , the time function, p , the circular frequency, C_1 to C_4 and D_1 and D_2 are the integration constants to be determined by boundary and initial conditions, respectively.

The moment and shear in the beam are

$$M = -EI \frac{\partial^2 y}{\partial x^2} \quad (35)$$

$$V = -EI \frac{\partial^3 y}{\partial x^3} \quad (36)$$

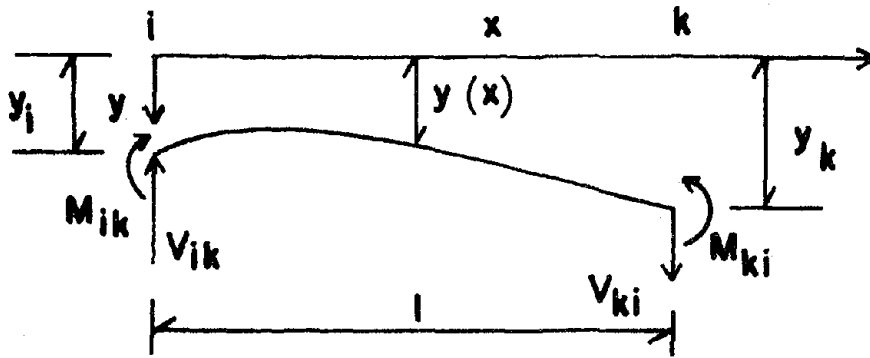
and in terms of shape function

$$\bar{M} = -EI \frac{\partial^2 X}{\partial x^2} \quad (37)$$

$$\bar{V} = -EI \frac{\partial^3 X}{\partial x^3} \quad (38)$$

Consider the case of a continuous beam with n spans, with $(n-1)$ intermediate supports. Fig. 3 shows a typical interior span $i \sim k$ with pulsating support settlements $y_i = \Delta_i \sin pt$ and $y_k = \Delta_k \sin pt$. The end condition to be satisfied may be interpreted for the shape function to be [14].

Fig. 3 - Boundary Conditions



$$x(0) = \Delta_i \quad (39)$$

$$x(l) = \Delta_k \quad (40)$$

$$x''(0) = -\frac{\bar{M}_{ik}}{EI} \quad (41)$$

$$x''(l) = -\frac{\bar{M}_{ki}}{EI} \quad (42)$$

Applying these four conditions, the four integral constants in Eq. (32) can be readily determined. Thus,

$$\begin{aligned} x(x) = & \frac{\bar{M}_{ik}}{2EI\lambda^2} \left[\frac{\sin \lambda(l-x)}{\sin \beta} - \frac{\sinh \lambda(l-x)}{\sinh \beta} \right] \\ & + \frac{\bar{M}_{ki}}{2EI\lambda^2} \left[\frac{\sin \lambda x}{\sin \beta} - \frac{\sinh \lambda x}{\sinh \beta} \right] \\ & + \frac{\Delta_i}{2} \left[\frac{\sin \lambda(l-x)}{\sin \beta} + \frac{\sinh \lambda(l-x)}{\sinh \beta} \right] \\ & + \frac{\Delta_k}{2} \left[\frac{\sin \lambda x}{\sin \beta} + \frac{\sinh \lambda x}{\sinh \beta} \right] \end{aligned} \quad (43)$$

in which $\beta = \lambda\ell$. The slopes $\Theta_i = X'(0)$, $\Theta_k = X'(\ell)$ take the following forms:

$$\Theta_i = \frac{\bar{M}_{ik}^{\ell}}{2EI} H_1(\beta) + \frac{\bar{M}_{ki}^{\ell}}{2EI} H_2(\beta) - \frac{\Delta_i}{2\ell} f_1(\beta) + \frac{\Delta_k}{2\ell} f_2(\beta) \quad (44)$$

$$\Theta_k = \frac{\bar{M}_{ki}^{\ell}}{2EI} H_2(\beta) - \frac{\bar{M}_{ik}^{\ell}}{2EI} H_1(\beta) - \frac{\Delta_i}{2\ell} f_2(\beta) + \frac{\Delta_k}{2\ell} f_1(\beta) \quad (45)$$

in which

$$H_1(\beta) = \frac{1}{\beta} (\coth \beta - \cot \beta) \quad (46)$$

$$H_2(\beta) = \frac{1}{\beta} (\operatorname{cosec} \beta - \operatorname{cosech} \beta) \quad (47)$$

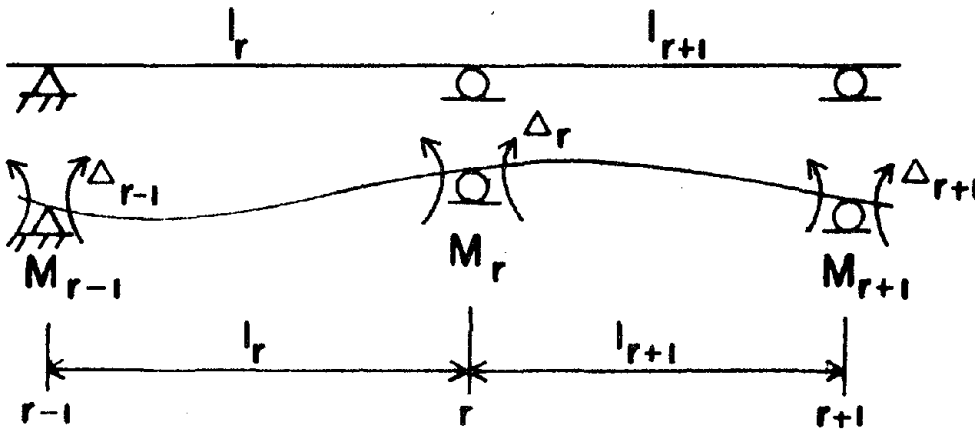
$$f_1(\beta) = \beta (\coth \beta + \cot \beta) \quad (48)$$

$$f_2(\beta) = \beta (\operatorname{cosec} \beta + \operatorname{cosech} \beta) \quad (49)$$

These quantities for various β values are given in Ref. [14].

Consider the two neighboring spans of a continuous beam as shown in Fig. 4.

Fig. 4 - Compatibility Conditions



Rewriting Eqs. (44) and (45) with appropriate notations given in Fig. 4 and making use of the condition of compatibility on the support 4, leads to the following equation:

$$\begin{aligned} & \bar{M}_{r-1} \left[\frac{l_r}{I_r} H_2(\beta_r) \right] + \bar{M}_r \left[\frac{l_r}{I_r} H_1(\beta_r) + \frac{l_{r+1}}{I_{r+1}} \right] + M_{r+1} \left[\frac{l_{r+1}}{I_{r+1}} H_2(\beta_{r+1}) \right] \\ & = E \left\{ \frac{f_2(\beta_r)}{l_r} \Delta_{r-1} - \left[\frac{f_1(\beta_{r+1})}{l_{r+1}} + \frac{f_1(\beta_r)}{I_r} \right] \Delta_r + \frac{f_2(\beta_{r+1})}{l_{r+1}} \Delta_{r+1} \right\} \end{aligned} \quad (50)$$

Multiplying Eq. (50) by a constant quantity, I_c , and introducing reduced lengths of the span ($l'_r = l_r \frac{I_c}{I_r}$, $l'_{r+1} = l_{r+1} \frac{I_c}{I_{r+1}}$, etc.) yields

$$\begin{aligned} & \bar{M}_{r-1} \left[l'_r H_2(\beta_r) \right] + \bar{M}_r \left[l'_r H_1(\beta_r) + l'_{r+1} H_1(\beta_{r+1}) \right] + \bar{M}_{r+1} \left[l'_{r+1} H_2(\beta_{r+1}) \right] \\ & = -EI_c \left\{ \frac{f_2(\beta_r)}{l_r} \Delta_{r-1} - \left[\frac{f_1(\beta_r)}{l_r} + \frac{f_1(\beta_{r+1})}{l_{r+1}} \right] \Delta_r + \frac{f_2(\beta_{r+1})}{l_{r+1}} \Delta_{r+1} \right\} \end{aligned} \quad (51)$$

in which $r = 1, 2, 3, \dots, n$. Eq. (51) is known as dynamic three moment equation. It is equivalent to the three moment equation used in static analysis. It should be noted that when $\Delta_{r-1}, \Delta_r, \Delta_{r+1}$ are equal to zero, i.e. no support settlements, eq. (51) reduces to the same one as given in Ref. [4,17].

In a particular case when p (circular frequency) approaches to zero (hence $\beta = 0$), quantities shown in Eqs. (46), (47), (48), and (49) take $H_1(0) = 2/3$, $H_2(0) = 1/3$, $f_1(0) = f_2(0) = 2$.

Substituting these limiting values into Eq. (50) and rearranging yields

$$\bar{M}_{r-1} \frac{\ell_r}{I_r} + 2\bar{M}_r \left(\frac{\ell_r}{I_r} + \frac{\ell_{r+1}}{I_{r+1}} \right) + \bar{M}_{r+1} \frac{\ell_{r+1}}{I_{r+1}} \quad (52)$$

$$= -6E \left[\frac{\Delta_{r-1}}{\ell_r} + \left(\frac{1}{\ell_r} + \frac{1}{\ell_{r+1}} \right) \Delta_r + \frac{\Delta_{r+1}}{\ell_{r+1}} \right]$$

Eq. (52) is a typical static three moment equation subjected to support settlements only.

The determination of the deflection, moment or shear at any point along the beam span necessitates the solution of the system of linear equations given in Eq. (50) or Eq. (51) as the first step. For any support settlement combination, the deflection amplitude at any location can then be determined by utilizing Eq. (43). Likewise, Eqs. (37) and (38) are used to determine the moment and shear at any point along the beam span are

$$\bar{M} = \frac{\bar{M}_{ik}}{2} \left[\frac{\sin \lambda(\ell-x)}{\sin \beta} + \frac{\sinh \lambda(\ell-x)}{\sinh \beta} \right] + \frac{\bar{M}_{ki}}{2} \left(\frac{\sin \lambda x}{\sin \beta} + \frac{\sinh \lambda x}{\sinh \beta} \right)$$

$$+ \frac{\Delta_i}{2} \left[\frac{\lambda^2 \sin \lambda(\ell-x)}{\sin \beta} - \frac{\lambda^2 \sinh \lambda(\ell-x)}{\sinh \beta} \right] EI + \frac{\Delta_k}{2} \left(\frac{\lambda^2 \sin \lambda x}{\sin \beta} - \frac{\lambda^2 \sinh \lambda x}{\sinh \beta} \right) EI \quad (53)$$

$$\bar{V} = \frac{\bar{M}_{ik}}{2} \left[\frac{\lambda \cos \lambda(\ell-x)}{\sin \beta} + \frac{\lambda \cosh \lambda(\ell-x)}{\sinh \beta} \right] + \frac{\bar{M}_{ki}}{2} \left(\frac{\lambda \cos \lambda x}{\sin \beta} + \frac{\lambda \cosh \lambda x}{\sinh \beta} \right)$$

$$+ \frac{\Delta_i}{2} \left[\frac{-\lambda^3 \cos \lambda(\ell-x)}{\sin \beta} + \frac{\lambda^3 \cosh \lambda(\ell-x)}{\sinh \beta} \right] EI + \frac{\Delta_k}{2} \left(\frac{\lambda^3 \cos \lambda x}{\sin \beta} - \frac{\lambda^3 \cosh \lambda x}{\sinh \beta} \right) EI \quad (54)$$

The fundamental circular frequency of the system can be determined using the dynamic three moment equation, Eq. (51) by equating the determinant for the unknown support moments to zero. For multispan continuous beams, Newton-Raphson type numerical method may be utilized.

D. Static Three Moment Equations

The homogeneous differential equation for flexure for an elastic beam with constant cross section is

$$EI \frac{\partial^4 y}{dx^4} = 0 \quad (55)$$

The solution of Eq. (55) is given by

$$y = C_1 x^3 + C_2 x^2 + C_3 x + C_4 \quad (56)$$

in which the integral constants C_1 to C_4 can be determined by the boundary conditions. The boundary conditions for the beam shown in Fig. 3 are

$$y(0) = \Delta_i \quad (57)$$

$$y(\ell) = \Delta_k \quad (58)$$

$$y''(0) = \frac{M_{ik}}{EI} \quad (59)$$

$$y''(\ell) = \frac{M_{ki}}{EI} \quad (60)$$

Applying these boundary conditions to Eq. (56) gives

$$y(x) = \left(\frac{x^3}{6\ell} - \frac{x^2}{2} - \frac{x\ell}{3} \right) \frac{M_{ik}}{EI} + \left(\frac{-x^3}{6\ell} + \frac{x\ell}{6} \right) \frac{M_{ki}}{EI} + \left(1 - \frac{x}{\ell} \right) \Delta_i + \left(\frac{x}{\ell} \right) \Delta_k \quad (61)$$

The moment and shear functions are

$$M(x) = \left(1 - \frac{x}{\ell}\right) M_{ik} + \left(\frac{x}{\ell}\right) M_{ki} \quad (62)$$

$$V(x) = \frac{M_{ik}}{\ell} + \frac{M_{ki}}{\ell} \quad (63)$$

Following a procedure similar to the one used in dynamic three moment equation, the static three moment equation is obtained as given in Eq. (52).

E. Limitation of Application

Recall Eqs. (29) and (50) which are used to determine the fundamental circular frequency of the system by setting the determinant for the unknown reactions or moments at each support equal to zero. However, the determinant does not have to vanish to satisfy the linear homogeneous equations, Eq. (29) or Eq. (50) for vibration modes composed of a series of the first vibration mode of an identical single span when the n spans are identical. Because of the symmetry, the reactions or moment at each interior support becomes zero and Eq. (29) and Eq. (50) are automatically satisfied regardless of the coefficient determinant. For these cases the frequencies are exactly equal to multiples of a simple beam frequencies and can thus be readily determined. Another approach to this difficulty is to use a slightly different span lengths within the range of numerical stability to avoid this phenomenon.

III. COMPUTER PROGRAMS

A. General Information

The formulations presented in Chapter II were programmed for use in a digital computer in ANSI Fortran program languages. A brief description of these computer programs and their input requirements is given in the next section of this chapter. Input and output examples are shown in Appendix I, while a program source listing is present in Appendix II. Two of the programs utilize the LINV3F subprogram developed by the International Mathematical and Statistical Library Company for the solution of systems of linear simultaneous equations and the determinant evaluation.

B. The Dynamic Fourier Series Program (DYNAMIC)

This program utilizes the Fourier series formulation presented in Chapter II. It has the capability of evaluating deflection, moment and shear induced by the pulsating intermediate bridge supports, as well as bridge natural frequencies. The program has three options. The first is to calculate the deflection, moment and shear. These functions may be evaluated for prescribed locations on the span over an input time range, or for specific instances of time over a distance range along the span. The second option is to determine the bridge natural frequencies only. When both the frequencies and the internal force functions are described, the third option must be used.

This program consists of a drive program and three subroutines in addition to the LINV3F subprogram of the IMSL programs library. The three subroutines are MAIN, FREQ and FX.

Subroutine MAIN consists of:

- 1) a read-write block, in which the options and parameters are read in and echoed.
- 2) a block for the determination of the loaded equivalent to the support settlements induced by the earthquake (see Eq. (26)). The LINV3F subprogram of the IMSL library is called for inversion in this block.
- 3) a block in which the functions are evaluated in accordance with Eq. (25) and its derivatives.
- 4) a block for the determination of the maximum values.
- 5) an output block for the functions evaluated.

When the second option is utilized, the MAIN subroutine calls FREQ subroutine and skips over the remaining blocks.

When the third option is chosen, MAIN calls FREQ before the execution of the remaining blocks. FREQ subroutine utilizes the Newton-Raphson method of search and successive convergence for the determination of the natural frequencies. This subroutine calls the FX subroutine to set up the matrix of coefficients, Eq. (29), for the evaluation of their determinant. The determinant evaluation is carried out by LINV3F subprograms.

It is very important that the input sequential order is strictly adhered to and consistent units are used throughout a problem. The input for this program is as follows:

CARD 1: HEAD

HEAD = a title composed of up to 80 alpha-numeric characters

CARD 2: NFREQ, NSPAN, NTERM, NPOINT, IOPT, IPLOT, IEQSP, TL, SM, E, XI,

DT, TINT, WF

NFREQ = the number of natural frequencies desired

NSPAN = the number of spans

NTERM = the number of terms to be summed in fourier series

NPOINT = the number of locations to be investigated

IOPT = 1, when the calculation of deflection, moment and shear
functions only is desired

= 2, when the calculation of the natural circular frequencies
only is desired

= 3, when the calculation of the deflection, moment and shear
functions as well as the natural frequencies is desired

IPLOT = 1, if an output file of data for the plotting is desired

= 0, if no such file is desired

IEQSP = 1, when the bridge consists of equal spans

= 0, when the bridge consists of unequal spans

TL = total length of the bridge

SM = mass per unit length

E = Young's modulus

XI = moment of inertia of the bridge cross section

DT = duration of earthquake

TINT = time increment within DT

WF = earthquake circular frequency

CARD 3: DEL(I)

DEL(I) = the amplitude of settlement of the ith support

CARD 4: XL(I)

XL(I) = the distance to the ith support measured from the left
abutment

CARD 5: X(J)

X(J) = the distance coordinate of the jth point along the span
measured from the left abutment

CARD 6: IMAX, EPS1

IMAX = the maximum number of iterations allowed for the determina-
tion of the natural frequencies

EPS1 = acceptable tolerance in the frequency computation

C. The Three Moment Equation Program (MOM3)

This program utilizes the dynamic three moment equation presented in the previous chapter. This program has the same three options available in the DYNAMIC program. It consists of a drive program and three subroutines. This program also utilizes the LINV3F subprogram of the IMSL library. The three subroutines of this program are MOMAIN, MOFREQ and MOFX.

- 1) an initialization and read-write block in which the options and parameters are read in and echoed out.

- 2) a block for the calculation of the moments at the supports in accordance with Eq. (50). The inversion is carried out by the use of LINV3F subprogram.
- 3) a block in which the functions are evaluated using, Eqs. (43), (53), and (54).
- 4) a block for the determination of the maximum values.
- 5) an output block for the function evaluated.

When the second option is chosen, the MOMAIN subroutine calls MOFREQ subroutine and skips over the last four blocks discussed above. When the third option is used, MOMAIN calls MOFREQ before the execution of the last four blocks. As in the DYNAMIC program, the MOFREQ subroutine utilizes the Newton-Raphson method for the determination of the natural frequencies. This subroutine then calls the MOFX to set up the matrix of coefficients of the moments for the evaluation of their determinant in accordance with Eq. (50). The determinant evaluation is performed by LINV3F program.

It is required that the input sequential order is strictly adhered to and consistent units are used through a problem. The input sequence and the meanings of the variables which are not explained previously are as follows:

CARD 1: HEAD

CARD 2: IOPT, NSPAN, IEQSP

Skip Card 3 if IOPT = 1

CARD 3: NFEQ, IMAX, EPS1

Skip Card 4 if IOPT = 2

CARD 4: ITYPE, NPOINT, WF, SXORT, XTINT

ITYPE = a four character alpha-numeric string that should be input as TIME when it is desired to have the variation of the functions with respect to time for specific locations on the bridge, or as DIST when it is desired to have the variation of the functions along the span at specific instances in time

NPOINT = the number of points along the span if ITYPE = TIME, or the number of points in time for which the functions are to be evaluated for specific points along the span if ITYPE = DIST

SXORT = the duration of an earthquake if ITYPE = TIME, or total span length if ITYPE = DIST

XTINT = time interval for the time-history if ITYPE = TIME, or distance interval along the span for which the functions are to be evaluated if ITYPE = DIST

CARD 5: SL(I), YM(I), SI(I), SSAM(2)

SL(I) = ith span, length

YM(I) = ith span, Young's modulus

SI(I) = ith span, moment of inertia

SSAM(I) = ith span, mass per unit length

Skip next two cards if IOPT = 2

CARD 6: DELMAX(J)

DELMAX(J) = the amplitude of the settlement of the jth support

CARD 7: XT(N)

XT(N) = the distance coordinate of nth point when ITYPE = TIME,

or

the time coordinate of nth point when ITYPE = DIST

D. The Static Three Moment Equation Program (MOM3ST)

This is a program to evaluate the deflection, moment, and shear along the span of a continuous beam subjected to static support settlements. It consists of a drive program and two subroutines. The drive program allocates memory spaces and has a flag to abort if the problem storage requirement exceeds the declared maximum. If the storage requirement is within allocated, the program calls STMAIN subroutine. This subroutine first sets up the coefficients of the unknown moments at the intermediate supports in accordance with Eq. (52). Then STMAIN calls SIMULE, which is an inversion subroutine, for the determination of these unknown moments. After the inversion, the control goes back to STMAIN. Next in STMAIN, is a loop for the determination of the deflection, moment and shear at the specified interval along the span. While these values are computed, a sorting is performed to determine and store the maximum values of each of these functions and the corresponding positions along the span.

The input sequence is as follows:

CARD 1: HEAD

CARD 2: NSPAN, TL, XINT

TL: = the total length of a continuous beam

XINT = the interval along the span for which the deflection,
moment and shear are desired to be computed

For each span, a card as follows is required:

CARD 3: SL(2), YM(I), SI(I), SSAM(I)

CARD 4: DELMAX(J)

E. Dynamic Amplification (SAMIR)

This program computes the maximum dynamic amplification factor as the ratios of the dynamic functions (deflection, moment and shear) to the corresponding maximum static values. The evaluation of these dynamic functions is carried out by the Fourier series method but the dynamic three moment equation may serve the same purpose as well. This program consists of a drive program and two subroutines. The primary function of the drive program is to allocate the storage requirements. Most of the calculations are performed in the MAIN program which calls SIMULE subroutine for the inversion of the coefficients matrix in accordance with Eq. (26). Maximum dynamic functions along the span for each forcing frequency and the dynamic amplification factors for

deflection, moment and shear are computed and these values may be plotted.

The input sequence and variables are given [2] as follows:

CARD 1: HEAD

CARD 2: NSPAN, NTERM, NWS, NINT, IPLOT

NWS = the number of natural frequencies within the range of the forcing frequencies.

NINT = the number of the fine subdivision on both sides of each natural frequency

IPLOT = 0, no plot is desired

= 1, output plots of the dynamic amplification is required

CARD 3: TL, SM, E, XI, XINT, WF1, WF2, WFINT, WSINT

WF1 = lower range of the forcing frequency

WF2 = upper

WFINT = the magnitude of the increment of the forcing frequency between WF1 and WF2

WSINT = the magnitude of the increment of the forcing frequency near the natural frequency

CARD 4: DEL(I)

CARD 5: XI(I)

Skip the following card if NWS = 0

CARD 6: WS(J)

WS: = system natural frequency

CARD 7: STDEF, STMOM, STSHR

STDEF = maximum static deflection

STMOM = maximum static moment

STSHR = maximum static shear

CARD 8: TOPDEF, TOPMOM, TOPSHR

TOPDEF = maximum value of dynamic amplification for deflections

TOPMOM = maximum value of dynamic amplification for moments

TOPSHR = maximum value of dynamic amplification for shear

IV. PARAMETER STUDIESA. An Example Problem

A two span concrete bridge of total length of 60 feet and width of 32 feet is considered as an example to illustrate the use of the computer programs. A span of this bridge has a length of 33 feet and the other a length of 27 feet. The bridge has a haunch at the intermediate support. This bridge is on Wisconsin State Highway I in Richland County near Interstate Highway 80. Although the actual bridge will behave in a three dimensional way, it seems appropriate to model as a two dimensional structure using equivalent cross sectional properties. Therefore, the equivalent cross sectional properties are evaluated [3] as follows: Moment of inertia; 92,850 in⁴, depth; 17 in. Considering the bridge to be of normal weight reinforced concrete, the mass per unit length, m , is 1.467 lb-sec²/in². This bridge is analyzed under the effects of three forcing frequencies. Although the acceleration rather than the earthquake frequency is usually documented for earthquakes [23], the forcing frequency in this mathematical model can be readily derived from the acceleration by estimating the support settlements under the effect of the acceleration. With both the acceleration and support settlement known, the forcing frequency can be determined by

$$w_f = \sqrt{\frac{a}{\Delta_i}} \quad (64)$$

in which w_f is the forcing frequency, a is the acceleration due to the earthquake, Δ_i , is the expected support settlement under the effect of the earthquake acceleration. The relationship of parameters in Eq. (64) is tabulated in Table 1.

a	a	Δ_i	ω_f
%g	in/sec ²	in	rad/sec
25	96.6	.2415	20
33	128.8	.0805	40
33	128.8	.0057	150

Table 1 - Forcing Frequencies used in Example Problems

The dynamic response to these three frequencies can be seen in Figs. 5 through 13. From these figures it can be observed that the deflected shape varies drastically with the variation of the forcing frequency, and that for high frequencies the bridge superstructure exhibits more than one bend along its span. With no such deflected shapes, the curvature is sharper and thus the higher moments and shears.

For the case of the forcing frequency of 40 rad/sec, the maximum deflection occurs at 540 inches (45 feet) from the left abutment. The maximum moment occurs at 568.8 inches (47.4 feet) from this abutment, while the maximum shear occurs at 410.4 inches (34.2 feet) from it. Incidentally the fundamental natural frequency of this bridge is computed as 31.43 rad/sec.

B. Fourier Series Vs. Dynamic Three Moment Equation

In general a very good correlation was observed between the results obtained from the Fourier series method and the results obtained from the three moment equation method. It was observed, however, that more terms have to be summed as the number of spans in the bridge analyzed increase, if the same accuracy is desired. For example, it was observed that the summation of fit terms gave reasonable results for the frequency computation as well as the deflection, moment and shear functions for two span bridges, whereas one hundred and two hundred terms were needed to achieve the same accuracy for three and four span bridges, respectively. As expected the convergence of the deflection function is more rapid due to the n^4 term in the denominator of the deflection expression. Similarly the convergence of the moment function is faster than that of the shear function because of n^2 term in the denominator of the moment expression as compared to n in the shear expression. To illustrate the convergence of these three functions, the example problem was analyzed twice for a forcing frequency of 40 rad/sec., once by summing two hundred terms and once by summing only the first fifty terms. The results obtained from the summation of two hundred terms are shown in Figs. 5 through 7 whereas the functions obtained by summing 50 terms are shown in Figs. 14 through 16. An examination of these curves reveals that there is no noticeable difference for the deflection and moment curves. The shear curve is smoother (indicating a better convergence) with the summation of two hundred terms, yet the differences in shear magnitude are very small. It should be noted that the dynamic three moment equation yields exact solution because of the shape function used being the exact solution of the governing differential equation.

Fig. 5 - Deflection (Two span, 33 ft-27 ft, $\omega_f = 40$ rad/sec, $\Delta = .0805$ in,
 $n = 200$ terms)

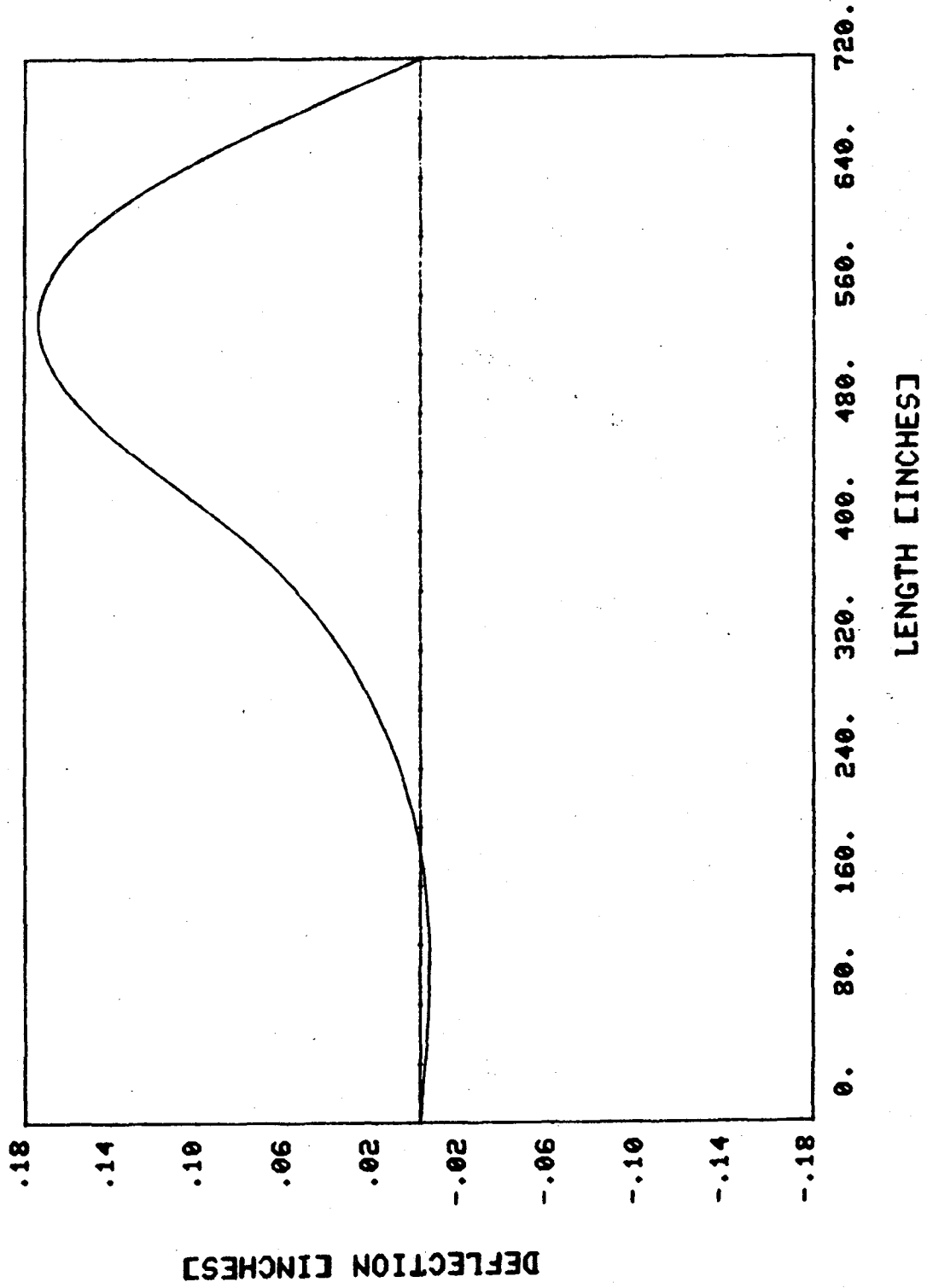


Fig. 6 - Moment (Two span, 33 ft-27 ft, $\omega_f = 40$ rad/sec, $\Delta = .0805$ in,
 $n = 200$ terms)

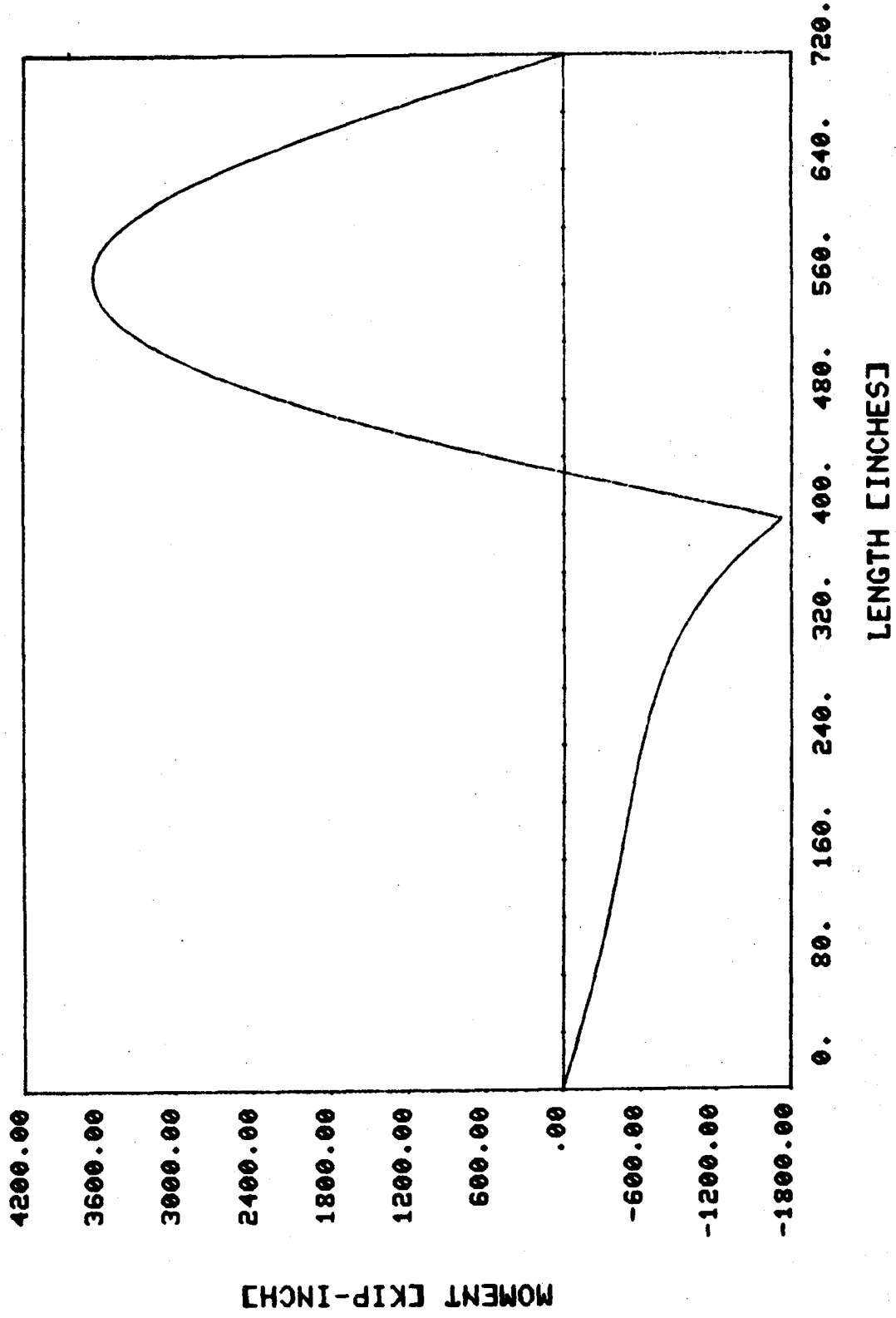


Fig. 7 - Shear (Two span, 33 ft-27 ft, $\omega_f = 40$ rad/sec, $\Delta = .0805$ in,
 $n = 200$ terms)

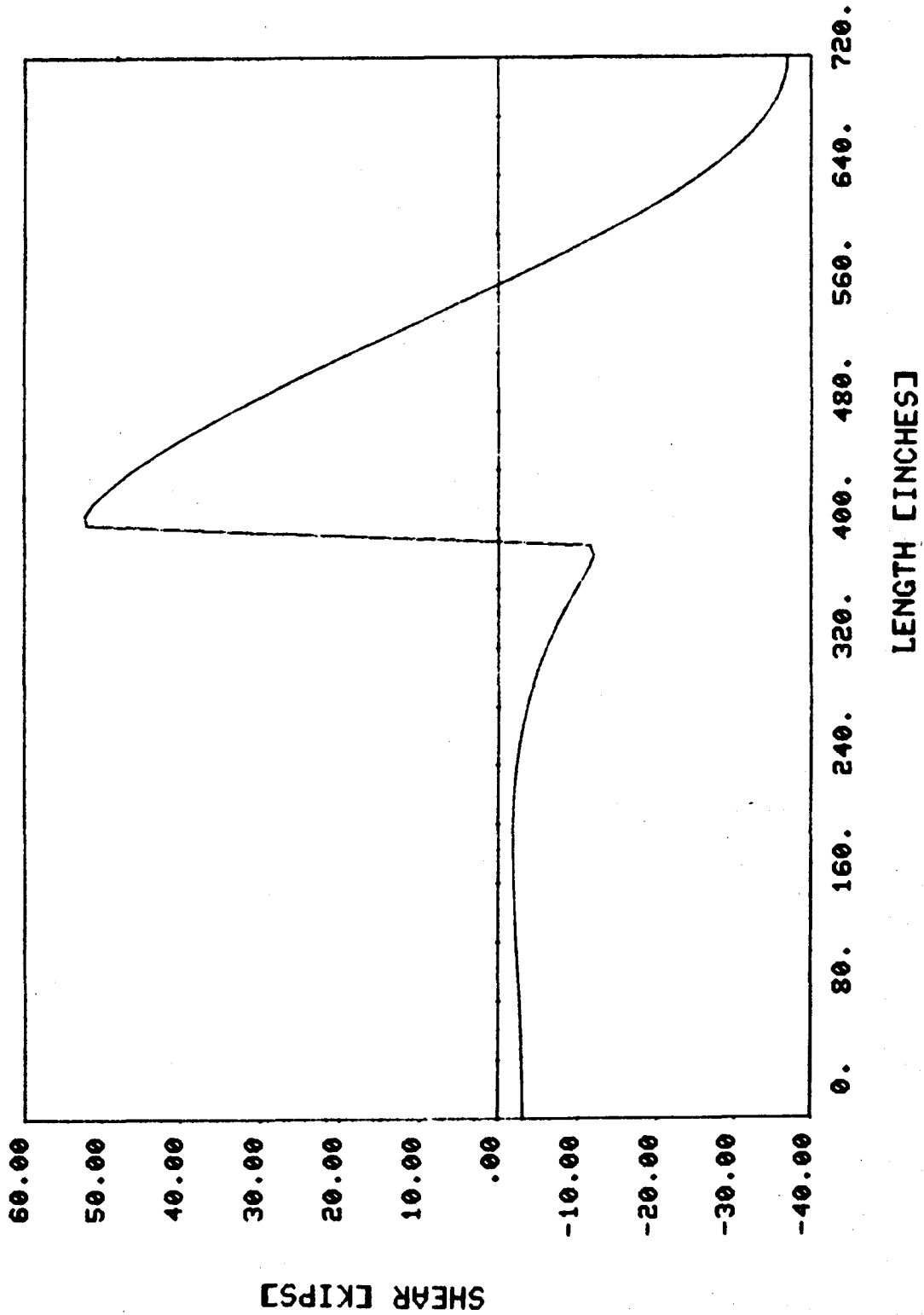


Fig. 8 - Deflection (Two span, 33 ft-27 ft, $\omega_f = 20$ rad/sec, $\Delta = .2415$ in,
 $n = 200$ terms)

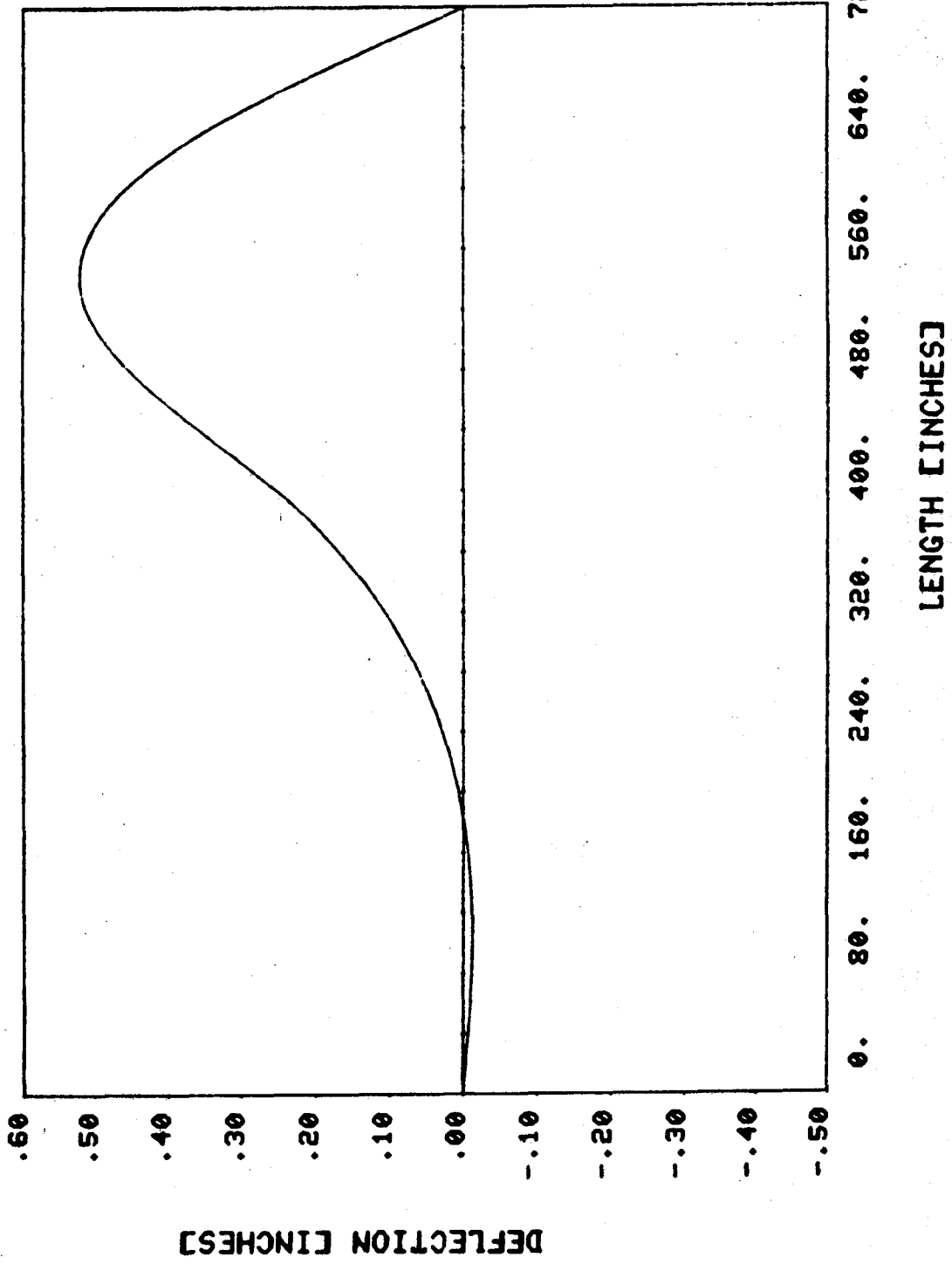


Fig. 9 - Moment (Two span, 33 ft-27 ft, $\omega_f = 20$ rad/sec, $\Delta = .2415$ in,
 $n = 200$ terms)

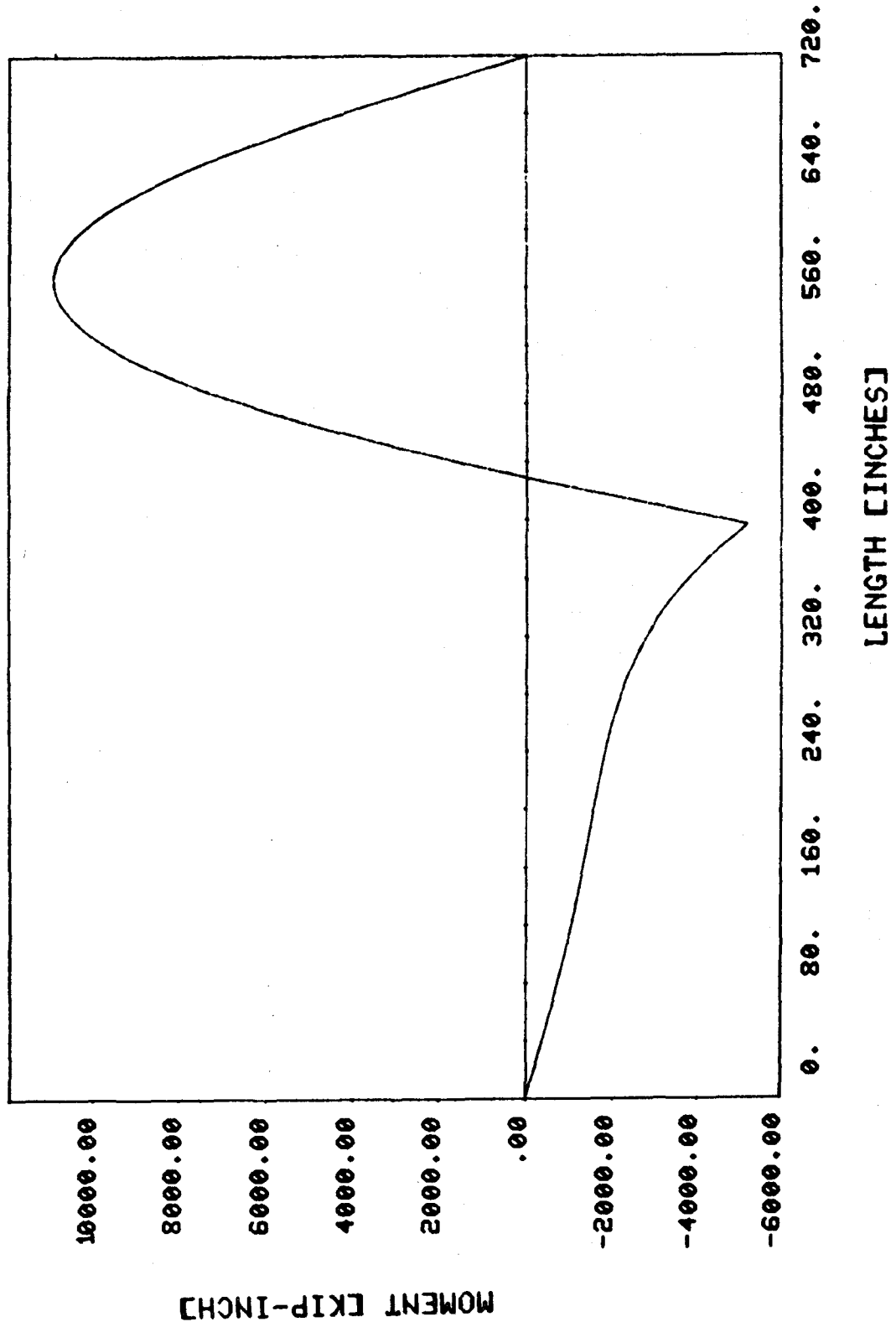


Fig. 10 - Shear (Two span, 33 ft-27 ft, $\omega_f = 20$ rad/sec, $\Delta = .2415$ in,
n = 200 terms)

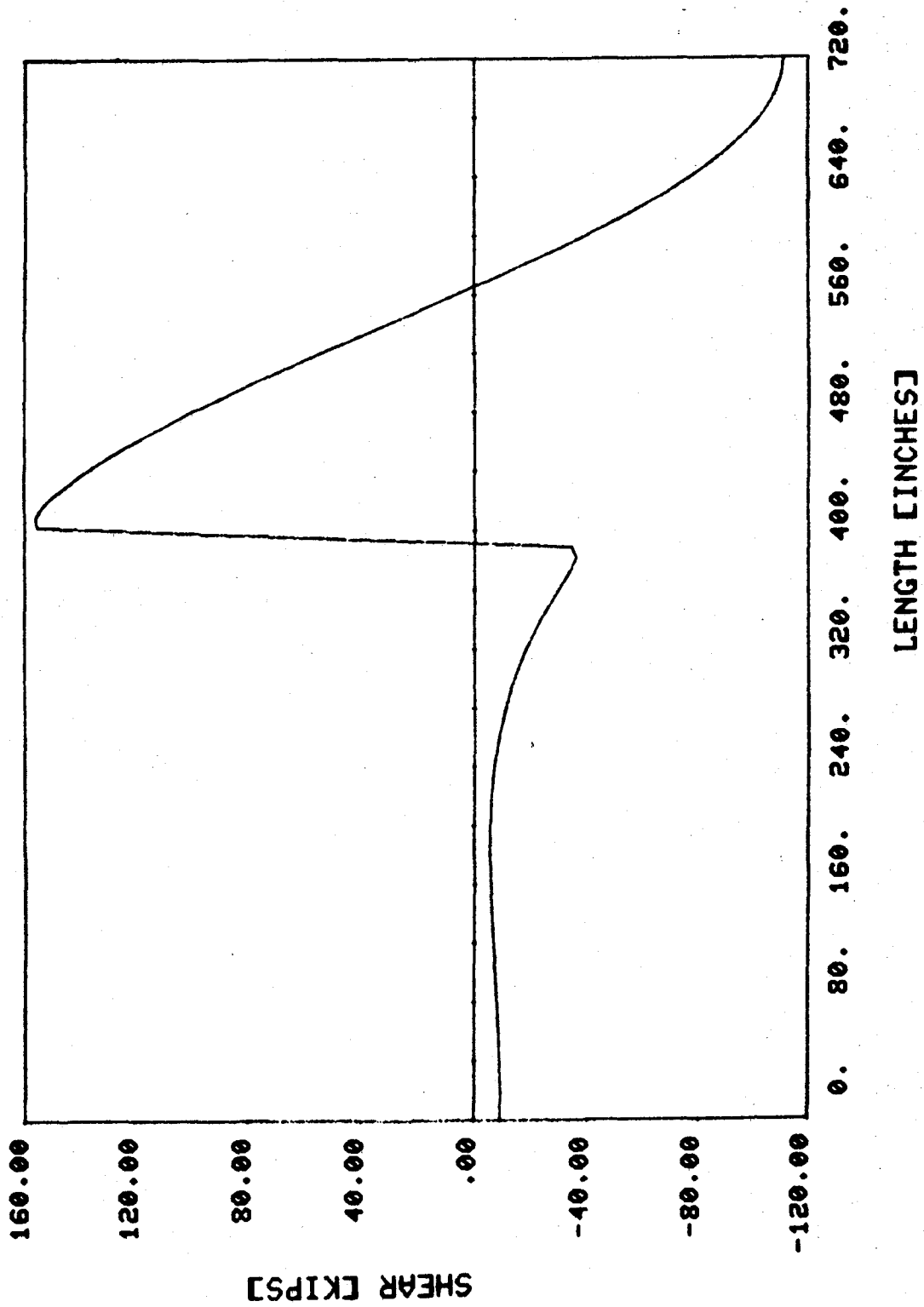


Fig. 11 - Deflection (Two span, 33 ft-27 ft, $\omega_f = 150$ rad/sec, $\Delta = .0057$ in,
 $n = 200$ terms)

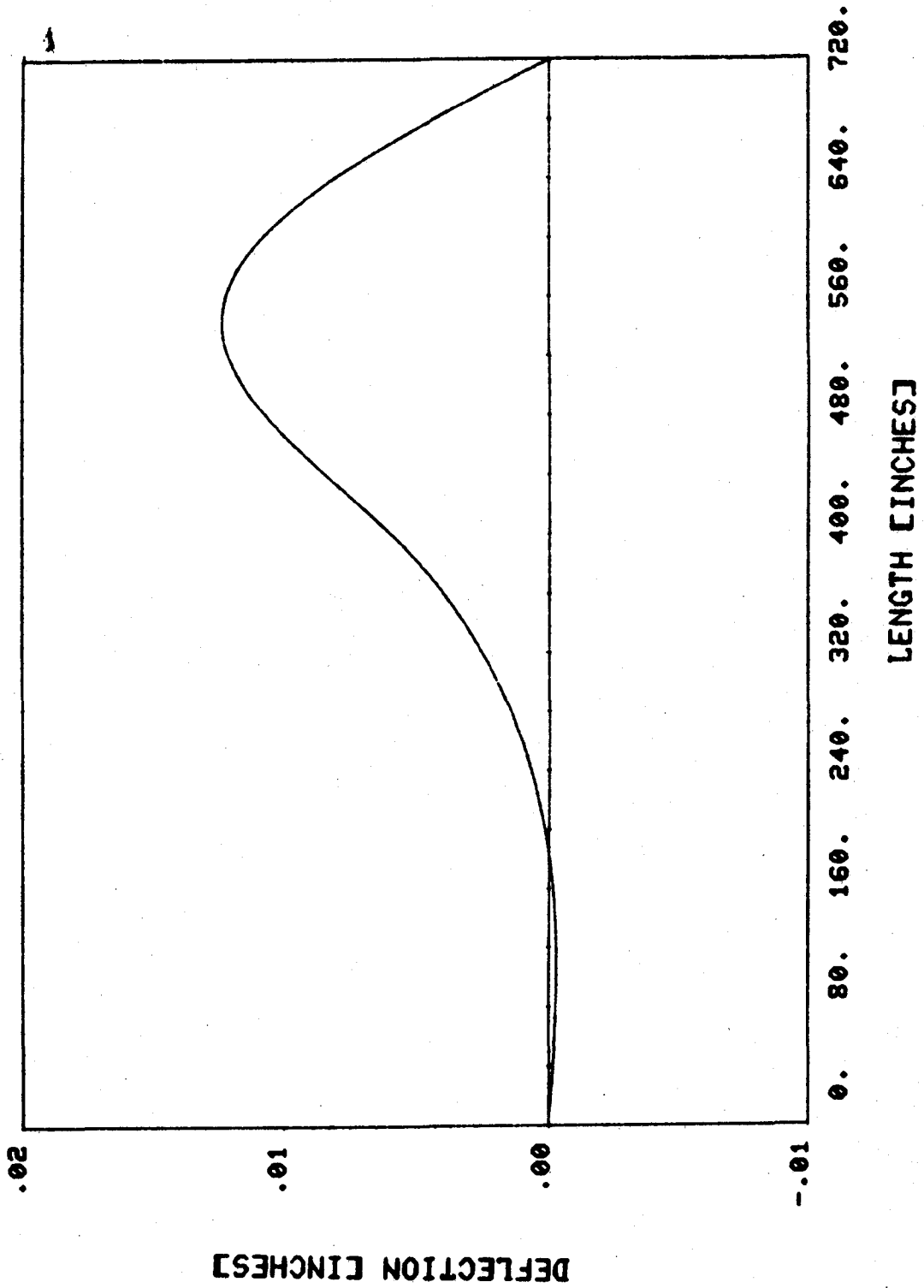


Fig. 12 - Moment (Two span, 33 ft-27 ft, $\omega_f = 150$ rad/sec, $\Delta = .0057$ in,
 $n = 200$ terms)

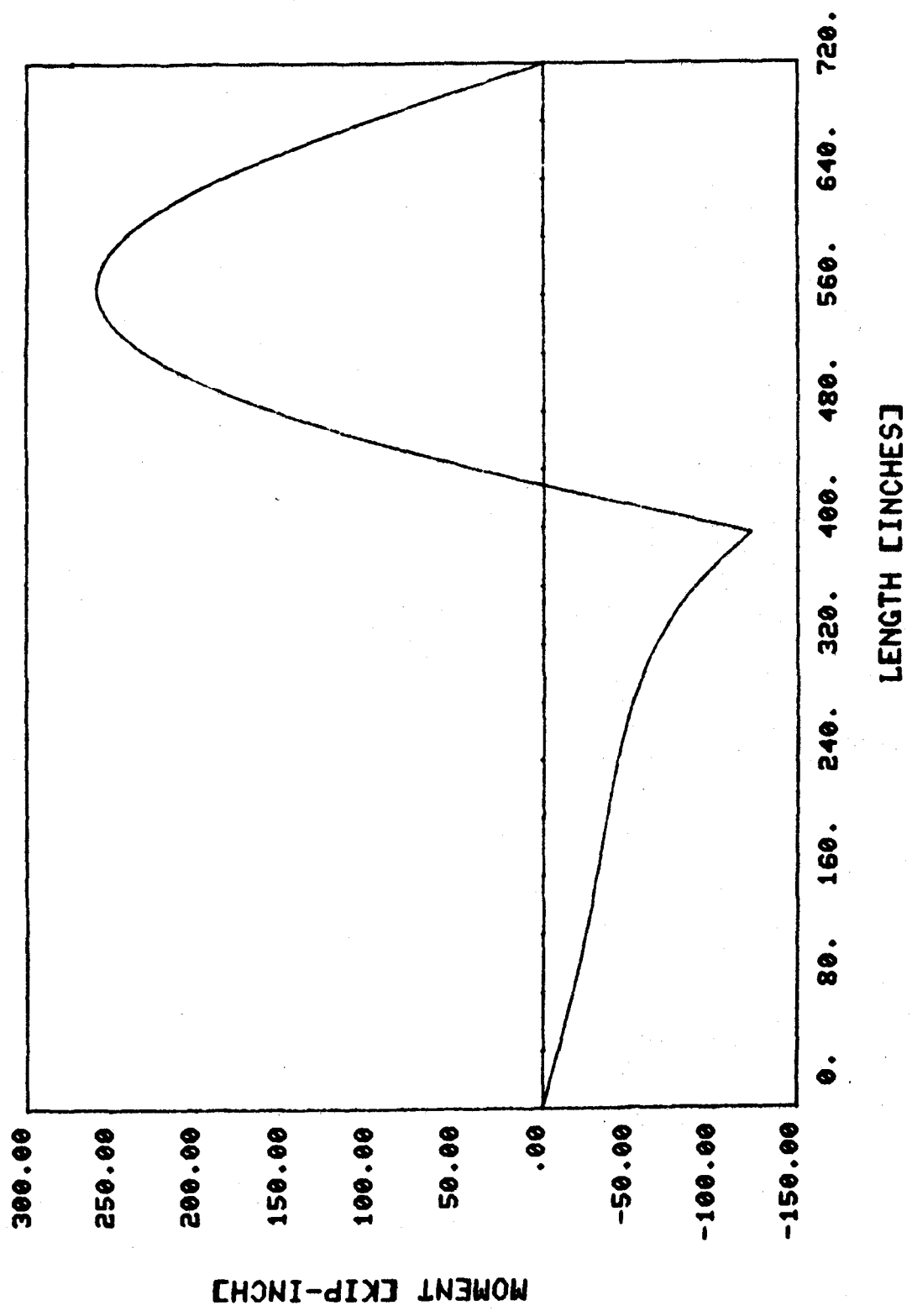


Fig. 13 - Shear (Two span, 33 ft-27 ft, $\omega_f = 150$ rad/sec, $\Delta = .0057$ in,
 $n = 200$ terms)

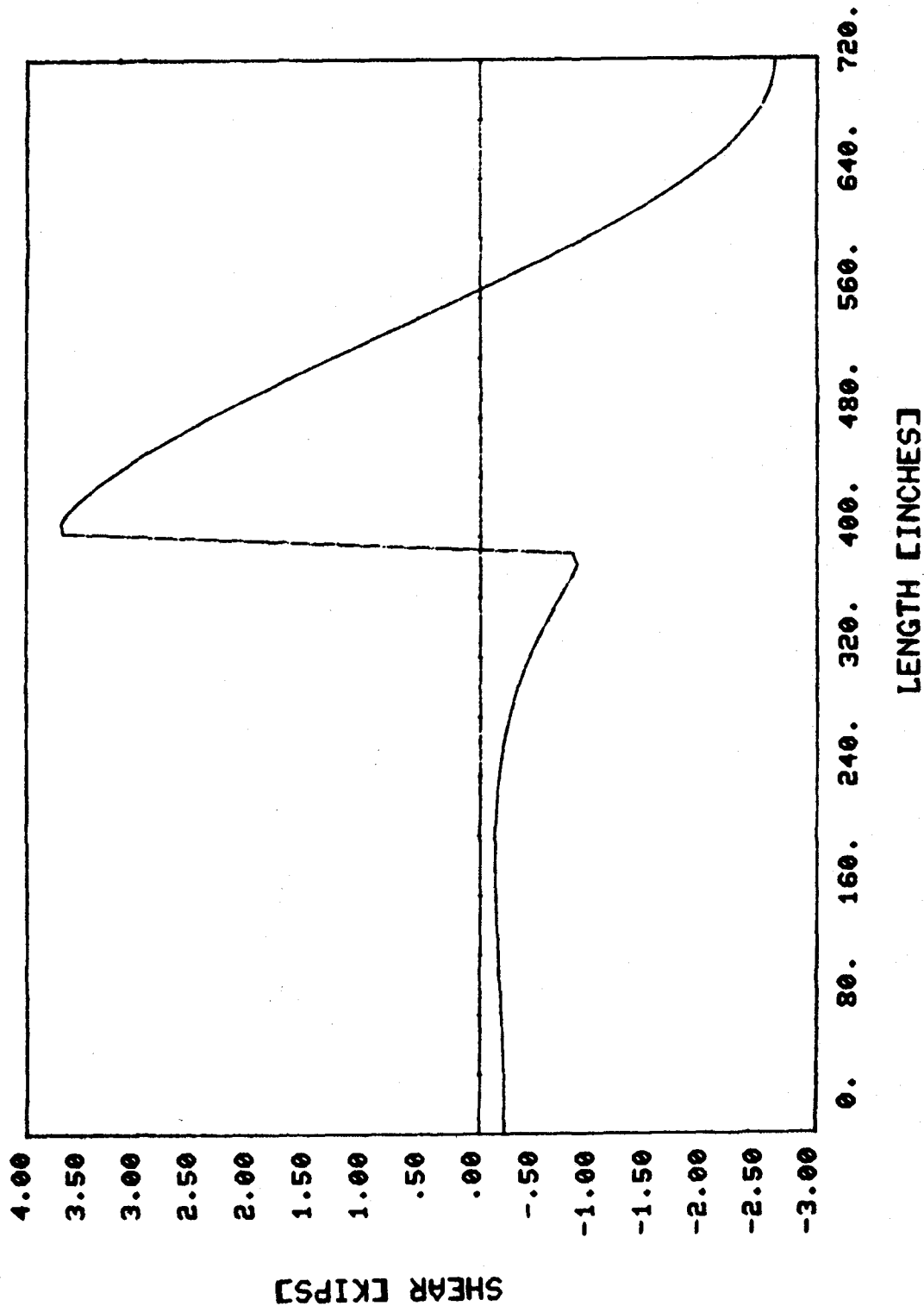


Fig. 14 - Deflection (Two span, 33 ft-27 ft, $\omega_f = 40$ rad/sec, $\Delta = .0805$ in,
 $n = 50$ terms)

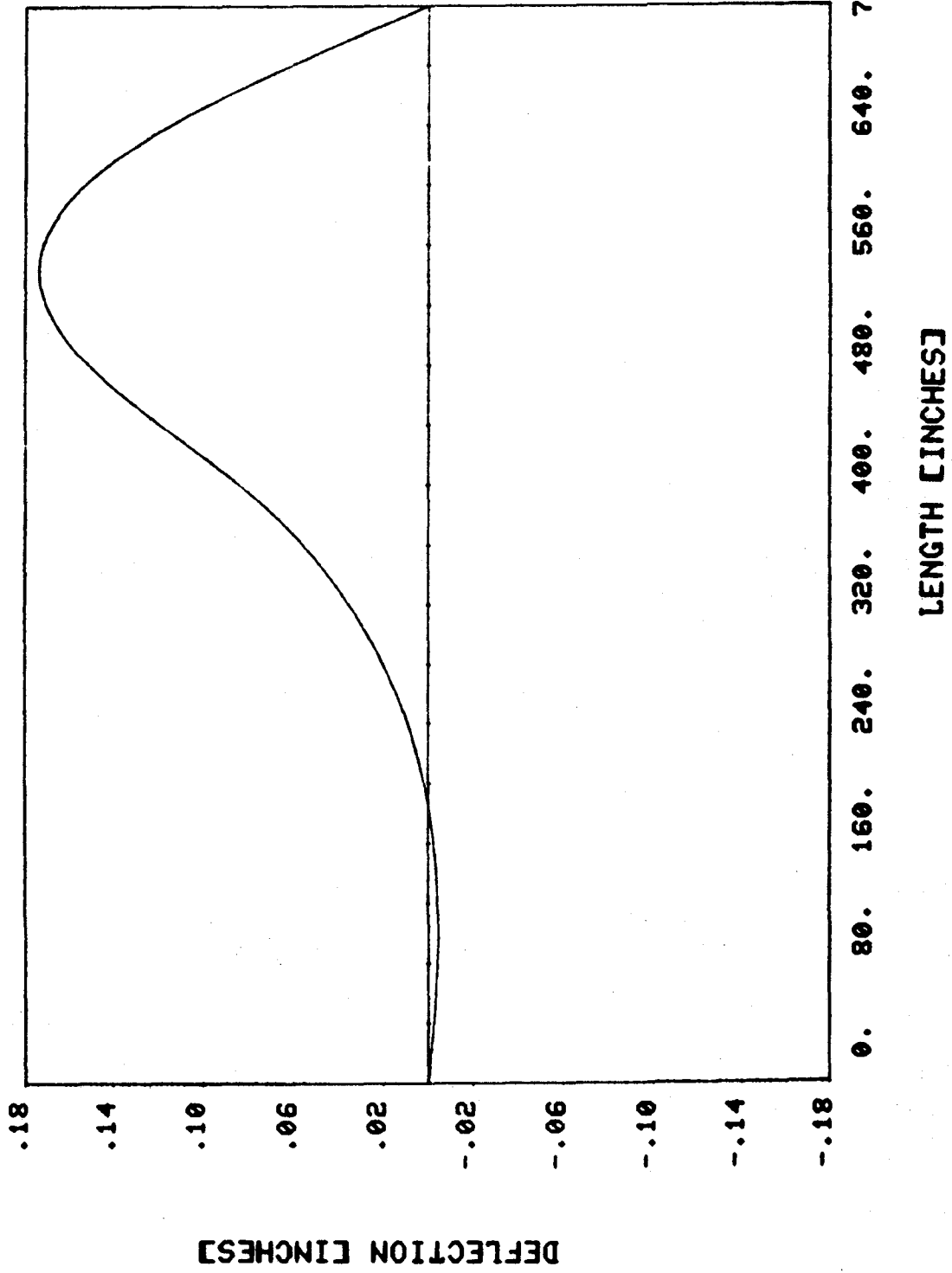


Fig. 15 - Moment (Two span, 33 ft-27 ft, $\omega_f = 40$ rad/sec, $\Delta = .0805$ in,
 $n = 50$ terms)

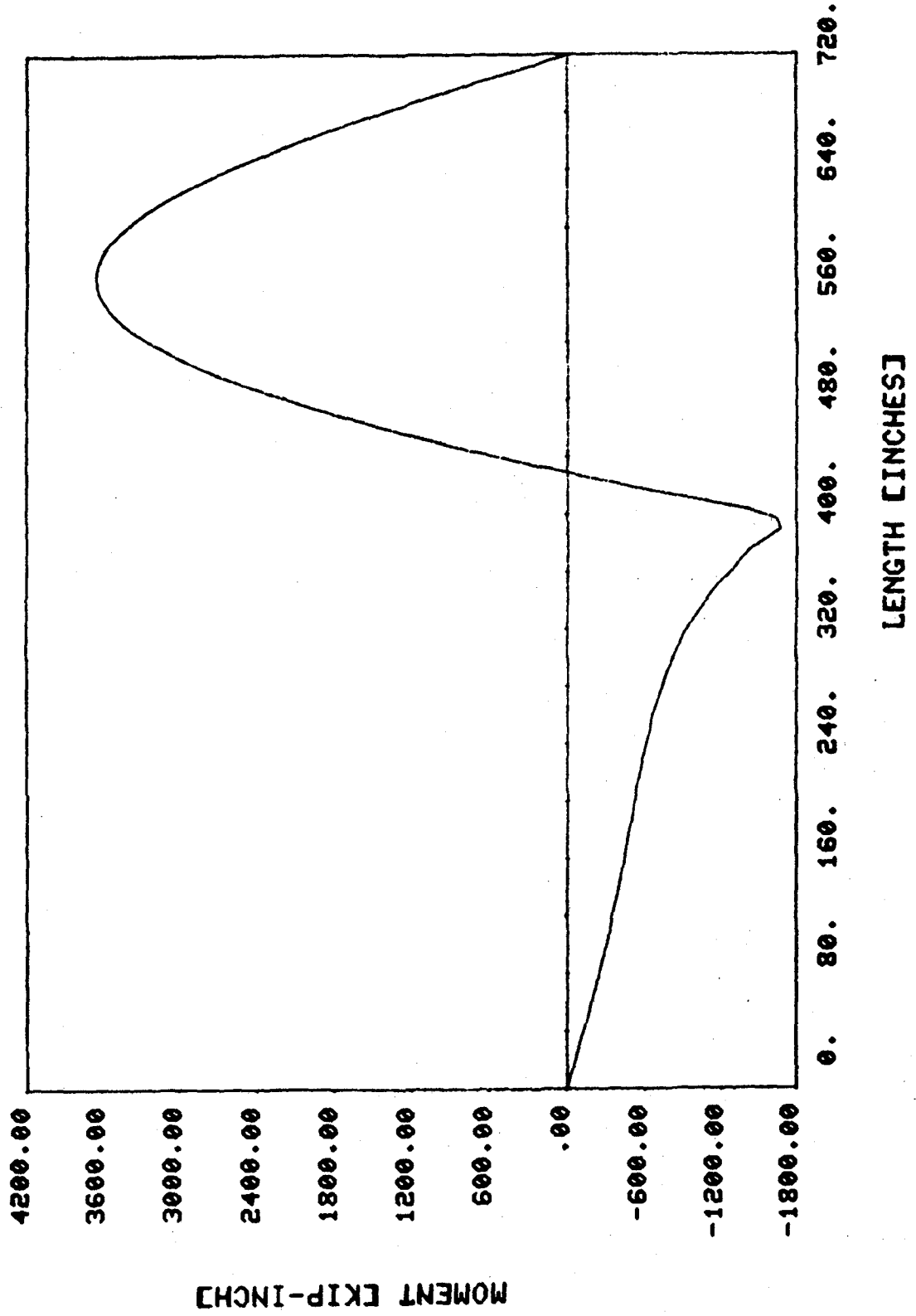
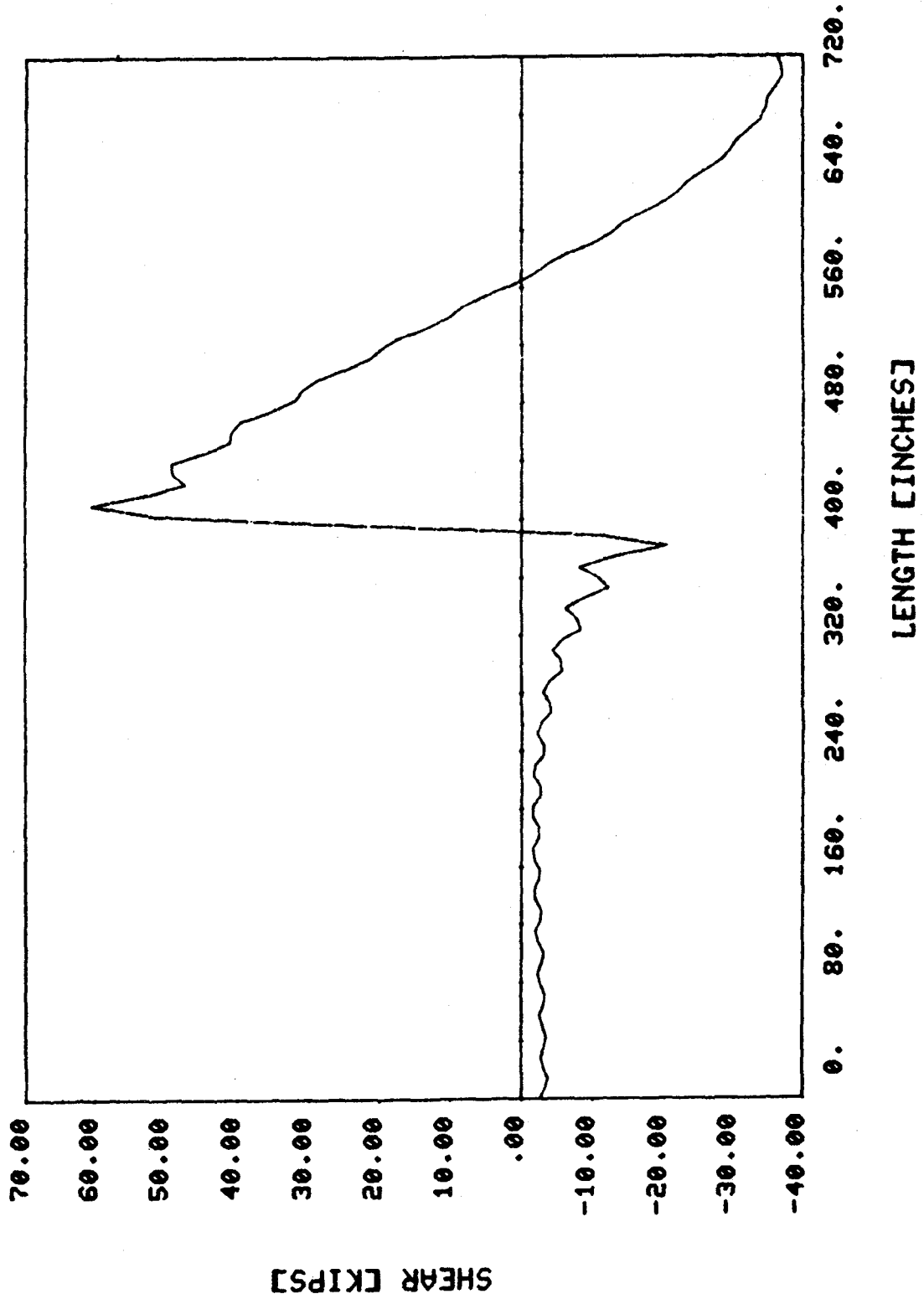


Fig. 16 - Shear (Two span, 33 ft-27 ft, $\omega_f = 40$ rad/sec, $\Delta = .0805$ in,
 $n = 50$ terms)



C. Development of Charts as Design Aids

With the computer programs developed, design aids for practical engineers can now be made. To illustrate this, sample charts were plotted for a two span bridge that has the intermediate support at a location fifty five percent of the total length from the left abutment. These charts can be used for bridges that have any combination of E, I, M, and L but that maintain a ratio of the intermediate support distance to the total length of .55. Similar charts for two span bridges with different intermediate support locations or for bridges with more than two spans can be readily developed by these programs.

It can be shown that the natural frequencies of a bridge can be expressed as

$$w_s = w_s' \sqrt{\frac{EI}{mL^4}} \quad (65)$$

in which w_s' is a dimensionless constant that depends only on the number of intermediate supports and their locations. Likewise the forcing frequency w_f may be transformed into non-dimensional entity as

$$w_f' = \frac{w_f}{\sqrt{\frac{EI}{mL^4}}}$$

With this new parameter as the abscissa, the dynamic amplification factor can now be plotted as explained in the SAMIR program. Once again these charts can be used for bridges with any combination of E, I, M, and L as long as they maintain the same value of the other two para-

meters; the number of intermediates and their relative locations. Figs. 17 through 46 are the dynamic amplification factors for deflection, moment and shear. The maximum static deflection, moment and shear along the bridge can be readily determined by the application of the static three moment equation, thus MOM3ST program or by any other readily available design tool. The information of the static response and the dynamic amplification factors leads to the dynamic analysis of bridges without actually conducting dynamic analysis. For example, using a static support settlement for the example bridge discussed in section A as .805 inches (see Table 1), the maximum static deflection is computed as .081045 inches, the maximum moment, 524300 lb-in (43.7 K-ft), the maximum shear 1,618.2 lbs (1.7182 kips). The output data from MOM3ST are given in Appendix 1.

The value of w_f' for an earthquake frequency of 40 rad/sec can be obtained by Eq. (66)

$$w_f' = \frac{40}{\frac{3,000,000 \times 92850}{1.467 \times (720)^4}} = 47.58$$

The dynamic amplification factor for the deflection is read off in Fig. 17 as 2.15 corresponding to w_f' value of 47.58. Thus the maximum dynamic deflection expected is $2.15 \times .081045 = 1.742$ inches. The exact value is computed as 1.738 by using either DYNAMIC or MOM3 program. Likewise the DAF for moment and shear are read off from the plots as 7.0 and 36.0, respectively. Thus the dynamic moment and shear are $7 \times 43.7 = 305.9$ k-ft and $36 \times 1.16182 = 58.255$ kips. The exact values of these functions are 304.1 k-ft and 55.457 kips as computed by MOM3 program.

Fig. 17 - DAF, Deflection (Two span, L-.6L)

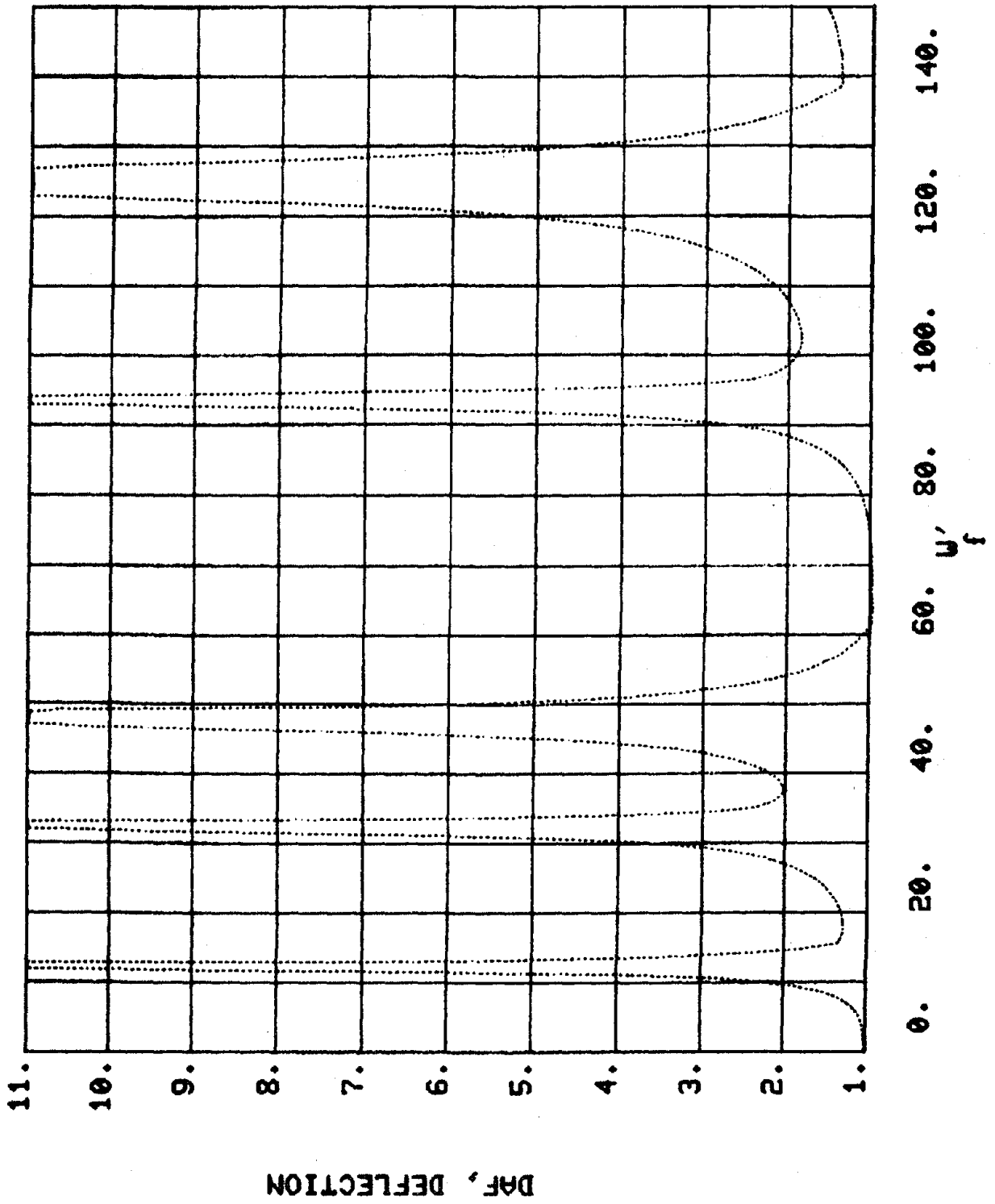


Fig. 18 - DAF, Moment (Two span, L-.6L)

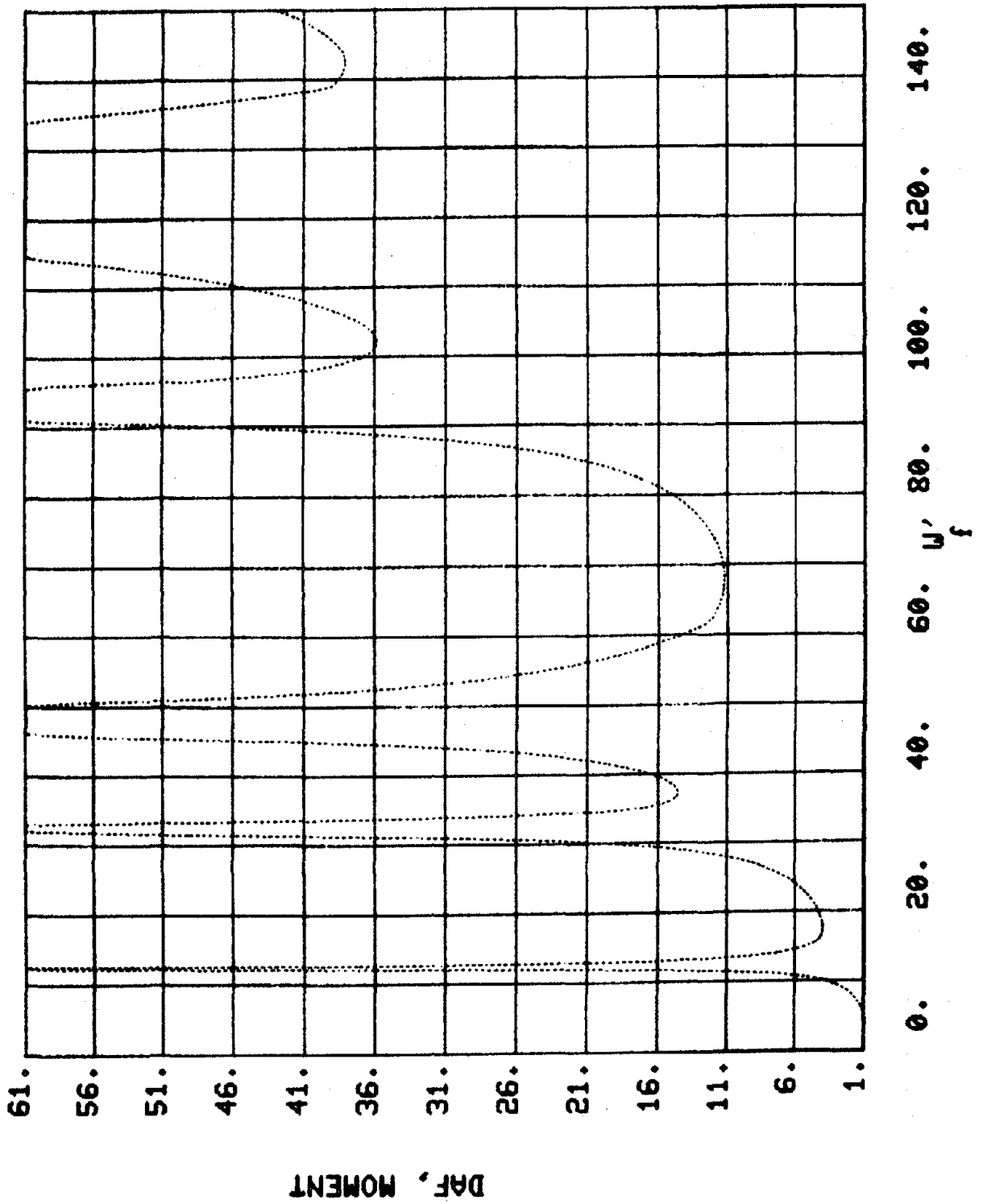


Fig. 19 - DAF, Shear (Two span, L-.6L)

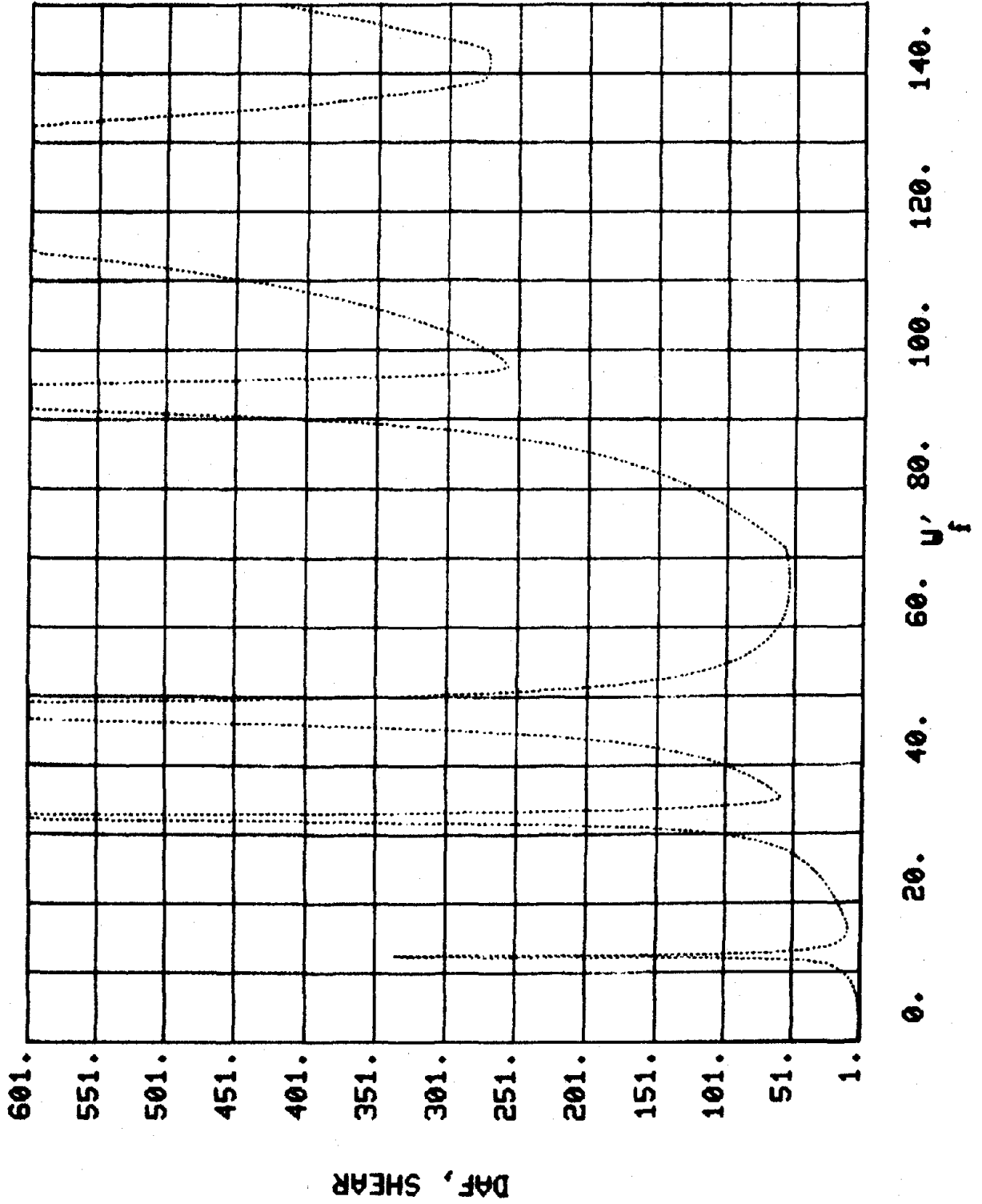


Fig. 20 - DAF, Deflection (Two span, L-.8182L)

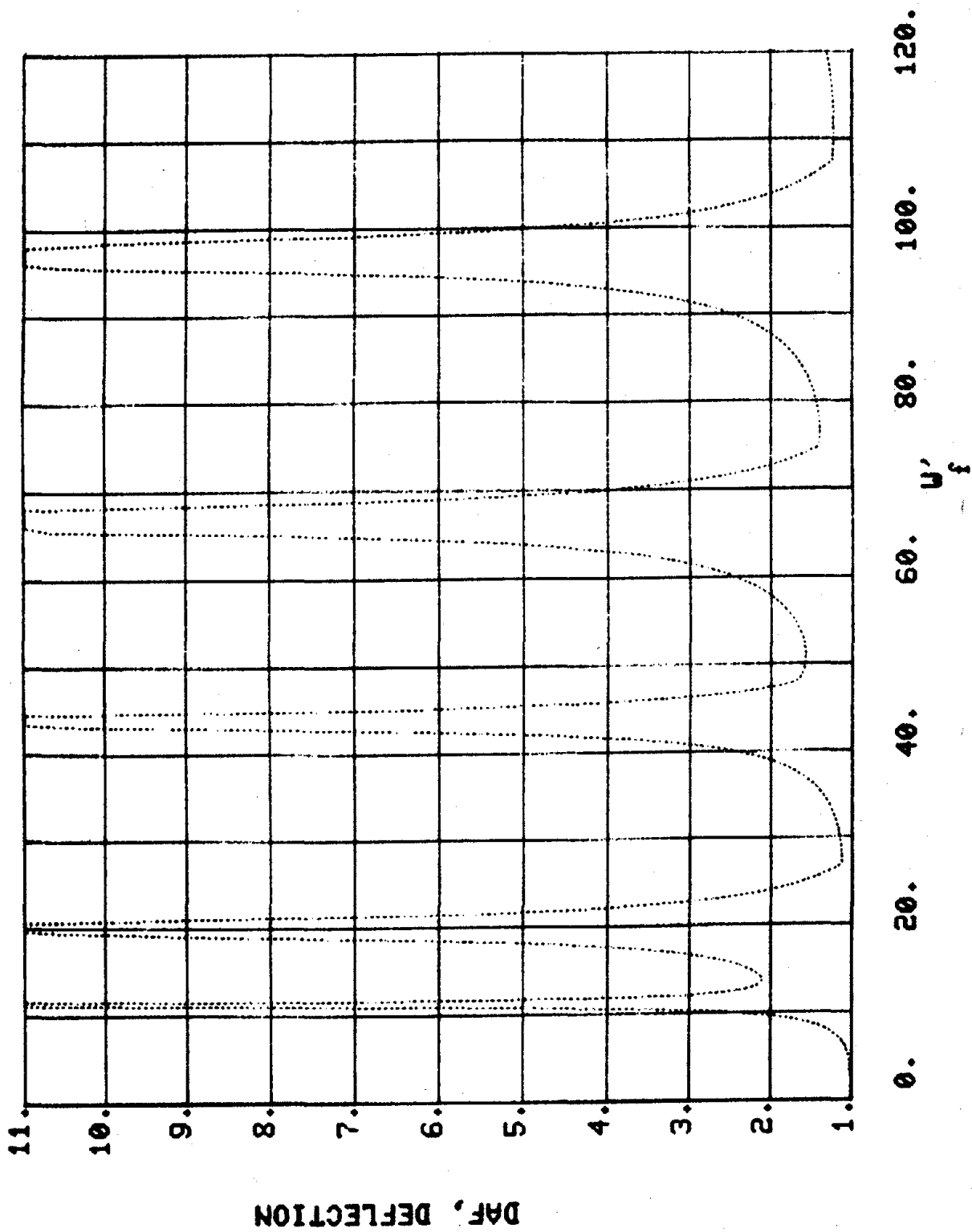


Fig. 21 - DAF, Moment (Two span, L-.8182L)

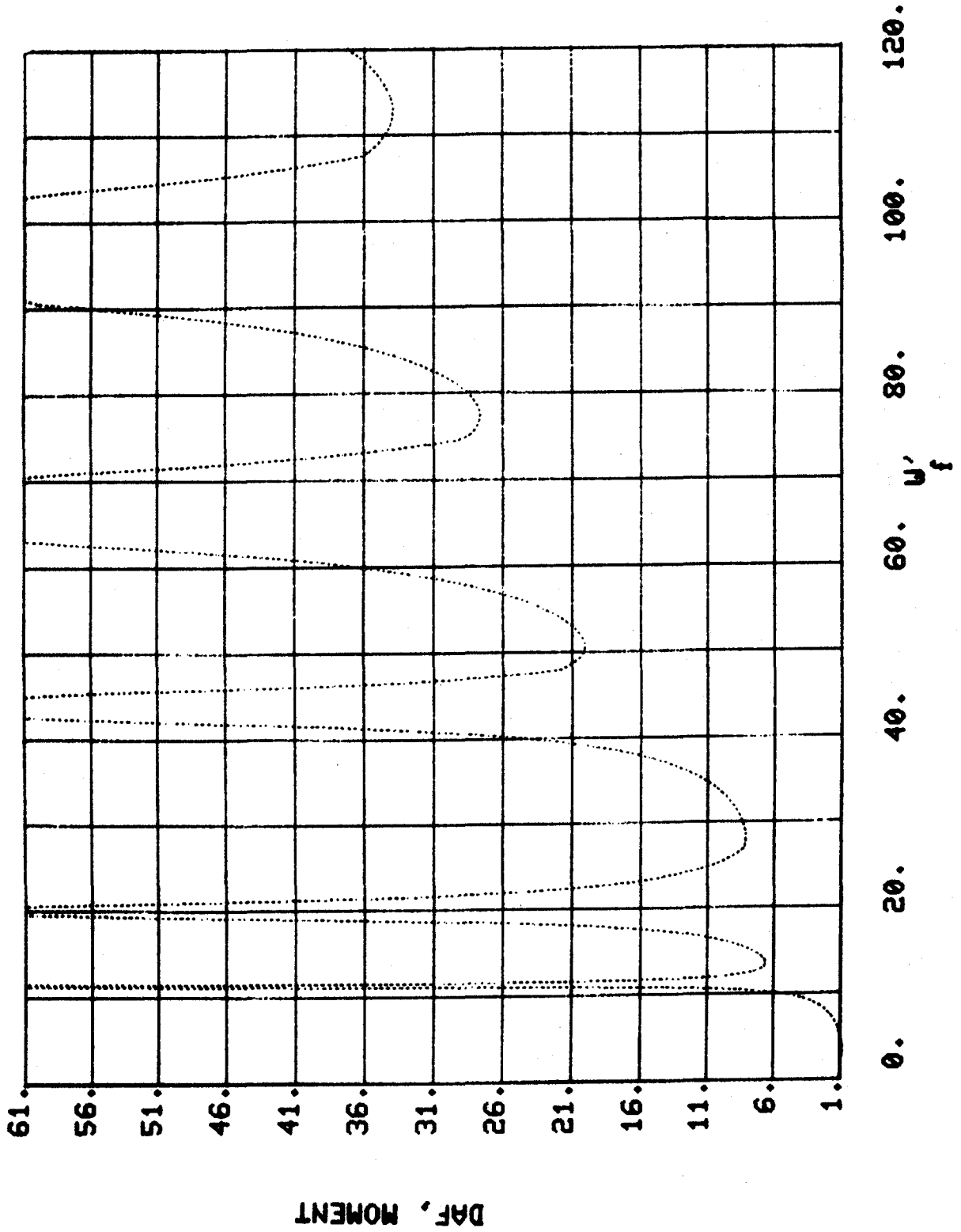


Fig. 22 - DAF, Shear (Two span, L-.8182L)

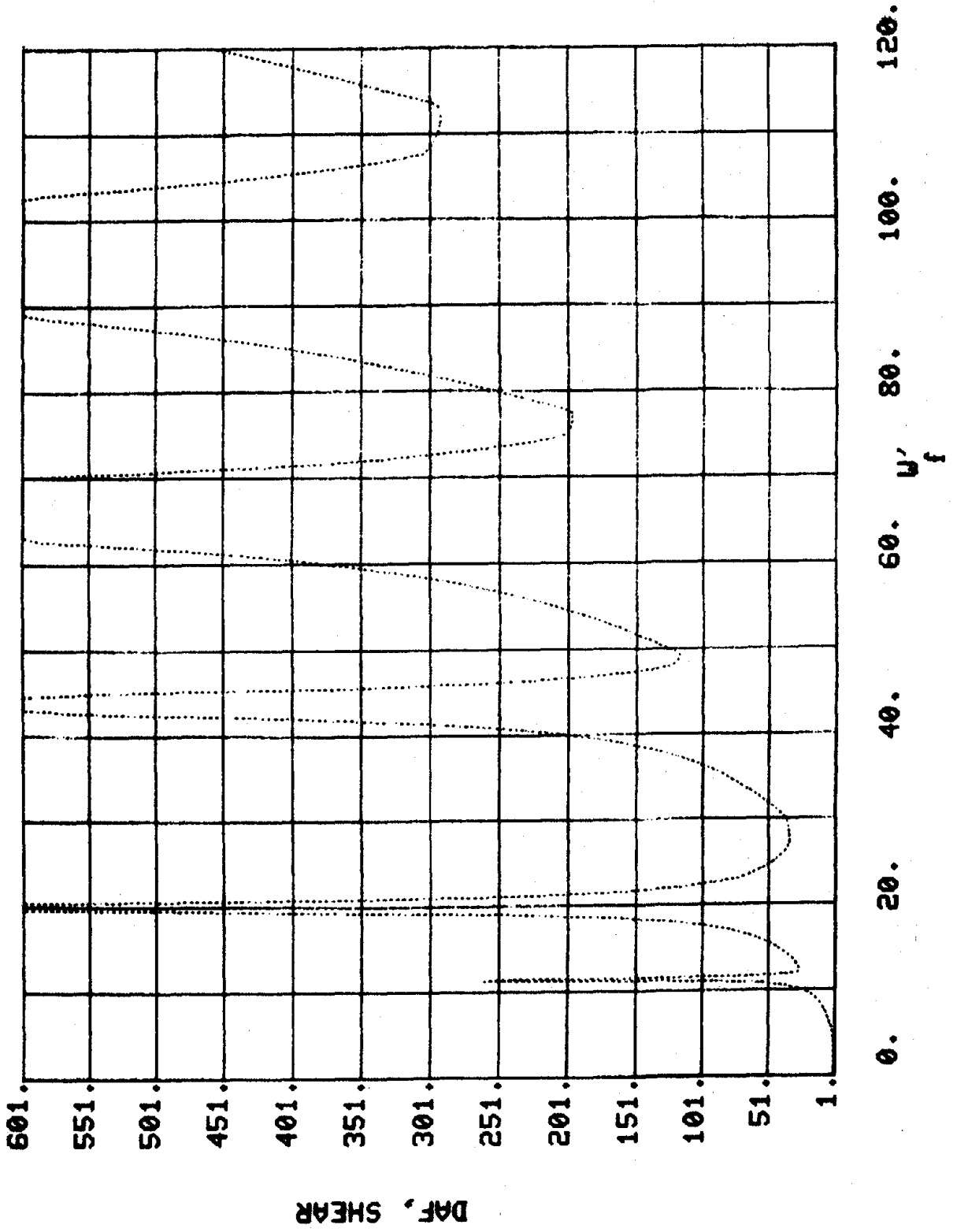


Fig. 23 - DAF, Deflection (Three span, .6L-L-.6L)

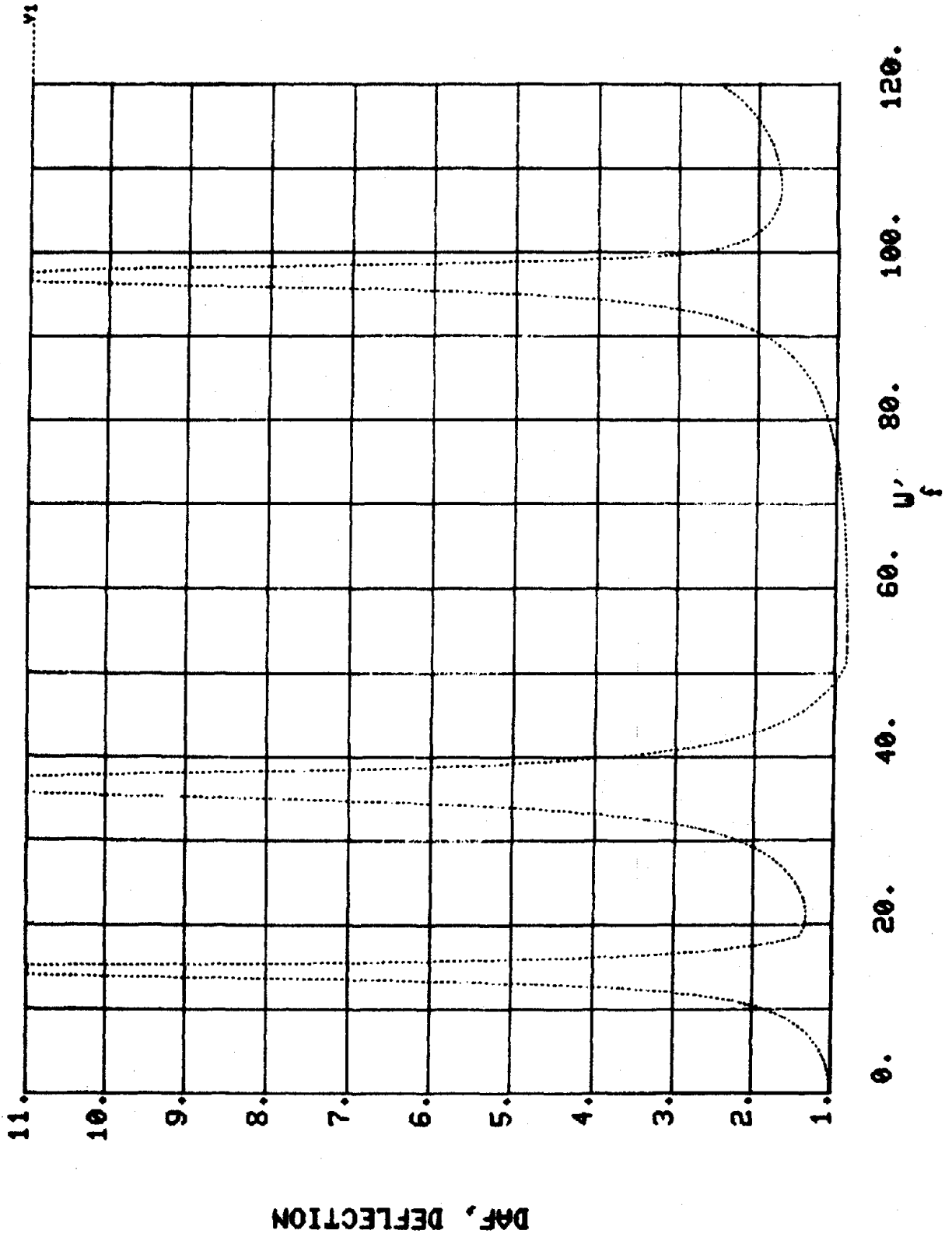


Fig. 24 - DAF, Moment (Three span, .6L-L-.6L)

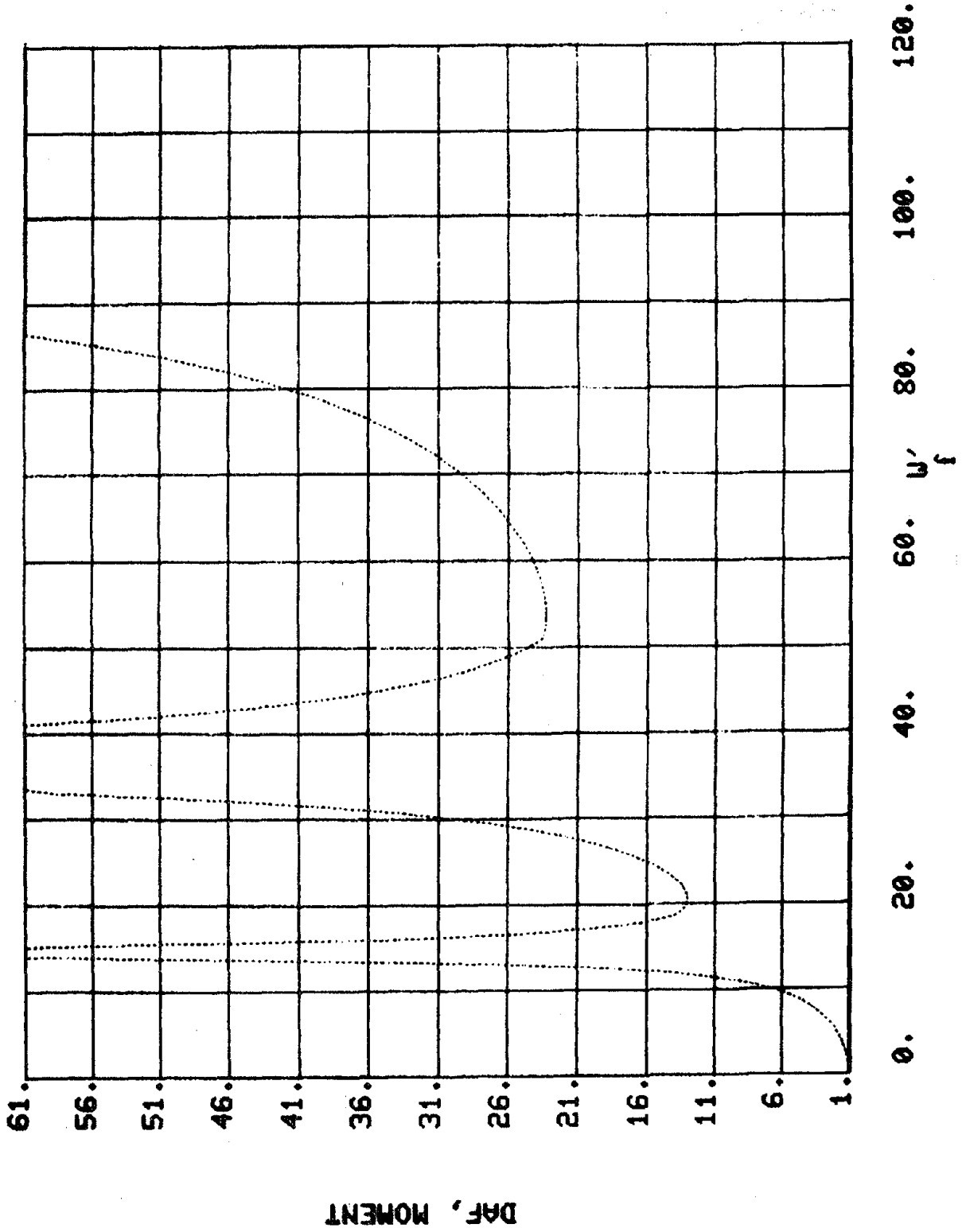


Fig. 25 - DAF, Shear (Three span, .6L-L-.6L)

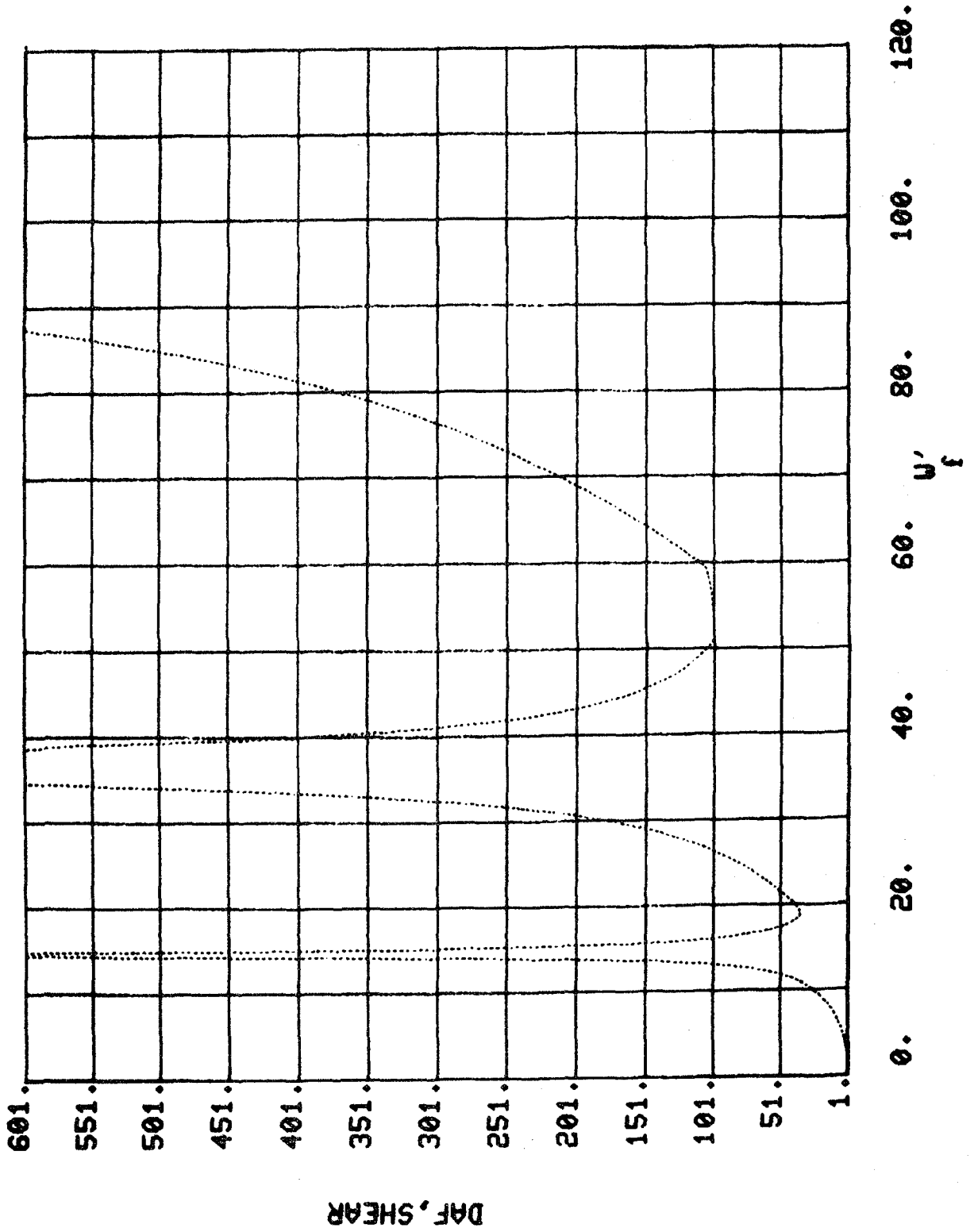


Fig. 26 - DAF, Deflection (Three span, .7L-L-.7L)

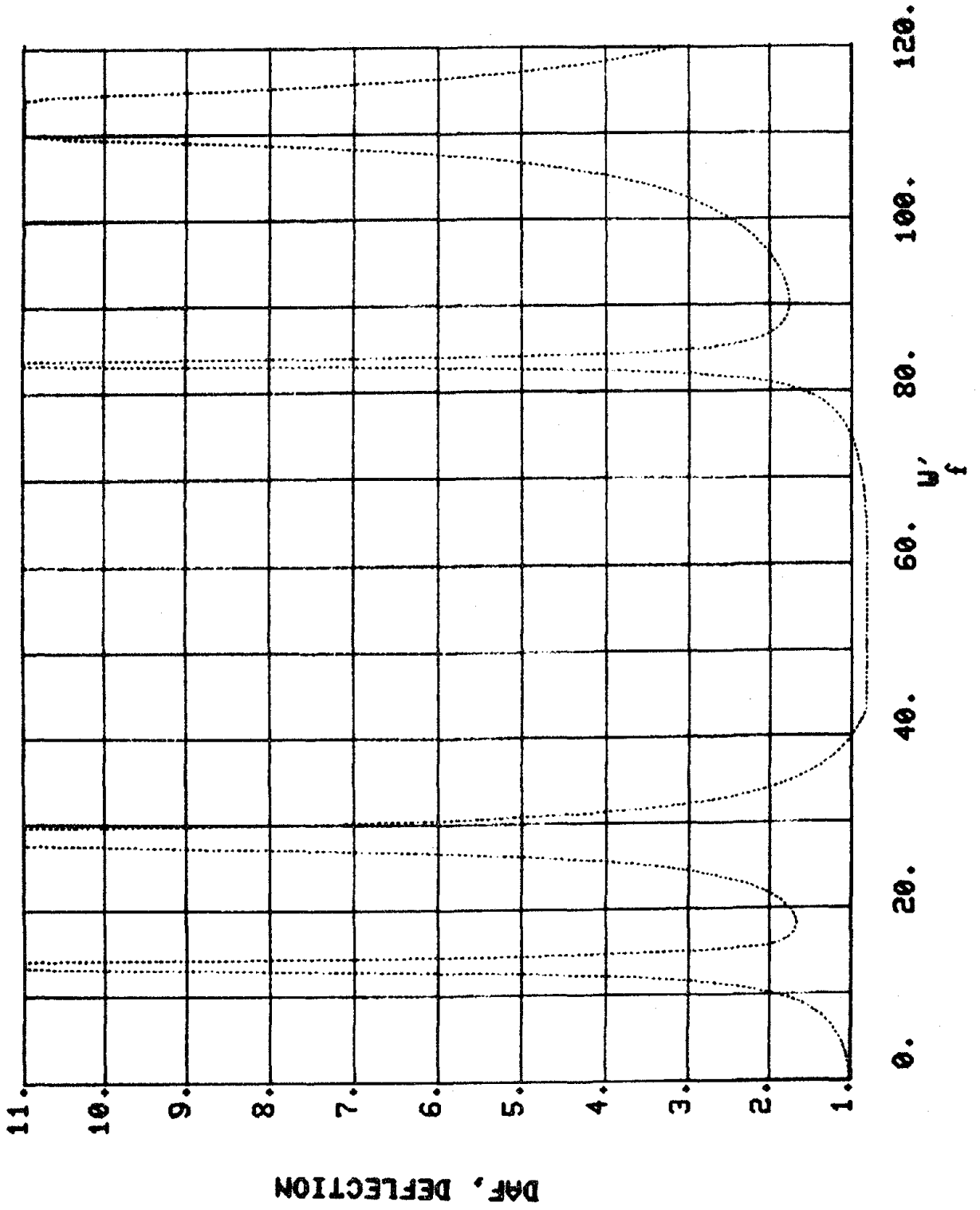


Fig. 27 - DAF, Moment (Three span, .7L-L-.7L)

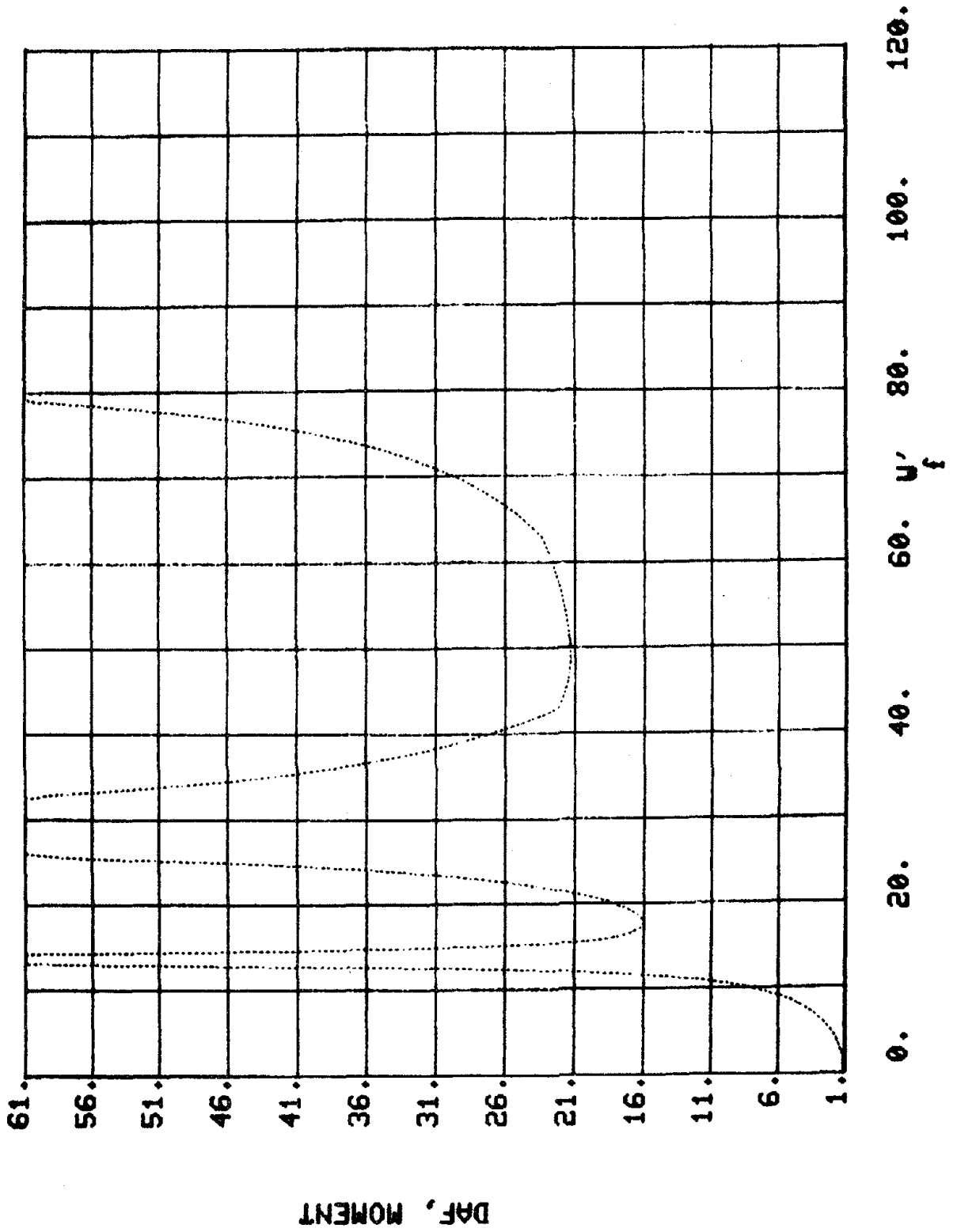


Fig. 28 - DAF, Shear (Three span, .7L-L-.7L)

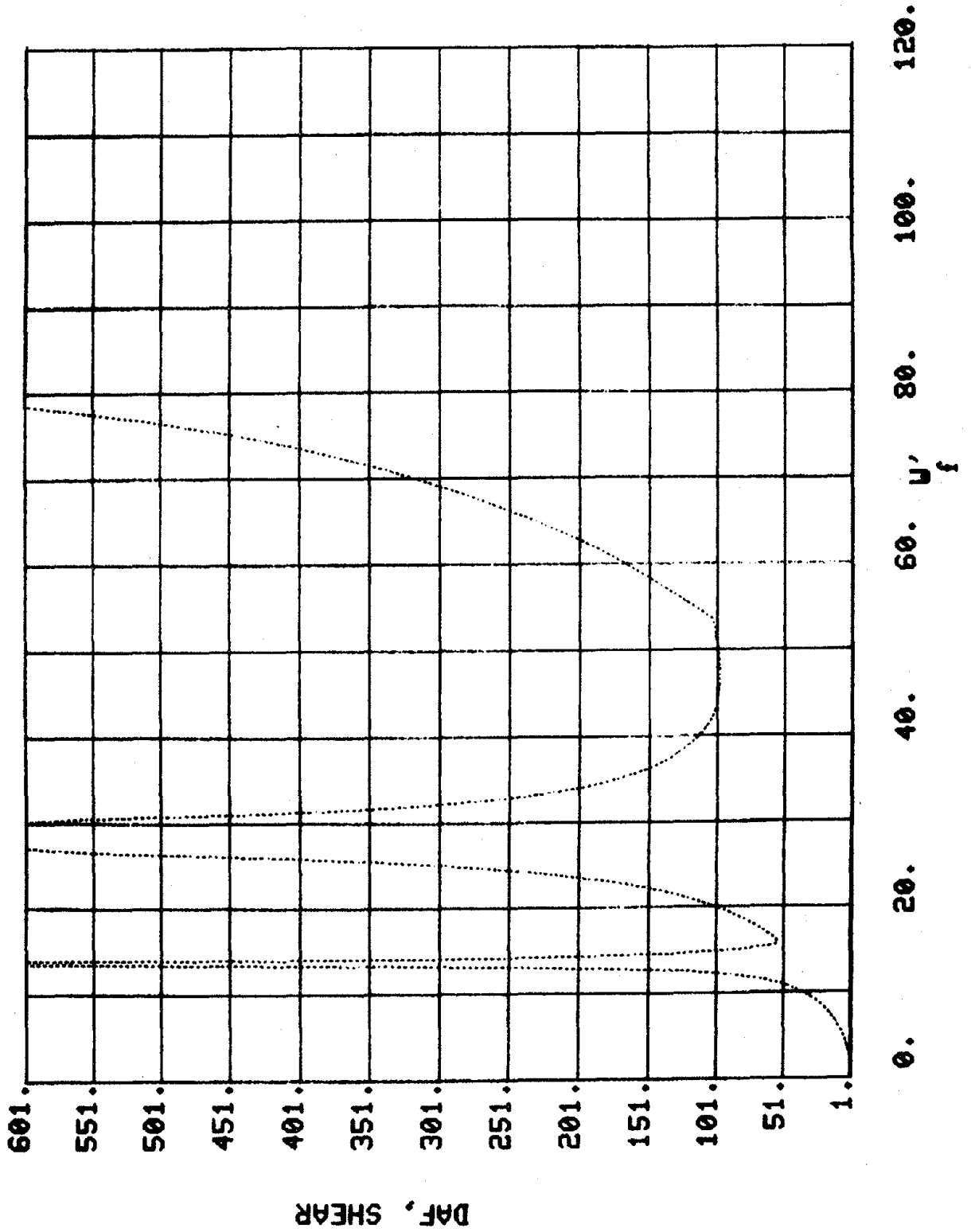


Fig. 29 - DAF, Deflection (Three span, .8L-L-.8L)

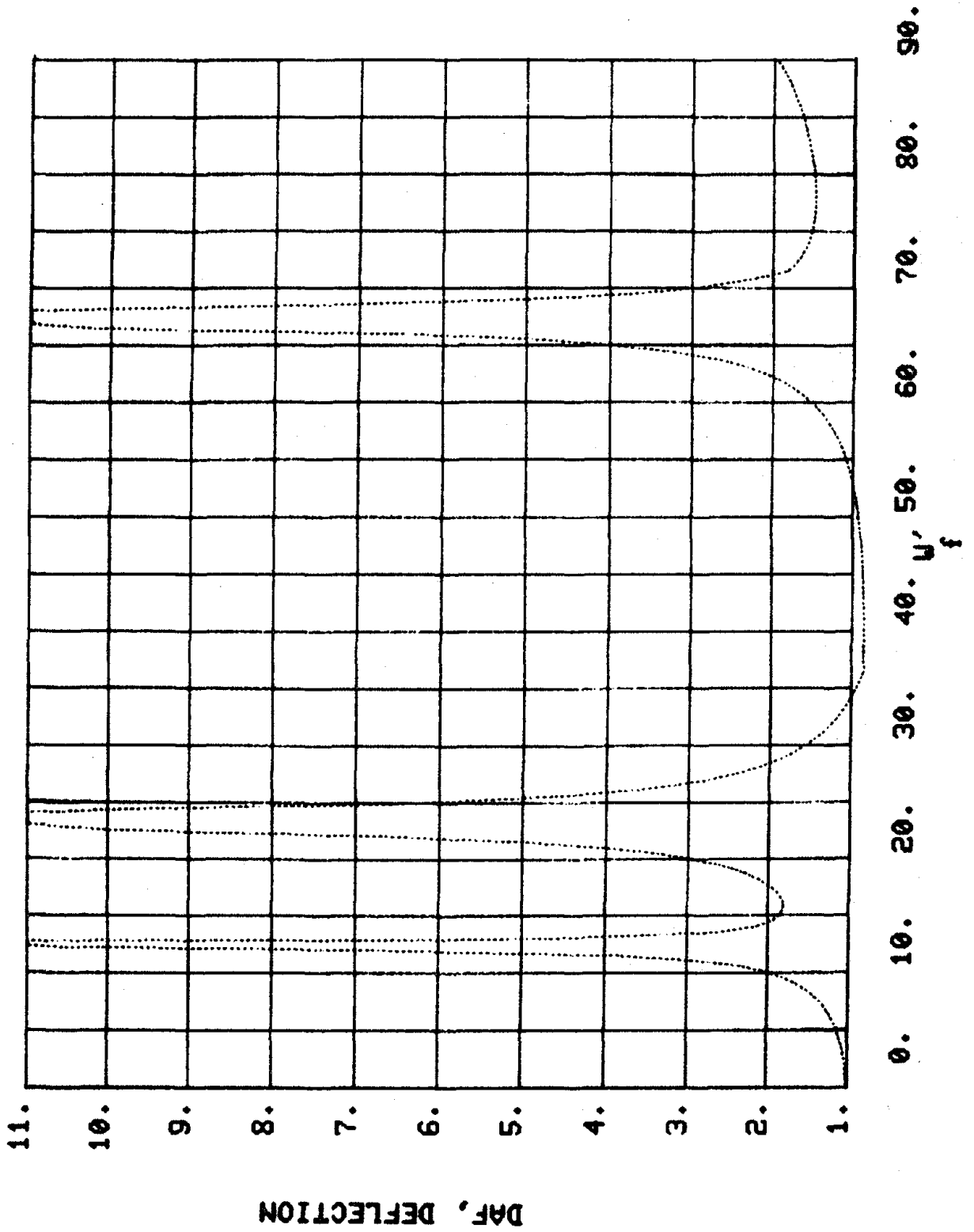


Fig. 30 - DAF, Moment (Three span, .8L-L-.8L)

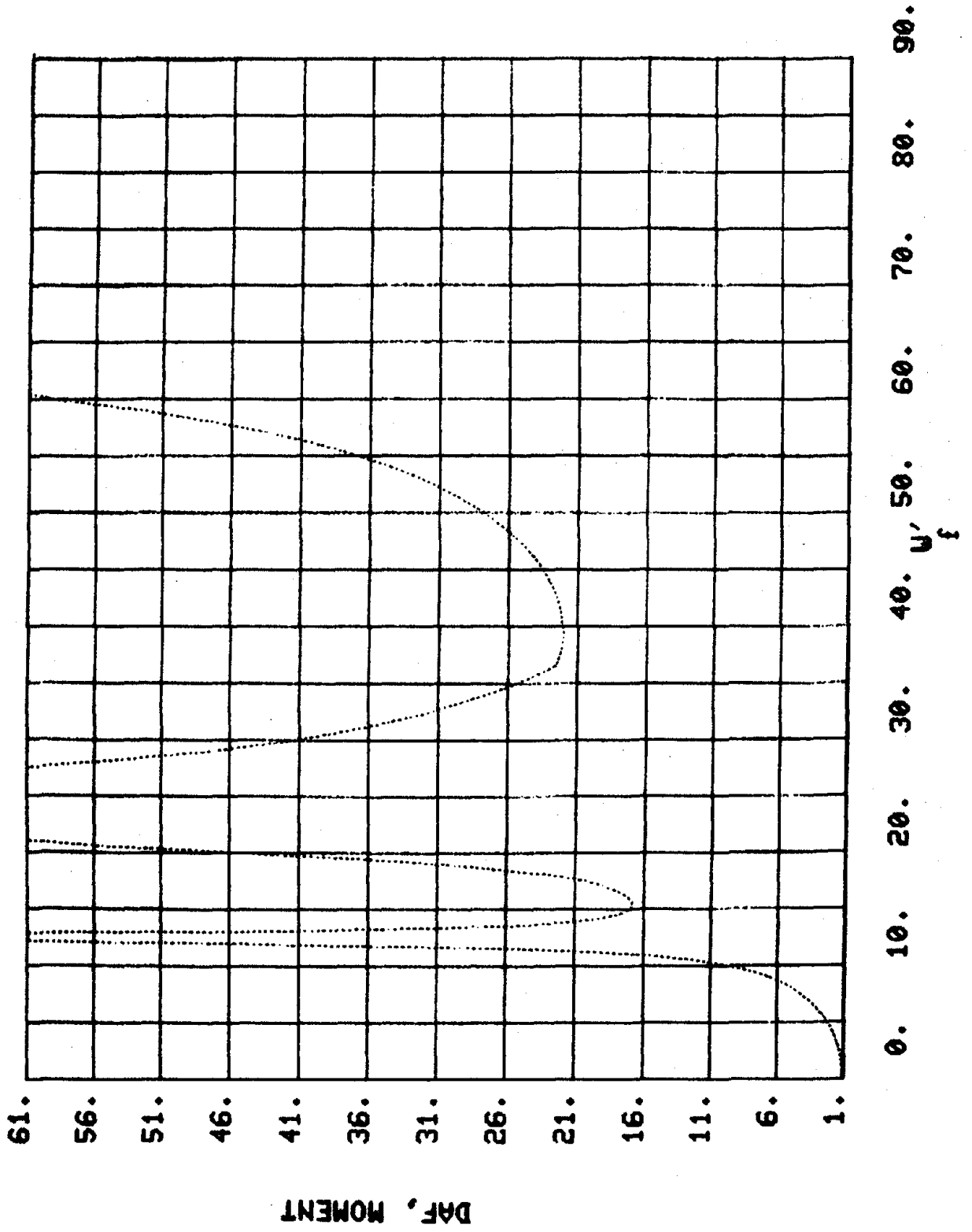


Fig. 31 - DAF, Shear (Three span, .8L-L-.8L)

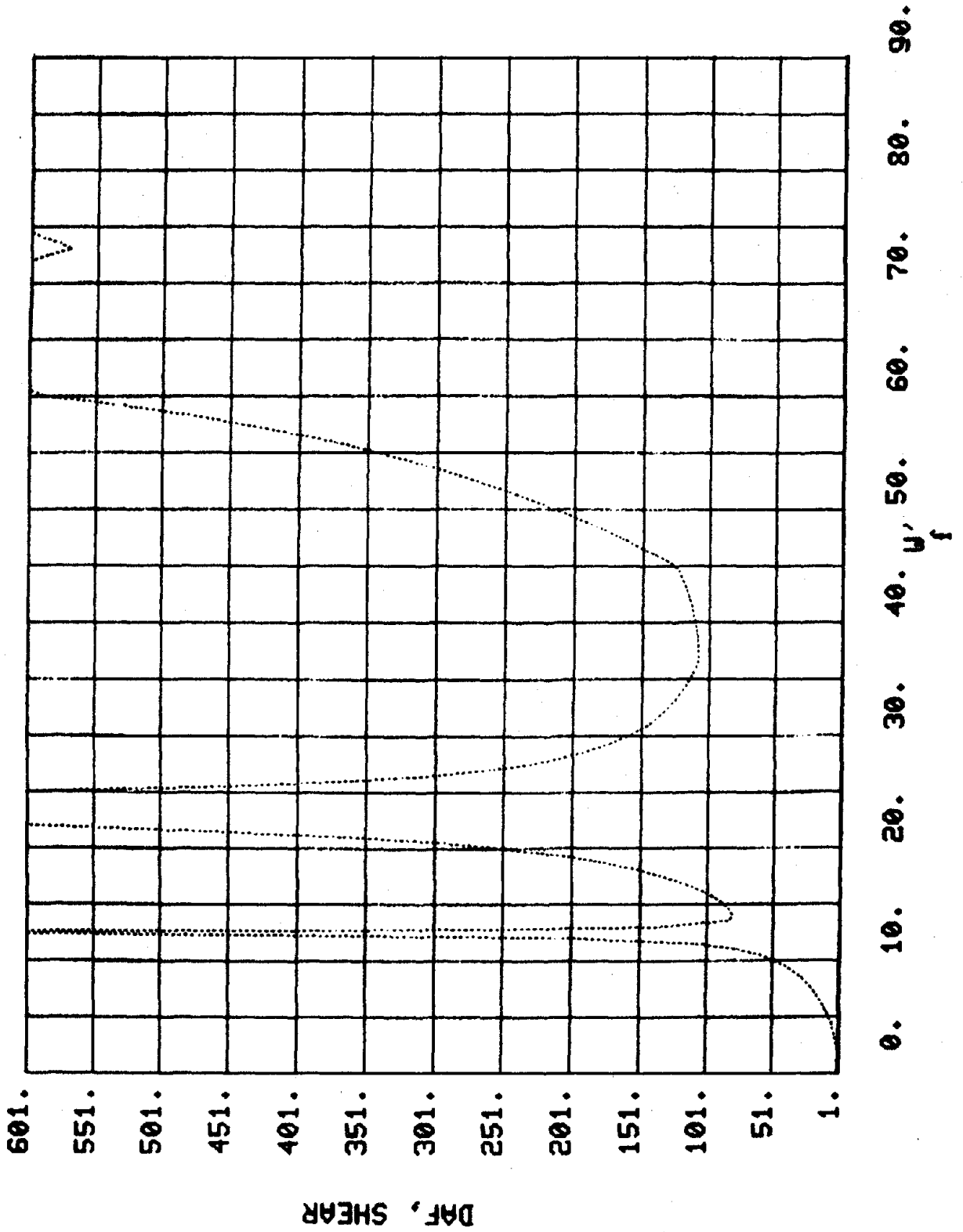


Fig. 32 - DAF, Deflection (Three span, .7L-L-.9L)

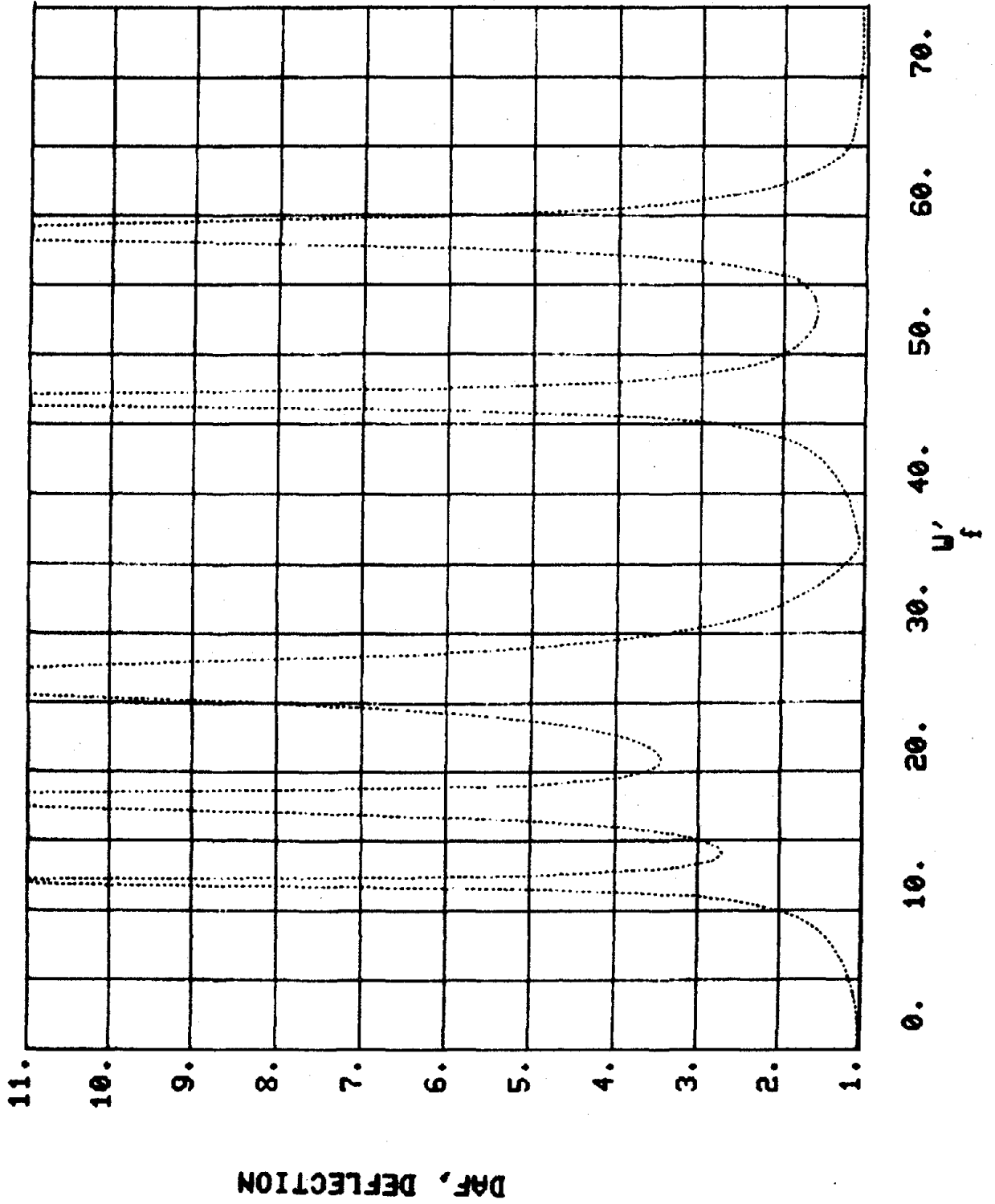


Fig. 33 - DAF, Moment (Three span, .7L-L-.9L)

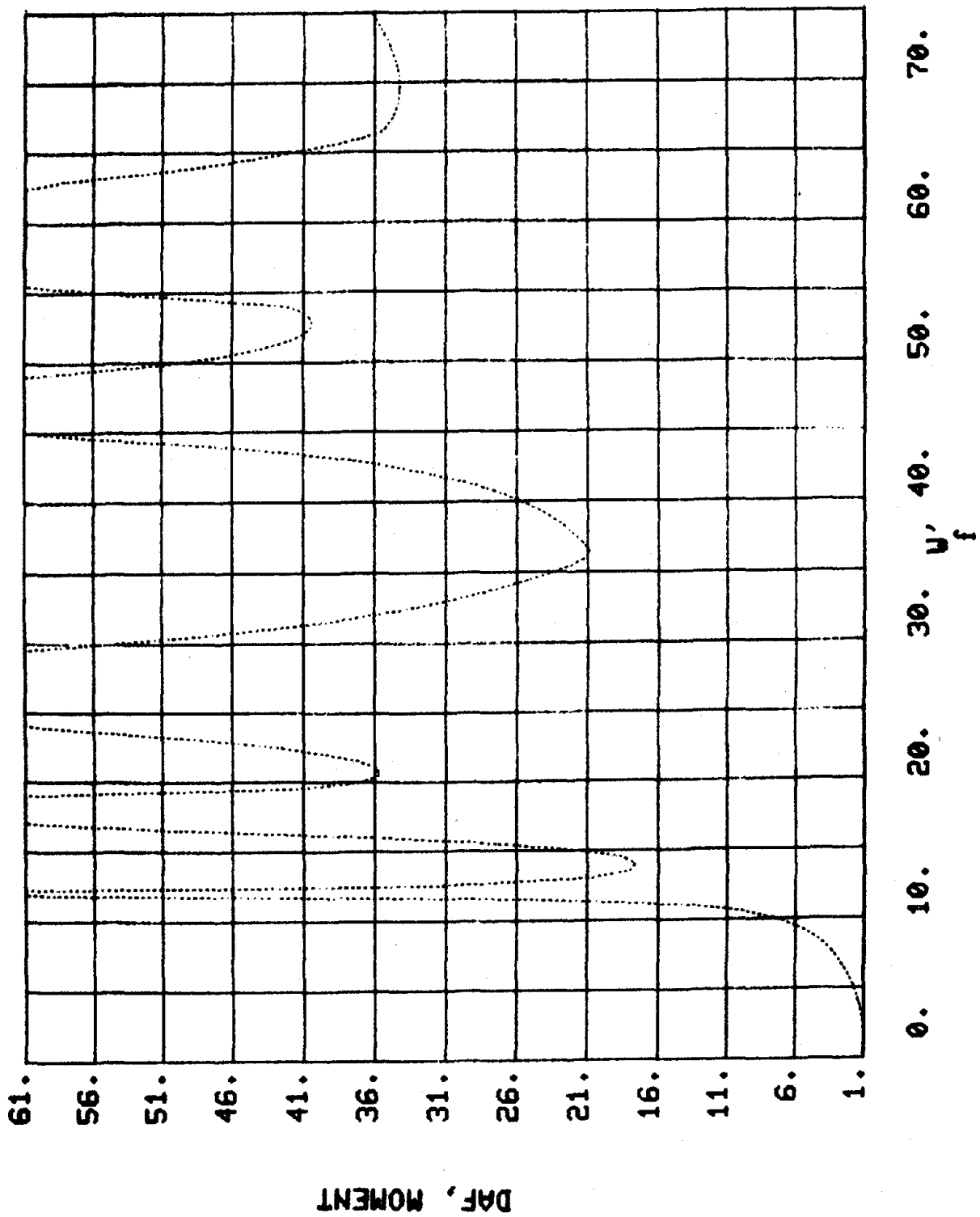


Fig. 34 - DAF, Shear (Three span, .7L-L-.9L)

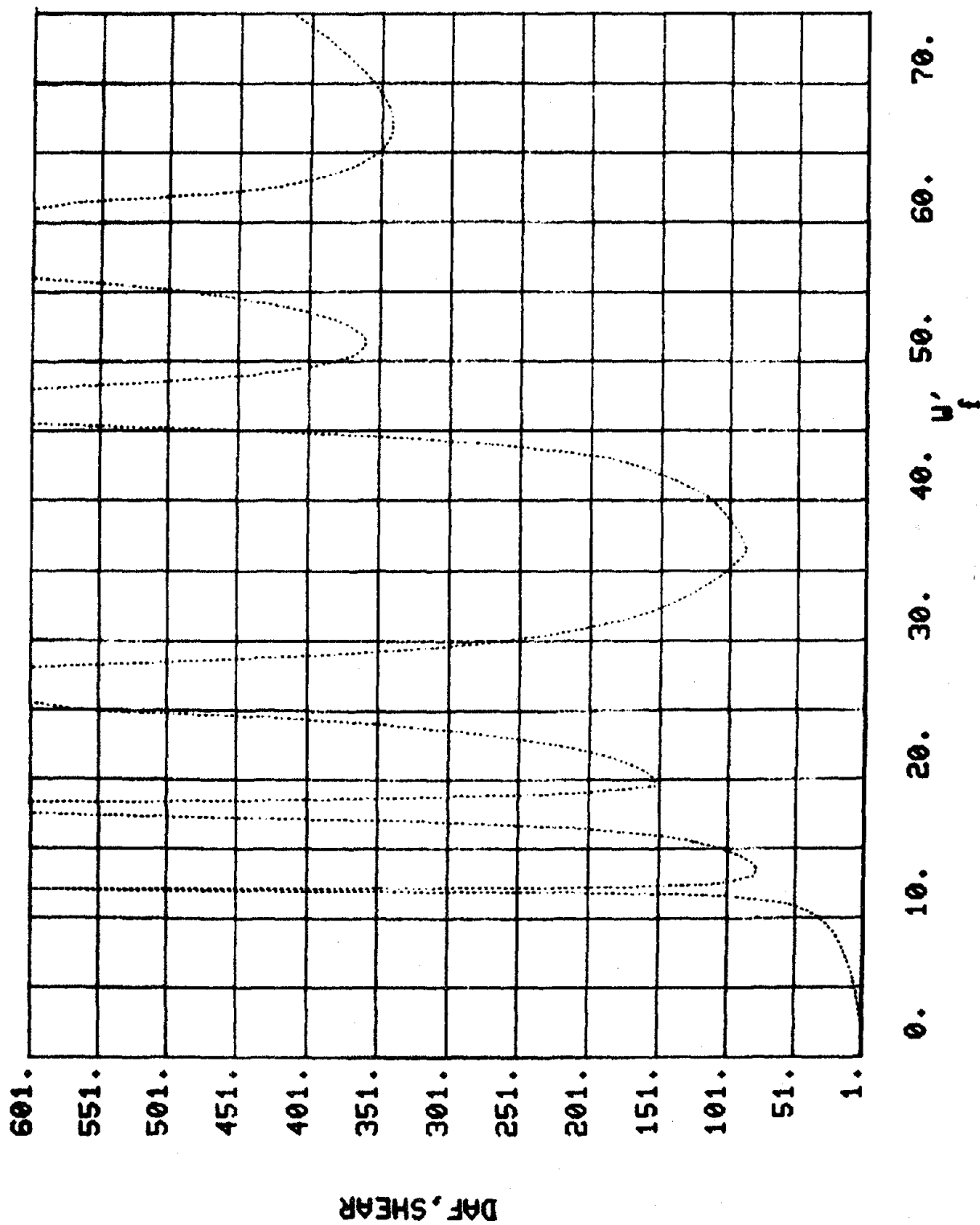


Fig. 35 - DAF, Deflection (Three span, .9L-L-.9L)

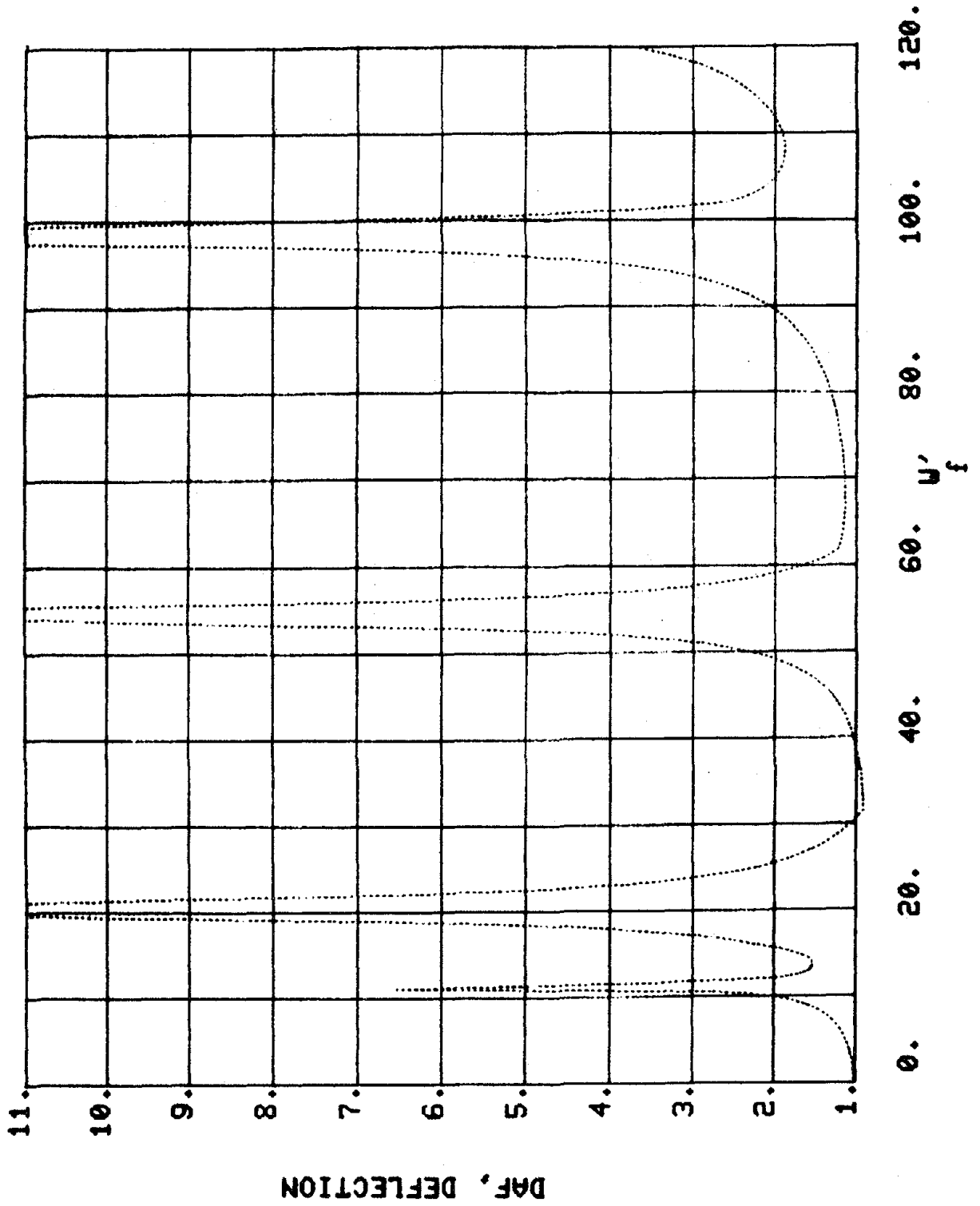


Fig. 36 - DAF, Moment (Three span, .9L-L-.9L)

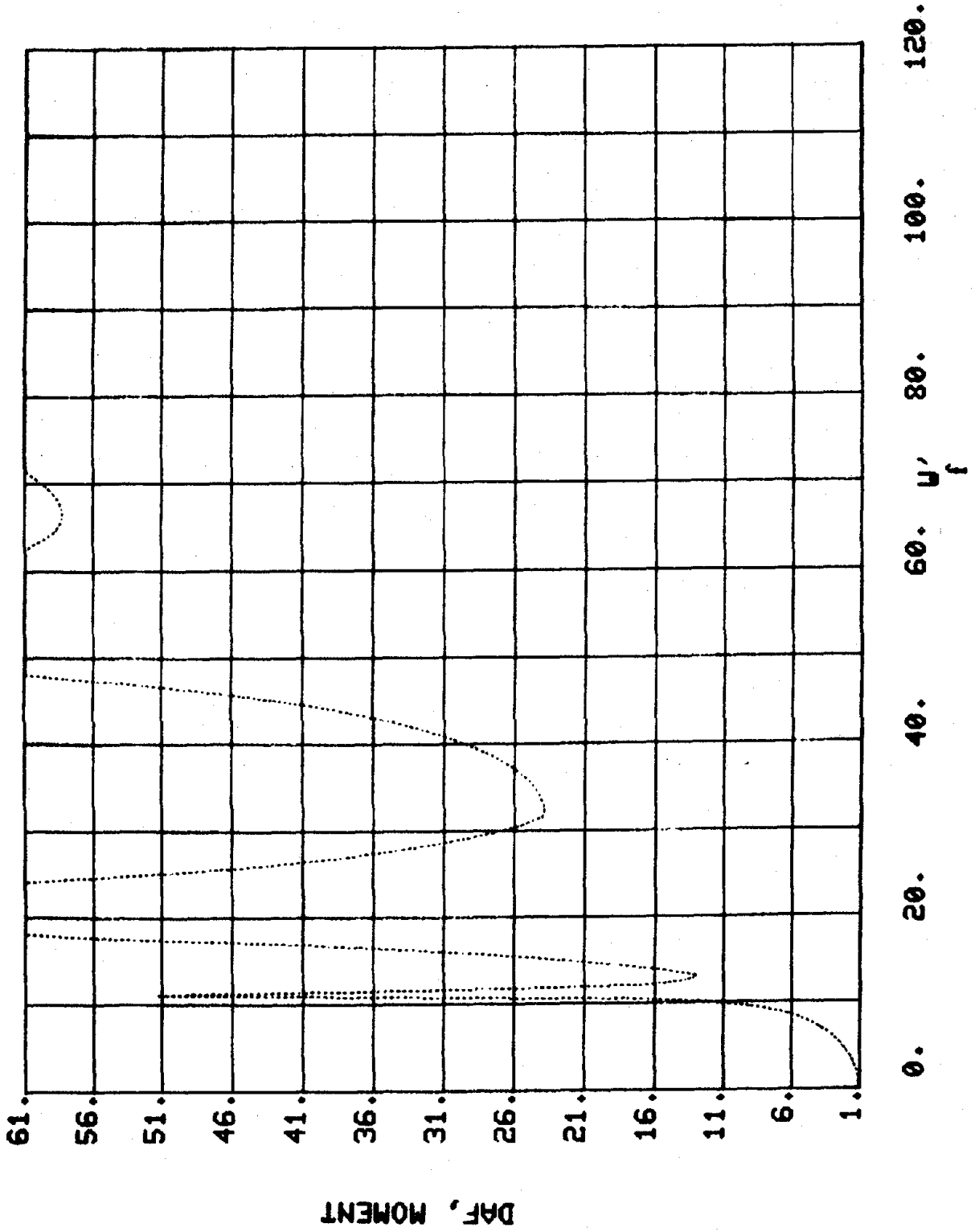


Fig. 37 - DAF, Shear (Three span, .9L-L-.9L)

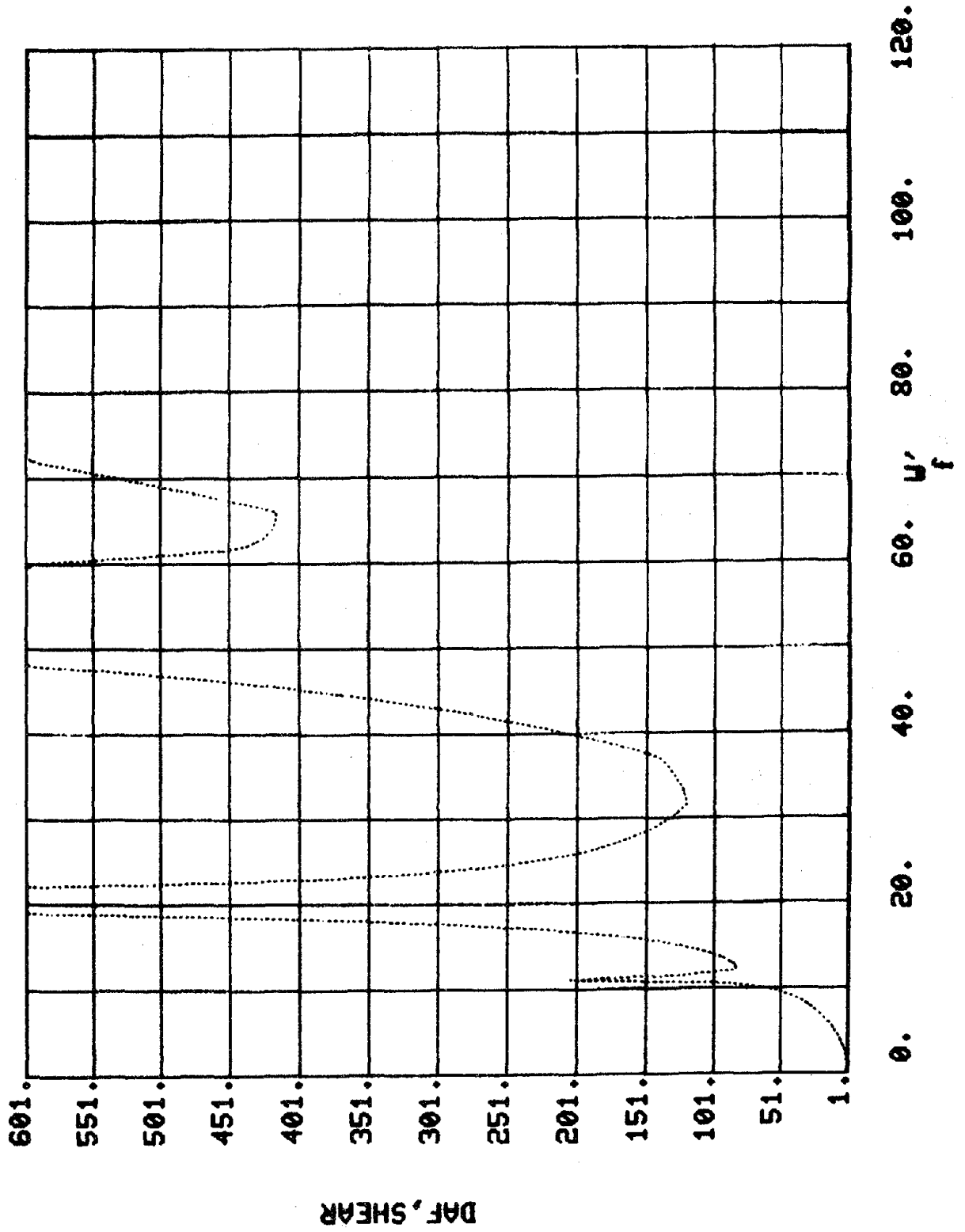


Fig. 38 - DAF, Deflection (Four span, .5L-L-L-.5L)

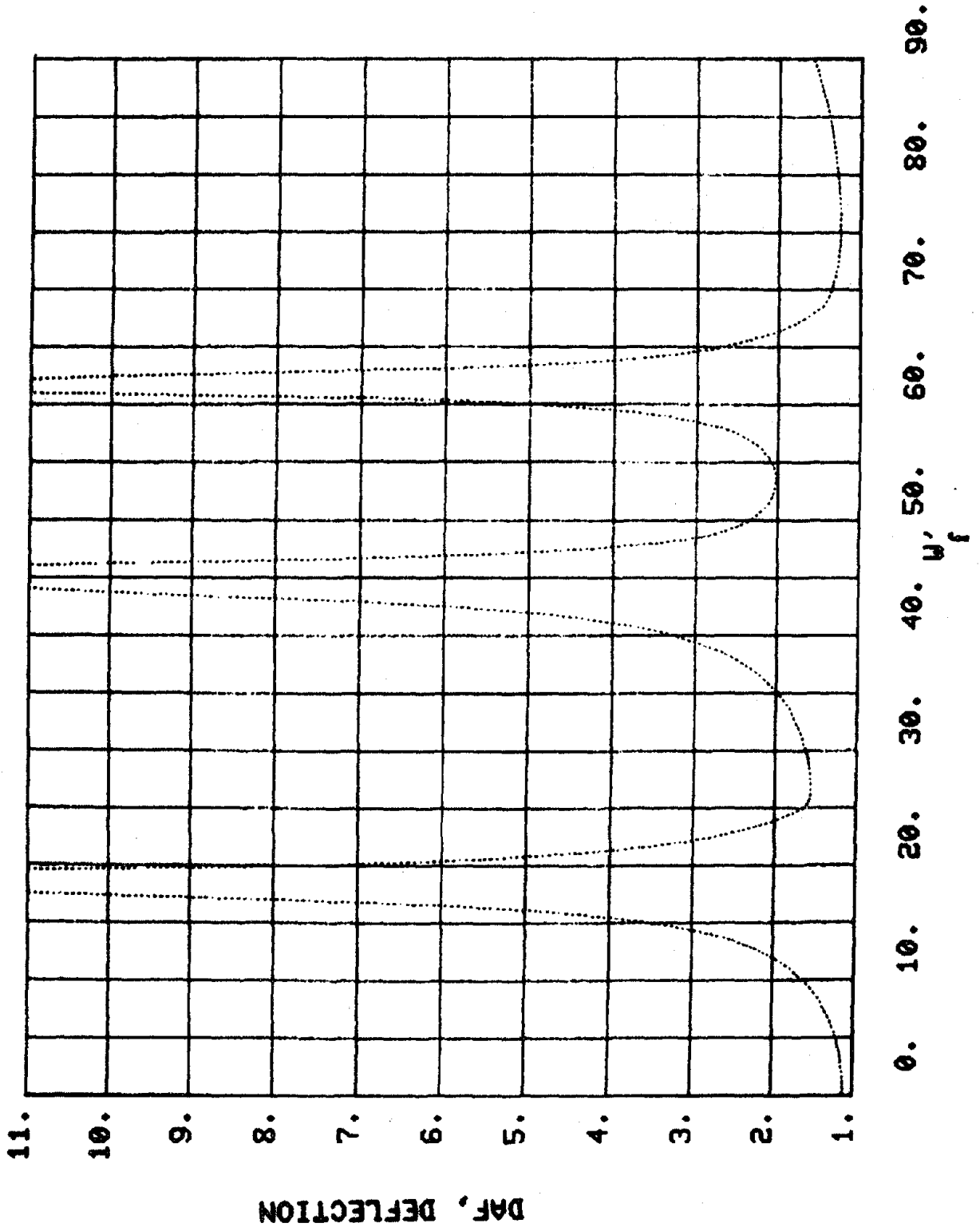


Fig. 39 - DAF, Moment (Four span, .5L-L-L-.5L)

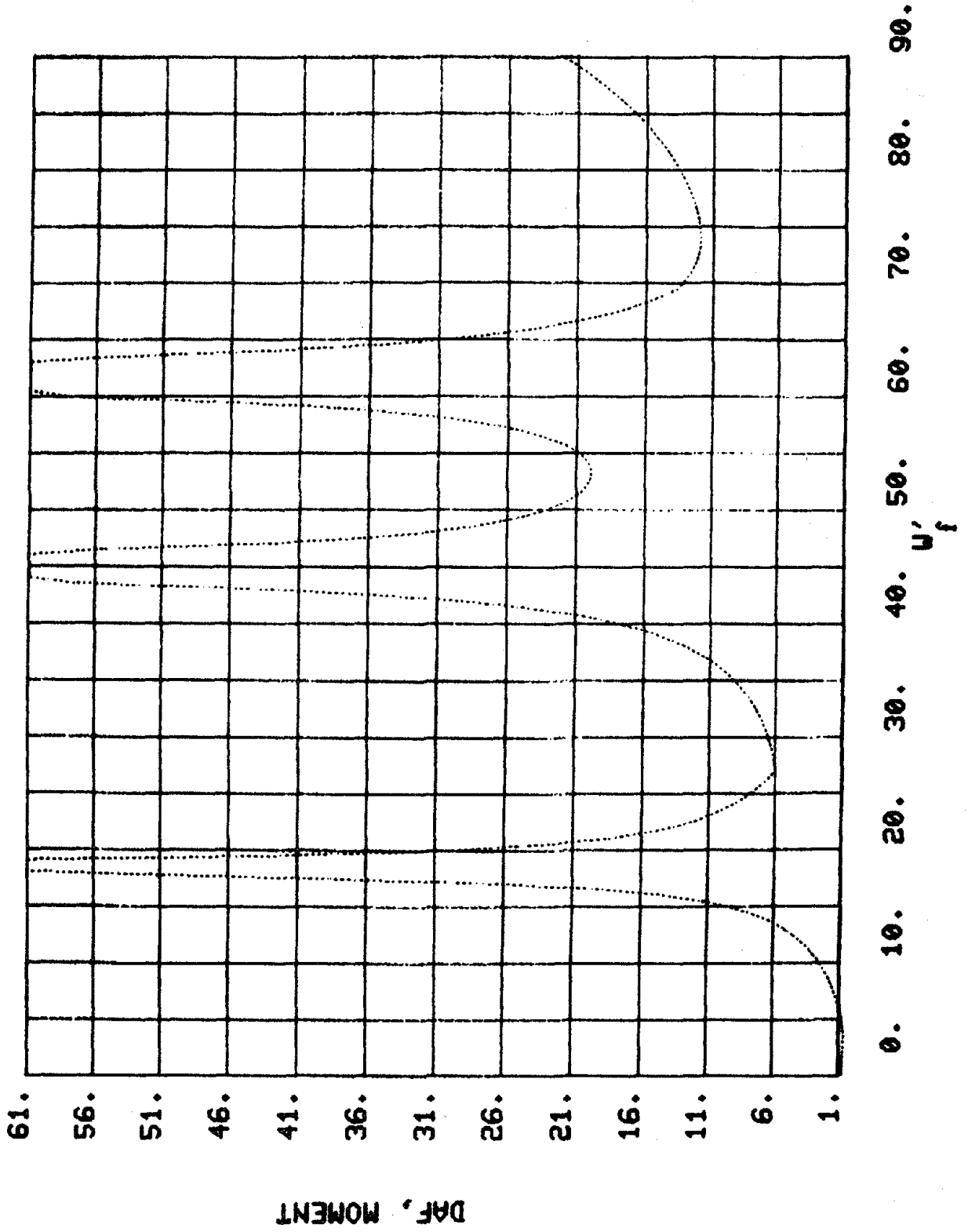


Fig. 40 - DAF, Shear (Four span, .5L-L-L-.5L)

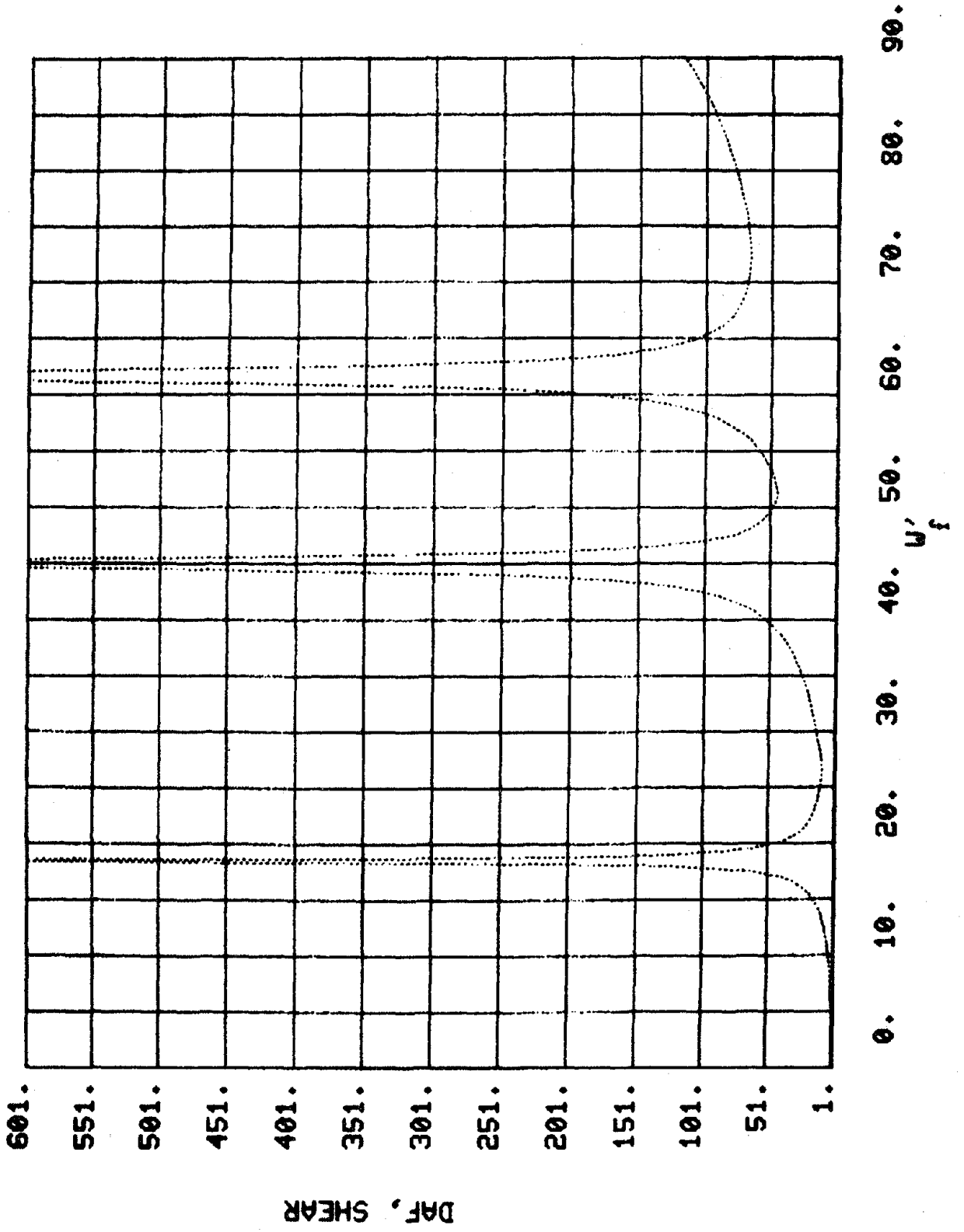


Fig. 41 - DAF, Deflection (Four span, .6L-L-L-.6L)

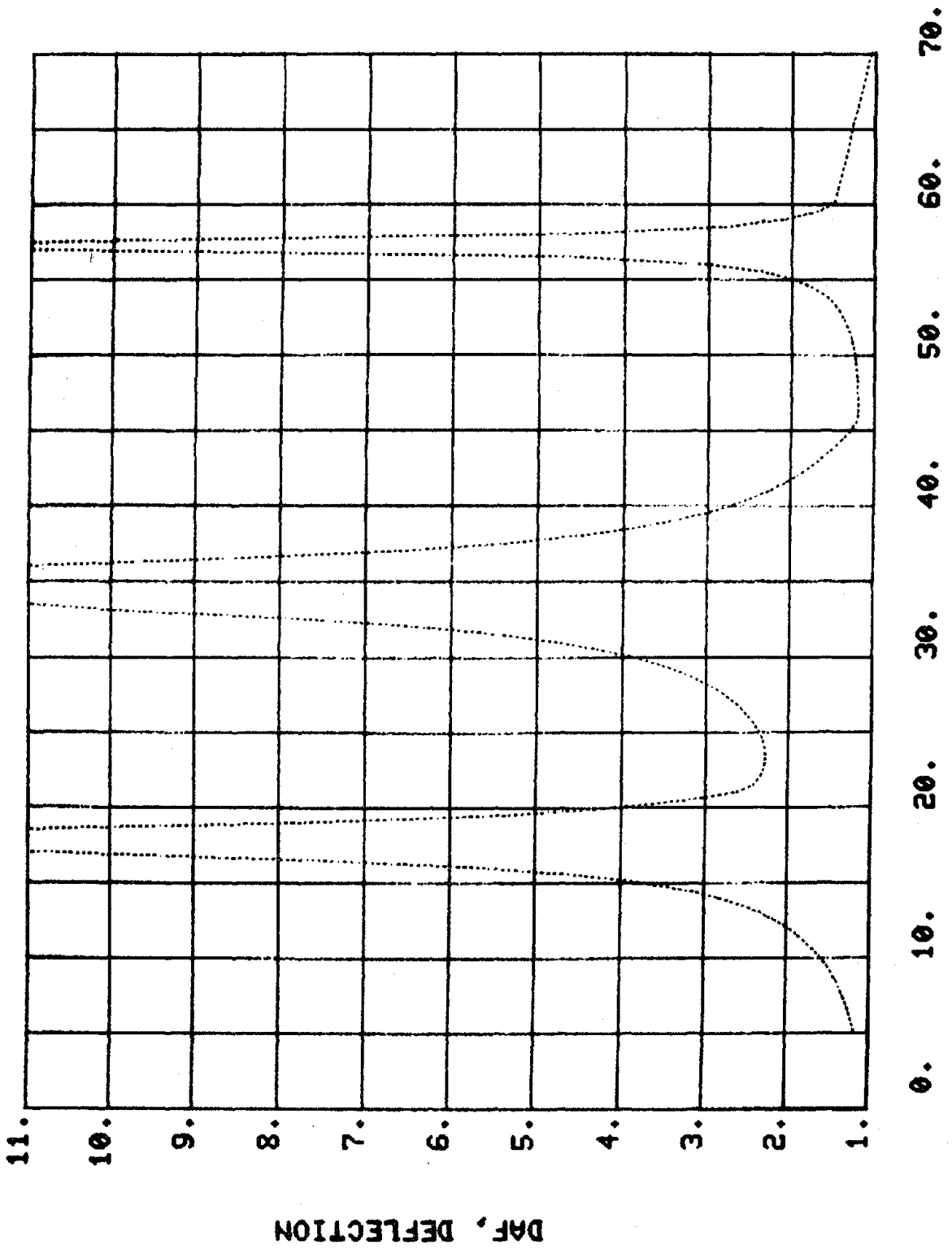


Fig. 41 - DAF, Moment (Four span, .6L-L-L-.6L)

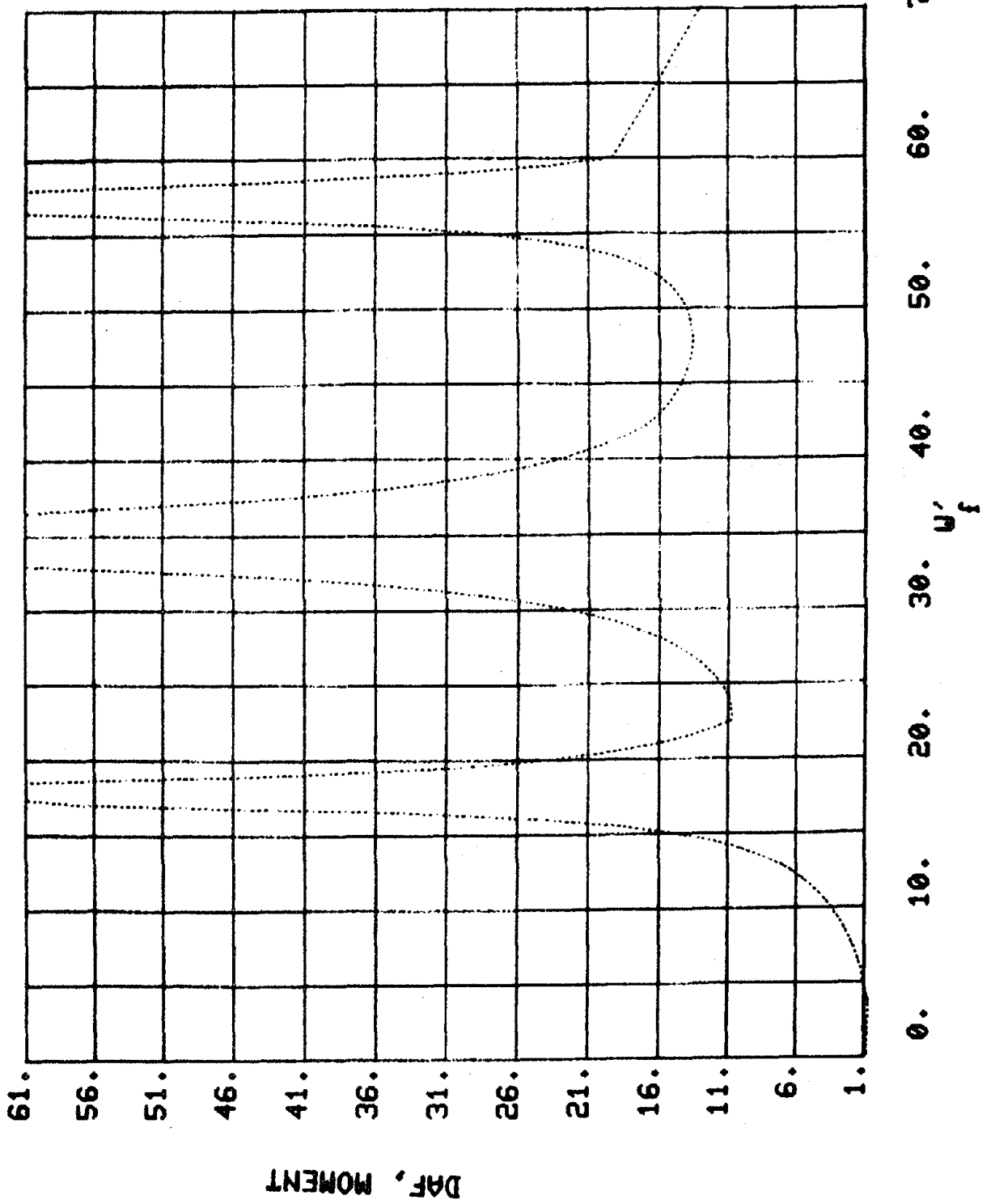


Fig. 41 - DAF, Shear (Four span, .6L-L-L-.6L)

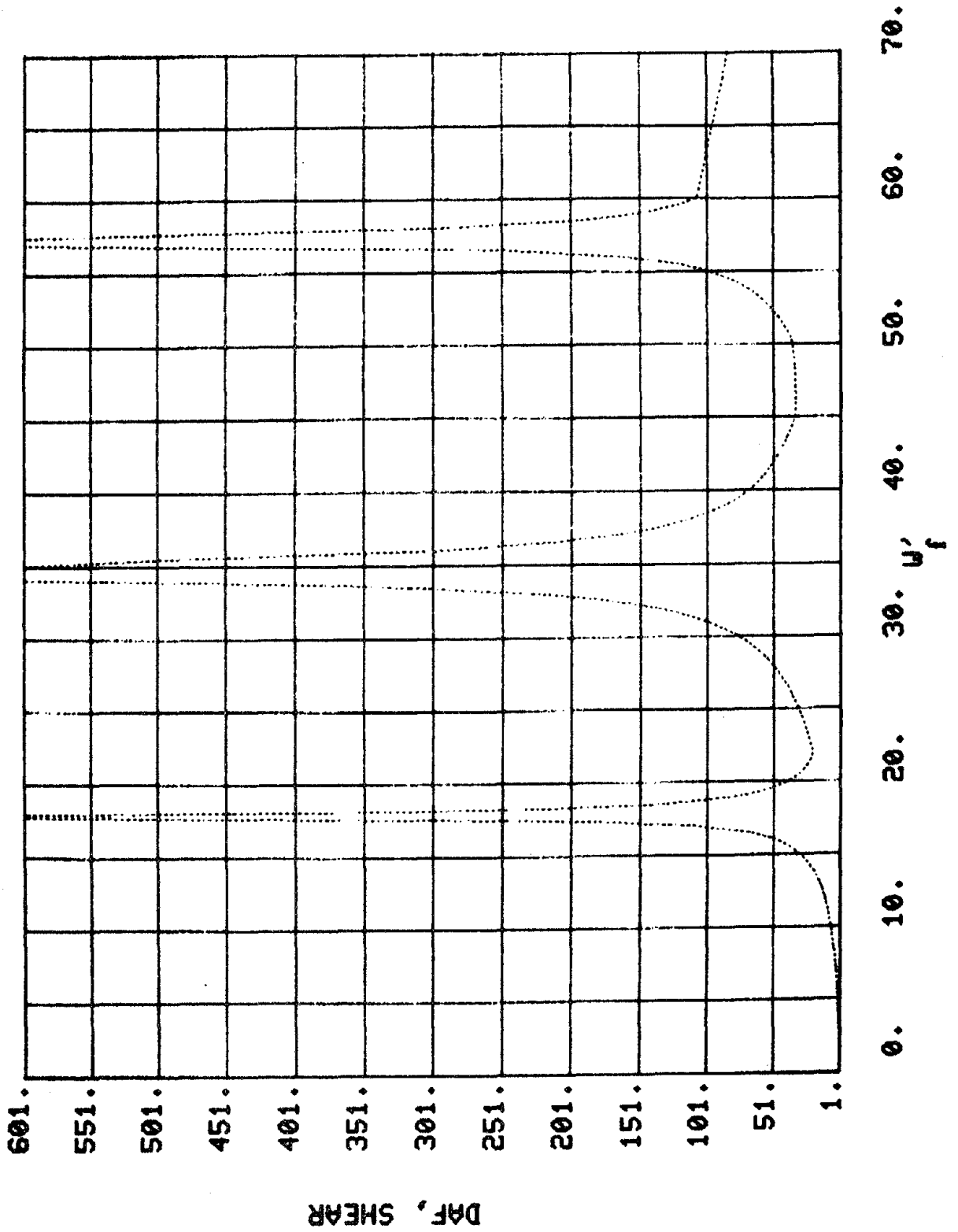


Fig. 44 - DAF, Deflection (Four span, .9L-L-L-.9L)

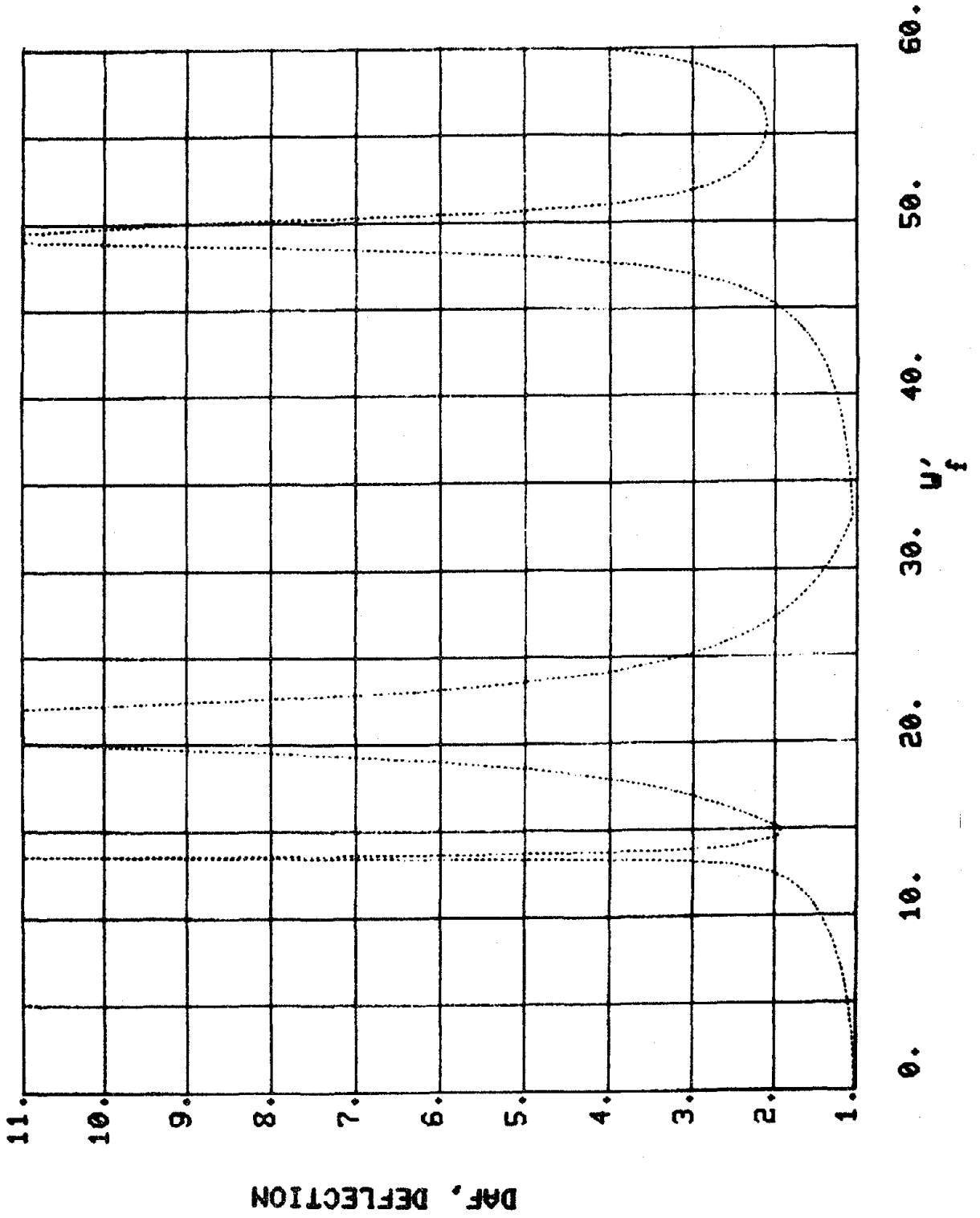


Fig. 44 - DAF, Moment (Four span, .9L-L-L-.9L)

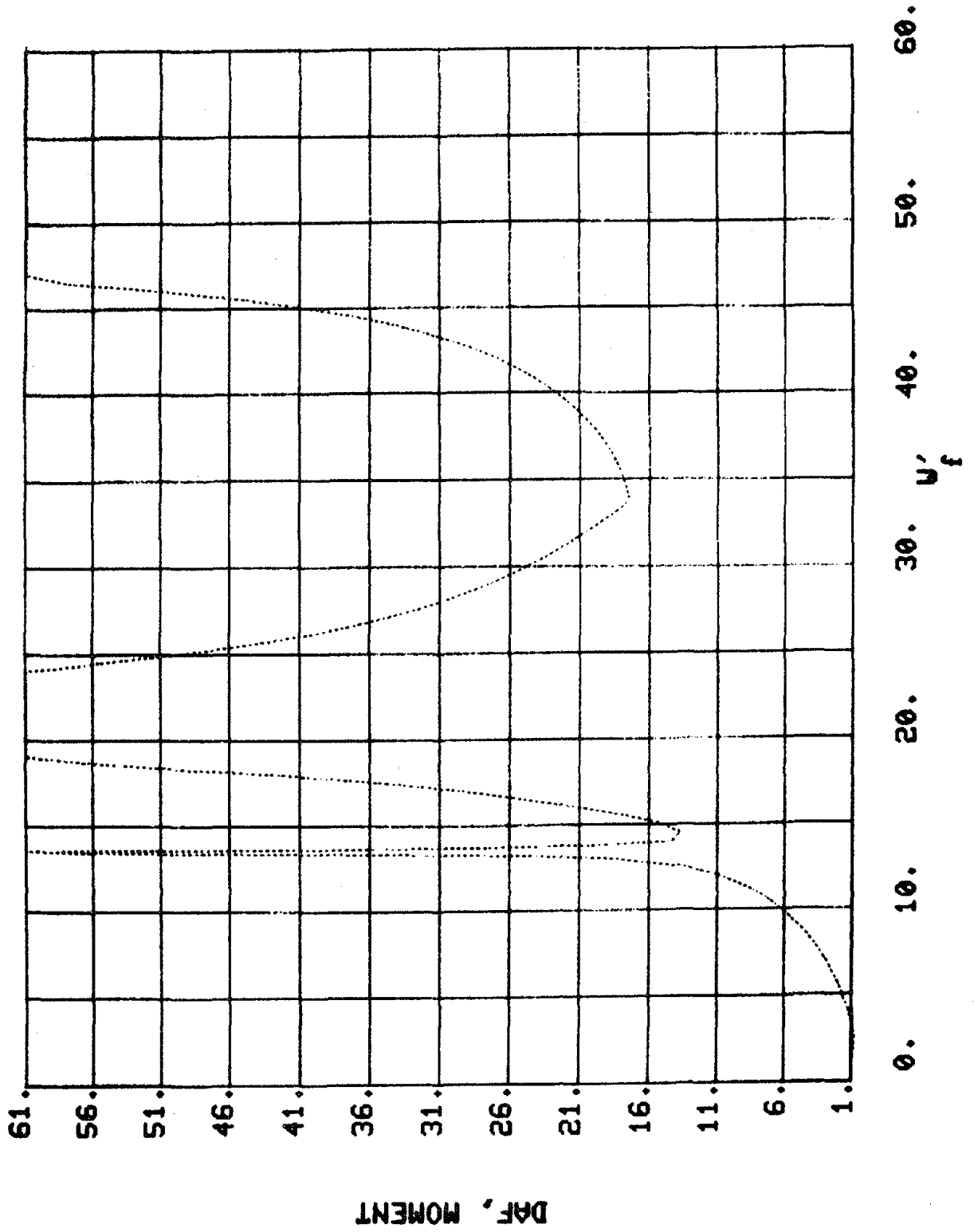
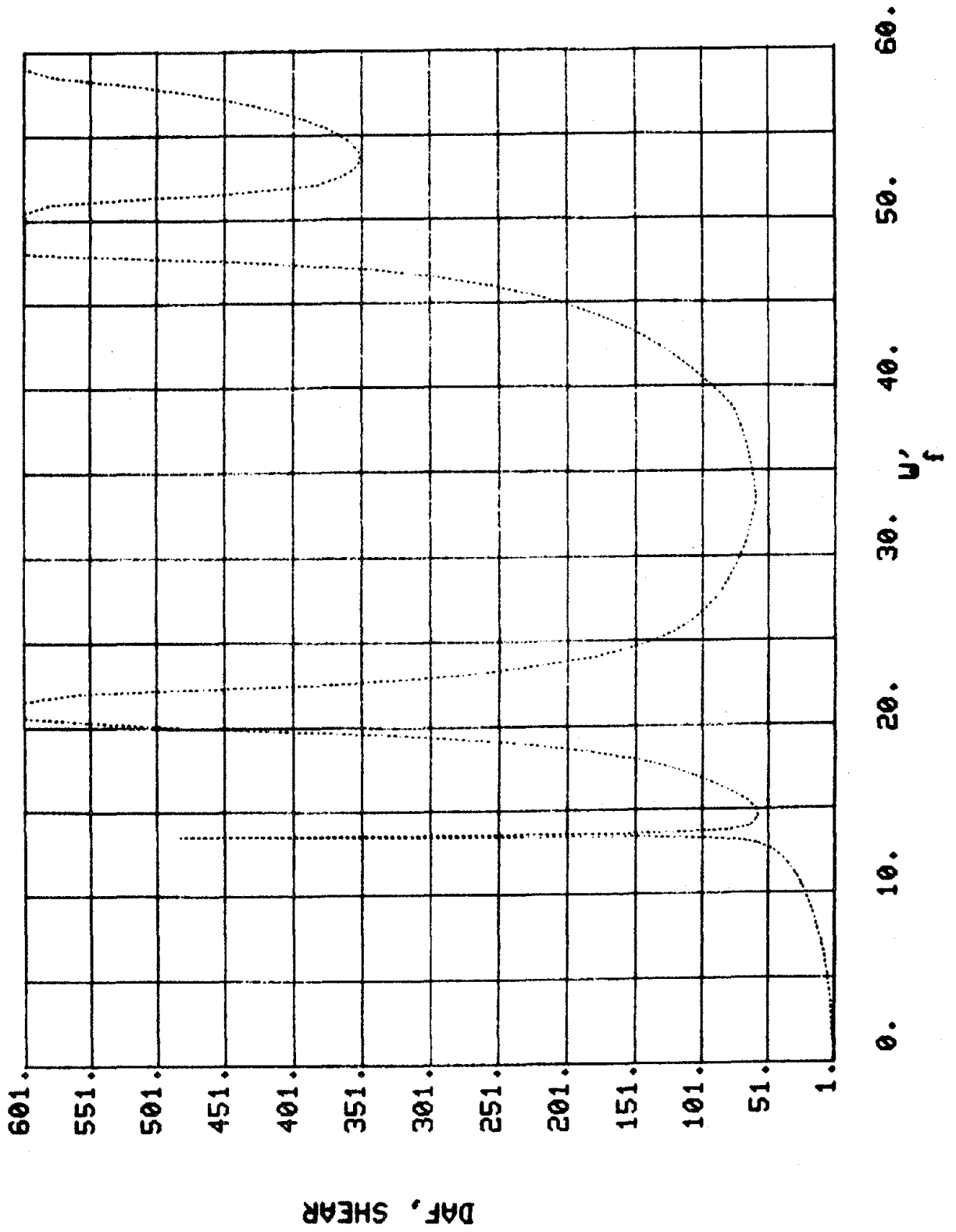


Fig. 44 - DAF, Shear (Four span, .9L-L-L-.9L)



V. SUMMARY AND CONCLUDING REMARKS

The dynamics of differential support settlements of highway bridges has been investigated. A governing differential equation due to pulsating support settlements has been derived. The deflection, moment, shear, velocity, and acceleration expressions were obtained in terms of Fourier series. Neglecting damping, these expressions were compared to the results obtained by the dynamic three moment equation. An excellent correlation was observed. The parameters in these expressions were non-dimensionalized as much as possible and various plots were obtained for the dynamic amplification factors versus forcing frequencies. An example bridge has been used to illustrate how these dynamic amplification factors can be properly used.

Experiences of the bridge performances during the San Fernando Earthquake [8,11,16] suggest that a majority of those bridges which collapsed were subjected to unusually high shear. This investigation seems to verify those observations made after the San Fernando Earthquake. As can be seen in the charts, the dynamic amplification factors for shear are quite high especially under higher frequencies.

Further investigations are recommended about the damping characteristics of bridge structures. The peaks in the dynamic amplification factors, indicating the dynamic resonances, may or may not be feasible since no structures are completely damping free and a considerable dynamic oscillation (duration of vibration) is normally required to develop a full resonance phenomenon. Another important point is the fact that an actual earthquake motion is by no means a simple harmonic motion. A study correlating the validity of using a simple harmonic

motion to represent a structural behavior due to an earthquake seems to be in order.

Finally, it is strongly recommended that accurate yield data collection regarding time dependent relative support settlements be devised and maintained for structures with a long horizontal dimension such as highway bridges. It is evident from this study that the dynamic support settlements have significant effects on the behavior of bridge superstructures.

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APPENDIX I

A. Sample input and output of the Dynamic Program

i) input

TWO SPAN BRIDGE 33FT-27FT, WF=40 RAD/SEC
 3,2,200,12,3,0,0,720.,1.46653,3000000.,92850.,.03927,.003927,40.,
 .0805,
 396.,
 0.,72.,144.,216.,288.,360.,396.,432.,504.,576.,648.,720.,
 648.,720.,
 1000,.000001,

ii) output

TWO SPAN BRIDGE 33FT-27FT, WF=40 RAD/SEC

BLANK COMMON ARRAY 2265

NATURAL FREQUENCIES AND PARTICULAR SOLUTION

DETERMINANT APPROACHING INFINITY FOR FREQ EQUAL TO .82974D+01
 DETERMINANT APPROACHING INFINITY FOR FREQ EQUAL TO .33190D+02
 DETERMINANT APPROACHING INFINITY FOR FREQ EQUAL TO .74676D+02
 FREQ (MODE 1) = .3143D+02 AFTER 138 ITRNS DET = -.3289D-08
 FREQ (MODE 2) = .5559D+02 AFTER 32 ITRNS DET = -.1447D-10
 FREQ (MODE 3) = .1218D+03 AFTER 39 ITRNS DET = -.5747D-08

TOTAL SPAN LENGTH	= .7200D+03
MASS DENSITY PER UNIT LENGTH	= .1467D+01
YOUNG'S MODULUS	= .3000D+07
MOMENT OF INERTIA	= .9285D+05

DURATION OF EARTHQUAKE	=	.3927D-01
TIME INCREMENT IN TIME HISTORY	=	.3927D-02
EARTHQUAKE (ANGULAR) VELOCITY	=	.4000D+02
NUMBER OF SPANS	=	2
NUMBER OF MODES IN FREE VIBRN	=	3
NUMBER OF TERMS TO BE SUMMED	=	200
NUMBER OF LOCATIONS (T HISTORY)	=	12

TIME HISTORY FOR THE POINT(X= .000D+00)

TIME	PART-DISP	PART-VELD	PART-ACCL	PART-MOMT	PART-SHRF
.000D+00	.000D+00	.000D+00	.000D+00	.000D+00	.000D+00
.393D-02	.000D+00	.000D+00	.000D+00	.000D+00	.487D+03
.785D-02	.000D+00	.000D+00	.000D+00	.000D+00	.961D+03
.118D-01	.000D+00	.000D+00	.000D+00	.000D+00	.141D+04
.157D-01	.000D+00	.000D+00	.000D+00	.000D+00	.183D+04
.196D-01	.000D+00	.000D+00	.000D+00	.000D+00	.220D+04
.236D-01	.000D+00	.000D+00	.000D+00	.000D+00	.252D+04
.275D-01	.000D+00	.000D+00	.000D+00	.000D+00	.277D+04
.314D-01	.000D+00	.000D+00	.000D+00	.000D+00	.296D+04
.353D-01	.000D+00	.000D+00	.000D+00	.000D+00	.307D+04
.383D-01	.000D+00	.000D+00	.000D+00	.000D+00	.311D+04

MAXIMUM ABSOLUTE VALUES

MAX ABS	.000D+00	.000D+00	.000D+00	.000D+00	.311D+04
CORR TIME	.393D-01	.393D-01	.393D-01	.393D-01	.393D-01

TIME HISTORY FOR THE POINT(X= .720D+02)

TIME	PART-DISP	PART-VELD	PART-ACCL	PART-MOMT	PART-SHRF
.000D+00	.000D+00	-.148D+00	.000D+00	.000D+00	.000D+00
.393D-02	-.578D-03	-.146D+00	.925D+00	-.348D+05	.432D+03
.785D-02	-.114D-02	-.141D+00	.183D+01	-.688D+05	.854D+03
.118D-01	-.168D-02	-.132D+00	.269D+01	-.101D+06	.125D+04
.157D-01	-.217D-02	-.120D+00	.348D+01	-.131D+06	.162D+04
.196D-01	-.261D-02	-.105D+00	.418D+01	-.157D+06	.195D+04
.236D-01	-.299D-02	-.869D-01	.479D+01	-.180D+06	.224D+04
.275D-01	-.329D-02	-.671D-01	.527D+01	-.198D+06	.246D+04
.314D-01	-.352D-02	-.457D-01	.563D+01	-.212D+06	.263D+04
.353D-01	-.365D-02	-.231D-01	.584D+01	-.220D+06	.273D+04
.393D-01	-.370D-02	.543D-06	.582D+01	-.223D+06	.276D+04

MAXIMUM ABSOLUTE VALUES

MAX ABS	.370D-02	.148D+00	.592D+01	.223D+06	.276D+04
CORR TIME	.393D-01	.000D+00	.393D-01	.393D-01	.393D-01

TIME HISTORY FOR THE POINT(X= .144D+03)

TIME	PART-DISP	PART-VELO	PART-ACCL	PART-MOMT	PART-SHRF
.000D+00	.000D+00	-.133D+00	.000D+00	.000D+00	.000D+00
.393D-02	-.519D-03	-.131D+00	.830D+00	-.633D+05	.324D+03
.785D-02	-.103D-02	-.126D+00	.164D+01	-.125D+06	.640D+03
.118D-01	-.151D-02	-.118D+00	.241D+01	-.184D+06	.941D+03
.157D-01	-.195D-02	-.107D+00	.312D+01	-.238D+06	.122D+04
.196D-01	-.235D-02	-.938D-01	.375D+01	-.286D+06	.147D+04
.236D-01	-.268D-02	-.780D-01	.429D+01	-.327D+06	.168D+04
.275D-01	-.296D-02	-.603D-01	.473D+01	-.360D+06	.185D+04
.314D-01	-.316D-02	-.410D-01	.505D+01	-.385D+06	.197D+04
.353D-01	-.328D-02	-.208D-01	.524D+01	-.399D+06	.205D+04
.393D-01	-.332D-02	.487D-06	.531D+01	-.404D+06	.207D+04

MAXIMUM ABSOLUTE VALUES

MAX ABS	.332D-02	.133D+00	.531D+01	.404D+06	.207D+04
CORR TIME	.393D-01	.000D+00	.393D-01	.393D-01	.393D-01

TIME HISTORY FOR THE POINT(X= .216D+03)

TIME	PART-DISP	PART-VELO	PART-ACCL	PART-MOMT	PART-SHRF
.000D+00	.000D+00	.181D+00	.000D+00	.000D+00	.000D+00
.393D-02	.709D-03	.179D+00	-.113D+01	-.866D+05	.315D+03
.785D-02	.140D-02	.172D+00	-.224D+01	-.171D+06	.622D+03
.118D-01	.206D-02	.161D+00	-.328D+01	-.251D+06	.914D+03
.157D-01	.266D-02	.147D+00	-.426D+01	-.325D+06	.118D+04
.196D-01	.320D-02	.128D+00	-.513D+01	-.391D+06	.142D+04
.236D-01	.367D-02	.107D+00	-.588D+01	-.448D+06	.163D+04
.275D-01	.404D-02	.823D-01	-.646D+01	-.493D+06	.179D+04
.314D-01	.431D-02	.560D-01	-.689D+01	-.526D+06	.192D+04
.353D-01	.447D-02	.283D-01	-.716D+01	-.547D+06	.198D+04
.393D-01	.453D-02	-.866D-08	-.725D+01	-.554D+06	.201D+04

MAXIMUM ABSOLUTE VALUES

MAX ABS	.453D-02	.181D+00	.725D+01	.554D+06	.201D+04
CORR TIME	.393D-01	.000D+00	.393D-01	.393D-01	.393D-01

TIME HISTORY FOR THE POINT(X= .288D+03)

TIME	PART-DISP	PART-VELO	PART-ACCL	PART-MOMT	PART-SHRF
.000D+00	.000D+00	.911D+00	.000D+00	.000D+00	.000D+00

.000D+00	.000D+00	.000D+00	-.070D+01	-.120D+06	.835D+03
.785D-02	.704D-02	.866D+00	-.113D+02	-.237D+06	.125D+04
.118D-01	.103D-01	.812D+00	-.165D+02	-.349D+06	.184D+04
.157D-01	.134D-01	.737D+00	-.214D+02	-.452D+06	.239D+04
.196D-01	.161D-01	.644D+00	-.258D+02	-.543D+06	.287D+04
.236D-01	.184D-01	.535D+00	-.295D+02	-.621D+06	.328D+04
.275D-01	.203D-01	.414D+00	-.325D+02	-.684D+06	.362D+04
.314D-01	.217D-01	.281D+00	-.347D+02	-.731D+06	.386D+04
.353D-01	.225D-01	.142D+00	-.360D+02	-.759D+06	.401D+04
.393D-01	.228D-01	-.335D-05	-.364D+02	-.768D+06	.406D+04

MAXIMUM ABSOLUTE VALUES

MAX ABS	.228D-01	.911D+00	.364D+02	.768D+06	.406D+04
CORR TIME	.393D-01	.000D+00	.393D-01	.393D-01	.393D-01

TIME HISTORY FOR THE POINT(X= .360D+03)

TIME	PART-DISP	PART-VELO	PART-ACCL	PART-MOMT	PART-SHRF
.000D+00	.000D+00	.223D+01	.000D+00	.000D+00	.000D+00
.393D-02	.872D-02	.220D+01	-.140D+02	-.199D+06	.156D+04
.785D-02	.172D-01	.212D+01	-.276D+02	-.394D+06	.307D+04
.118D-01	.253D-01	.199D+01	-.405D+02	-.579D+06	.452D+04
.157D-01	.328D-01	.180D+01	-.524D+02	-.749D+06	.585D+04
.196D-01	.394D-01	.158D+01	-.631D+02	-.901D+06	.703D+04
.236D-01	.451D-01	.131D+01	-.722D+02	-.103D+07	.805D+04
.275D-01	.497D-01	.101D+01	-.795D+02	-.114D+07	.886D+04
.314D-01	.530D-01	.689D+00	-.848D+02	-.121D+07	.846D+04
.353D-01	.551D-01	.349D+00	-.881D+02	-.126D+07	.983D+04
.393D-01	.557D-01	-.819D-05	-.892D+02	-.127D+07	.995D+04

MAXIMUM ABSOLUTE VALUES

MAX ABS	.557D-01	.223D+01	.892D+02	.127D+07	.995D+04
CORR TIME	.393D-01	.000D+00	.393D-01	.393D-01	.393D-01

TIME HISTORY FOR THE POINT(X= .396D+03)

TIME	PART-DISP	PART-VELO	PART-ACCL	PART-MOMT	PART-SHRF
.000D+00	.000D+00	.322D+01	.000D+00	.000D+00	.000D+00
.393D-02	.126D-01	.318D+01	-.201D+02	-.270D+06	-.316D+04
.785D-02	.249D-01	.306D+01	-.398D+02	-.534D+06	-.625D+04
.118D-01	.365D-01	.287D+01	-.585D+02	-.785D+06	-.918D+04
.157D-01	.473D-01	.261D+01	-.757D+02	-.102D+07	-.119D+05
.196D-01	.569D-01	.228D+01	-.911D+02	-.122D+07	-.143D+05
.236D-01	.651D-01	.189D+01	-.104D+03	-.140D+07	-.164D+05
.275D-01	.717D-01	.146D+01	-.115D+03	-.154D+07	-.180D+05

.314D-01	.786D-01	.995D+00	-.122D+03	-.164D+07	-.192D+05
.353D-01	.795D-01	.504D+00	-.127D+03	-.171D+07	-.200D+05
.393D-01	.805D-01	-.118D-04	-.129D+03	-.173D+07	-.202D+05

MAXIMUM ABSOLUTE VALUES

MAX ABS	.805D-01	.322D+01	.129D+03	.173D+07	.202D+05
CORR TIME	.393D-01	.000D+00	.393D-01	.393D-01	.393D-01

TIME HISTORY FOR THE POINT(X= .432D+03)

TIME	PART-DISP	PART-VELD	PART-ACCL	PART-MOMT	PART-SHRF
.000D+00	.000D+00	.446D+01	.000D+00	.000D+00	.000D+00
.393D-02	.174D-01	.440D+01	-.279D+02	.241D+05	-.751D+04
.785D-02	.344D-01	.424D+01	-.551D+02	.476D+05	-.148D+05
.118D-01	.506D-01	.397D+01	-.810D+02	.699D+05	-.218D+05
.157D-01	.655D-01	.361D+01	-.105D+03	.906D+05	-.282D+05
.196D-01	.788D-01	.315D+01	-.126D+03	.109D+06	-.340D+05
.236D-01	.902D-01	.262D+01	-.144D+03	.125D+06	-.389D+05
.275D-01	.993D-01	.202D+01	-.159D+03	.137D+06	-.428D+05
.314D-01	.106D+00	.138D+01	-.170D+03	.147D+06	-.457D+05
.353D-01	.110D+00	.687D+00	-.176D+03	.152D+06	-.474D+05
.393D-01	.111D+00	-.164D-04	-.178D+03	.154D+06	-.480D+05

MAXIMUM ABSOLUTE VALUES

MAX ABS	.111D+00	.446D+01	.178D+03	.154D+06	.480D+05
CORR TIME	.393D-01	.000D+00	.393D-01	.393D-01	.393D-01

TIME HISTORY FOR THE POINT(X= .504D+03)

TIME	PART-DISP	PART-VELD	PART-ACCL	PART-MOMT	PART-SHRF
.000D+00	.000D+00	.858D+01	.000D+00	.000D+00	.000D+00
.393D-02	.257D-01	.649D+01	-.411D+02	.448D+06	-.388D+04
.785D-02	.508D-01	.625D+01	-.813D+02	.886D+06	-.766D+04
.118D-01	.746D-01	.586D+01	-.119D+03	.130D+07	-.112D+05
.157D-01	.966D-01	.532D+01	-.155D+03	.168D+07	-.146D+05
.196D-01	.116D+00	.465D+01	-.186D+03	.203D+07	-.175D+05
.236D-01	.133D+00	.386D+01	-.213D+03	.232D+07	-.200D+05
.275D-01	.146D+00	.299D+01	-.234D+03	.255D+07	-.221D+05
.314D-01	.156D+00	.203D+01	-.250D+03	.273D+07	-.236D+05
.353D-01	.162D+00	.103D+01	-.260D+03	.283D+07	-.245D+05
.393D-01	.164D+00	-.242D-04	-.263D+03	.287D+07	-.248D+05

MAXIMUM ABSOLUTE VALUES

MAX ABS	.164D+00	.658D+01	.263D+03	.287D+07	.248D+05
CORR TIME	.393D-01	.000D+00	.393D-01	.393D-01	.393D-01

TIME HISTORY FOR THE POINT(X= .576D+03)

TIME	PART-DISP	PART-VELD	PART-ACCL	PART-MOMT	PART-SHRF
.000D+00	.000D+00	.668D+01	.000D+00	.000D+00	.000D+00
.393D-02	.261D-01	.660D+01	-.418D+02	.568D+06	.631D+03
.785D-02	.516D-01	.635D+01	-.826D+02	.112D+07	.125D+04
.118D-01	.758D-01	.595D+01	-.121D+03	.165D+07	.183D+04
.157D-01	.982D-01	.540D+01	-.157D+03	.213D+07	.237D+04
.196D-01	.118D+00	.472D+01	-.189D+03	.257D+07	.285D+04
.236D-01	.135D+00	.393D+01	-.216D+03	.294D+07	.326D+04
.275D-01	.149D+00	.303D+01	-.238D+03	.324D+07	.359D+04
.314D-01	.159D+00	.206D+01	-.254D+03	.345D+07	.384D+04
.353D-01	.165D+00	.105D+01	-.264D+03	.359D+07	.398D+04
.393D-01	.167D+00	-.245D-04	-.267D+03	.363D+07	.403D+04

MAXIMUM ABSOLUTE VALUES

MAX ABS	.167D+00	.668D+01	.267D+03	.363D+07	.403D+04
CORR TIME	.393D-01	.000D+00	.393D-01	.393D-01	.393D-01

TIME HISTORY FOR THE POINT(X= .648D+03)

TIME	PART-DISP	PART-VELD	PART-ACCL	PART-MOMT	PART-SHRF
.000D+00	.000D+00	.421D+01	.000D+00	.000D+00	.000D+00
.393D-02	.165D-01	.415D+01	-.263D+02	.381D+06	.435D+04
.785D-02	.325D-01	.400D+01	-.520D+02	.752D+06	.860D+04
.118D-01	.477D-01	.375D+01	-.764D+02	.110D+07	.126D+05
.157D-01	.618D-01	.340D+01	-.989D+02	.143D+07	.164D+05
.196D-01	.744D-01	.297D+01	-.119D+03	.172D+07	.187D+05
.236D-01	.851D-01	.247D+01	-.136D+03	.197D+07	.225D+05
.275D-01	.937D-01	.191D+01	-.150D+03	.217D+07	.248D+05
.314D-01	.100D+00	.130D+01	-.160D+03	.231D+07	.265D+05
.353D-01	.104D+00	.658D+00	-.166D+03	.240D+07	.275D+05
.393D-01	.105D+00	-.155D-04	-.168D+03	.243D+07	.278D+05

MAXIMUM ABSOLUTE VALUES

MAX ABS	.105D+00	.421D+01	.168D+03	.243D+07	.278D+05
CORR TIME	.393D-01	.000D+00	.393D-01	.393D-01	.393D-01

TIME HISTORY FOR THE POINT(X= .720D+03)

TIME	PART-DISP	PART-VELD	PART-ACCL	PART-MOMT	PART-SHRF
.000D+00	.000D+00	.863D-14	.000D+00	.000D+00	.000D+00
.393D-02	.337D-16	.852D-14	-.540D-13	.647D-09	.580D+04
.785D-02	.666D-16	.820D-14	-.107D-12	.128D-08	.114D+05
.118D-01	.979D-16	.769D-14	-.157D-12	.188D-08	.168D+05
.157D-01	.127D-15	.698D-14	-.203D-12	.243D-08	.218D+05
.196D-01	.152D-15	.610D-14	-.244D-12	.292D-08	.262D+05
.236D-01	.174D-15	.507D-14	-.279D-12	.334D-08	.300D+05
.275D-01	.192D-15	.392D-14	-.307D-12	.368D-08	.330D+05
.314D-01	.205D-15	.267D-14	-.328D-12	.393D-08	.352D+05
.353D-01	.213D-15	.135D-14	-.341D-12	.408D-08	.366D+05
.393D-01	.216D-15	-.317D-19	-.345D-12	.413D-08	.370D+05

MAXIMUM ABSOLUTE VALUES

MAX ABS	.216D-15	.863D-14	.345D-12	.413D-08	.370D+05
CORR TIME	.393D-01	.000D+00	.393D-01	.393D-01	.393D-01
STOP					

B) Sample input and output of the MOM3 Program

i) input

TWO SPAN BRIDGE 33FT-27FT, WF=40 RAD/SEC
 3,2,0,
 5,1000,.00001,
 DIST 1,40.,720.,7.2,
 396.,3000000.,92850.,1.46653,
 324.,3000000.,92850.,1.46653,
 .0805,
 999.,

ii) output

TWO SPAN BRIDGE 33FT-27FT, WF=40 RAD/SEC

BLANK COMMON SIZE 462

DETERMINANT APPROACHING INFINITY FOR FREQ EQUAL TO .27429D+02

DETERMINANT APPROACHING INFINITY FOR FREQ EQUAL TO .40875D+02

DETERMINANT APPROACHING INFINITY FOR FREQ EQUAL TO .10972D+03

DETERMINANT APPROACHING INFINITY FOR FREQ EQUAL TO .16390D+03

DETERMINANT APPROACHING INFINITY FOR FREQ EQUAL TO .24686D+03

FREQ (MODE 1) = .3143D+02 AFTER 21 ITRNS DET = .7750D-06

FREQ (MODE 2) = .5559D+02 AFTER 21 ITRNS DET = .5134D-06

FREQ (MODE 3) = .1218D+03 AFTER 9 ITRNS DET = .4101D-05

FREQ (MODE 4) = .1858D+03 AFTER 7 ITRNS DET = .4475D-05

FREQ (MODE 5) = .2697D+03 AFTER 4 ITRNS DET = .4176D-05

NSPAN = 2
 FORCING FREQUENCY(RAD/SEC) = .4000D+02
 TOTAL SPAN LENGTH = .7200D+03

DITANCE INERVAL ALONG THE SPAN = .7200D+01
 NO. OF POINTS IN TIME = 1

FOR THE SPAN NO. 1

THE SPAN LENGTH IS = .3960D+03
 YOUNGS MODULUS IS = .3000D+07
 THE MOMENT OF INERTIA IS = .9285D+05
 THE MASS PER UNIT LENGTH IS = .1467D+01

FOR THE SPAN NO. 2

THE SPAN LENGTH IS = .3240D+03
 YOUNGS MODULUS IS = .3000D+07
 THE MOMENT OF INERTIA IS = .9285D+05
 THE MASS PER UNIT LENGTH IS = .1467D+01

FOR THE SUPPORT NO. 1

THE AMPLITUDE OF SETTLEMENT IS = .8050D-01

MAXIMUM PARTICULAR RESPONSE ALONG THE SPAN (T= .39270D-01)

DIST -----	PART DISP -----	PART MOM -----	PART SHEAR -----
.00000D+00	.00000D+00	.00000D+00	-.32101D+04
.72000D+01	-.43986D-03	-.23103D+05	-.32063D+04
.14400D+02	-.87541D-03	-.46153D+05	-.31852D+04
.21600D+02	-.13024D-02	-.69097D+05	-.31768D+04
.28800D+02	-.17165D-02	-.91882D+05	-.31513D+04
.36000D+02	-.21135D-02	-.11446D+06	-.31189D+04
.43200D+02	-.24892D-02	-.13678D+06	-.30800D+04
.50400D+02	-.28395D-02	-.15880D+06	-.30349D+04
.57600D+02	-.31602D-02	-.18047D+06	-.29842D+04
.64800D+02	-.34473D-02	-.20176D+06	-.29283D+04
.72000D+02	-.36969D-02	-.22263D+06	-.28679D+04
.79200D+02	-.39051D-02	-.24305D+06	-.28037D+04
.86400D+02	-.40681D-02	-.26299D+06	-.27362D+04
.93600D+02	-.41821D-02	-.28244D+06	-.26665D+04
.10080D+03	-.42435D-02	-.30138D+06	-.25952D+04
.10800D+03	-.42489D-02	-.31981D+06	-.25234D+04
.11520D+03	-.41948D-02	-.33772D+06	-.24520D+04
.12240D+03	-.40778D-02	-.35512D+06	-.23820D+04
.12960D+03	-.38947D-02	-.37203D+06	-.23146D+04
.13680D+03	-.36424D-02	-.38846D+06	-.22508D+04
.14400D+03	-.33178D-02	-.40445D+06	-.21919D+04
.15120D+03	-.29180D-02	-.42004D+06	-.21381D+04
.15840D+03	-.24400D-02	-.43528D+06	-.20938D+04

.16560D+03	-.18809D-02	-.45021D+06	-.20572D+04
.17280D+03	-.12381D-02	-.46492D+06	-.20307D+04
.18000D+03	-.50880D-03	-.47948D+06	-.20158D+04
.18720D+03	.30976D-03	-.49398D+06	-.20140D+04
.19440D+03	.12203D-02	-.50852D+06	-.20268D+04
.20160D+03	.22254D-02	-.52321D+06	-.20558D+04
.20880D+03	.33279D-02	-.53817D+06	-.21025D+04
.21600D+03	.45306D-02	-.55353D+06	-.21688D+04
.22320D+03	.58363D-02	-.56945D+06	-.22562D+04
.23040D+03	.72480D-02	-.58607D+06	-.23666D+04
.23760D+03	.87688D-02	-.60358D+06	-.25017D+04
.24480D+03	.10402D-01	-.62216D+06	-.26635D+04
.25200D+03	.12151D-01	-.64201D+06	-.28538D+04
.25920D+03	.14019D-01	-.66333D+06	-.30747D+04
.26640D+03	.16011D-01	-.68636D+06	-.33282D+04
.27360D+03	.18131D-01	-.71134D+06	-.36164D+04
.28080D+03	.20383D-01	-.73853D+06	-.38416D+04
.28800D+03	.22773D-01	-.76819D+06	-.43059D+04
.29520D+03	.25305D-01	-.80063D+06	-.47119D+04
.30240D+03	.27987D-01	-.83615D+06	-.51618D+04
.30960D+03	.30824D-01	-.87507D+06	-.56584D+04
.31680D+03	.33824D-01	-.91775D+06	-.62042D+04
.32400D+03	.36995D-01	-.96454D+06	-.68022D+04
.33120D+03	.40346D-01	-.10158D+07	-.74553D+04
.33840D+03	.43886D-01	-.10720D+07	-.81665D+04
.34560D+03	.47625D-01	-.11336D+07	-.89393D+04
.35280D+03	.51576D-01	-.12009D+07	-.97769D+04
.36000D+03	.55750D-01	-.12745D+07	-.10683D+05
.36720D+03	.60161D-01	-.13549D+07	-.11662D+05
.37440D+03	.64825D-01	-.14426D+07	-.12717D+05
.38160D+03	.69757D-01	-.15383D+07	-.13854D+05
.38880D+03	.74976D-01	-.16423D+07	-.15076D+05
.39600D+03	.80500D-01	-.17556D+07	-.16389D+05
.40320D+03	.86335D-01	-.13511D+07	.55457D+05
.41040D+03	.92422D-01	-.95722D+06	.53948D+05
.41760D+03	.98686D-01	-.57455D+06	.52333D+05
.42480D+03	.10506D+00	-.20388D+06	.50612D+05
.43200D+03	.11147D+00	.15401D+06	.48783D+05
.43920D+03	.11785D+00	.49834D+06	.46846D+05
.44640D+03	.12414D+00	.82834D+06	.44802D+05
.45360D+03	.13027D+00	.11432D+07	.42653D+05
.46080D+03	.13620D+00	.14423D+07	.40401D+05
.46800D+03	.14185D+00	.17248D+07	.38052D+05
.47520D+03	.14719D+00	.19900D+07	.35610D+05
.48240D+03	.15215D+00	.22374D+07	.33081D+05
.48960D+03	.15670D+00	.24662D+07	.30471D+05
.49680D+03	.16079D+00	.26760D+07	.27789D+05
.50400D+03	.16438D+00	.28662D+07	.25041D+05
.51120D+03	.16744D+00	.30364D+07	.22238D+05
.51840D+03	.16993D+00	.31863D+07	.19387D+05
.52560D+03	.17183D+00	.33155D+07	.16499D+05

.53280D+03	.17311D+00	.34238D+07	.13585D+05
.54000D+03	.17376D+00	.35111D+07	.10654D+05
.54720D+03	.17375D+00	.35772D+07	.77172D+04
.55440D+03	.17308D+00	.36222D+07	.47864D+04
.56160D+03	.17174D+00	.36462D+07	.18727D+04
.56880D+03	.16971D+00	.36493D+07	-.10126D+04
.57600D+03	.16701D+00	.36317D+07	-.38579D+04
.58320D+03	.16363D+00	.35938D+07	-.66519D+04
.59040D+03	.15959D+00	.35361D+07	-.93832D+04
.59760D+03	.15488D+00	.34589D+07	-.12041D+05
.60480D+03	.14954D+00	.33629D+07	-.14613D+05
.61200D+03	.14356D+00	.32487D+07	-.17090D+05
.61920D+03	.13699D+00	.31171D+07	-.19460D+05
.62640D+03	.12983D+00	.29688D+07	-.21715D+05
.63360D+03	.12212D+00	.28047D+07	-.23844D+05
.64080D+03	.11389D+00	.26257D+07	-.25838D+05
.64800D+03	.10517D+00	.24329D+07	-.27689D+05
.65520D+03	.95995D-01	.22274D+07	-.29389D+05
.66240D+03	.86409D-01	.20101D+07	-.30931D+05
.66960D+03	.76448D-01	.17824D+07	-.32307D+05
.67680D+03	.66156D-01	.15453D+07	-.33512D+05
.68400D+03	.55576D-01	.13002D+07	-.34540D+05
.69120D+03	.44754D-01	.10484D+07	-.35388D+05
.69840D+03	.33738D-01	.79106D+06	-.36052D+05
.70560D+03	.22574D-01	.52966D+06	-.36527D+05
.71280D+03	.11312D-01	.26552D+06	-.36814D+05
.72000D+03	.89370D-16	.20981D-08	-.36909D+05

MAX VALUES
AT DIST

.17376D+00	.36493D+07	.55457D+05
.54000D+03	.56880D+03	.40320D+03

STOP

C) Sample input and output of the MOMST Program

i) input

TWO SPAN BRIDGE 33FT-27FT
 2,720.,7.2,
 396.,3000000.,92850.,1.46653,
 324.,3000000.,92850.,1.46653,
 .0805,

ii) output

TWO SPAN BRIDGE 33FT-27FT

BLANK COMMON SIZE 634

NSPAN	=	2
TOTAL SPAN LENGTH	=	.7200D+03
X INTERVAL ALONG THE SPAN	=	.7200D+01

FOR THE SPAN NO. 1

THE SPAN LENGTH IS	=	.3960D+03
YOUNGS MODULUS IS	=	.3000D+07
THE MOMENT OF INERTIA IS	=	.9285D+05
THE MASS PER UNIT LENGTH IS	=	.1467D+01

FOR THE SPAN NO. 2

THE SPAN LENGTH IS	=	.3240D+03
YOUNGS MODULUS IS	=	.3000D+07
THE MOMENT OF INERTIA IS	=	.9285D+05
THE MASS PER UNIT LENGTH IS	=	.1467D+01

FOR THE SUPPORT NO. 1

THE SETTLEMENT IS	=	.8050D-01
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DIST -----	DISP -----	MOM -----	SHEAR -----
.00000E+00	.00000E+00	.00000E+00	.13240E+04
.72000E+01	.23578E-02	.95327E+04	.13240E+04
.14400E+02	.47138E-02	.19065E+05	.13240E+04
.21600E+02	.70663E-02	.28598E+05	.13240E+04
.28800E+02	.94134E-02	.38131E+05	.13240E+04
.36000E+02	.11753E-01	.47864E+05	.13240E+04
.43200E+02	.14085E-01	.57196E+05	.13240E+04
.50400E+02	.16405E-01	.66729E+05	.13240E+04
.57600E+02	.18713E-01	.76262E+05	.13240E+04
.64800E+02	.21007E-01	.85795E+05	.13240E+04
.72000E+02	.23285E-01	.95327E+05	.13240E+04
.79200E+02	.25545E-01	.10486E+06	.13240E+04
.86400E+02	.27786E-01	.11439E+06	.13240E+04
.93600E+02	.30005E-01	.12393E+06	.13240E+04
.10080E+03	.32202E-01	.13346E+06	.13240E+04
.10800E+03	.34373E-01	.14299E+06	.13240E+04
.11520E+03	.36518E-01	.15252E+06	.13240E+04
.12240E+03	.38635E-01	.16206E+06	.13240E+04
.12960E+03	.40721E-01	.17159E+06	.13240E+04
.13680E+03	.42775E-01	.18112E+06	.13240E+04
.14400E+03	.44796E-01	.19065E+06	.13240E+04
.15120E+03	.46781E-01	.20019E+06	.13240E+04
.15840E+03	.48729E-01	.20972E+06	.13240E+04
.16560E+03	.50638E-01	.21925E+06	.13240E+04
.17280E+03	.52506E-01	.22879E+06	.13240E+04
.18000E+03	.54332E-01	.23832E+06	.13240E+04
.18720E+03	.56113E-01	.24785E+06	.13240E+04
.19440E+03	.57848E-01	.25738E+06	.13240E+04
.20160E+03	.59535E-01	.26692E+06	.13240E+04
.20880E+03	.61173E-01	.27645E+06	.13240E+04
.21600E+03	.62759E-01	.28598E+06	.13240E+04
.22320E+03	.64292E-01	.29551E+06	.13240E+04
.23040E+03	.65770E-01	.30505E+06	.13240E+04
.23760E+03	.67191E-01	.31458E+06	.13240E+04
.24480E+03	.68553E-01	.32411E+06	.13240E+04
.25200E+03	.69855E-01	.33365E+06	.13240E+04
.25920E+03	.71095E-01	.34318E+06	.13240E+04
.26640E+03	.72272E-01	.35271E+06	.13240E+04
.27360E+03	.73382E-01	.36224E+06	.13240E+04
.28080E+03	.74425E-01	.37178E+06	.13240E+04
.28800E+03	.75399E-01	.38131E+06	.13240E+04
.29520E+03	.76302E-01	.39084E+06	.13240E+04
.30240E+03	.77133E-01	.40037E+06	.13240E+04
.30960E+03	.77888E-01	.40991E+06	.13240E+04
.31680E+03	.78568E-01	.41944E+06	.13240E+04
.32400E+03	.79169E-01	.42897E+06	.13240E+04
.33120E+03	.79691E-01	.43851E+06	.13240E+04
.33840E+03	.80131E-01	.44804E+06	.13240E+04
.34560E+03	.80488E-01	.45757E+06	.13240E+04

.35280E+03	.80759E-01	.46710E+06	.13240E+04
.36000E+03	.80944E-01	.47664E+06	.13240E+04
.36720E+03	.81039E-01	.48617E+06	.13240E+04
.37440E+03	.81045E-01	.49570E+06	.13240E+04
.38160E+03	.80958E-01	.50523E+06	.13240E+04
.38880E+03	.80777E-01	.51477E+06	.13240E+04
.39600E+03	.80500E-01	.52430E+06	.13240E+04
.40320E+03	.80126E-01	.51265E+06	-.16182E+04
.41040E+03	.79657E-01	.50100E+06	-.16182E+04
.41760E+03	.79095E-01	.48935E+06	-.16182E+04
.42480E+03	.78442E-01	.47770E+06	-.16182E+04
.43200E+03	.77699E-01	.46604E+06	-.16182E+04
.43920E+03	.76870E-01	.45439E+06	-.16182E+04
.44640E+03	.75957E-01	.44274E+06	-.16182E+04
.45360E+03	.74961E-01	.43109E+06	-.16182E+04
.46080E+03	.73884E-01	.41944E+06	-.16182E+04
.46800E+03	.72730E-01	.40779E+06	-.16182E+04
.47520E+03	.71500E-01	.39614E+06	-.16182E+04
.48240E+03	.70196E-01	.38449E+06	-.16182E+04
.48960E+03	.68821E-01	.37284E+06	-.16182E+04
.49680E+03	.67376E-01	.36118E+06	-.16182E+04
.50400E+03	.65864E-01	.34953E+06	-.16182E+04
.51120E+03	.64286E-01	.33788E+06	-.16182E+04
.51840E+03	.62647E-01	.32623E+06	-.16182E+04
.52560E+03	.60946E-01	.31458E+06	-.16182E+04
.53280E+03	.59187E-01	.30293E+06	-.16182E+04
.54000E+03	.57371E-01	.29128E+06	-.16182E+04
.54720E+03	.55501E-01	.27963E+06	-.16182E+04
.55440E+03	.53579E-01	.26798E+06	-.16182E+04
.56160E+03	.51607E-01	.25632E+06	-.16182E+04
.56880E+03	.49588E-01	.24467E+06	-.16182E+04
.57600E+03	.47523E-01	.23302E+06	-.16182E+04
.58320E+03	.45415E-01	.22137E+06	-.16182E+04
.59040E+03	.43265E-01	.20972E+06	-.16182E+04
.59760E+03	.41076E-01	.19807E+06	-.16182E+04
.60480E+03	.38851E-01	.18642E+06	-.16182E+04
.61200E+03	.36591E-01	.17477E+06	-.16182E+04
.61920E+03	.34298E-01	.16312E+06	-.16182E+04
.62640E+03	.31975E-01	.15146E+06	-.16182E+04
.63360E+03	.29624E-01	.13981E+06	-.16182E+04
.64080E+03	.27247E-01	.12816E+06	-.16182E+04
.64800E+03	.24846E-01	.11651E+06	-.16182E+04
.65520E+03	.22423E-01	.10486E+06	-.16182E+04
.66240E+03	.19981E-01	.93209E+05	-.16182E+04
.66960E+03	.17521E-01	.81558E+05	-.16182E+04
.67680E+03	.15046E-01	.69907E+05	-.16182E+04
.68400E+03	.12558E-01	.58256E+05	-.16182E+04
.69120E+03	.10060E-01	.46604E+05	-.16182E+04
.69840E+03	.75524E-02	.34953E+05	-.16182E+04
.70560E+03	.50385E-02	.23302E+05	-.16182E+04
.71280E+03	.25203E-02	.11651E+05	-.16182E+04
.72000E+03	.17875E-16	.11642E-09	-.16182E+04

MAX VALUES
AT DIST

.81045E-01	.52430E+06	-.16182E+04
.37440E+03	.39600E+03	.72000E+03

D) Sample input and output of the SAMIR Program

i) input

THREE SPAN BRIDGE, .8L-L.8L
 3,200,5,5,1,
 2.6,1.,1.,1.,.01,0.,90.,.5,.1,
 1.,1.,
 .8,1.8,
 12.49,19.06,23.47,47.79,67.37,
 1.2038,1.6304,2.038,
 15.,90.,600.,

ii) output

THREE SPAN BRIDGE, .8L-L.8L

BLANK COMMON ARRAY 6036

NUMBER OF SPANS	=	3
TOTAL SPAN LENGTH	=	.2600D+01
MASS DENSITY PER UNIT LENGTH	=	.1000D+01
YOUNG'S MODULUS	=	.1000D+01
MOMENT OF INERTIA	=	.1000D+01
NUMBER OF TERMS TO BE SUMMED	=	200

SUPPORT DISTANCES FROM LEFT ABUTEMENT

SUPPORT NO. 1	=	.8000D+00
SUPPORT NO. 2	=	.1800D+01

SUPPORT SETTLEMENTS

SUPPORT NO. 1	=	.1000D+01
SUPPORT NO. 2	=	.1000D+01

MAXIMUM DYNAMIC AMPLIFICATION FACTORS FOR DEFLECTION FUNCTION

FREQUENCY

.00000E+00	.10000E+01
.50000E+00	.10011E+01
.10000E+01	.10044E+01
.15000E+01	.10099E+01
.20000E+01	.10178E+01
.25000E+01	.10281E+01
.30000E+01	.10410E+01
.35000E+01	.10568E+01
.40000E+01	.10757E+01
.45000E+01	.10982E+01
.50000E+01	.11246E+01
.55000E+01	.11557E+01
.60000E+01	.11923E+01
.65000E+01	.12354E+01
.70000E+01	.12866E+01
.75000E+01	.13480E+01
.80000E+01	.14226E+01
.85000E+01	.15150E+01
.90000E+01	.16321E+01
.95000E+01	.17858E+01
.10000E+02	.19771E+01
.10500E+02	.23082E+01
.11000E+02	.28176E+01
.11500E+02	.38224E+01
.11990E+02	.67097E+01
.12000E+02	.68281E+01
.12090E+02	.81585E+01
.12190E+02	.10566E+02
.12290E+02	.15364E+02
.12390E+02	.29652E+02
.12500E+02	.32122E+03
.12590E+02	.28327E+02
.12690E+02	.13604E+02
.12790E+02	.87250E+01
.12890E+02	.62879E+01
.12990E+02	.48248E+01
.13000E+02	.47100E+01
.13500E+02	.26326E+01
.14000E+02	.21590E+01
.14500E+02	.19481E+01
.15000E+02	.18468E+01
.15500E+02	.18055E+01
.16000E+02	.18027E+01
.16500E+02	.18303E+01
.17000E+02	.18839E+01
.17500E+02	.19644E+01
.18000E+02	.20736E+01
.18500E+02	.22167E+01
.18560E+02	.22366E+01
.18660E+02	.22712E+01
.18760E+02	.23077E+01
.18860E+02	.23461E+01
.18960E+02	.23865E+01
.19000E+02	.24033E+01
.19160E+02	.24739E+01

.19260E+02	.25212E+01
.19360E+02	.25712E+01
.19460E+02	.26243E+01
.19500E+02	.26464E+01
.19540E+02	.26804E+01
.20000E+02	.29693E+01
.20500E+02	.34109E+01
.21000E+02	.40402E+01
.21500E+02	.50026E+01
.22000E+02	.66317E+01
.22500E+02	.99680E+01
.22970E+02	.19259E+02
.23000E+02	.20490E+02
.23070E+02	.24087E+02
.23170E+02	.32180E+02
.23270E+02	.48532E+02
.23370E+02	.98952E+02
.23500E+02	.27902E+03
.23570E+02	.91124E+02
.23670E+02	.46397E+02
.23770E+02	.31098E+02
.23870E+02	.23373E+02
.23970E+02	.18715E+02
.24000E+02	.17658E+02
.24500E+02	.90819E+01
.25000E+02	.61010E+01
.25500E+02	.46083E+01
.26000E+02	.37329E+01
.26500E+02	.31449E+01
.27000E+02	.27224E+01
.27500E+02	.24040E+01
.28000E+02	.21553E+01
.28500E+02	.19556E+01
.29000E+02	.17918E+01
.29500E+02	.16549E+01
.30000E+02	.15388E+01
.30500E+02	.14391E+01
.31000E+02	.13525E+01
.31500E+02	.12767E+01
.32000E+02	.12096E+01
.32500E+02	.11500E+01
.33000E+02	.10965E+01
.33500E+02	.10484E+01
.34000E+02	.10048E+01
.34500E+02	.96517E+00
.35000E+02	.92893E+00
.35500E+02	.89569E+00
.36000E+02	.86508E+00
.36500E+02	.83680E+00
.37000E+02	.83741E+00
.37500E+02	.83802E+00
.38000E+02	.83862E+00
.38500E+02	.83950E+00
.39000E+02	.84055E+00
.39500E+02	.84161E+00
.40000E+02	.84268E+00
.40500E+02	.84377E+00

.41000E+02	.84503E+00
.41500E+02	.84668E+00
.42000E+02	.84837E+00
.42500E+02	.85010E+00
.43000E+02	.85198E+00
.43500E+02	.85441E+00
.44000E+02	.85692E+00
.44500E+02	.85950E+00
.45000E+02	.86270E+00
.45500E+02	.86618E+00
.46000E+02	.86978E+00
.46500E+02	.87413E+00
.47000E+02	.87883E+00
.47290E+02	.88164E+00
.47390E+02	.88262E+00
.47490E+02	.88363E+00
.47500E+02	.88375E+00
.47590E+02	.88481E+00
.47690E+02	.88600E+00
.47890E+02	.88840E+00
.47990E+02	.88962E+00
.48000E+02	.88974E+00
.48090E+02	.89084E+00
.48190E+02	.89208E+00
.48290E+02	.89333E+00
.48500E+02	.89599E+00
.49000E+02	.90298E+00
.49500E+02	.91083E+00
.50000E+02	.91906E+00
.50500E+02	.92879E+00
.51000E+02	.93905E+00
.51500E+02	.95053E+00
.52000E+02	.96320E+00
.52500E+02	.97680E+00
.53000E+02	.99238E+00
.53500E+02	.10090E+01
.54000E+02	.10277E+01
.54500E+02	.10482E+01
.55000E+02	.10705E+01
.55500E+02	.10958E+01
.56000E+02	.11233E+01
.56500E+02	.11542E+01
.57000E+02	.11885E+01
.57500E+02	.12263E+01
.58000E+02	.12697E+01
.58500E+02	.13179E+01
.59000E+02	.13723E+01
.59500E+02	.14349E+01
.60000E+02	.15057E+01
.60500E+02	.15879E+01
.61000E+02	.16839E+01
.61500E+02	.17960E+01
.62000E+02	.19306E+01
.62500E+02	.20935E+01
.63000E+02	.22933E+01
.63500E+02	.25467E+01
.64000E+02	.28764E+01

.64500E+02	.33201E+01
.65000E+02	.39526E+01
.65500E+02	.49266E+01
.66000E+02	.66083E+01
.66500E+02	.10216E+02
.66870E+02	.17526E+02
.66970E+02	.21808E+02
.67000E+02	.23541E+02
.67070E+02	.28922E+02
.67170E+02	.43062E+02
.67270E+02	.84807E+02
.67470E+02	.88453E+02
.67500E+02	.67579E+02
.67570E+02	.43527E+02
.67670E+02	.28798E+02
.67770E+02	.21478E+02
.67870E+02	.17100E+02
.68000E+02	.13495E+02
.68500E+02	.73820E+01
.69000E+02	.50267E+01
.69500E+02	.37769E+01
.70000E+02	.30015E+01
.70500E+02	.24771E+01
.71000E+02	.20965E+01
.71500E+02	.18072E+01
.72000E+02	.17189E+01
.72500E+02	.16604E+01
.73000E+02	.16143E+01
.73500E+02	.15776E+01
.74000E+02	.15483E+01
.74500E+02	.15249E+01
.75000E+02	.15064E+01
.75500E+02	.14918E+01
.76000E+02	.14807E+01
.76500E+02	.14725E+01
.77000E+02	.14669E+01
.77500E+02	.14635E+01
.78000E+02	.14621E+01
.78500E+02	.14626E+01
.79000E+02	.14649E+01
.79500E+02	.14687E+01
.80000E+02	.14748E+01
.80500E+02	.14824E+01
.81000E+02	.14915E+01
.81500E+02	.15019E+01
.82000E+02	.15137E+01
.82500E+02	.15268E+01
.83000E+02	.15414E+01
.83500E+02	.15574E+01
.84000E+02	.15748E+01
.84500E+02	.15937E+01
.85000E+02	.16151E+01
.85500E+02	.16383E+01
.86000E+02	.16633E+01
.86500E+02	.16901E+01
.87000E+02	.17189E+01
.87500E+02	.17498E+01

.88000E+02	.17829E+01
.88500E+02	.18185E+01
.89000E+02	.18567E+01
.89500E+02	.18987E+01
.90000E+02	.19444E+01

MAXIMUM DYNAMIC AMPLIFICATION FACTORS FOR MOMENT FUNCTION

FREQUENCY

.00000E+00	.10005E+01
.50000E+00	.10103E+01
.10000E+01	.10411E+01
.15000E+01	.10930E+01
.20000E+01	.11667E+01
.25000E+01	.12633E+01
.30000E+01	.13842E+01
.35000E+01	.15314E+01
.40000E+01	.17073E+01
.45000E+01	.19153E+01
.50000E+01	.21594E+01
.55000E+01	.24450E+01
.60000E+01	.27791E+01
.65000E+01	.31709E+01
.70000E+01	.36330E+01
.75000E+01	.41829E+01
.80000E+01	.48455E+01
.85000E+01	.56579E+01
.90000E+01	.66780E+01
.95000E+01	.80017E+01
.10000E+02	.98007E+01
.10500E+02	.12418E+02
.11000E+02	.16651E+02
.11500E+02	.24909E+02
.11990E+02	.48427E+02
.12000E+02	.49389E+02
.12090E+02	.60190E+02
.12190E+02	.79723E+02
.12290E+02	.11862E+03
.12390E+02	.23438E+03
.12500E+02	.26077E+04
.12590E+02	.23521E+03
.12690E+02	.11591E+03
.12790E+02	.76355E+02
.12890E+02	.56581E+02
.12990E+02	.44695E+02
.13000E+02	.43761E+02
.13500E+02	.24695E+02
.14000E+02	.19774E+02
.14500E+02	.17748E+02
.15000E+02	.16942E+02
.15500E+02	.16820E+02
.16000E+02	.17160E+02
.16500E+02	.17877E+02

.17000E+02	.18935E+02
.17500E+02	.20362E+02
.18000E+02	.23005E+02
.18500E+02	.26969E+02
.18560E+02	.27495E+02
.18660E+02	.28399E+02
.18760E+02	.29338E+02
.18860E+02	.30317E+02
.18960E+02	.31336E+02
.19000E+02	.31756E+02
.19160E+02	.33511E+02
.19260E+02	.34672E+02
.19360E+02	.35888E+02
.19460E+02	.37163E+02
.19500E+02	.37690E+02
.19560E+02	.38500E+02
.20000E+02	.45275E+02
.20500E+02	.55354E+02
.21000E+02	.69448E+02
.21500E+02	.90627E+02
.22000E+02	.12614E+03
.22500E+02	.19826E+03
.22970E+02	.39835E+03
.23000E+02	.42481E+03
.23070E+02	.50215E+03
.23170E+02	.67606E+03
.23270E+02	.10274E+04
.23370E+02	.21105E+04
.23500E+02	.60086E+04
.23570E+02	.19724E+04
.23670E+02	.10115E+04
.23770E+02	.68282E+03
.23870E+02	.51682E+03
.23970E+02	.41668E+03
.24000E+02	.39395E+03
.24500E+02	.20921E+03
.25000E+02	.14479E+03
.25500E+02	.11196E+03
.26000E+02	.92016E+02
.26500E+02	.78584E+02
.27000E+02	.68898E+02
.27500E+02	.61564E+02
.28000E+02	.55802E+02
.28500E+02	.51142E+02
.29000E+02	.47285E+02
.29500E+02	.44028E+02
.30000E+02	.41233E+02
.30500E+02	.38801E+02
.31000E+02	.36657E+02
.31500E+02	.34746E+02
.32000E+02	.33027E+02
.32500E+02	.31466E+02
.33000E+02	.30038E+02
.33500E+02	.28721E+02
.34000E+02	.27498E+02
.34500E+02	.26356E+02
.35000E+02	.25283E+02

.35500E+02	.24280E+02
.36000E+02	.23374E+02
.36500E+02	.22552E+02
.37000E+02	.22375E+02
.37500E+02	.22232E+02
.38000E+02	.22122E+02
.38500E+02	.22052E+02
.39000E+02	.22012E+02
.39500E+02	.21999E+02
.40000E+02	.22011E+02
.40500E+02	.22048E+02
.41000E+02	.22108E+02
.41500E+02	.22191E+02
.42000E+02	.22309E+02
.42500E+02	.22450E+02
.43000E+02	.22614E+02
.43500E+02	.22801E+02
.44000E+02	.23010E+02
.44500E+02	.23243E+02
.45000E+02	.23499E+02
.45500E+02	.23788E+02
.46000E+02	.24108E+02
.46500E+02	.24455E+02
.47000E+02	.24829E+02
.47290E+02	.25059E+02
.47390E+02	.25141E+02
.47490E+02	.25224E+02
.47500E+02	.25232E+02
.47590E+02	.25308E+02
.47690E+02	.25393E+02
.47890E+02	.25568E+02
.47990E+02	.25657E+02
.48000E+02	.25666E+02
.48090E+02	.25748E+02
.48190E+02	.25839E+02
.48290E+02	.25932E+02
.48500E+02	.26132E+02
.49000E+02	.26633E+02
.49500E+02	.27176E+02
.50000E+02	.27768E+02
.50500E+02	.28403E+02
.51000E+02	.29083E+02
.51500E+02	.29814E+02
.52000E+02	.30600E+02
.52500E+02	.31444E+02
.53000E+02	.32354E+02
.53500E+02	.33335E+02
.54000E+02	.34395E+02
.54500E+02	.35561E+02
.55000E+02	.36828E+02
.55500E+02	.38206E+02
.56000E+02	.39710E+02
.56500E+02	.41357E+02
.57000E+02	.43166E+02
.57500E+02	.45163E+02
.58000E+02	.47375E+02
.58500E+02	.49839E+02

.59000E+02	.52601E+02
.59500E+02	.55741E+02
.60000E+02	.59311E+02
.60500E+02	.63402E+02
.61000E+02	.68134E+02
.61500E+02	.73672E+02
.62000E+02	.80237E+02
.62500E+02	.88147E+02
.63000E+02	.97860E+02
.63500E+02	.11007E+03
.64000E+02	.12590E+03
.64500E+02	.14722E+03
.65000E+02	.17751E+03
.65500E+02	.22402E+03
.66000E+02	.30444E+03
.66500E+02	.47694E+03
.66870E+02	.82554E+03
.66970E+02	.10298E+04
.67000E+02	.11124E+04
.67070E+02	.13690E+04
.67170E+02	.20434E+04
.67270E+02	.40344E+04
.67470E+02	.42291E+04
.67500E+02	.32336E+04
.67570E+02	.20864E+04
.67670E+02	.13839E+04
.67770E+02	.10348E+04
.67870E+02	.82605E+03
.68000E+02	.65411E+03
.68500E+02	.36222E+03
.69000E+02	.24961E+03
.69500E+02	.18988E+03
.70000E+02	.15283E+03
.70500E+02	.12758E+03
.71000E+02	.10926E+03
.71500E+02	.95329E+02
.72000E+02	.90381E+02
.72500E+02	.86975E+02
.73000E+02	.84335E+02
.73500E+02	.82281E+02
.74000E+02	.80687E+02
.74500E+02	.79465E+02
.75000E+02	.78547E+02
.75500E+02	.77886E+02
.76000E+02	.77444E+02
.76500E+02	.77191E+02
.77000E+02	.77107E+02
.77500E+02	.77174E+02
.78000E+02	.77377E+02
.78500E+02	.77708E+02
.79000E+02	.78156E+02
.79500E+02	.78718E+02
.80000E+02	.79387E+02
.80500E+02	.80160E+02
.81000E+02	.81037E+02
.81500E+02	.82015E+02
.82000E+02	.83095E+02

.82500E+02	.84278E+02
.83000E+02	.85565E+02
.83500E+02	.86959E+02
.84000E+02	.88464E+02
.84500E+02	.90085E+02
.85000E+02	.91825E+02
.85500E+02	.93692E+02
.86000E+02	.95692E+02
.86500E+02	.97834E+02
.87000E+02	.10013E+03
.87500E+02	.10258E+03
.88000E+02	.10521E+03
.88500E+02	.10803E+03
.89000E+02	.11105E+03
.89500E+02	.11429E+03
.90000E+02	.11778E+03

MAXIMUM DYNAMIC AMPLIFICATION FACTORS FOR SHEAR FUNCTION

FREQUENCY

.00000E+00	.10584E+01
.50000E+00	.10065E+01
.10000E+01	.10246E+01
.15000E+01	.10547E+01
.20000E+01	.11610E+01
.25000E+01	.18602E+01
.30000E+01	.27271E+01
.35000E+01	.37696E+01
.40000E+01	.49984E+01
.45000E+01	.64266E+01
.50000E+01	.80711E+01
.55000E+01	.99536E+01
.60000E+01	.12102E+02
.65000E+01	.14553E+02
.70000E+01	.17356E+02
.75000E+01	.20577E+02
.80000E+01	.24314E+02
.85000E+01	.28705E+02
.90000E+01	.33969E+02
.95000E+01	.40463E+02
.10000E+02	.48823E+02
.10500E+02	.60310E+02
.11000E+02	.77859E+02
.11500E+02	.11029E+03
.11990E+02	.19870E+03
.12000E+02	.20227E+03
.12090E+02	.24225E+03
.12190E+02	.31425E+03
.12290E+02	.45712E+03
.12390E+02	.88133E+03
.12500E+02	.95204E+04
.12590E+02	.83632E+03
.12690E+02	.39915E+03

.12790E+02	.25382E+03
.12890E+02	.18088E+03
.12990E+02	.13734E+03
.13000E+02	.13396E+03
.13500E+02	.83851E+02
.14000E+02	.81431E+02
.14500E+02	.83790E+02
.15000E+02	.88535E+02
.15500E+02	.94954E+02
.16000E+02	.10284E+03
.16500E+02	.11222E+03
.17000E+02	.12325E+03
.17500E+02	.13623E+03
.18000E+02	.15159E+03
.18500E+02	.16997E+03
.18560E+02	.17243E+03
.18660E+02	.17664E+03
.18760E+02	.18103E+03
.18860E+02	.18561E+03
.18960E+02	.19038E+03
.19000E+02	.19234E+03
.19160E+02	.20056E+03
.19260E+02	.20600E+03
.19360E+02	.21170E+03
.19460E+02	.21767E+03
.19500E+02	.22014E+03
.19560E+02	.22394E+03
.20000E+02	.25567E+03
.20500E+02	.30277E+03
.21000E+02	.36843E+03
.21500E+02	.46674E+03
.22000E+02	.63102E+03
.22500E+02	.96357E+03
.22970E+02	.18841E+04
.23000E+02	.20058E+04
.23070E+02	.23613E+04
.23170E+02	.31607E+04
.23270E+02	.47753E+04
.23370E+02	.97526E+04
.23500E+02	.27554E+05
.23570E+02	.90078E+04
.23670E+02	.45924E+04
.23770E+02	.30817E+04
.23870E+02	.23187E+04
.23970E+02	.18583E+04
.24000E+02	.17538E+04
.24500E+02	.90346E+03
.25000E+02	.60597E+03
.25500E+02	.45355E+03
.26000E+02	.36029E+03
.26500E+02	.29691E+03
.27000E+02	.25811E+03
.27500E+02	.22967E+03
.28000E+02	.20980E+03
.28500E+02	.19379E+03
.29000E+02	.18061E+03
.29500E+02	.16954E+03

.30000E+02	.16010E+03
.30500E+02	.15196E+03
.31000E+02	.14537E+03
.31500E+02	.13993E+03
.32000E+02	.13513E+03
.32500E+02	.13085E+03
.33000E+02	.12702E+03
.33500E+02	.12356E+03
.34000E+02	.12042E+03
.34500E+02	.11756E+03
.35000E+02	.11494E+03
.35500E+02	.11263E+03
.36000E+02	.11074E+03
.36500E+02	.10900E+03
.37000E+02	.10882E+03
.37500E+02	.10881E+03
.38000E+02	.10896E+03
.38500E+02	.10926E+03
.39000E+02	.10970E+03
.39500E+02	.11028E+03
.40000E+02	.11100E+03
.40500E+02	.11185E+03
.41000E+02	.11283E+03
.41500E+02	.11394E+03
.42000E+02	.11518E+03
.42500E+02	.11655E+03
.43000E+02	.11806E+03
.43500E+02	.11970E+03
.44000E+02	.12148E+03
.44500E+02	.12341E+03
.45000E+02	.12636E+03
.45500E+02	.13423E+03
.46000E+02	.14228E+03
.46500E+02	.15052E+03
.47000E+02	.15897E+03
.47290E+02	.16398E+03
.47390E+02	.16572E+03
.47490E+02	.16747E+03
.47500E+02	.16765E+03
.47590E+02	.16924E+03
.47690E+02	.17101E+03
.47890E+02	.17459E+03
.47990E+02	.17639E+03
.48000E+02	.17657E+03
.48090E+02	.17821E+03
.48190E+02	.18003E+03
.48290E+02	.18187E+03
.48500E+02	.18576E+03
.49000E+02	.19524E+03
.49500E+02	.20503E+03
.50000E+02	.21516E+03
.50500E+02	.22566E+03
.51000E+02	.23656E+03
.51500E+02	.24790E+03
.52000E+02	.25972E+03
.52500E+02	.27206E+03
.53000E+02	.28498E+03

.53500E+02	.29854E+03
.54000E+02	.31281E+03
.54500E+02	.32786E+03
.55000E+02	.34378E+03
.55500E+02	.36069E+03
.56000E+02	.37870E+03
.56500E+02	.39797E+03
.57000E+02	.41866E+03
.57500E+02	.44099E+03
.58000E+02	.46522E+03
.58500E+02	.49165E+03
.59000E+02	.52068E+03
.59500E+02	.55278E+03
.60000E+02	.58857E+03
.60500E+02	.62885E+03
.61000E+02	.67465E+03
.61500E+02	.72738E+03
.62000E+02	.78894E+03
.62500E+02	.86204E+03
.63000E+02	.95062E+03
.63500E+02	.10606E+04
.64000E+02	.12016E+04
.64500E+02	.13897E+04
.65000E+02	.16545E+04
.65500E+02	.20575E+04
.66000E+02	.27498E+04
.66500E+02	.42285E+04
.66870E+02	.72092E+04
.66970E+02	.89541E+04
.67000E+02	.96601E+04
.67070E+02	.11852E+05
.67170E+02	.17611E+05
.67270E+02	.34612E+05
.67470E+02	.35942E+05
.67500E+02	.27441E+05
.67570E+02	.17646E+05
.67670E+02	.11646E+05
.67770E+02	.86646E+04
.67870E+02	.68808E+04
.68000E+02	.54112E+04
.68500E+02	.29131E+04
.69000E+02	.19456E+04
.69500E+02	.14295E+04
.70000E+02	.11202E+04
.70500E+02	.91114E+03
.71000E+02	.75801E+03
.71500E+02	.64639E+03
.72000E+02	.61133E+03
.72500E+02	.58861E+03
.73000E+02	.57120E+03
.73500E+02	.57958E+03
.74000E+02	.59186E+03
.74500E+02	.60476E+03
.75000E+02	.61822E+03
.75500E+02	.63224E+03
.76000E+02	.64681E+03
.76500E+02	.66192E+03

.77000E+02	.67758E+03
.77500E+02	.69380E+03
.78000E+02	.71060E+03
.78500E+02	.72799E+03
.79000E+02	.74599E+03
.79500E+02	.76462E+03
.80000E+02	.78393E+03
.80500E+02	.80393E+03
.81000E+02	.82467E+03
.81500E+02	.84618E+03
.82000E+02	.86851E+03
.82500E+02	.89171E+03
.83000E+02	.91583E+03
.83500E+02	.94093E+03
.84000E+02	.96707E+03
.84500E+02	.99433E+03
.85000E+02	.10228E+04
.85500E+02	.10525E+04
.86000E+02	.10836E+04
.86500E+02	.11163E+04
.87000E+02	.11505E+04
.87500E+02	.11864E+04
.88000E+02	.12243E+04
.88500E+02	.12642E+04
.89000E+02	.13064E+04
.89500E+02	.13511E+04
.90000E+02	.13985E+04

APPENDIX II


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C DYNAMIC
C COMPUTER PROGRAM TO INVESTIGATE THE DYNAMIC RESPONSE
C OF MULTISPAN GIRDERS SUBJECTED TO PULSATING SUPPORT
C SETTLEMENTS INDUCED BY VERTICAL EARTHQUAKE MOTION
C *****
C PROGRAMMED BY DR. C. YOO
C AUGUST 1979
C ON XEROX SIGMA 9, MARQUETTE UNIVERSITY
C *****
C INPUT VARIABLES
C HEAD=HEADING (UP TO 80 ALPHANUMERIC CHARACTER)
C NFRQ=NUMBER OF MODES IN FREE VIBRATION CONSIDERED
C IMAX=MAXIMUM NUMBER OF ITERATIONS IN NATURAL FREQUENCY ROUTINE
C EPS1=TOLERANCE IN NATURAL FREQUENCY ROUTINE
C NSPAN=NUMBER OF SPANS (2 OR MORE)
C NTERM=NUMBER OF TERMS TO BE SUMMED FOR INFINITE
C SERIES (3 OR MORE)
C NPOINT=NUMBER OF LOCATIONS ALONG THE GIRDER LENGTH
C FOR WHICH THE TIME HISTORY IS SOUGHT
C IOPT=1: PARTICULAR SOLUTION
C =2: COMPUTATION OF NATURAL CIRCULAR FREQUENCIES ONLY
C IPLT=0 IF NO PLOT DATA FILE IS DESIRED
C =1 OR ANY INTEGER IF A PLOT DATA FILE IS DESIRED
C IEQSP=0: THERE ARE NO EQUAL SPANS
C =1 OR ANY INTEGER IF THERE ARE EQUAL SPANS
C TL=TOTAL SPAN LENGTH
C SM=MASS DENSITY PER UNIT LENGTH OF THE GIRDER
C E=YOUNG'S MODULUS
C XI=MOMENT OF INERTIA OF THE GIRDER
C DT=DURATION OF EARTHQUAKE
C TINT=TIME INCREMENT FOR THE TIME HISTORY
C WF=ANGULAR (CIRCULAR) VELOCITY OF EARTHQUAKE (RAD/SEC)
C DEL(I)=MAXIMUM AMPLITUDE OF SUPPORT SETTLEMENTS RELATIVE
C TO BOTH END ABUTMENTS
C XL(I)=X-COORDINATE OF INTERNAL SUPPORT
C X(I)=X-COORDINATE ON THE GIRDER FOR WHICH THE TIME
C HISTORY IS SOUGHT
C ..... X
C .
C .
C .
C .
C .
C Y
C INPUT SEQUENCE
C NO CARDS VARIABLES FORMAT
C 1 HEAD 20A4
C 1 NFRQ,NSPAN,NTERM,NPOINT,IOPT,IPLT, 7I5,7G10.0
C IEQSP,TL,SM,E,XI,DT,TINT,WF
C 1 DEL(I) 10G10.0
C 1 XL(I) 10G10.0
C IF(IOPT.EQ.1) SKIP THE FOLLOWING CARD
C 1 X(I) 10G10.0
C IF(IOPT.EQ.1) SKIP THE FOLLOWING CARD
C 1 IMAX,EPS1 1I0,G10.0
C
C IMPLICIT DOUBLE PRECISION(A-H,Q-Z)
C REAL HEAD(20),AA(10000)
C DIMENSION A(5000),LKI(10000)
C COMMON AA
C EQUIVALENCE (AA(1),A(1)),(AA(1),LKI(1))
C COMMON/SETUP/TL,SM,E,XI,DT,TINT,WF,EPS1,PAI,EX,CX,NSPAN,NTERM,
C 1 NPOINT,NS1,NT,IN,IG,IO2,IOPT,IPLT,IMAX,NFRQ,NTERM1,IEQSP
C IN=5
C IO=6
C IO2=3
C IDS=2
C DYNAA0001
C DYNAA0002
C DYNAA0003
C DYNAA0004
C DYNAA0005
C DYNAA0006
C DYNAA0007
C DYNAA0008
C DYNAA0009
C DYNAA0010
C DYNAA0011
C DYNAA0012
C DYNAA0013
C DYNAA0014
C DYNAA0015
C DYNAA0016
C DYNAA0017
C DYNAA0018
C DYNAA0019
C DYNAA0020
C DYNAA0021
C DYNAA0022
C DYNAA0023
C DYNAA0024
C DYNAA0025
C DYNAA0026
C DYNAA0027
C DYNAA0028
C DYNAA0029
C DYNAA0030
C DYNAA0031
C DYNAA0032
C DYNAA0033
C DYNAA0034
C DYNAA0035
C DYNAA0036
C DYNAA0037
C DYNAA0038
C DYNAA0039
C DYNAA0040
C DYNAA0041
C DYNAA0042
C DYNAA0043
C DYNAA0044
C DYNAA0045
C DYNAA0046
C DYNAA0047
C DYNAA0048
C DYNAA0049
C DYNAA0050
C DYNAA0051
C DYNAA0052
C DYNAA0053
C DYNAA0054
C DYNAA0055
C DYNAA0056
C DYNAA0057
C DYNAA0058
C DYNAA0059
C DYNAA0060
C DYNAA0061
C DYNAA0062
C DYNAA0063
C DYNAA0064
C DYNAA0065
C DYNAA0066
C DYNAA0067
C DYNAA0068
C DYNAA0069
C DYNAA0070
C DYNAA0071
C DYNAA0072
C DYNAA0073
C DYNAA0074
C DYNAA0075

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```

998 READ(IN,500,END=999) HEAD
500 FORMAT(20A4)
WRITE(ID,600) HEAD
600 FORMAT(1H1,20A4)
READ(IN,501) NFRQ,NSPAN,NTERM,NPOINT,IOPT,IPLLOT,IEGSP,TL,SM,E,
1 XI,DT,TINT,WF
501 FORMAT(7I5,7G10.0)
NS1=NSPAN-1
NTERM1=NTERM+20
C
NT=DT/TINT+1
IF(NT.LT.2) NT=2
IF(NS1.EQ.0) GO TO 998
N2=1+NS1
N3=N2+NS1
N4=N3+NPOINT
N5=N4+NT
N6=N5+NT
N7=N6+NT
N8=N7+NT
N9=N8+NT
N10=N9+NT
N11=N10+NTERM
N12=N11+NTERM
N13=N12+NS1*NTERM1
N14=N13+NTERM
N15=N14+NTERM1
N16=N15+NFRQ
N17=N16+NFRQ
N18=N17+NS1
N19=N18+NS1*NS1
N20=(N19+NS1*2)*IDS
NSIZE=N20+NFRQ
IF(NSIZE.GT.10000) GO TO 997
WRITE(ID,601) NSIZE
601 FORMAT(///', 'BLANK COMMON ARRAY',I7)
CALL MAIN(A(1),A(N2),A(N3),A(N4),A(N5),A(N6),A(N7),
1 A(N8),A(N9),A(N10),A(N11),A(N12),A(N13),A(N14),A(N15),
2 A(N16),A(N17),A(N18),A(N19),LKI(N20),HEAD)
GO TO 998
997 WRITE(ID,602) NSIZE
602 FORMAT(///', 'YOUR BLANK COMMON ARRAY MUST BE',I7)
GO TO 998
998 WRITE(ID,603) NSPAN
603 FORMAT(///', 'THIS IS NOT A MULTISPAN GIRDER',I5)
GO TO 998
999 STOP
END
SUBROUTINE MAIN(DEL,XL,X,PY,PYT,PM,PG,TT,XTC,XT,
1 XN,DN,P,FRQ,DET,RP,RCOEF,WKAREA,ITN,HEAD)
IMPLICIT DOUBLE PRECISION(A-H,O-Z)
REAL HEAD(20)
DIMENSION DEL(NS1),XL(NS1),X(NPOINT),PY(NT),PYT(NT),PYT2(NT),
1 PM(NT),PG(NT),TT(NT),XTC(NTERM),XT(NTERM),XN(NS1,NTERM1),
2 DN(NTERM),P(NTERM1),FRQ(NFRQ),DET(NFRQ),RP(NS1),RCOEF(NS1,NS1),
3 WKAREA(1),ITN(NFRQ)
COMMON/SETUP/TL,SM,E,XI,DT,TINT,WF,EPS1,PAI,EX,CX,NSPAN,NTERM,
1 NPOINT,NS1,NT,IN,IO,IOZ,IOPT,IPLLOT,IMAX,NFRQ,NTERM1,IEGSP
C
COMMON FACTORS
EX=E*XI/(SM*TL**4)
WF2=WF**2
PAI=DATAN(1.00)*4.
CX=DSQRT(2./(SM*TL))
GO TO (201,202,203),IOPT
201 WRITE(ID,701)
701 FORMAT(///', 'PARTICULAR SOLUTION'///)
GO TO 205
203 WRITE(ID,702)
702 FORMAT(///', 'NATURAL FREQUENCIES AND PARTICULAR SOLUTION'///)
GO TO 205
202 WRITE(ID,703)
703 FORMAT(///', 'COMPUTATION OF NATURAL CIRCULAR FREQUENCIES ONLY'///)
205 READ(IN,500) (DEL(I),I=1,NS1)
DYNAA0078
DYNAA0077
DYNAA0078
DYNAA0079
DYNAA0080
DYNAA0081
DYNAA0082
DYNAA0083
DYNAA0084
DYNAA0085
DYNAA0086
DYNAA0087
DYNAA0088
DYNAA0089
DYNAA0090
DYNAA0091
DYNAA0092
DYNAA0093
DYNAA0094
DYNAA0095
DYNAA0096
DYNAA0097
DYNAA0098
DYNAA0099
DYNAA0100
DYNAA0101
DYNAA0102
DYNAA0103
DYNAA0104
DYNAA0105
DYNAA0106
DYNAA0107
DYNAA0108
DYNAA0109
DYNAA0110
DYNAA0111
DYNAA0112
DYNAA0113
DYNAA0114
DYNAA0115
DYNAA0116
DYNAA0117
DYNAA0118
DYNAA0119
DYNAA0120
DYNAA0121
DYNAA0122
DYNAA0123
DYNAA0124
DYNAA0125
DYNAA0126
DYNAA0127
DYNAA0128
DYNAA0129
DYNAA0130
DYNAA0131
DYNAA0132
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DYNAA0144
DYNAA0145
DYNAA0146
DYNAA0147
DYNAA0148
DYNAA0149
DYNAA0150

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500 FORMAT(10G10.0)
    READ(IN,500) (XL(I),I=1,NS1)
    IF(IOPT.EQ.2) GO TO 704
    READ(IN,500) (X(I),I=1,NPOINT)
    IF(IOPT.EQ.1) GO TO 705
704 READ(IN,501) IMAX,EP51
501 FORMAT(I10,G10.0)
705 DO 215 N=1,NTERM1
    PAIN=N*PAI
    PAINTL=PAIN/TL
    P(N)=DSQRT(PAIN**4*EX)
    DO 215 I=1,NS1
215 XN(I,N)=CX*DSIN(PAINTL*XL(I))
    IF(IOPT.EQ.1) GO TO 208
    CALL FREQ(P,XN,FRQ,DET,ITN,RCOEF,WKAREA)
    IF(IOPT.EQ.2) GO TO 59

C
C
C
ECHO PRINTING OF PARAMETERS

208 WRITE(IO,699) TL,SM,E,XI,DT,TINT,WF,NSPAN,NFRQ,NTERM,NPOINT
699 FORMAT(//'0','TOTAL SPAN LENGTH           =',D10.4/
1      ' ', 'MASS DENSITY PER UNIT LENGTH      =',D10.4/
2      ' ', 'YOUNG'S MODULUS                   =',D10.4/
3      ' ', 'MOMENT OF INERTIA                 =',D10.4/
4      ' ', 'DURATION OF EARTHQUAKE            =',D10.4/
5      ' ', 'TIME INCREMENT IN TIME HISTORY     =',D10.4/
6      ' ', 'EARTHQUAKE (ANGULAR) VELOCITY     =',D10.4/
8      ' ', 'NUMBER OF SPANS                   =',I10/
8      ' ', 'NUMBER OF MODES IN FREE VIBRN      =',I10/
9      ' ', 'NUMBER OF TERMS TO BE SUMMED      =',I10/
A      ' ', 'NUMBER OF LOCATIONS (T HISTORY)    =',I10)
    DO 150 I1=1,NS1
    DO 150 I2=1,NS1
150 RCOEF(I1,I2)=0.
    DO 100 N=1,NTERM
    PAIN=PAI*N
    PAINTL=PAIN/TL
    DEN=P(N)**2-WF2
    ADEN=DABS(DEN)
    IF(ADEN.LT..0001) GO TO 100
    DO 100 I1=1,NS1
    DO 100 I2=1,NS1
    RCOEF(I1,I2)=RCOEF(I1,I2)+XN(I1,N)*XN(I2,N)/DEN
100 CONTINUE
    DO 105 I=1,NS1
105 RP(I)=DEL(I)
    D1=-1.
    CALL LINV3F(RCOEF,RP,2,NS1,NS1,D1,D2,WKAREA,IER)
    DO 50 NP=1,NPOINT
    DO 51 K=1,NT
    PY(K)=0.
    PYT(K)=0.
    PYT2(K)=0.
    PM(K)=0.
    PG(K)=0.
211 T=(K-1)*TINT
    TT(K)=T
    DO 45 N=1,NTERM
    SWT=WF*T
    SWTSIN=DSIN(SWT)
    SWTCOS=DCOS(SWT)
    PAIN=N*PAI
    PAINTL=PAIN/TL
    DEN=P(N)**2-WF2
    ADEN=DABS(DEN)
    IF(ADEN.LT..0001) GO TO 45
    DN(N)=0.
    DO 44 I=1,NS1
44 DN(N)=DN(N)+RP(I)*XN(I,N)/DEN
42 XTC(N)=CX*DCOS(PAINTL*X(NP))
    XT(N)=CX*DSIN(PAINTL*X(NP))
    PY(K)=PY(K)+XT(N)*DN(N)*SWTSIN
    PYT(K)=PYT(K)+XT(N)*DN(N)*WF*SWTCOS
    PYT2(K)=PYT2(K)+XT(N)*DN(N)*(-WF**2*SWTSIN)
    PM(K)=PM(K)+(PAINTL**2*XT(N)*DN(N)*SWTSIN)*E*XI
DYNA0151
DYNA0152
DYNA0153
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DYNA0156
DYNA0157
DYNA0158
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DYNA0160
DYNA0161
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DYNA0200
DYNA0201
DYNA0202
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DYNA0210
DYNA0211
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DYNA0215
DYNA0216
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DYNA0218
DYNA0219
DYNA0220
DYNA0221
DYNA0222
DYNA0223
DYNA0224
DYNA0225

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      PG(K)=PG(K)-(PAINTL**3*XTC(N)*DN(N)*SWTSIN)*E*XI
45  CONTINUE
51  CONTINUE
      WRITE(ID,700)
700 FORMAT(//)
      WRITE(ID,605) X(NP)
805 FORMAT(1H1,'TIME HISTORY FOR THE POINT( X='',.D9.3,' ')/')
      WRITE(ID,606)
806 FORMAT(' ', ' TIME      PART-DISP PART-VELD PART-ACCL PART-MOMT'
1 ' PART-SHRF')
      APY=0.
      APYT=0.
      APYT2=0.
      APM=0.
      APG=0.
      DO 52 I=1,NT
      IF(APY-DABS(PY(I))) 62,62,63
62  TPY=TT(I)
      APY=DABS(PY(I))
63  IF(APYT-DABS(PYT(I))) 66,66,67
66  TPYT=TT(I)
      APYT=DABS(PYT(I))
67  IF(APYT2-DABS(PYT2(I))) 70,70,71
70  TPYT2=TT(I)
      APYT2=DABS(PYT2(I))
71  IF(APM-DABS(PM(I))) 74,74,75
74  TPM=TT(I)
      APM=DABS(PM(I))
75  IF(APG-DABS(PG(I))) 78,78,79
78  TPQ=TT(I)
      APG=DABS(PG(I))
79  CONTINUE
52  WRITE(ID,607) TT(I),PY(I),PYT(I),PYT2(I),PM(I),PG(I)
      WRITE(ID,610)
607 FORMAT(6D10.3)
610 FORMAT(' ', 'MAXIMUM ABSOLUTE VALUES')
      WRITE(ID,611) APY,APYT,APYT2,APM,APG
611 FORMAT(' MAX ABS ',5D10.3)
      WRITE(ID,612) TPY,TPYT,TPYT2,TPM,TPQ
612 FORMAT(' CORR TIME',5D10.3)
50  CONTINUE
59  IF(IPL0T.EQ.0)GO TO 999
      WRITE(ID2,800)HEAD
800 FORMAT(20A4)
      WRITE(ID2,801)NFRQ,NS1,NTERM
801 FORMAT(3I5)
      WRITE(ID2,802)TL,E,XI,WF,CX
802 FORMAT(3D25.16)
      WRITE(ID2,802)(P(N),N=1,NTERM)
      WRITE(ID2,802)(DN(N),N=1,NTERM)
      DO 7 I=1,NS1
      WRITE(ID2,802)(XN(I,N),N=1,NTERM)
      WRITE(ID2,803)(XL(I),I=1,NS1)
803 FORMAT(5E15.5)
999 RETURN
      END
      SUBROUTINE FREQ(P,XN,FRQ,DET,ITN,RCDEF,WKAREA)
      IMPLICIT DOUBLE PRECISION(A-H,O-Z)
      DIMENSION P(1),DET(1),FRQ(1),ITN(1),XN(NS1,1),RCDEF(NS1,1),
*          WKAREA(1)
      COMMON/SETUP/TL,SM,E,XI,DT,TINT,WF,EPS1,PAI,EX,CX,NSPAN,NTERM,
1 NPOINT,NS1,NT,IN,IO,IO2,IOPT,IPL0T,IMAX,NFRQ,NTERM1,IEGSP
      FMAX=1.D+4
      DO 53 I=1,NFRQ
      FRQ(I)=0.0
      DET(I)=0.0
53  ITN(I)=0
      IF(IEGSP.EQ.0) GO TO 70
      DO 72 I=1,NFRQ
      JFRQ=1+(I-1)*NSPAN
      IF(JFRQ.GT.NFRQ) GO TO 73
72  FRQ(JFRQ)=(NSPAN*PAI*I)**2*DSGRT(E*XI/(SM*TL**4))
73  CONTINUE
70  DO 74 I=1,NFRQ
      IF(FRQ(I).GT.0.0) GO TO 74
      DYNA0226
      DYNA0227
      DYNA0228
      DYNA0229
      DYNA0230
      DYNA0231
      DYNA0232
      DYNA0233
      DYNA0234
      DYNA0235
      DYNA0236
      DYNA0237
      DYNA0238
      DYNA0239
      DYNA0240
      DYNA0241
      DYNA0242
      DYNA0243
      DYNA0244
      DYNA0245
      DYNA0246
      DYNA0247
      DYNA0248
      DYNA0249
      DYNA0250
      DYNA0251
      DYNA0252
      DYNA0253
      DYNA0254
      DYNA0255
      DYNA0256
      DYNA0257
      DYNA0258
      DYNA0259
      DYNA0260
      DYNA0261
      DYNA0262
      DYNA0263
      DYNA0264
      DYNA0265
      DYNA0266
      DYNA0267
      DYNA0268
      DYNA0269
      DYNA0270
      DYNA0271
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      DYNA0279
      DYNA0280
      DYNA0281
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      DYNA0283
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      DYNA0289
      DYNA0290
      DYNA0291
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      DYNA0294
      DYNA0295
      DYNA0296
      DYNA0297
      DYNA0298
      DYNA0299
      DYNA0300

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      IF(I.EQ.1) GO TO 75
      X1=FRQ(I-1)*1.05
      GO TO 76
75  X1=P(1)
76  DELX=X1/50.
      ITR=1
32  CALL FX(FX1,X1,P,XN,RCOEF,WKAREA)
34  X2=X1+DELX
      CALL FX(FX2,X2,P,XN,RCOEF,WKAREA)
      IF(FX1*FX2) 37,35,38
35  XD=X2
      DET(I)=FX2
50  FRQ(I)=XD
      ITN(I)=ITR
      GO TO 74
38  ITR=ITR+1
      IF(ITR.GT.IMAX) GO TO 901
      X1=X2
      FX1=FX2
      GO TO 34
37  IF(DABS(FX2).GT.FMAX*DABS(FX1)) GO TO 60
      IF(DABS(FX1).GT.FMAX*DABS(FX2)) GO TO 60
      X3=(X1*FX2-X2*FX1)/(FX2-FX1)
      GO TO 61
60  X3=(X1+X2)/2.
61  CALL FX(FX3,X3,P,XN,RCOEF,WKAREA)
      IF(DABS(FX3).GT.EPS1) GO TO 39
      XD=X3
      DET(I)=FX3
      GO TO 50
39  IF(DABS(FX3).LT.FMAX) GO TO 40
      WRITE(ID,600) X3
600 FORMAT('/' ' ', 'DETERMINANT APPROACHING INFINITY FOR FREQ EQUAL TO '
1  ',D12.5)
      X1=X3+DELX
      ITR=0
      GO TO 32
40  IF(FX1*FX3) 42,44,41
44  XD=X3
      DET(I)=FX3
      GO TO 50
41  ITR=ITR+1
      IF(ITR.GT.IMAX) GO TO 901
      X1=X3
      FX1=FX3
      GO TO 37
42  ITR=ITR+1
      IF(ITR.GT.IMAX) GO TO 901
      X2=X3
      FX2=FX3
      GO TO 37
901 WRITE(ID,601) ITR, EPS1
601 FORMAT('/' ' ', 'NO CONVERGENCE INCREASE IMAX', I6, ' OR '
1  ' EPS1', D12.4)
74  CONTINUE
      DO 79 I=1, NFRQ
79  WRITE(ID,602) I, FRQ(I), ITN(I), DET(I)
602 FORMAT('/' ' ', 'FREQ (MODE', I2, ' ) = ', D9.4, ' AFTER', I5, ' ITRNS',
1  ' DET = ', D10.4)
      RETURN
      END
      SUBROUTINE FX(DETER, WS, P, XN, RCOEF, WKAREA)
      IMPLICIT DOUBLE PRECISION(A-H, O-Z)
      DIMENSION P(1), XN(NS1,1), RCOEF(NS1,1), WKAREA(1)
      COMMON/SETUP/TL, SM, E, XI, DT, TINT, WF, EPS1, PAI, EX, CX, NSPAN, NTERM,
1  NPOINT, NS1, NT, IN, IO, IO2, IOPT, IPLOT, IMAX, NFRQ, NTERM1, IEQSP
      DO 7 I1=1, NS1
      DO 7 I2=1, NS1
7  RCOEF(I1, I2)=0.
      DO 10 N=1, NTERM1
      PAIN=PAI*N
      PAINTL=PAIN/TL
      DEN=P(N)**2-WS**2
      IF(DABS(DEN).LT..1D-50) GO TO 10
      DO 10 I1=1, NS1

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DYNA0301
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DYNA0321
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DYNA0363
DYNA0364
DYNA0365
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DYNA0368
DYNA0369
DYNA0370
DYNA0371
DYNA0372
DYNA0373
DYNA0374
DYNA0375

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DO 10 I2=1,NS1
RCOEF(I1,I2)=RCOEF(I1,I2)+XN(I1,N)*XN(I2,N)/DEN
10 CONTINUE
D1=0.
CALL LINV3F(RCOEF,B,4,NS1,NS1,D1,D2,WKAREA,IER)
IF(D2.LT.200.)GO TO 12
WRITE(IO,600)MS
600 FORMAT(' DETERMINANT APPROACING INFINITY FOR FREQ EQUAL TO '
1 D12.5)
GO TO 13
12 DETER=D1*2**D2
13 IF(IER.NE.65.)GO TO 15
WRITE(IO,555)
555 FORMAT('/' A SINGULAR MATRIX WAS ENCOUNTERED. '/')
15 RETURN
END
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DYNA0376
DYNA0377
DYNA0378
DYNA0379
DYNA0380
DYNA0381
DYNA0382
DYNA0383
DYNA0384
DYNA0385
DYNA0386
DYNA0387
DYNA0388
DYNA0389
DYNA0390
DYNA0391
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C MOM3
C A COMPUTER PROGRAM TO EVALUATE THE FORCED VIBRATION DISP- * MOM30001
C LACEMENTS-MOMENT AND SHEAR FOR A CONTINUOUS BEAM,USING * MOM30002
C THE DYNAMIC THREE MOMENT EQUATION. * MOM30003
C * MOM30004
C ***** * MOM30005
C * MOM30006
C PROGRAMMED SPRING 1980 FOR XEROX SIGMA 9 * MOM30007
C AT MARQUETTE UNIVERSITY * MOM30008
C * MOM30009
C ***** * MOM30010
C * MOM30011
C DEFINITION OF VARIABLES : * MOM30012
C * MOM30013
C HEAD = TITLE COMPOSED OF UP TO 80 ALPHANUMERIC CHARACTERS * MOM30014
C NSPAN = THE NO. OF SPANS * MOM30015
C WF= FORCING FREQUENCY (RAD/SEC) * MOM30016
C SL(I) = SPAN LENGTH OF SPAN I * MOM30017
C YM(I) = YOUNG'S MODULUS OF SPAN I * MOM30018
C SI(I) = MOMENT OF INERTIA OF SPAN I * MOM30019
C SSAM(I) = MASS OF SPAN I * MOM30020
C DELMAX(J) = AMPLITUDE OF VIBRATION OF SUPPORT J * MOM30021
C ITYPE = ALPHANUMERIC STRING OF FOUR CHARACTERS THAT SHOULD * MOM30022
C BE INPUT AS ' TIME ' WHEN THE DISPLACEMENT,MOMENT, * MOM30023
C SHEAR AS A FUNCTION OF TIME AT SPECIFIC POINTS ALONG * MOM30024
C THE BEAM SPAN.ITYDE SHOULD BE INPUT AS ' DIST ' WHEN * MOM30025
C THESE FUNCTIONS ARE DESIRED TO BE EVALUATED W.R.T * MOM30026
C DISTANCE ALONG THE SPAN AT CERTAIN POINTS IN TIME * MOM30027
C * MOM30028
C THE FOLLOWING DEFINITIONS HOLD WHEN ITYPE=TIME * MOM30029
C * MOM30030
C NPOINT = NO. OF POINTS WHOSE TIME HISTORY ARE SOUGHT * MOM30031
C XT(N) = THE X COORDINATE OF THE POINT N WHERE * MOM30032
C THE TIME HISTORY IS SOUGHT * MOM30033
C SXORT = DURATION OF EARTHQUAKE * MOM30034
C XTINT = TIME INTERVAL FOR THE TIME HISTORY * MOM30035
C * MOM30036
C THE FOLLOWING DEFINITIONS HOLD WHEN ITYPE=DIST * MOM30037
C * MOM30038
C NPOINT = THE NUMBER OF POINTS IN TIME FOR WHICH THE * MOM30039
C FUNCTIONS ARE TO BE EVALUATED FOR SPECIFIC POINTS * MOM30040
C ALONG THE BEAM SPAN. * MOM30041
C XT(N) = THE TIME COORDINATE OF POINT N * MOM30042
C IF XT(N) IS INPUT AS 999, THEN THE PROGRAM * MOM30043
C AUTOMATICALLY CHOSSES THE TIME FOR MAX PARTICULAR * MOM30044
C RESPONSE  $T=PI/(2.*WF)$  * MOM30045
C * MOM30046
C SXORT = TOTAL SPAN LENGTH * MOM30047
C XTINT = DISTANCE INTERVAL ALONG THE SPAN FOR WHICH * MOM30048
C THE FUNCTIONS ARE TO BE EVALUATED * MOM30049
C ***** * MOM30050
C * MOM30051
C INPUT CARDS SEQUENCE : * MOM30052
C * MOM30053
C NO. OF * MOM30054
C CARDS VARIABLES FORMAT * MOM30055
C * MOM30056
C 1 HEAD (20A4) * MOM30057
C 1 IOPT,NSPAN,IEQSP (3I5) * MOM30058
C 1 NFRQ,IMAX,EPS1 (2I5,G10.0) * MOM30059
C 1 ITYPE,NPOINT,WF,SXORT,XTINT (A4,I5,3G10.0) * MOM30060
C NSPAN SL(I),YM(I),SI(I),SSAM(I) (4G10.0) * MOM30061
C NSPAN-1 DELMAX(J) (10G10.0) * MOM30062
C NPOINT XT(N) (10G10.0) * MOM30063
C * MOM30064
C ***** * MOM30065
C * MOM30066
C IMPLICIT DOUBLE PRECISION(A-H,O-Z) * MOM30067
C DIMENSION HEAD(20),NN(2000) * MOM30068
C COMMON /SETUP/ WF,SXORT,XTINT,EPS1,PI,NSPAN,NSUP,NPOINT,NXORT, * MOM30069
C * NFRQ,IMAX,IN,IO,IOPT,ITYPE,IEQSP * MOM30070
C COMMON A(1000) * MOM30071
C EQUIVALENCE(A(1),NN(1)) * MOM30072
C IN=5 * MOM30073
C IO=6 * MOM30074

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999 READ(IN,500,END=997) HEAD
500 FORMAT(20A4)
WRITE(IO,600)HEAD
600 FORMAT(1H1,///',20A4)
READ(IN,501)IOPT,NSPAN,IEQSP
501 FORMAT(3I5)
IF(IOPT.EQ.1)GO TO 105
READ(IN,503)NFRQ,IMAX,EPS1
503 FORMAT(2I5,G10.0)
NXORT=0
NPOINT=0
IF(IOPT.EQ.2)GO TO 110
105 READ(IN,504) ITYPE,NPOINT,WF, SXORT,XTINT
504 FORMAT(A4, I5,3G10.0)
NXORT= SXORT/XTINT+1
110 NSUP=NSPAN-1
L2=1+NSPAN
L3=L2+NSPAN
L4=L3+NSPAN
L5=L4+NSPAN
L6=L5+NSUP
L7=L6+NSUP
L8=L7+NSUP
L9=L8+NPOINT
L10=L9+NXORT
L11=L10+NXORT
L12=L11+NSUP**2
L13=L12+NFRQ
L14=L13+NFRQ
L15=2*(L14+2*NSUP)+1
NSIZE=L15+NFRQ
WRITE(IO,602) NSIZE
602 FORMAT(///', 'BLANK COMMON SIZE',I6)
IF(NSIZE.GT.2000) GO TO 998
CALL MOMAIN(A(1),A(L2),A(L3),A(L4),A(L5),A(L6),A(L7),A(L8),A(L9),
1 A(L10),A(L11),A(L12),A(L13),A(L14),NN(L15))
GO TO 999
998 WRITE(IO,601) NSIZE
601 FORMAT(///', 'INCREASE BLANK COMMON ARRAY UP TO ',I6)
997 STOP
END
SUBROUTINE MOMAIN(SL,YM,SI,SSAM,DELMAX,TH,VECTOR,XT,X,T,
1 ALPHA,FRQ,DET,WKAREA,ITN)
IMPLICIT DOUBLE PRECISION(A-H,O-Z)
DIMENSION SL(NSPAN),YM(NSPAN),SI(NSPAN),SSAM(NSPAN),
1 DELMAX(NSUP),TH(NSUP),VECTOR(NSUP),XT(NPOINT),
2 ALPHA(NSUP,NSUP),X(NXORT),T(NXORT),FRQ(NFRQ),DET(NFRQ),
3 WKAREA(2*NSUP),ITN(NFRQ)
COMMON /SETUP/ WF, SXORT,XTINT,EPS1,PI,NSPAN,NSUP,NPOINT,NXORT,
* NFRQ,IMAX,IN,IO,IOPT,ITYPE,IEQSP
C
C INITIALIZATION, READ WRITE BLOCK
C
FI=2.*DASIN(1.D0)
DO 5 I=1,NSPAN
5 READ(IN,503)SL(I),YM(I),SI(I),SSAM(I)
IF(IOPT.EQ.2)GO TO 177
DO 10 I=1,NSUP
10 READ(IN,503)DELMAX(I)
READ (IN,503)(XT(I),I=1,NPOINT)
503 FORMAT(10G10.3)
177 IF(IOPT.EQ.1)GO TO 7
CALL MOFREQ(FRQ,DET,ITN,ALPHA,VECTOR,WKAREA,SL,YM,SSAM,SI)
IF(IOPT.EQ.2) GO TO 17
7 IF(ITYPE.NE.4HDIST)GO TO 15
WRITE(IO,601)NSPAN,WF, SXORT,XTINT,NPOINT
601 FORMAT(/// NSPAN =I10,/,
* / FORCING FREQUENCY(RAD/SEC) =D10.4,/,
* / TOTAL SPAN LENGTH =D10.4,/,
* / DISTANCE INERVAL ALONG THE SPAN =D10.4,/,
* / NO. OF POINTS IN TIME =I10)
GO TO 17
15 WRITE(IO,602)NSPAN,WF, SXORT,XTINT,NPOINT
602 FORMAT(/// NSPAN =I10,/,
* / FORCING FREQUENCY(RAD/SEC) =D10.4,/,
* / DURATION OF EARTHQUAKE =D10.4,/,

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MOM30075
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*          TIME INCREMENT IN TIME HISTORY      =D10.4,/,
*          NO. OF POINTS WHERE HISTORY SOUGHT =I10)
17 DO 10 I=1,NSPAN
20 WRITE(IO,603)I,SL(I),YM(I),SI(I),SSAM(I)
603 FORMAT(/' FOR THE SPAN NO. 'I2//,
*          THE SPAN LENGTH IS                  =D10.4,/,
*          YOUNGS MODULUS IS                  =D10.4,/,
*          THE MOMENT OF INERTIA IS           =D10.4,/,
*          THE MASS PER UNIT LENGTH IS       =D10.4)
IF(IOPT.EQ.2)GO TO 999
DO 25 I=1,NSUP
25 WRITE(IO,604)I,DELMAX(I)
604 FORMAT(/' FOR THE SUPPORT NO. 'I2//,
*          THE AMPLITUDE OF SETTLEMENT IS    =D10.4)
C
C          CALCULATION OF THE DYNAMIC MOMENTS AT THE INTERIOR SUPPORTS
C
DO 125 I=1,NSUP
DO 125 J=1,NSUP
125 ALPHA(I,J)=0.
DO 150 IR=1,NSUP
GAMMA=0.
GAMMAL=0.
GAMMAR=0.
IRR=IR+1
IRL=IR-1
BR=(SSAM(IR)*WF**2/(YM(IR)*SI(IR)))*.25*SL(IR)
H1BR=(DCOSH(BR)/DSINH(BR)-DCOS(BR)/DSIN(BR))/BR
F1BR=(DCOSH(BR)/DSINH(BR)+DCOS(BR)/DSIN(BR))*BR
BR1=(SSAM(IRR)*WF**2/(YM(IRR)*SI(IRR)))*.25*SL(IRR)
H1BR1=(DCOSH(BR1)/DSINH(BR1)-DCOS(BR1)/DSIN(BR1))/BR1
F1BR1=(DCOSH(BR1)/DSINH(BR1)+DCOS(BR1)/DSIN(BR1))*BR1
ALPHA(IR,IR)=SL(IR)*H1BR/SI(IR)+SL(IRR)*H1BR1/SI(IRR)
GAMMA=(F1BR/SL(IR)+F1BR1/SL(IRR))*DELMAX(IR)*YM(IR)
IF(NSUP.EQ.1)GO TO 140
H2BR=(1./DSIN(BR)-1./DSINH(BR))/BR
F2BR=(1./DSIN(BR)+1./DSINH(BR))*BR
H2BR1=(1./DSIN(BR1)-1./DSINH(BR1))/BR1
F2BR1=(1./DSIN(BR1)+1./DSINH(BR1))*BR1
IF(IR.EQ.1)GO TO 135
130 ALPHA(IR,IRL)=H2BR*SL(IR)/SI(IR)
GAMMAL=-F2BR*DELMAX(IRL)*YM(IR)/SL(IR)
IF(IR.EQ.NSUP) GO TO 140
135 ALPHA(IR,IRR)=H2BR1*SL(IRR)/SI(IRR)
GAMMAR=-F2BR1*DELMAX(IRR)*YM(IRR)/SL(IRR)
140 VECTOR(IR)=GAMMAL+GAMMA+GAMMAR
150 CONTINUE
IF(NSUP.GT.1) GO TO 175
TM(1)=VECTOR(1)/ALPHA(1,1)
GO TO 200
175 DO 180 I=1,NSUP
180 TM(I)=VECTOR(I)
D1=-1.
CALL LINV3F(ALPHA,TM,2,NSUP,NSUP,D1,D2,WKAREA,IER)
IF(IER.EQ.65)GO TO 500
C
C          CALCULATION OF THE DISP,MOM,SHEAR FUNCTIONS
C
200 DO 400 NP=1,NPOINT
YMAX=0.
RMMAX=0.
QMAX=0.
IF(ITYPE.NE.4HTIME)GO TO 210
DO 205 I=1,NXORT
X(I)=XT(NP)
T(I)=(I-1)*XTINT
205 CONTINUE
WRITE(IO,605)XT(NP)
605 FORMAT(1H1/' TIME HISTORY FOR THE POINT ( X='G10.3,')')
GO TO 230
210 IF(XT(NP).EQ.999.)GO TO 220
DO 215 I=1,NXDRT
X(I)=(I-1)*XTINT
T(I)=XT(NP)
215 CONTINUE

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MOM30225

	WRITE(IO,604)XT(NP)	MOM30226
606	FORMAT(1H1// ' PARTICULAR RESPONSE ALONG THE SPAN AT T='D12.5)	MOM30227
	GO TO 230	MOM30228
220	DO 225 I=1,NXORT	MOM30229
	X(I)=(I-1)*XTINT	MOM30230
	T(I)=PI/(2.*WF)	MOM30231
225	CONTINUE	MOM30232
	WRITE(IO,607)T(1)	MOM30233
607	FORMAT(1H1// ' MAXIMUM PARTICULAR RESPONSE ALONG THE SPAN (T='	MOM30234
	* D12.5, /)')	MOM30235
230	WRITE(IO,608)ITYPE	MOM30236
608	FORMAT(//15X,A4,9X,'PART DISP',6X,'PART MOM',6X,'PART SHEAR',/	MOM30237
	* 15X,4(1H-),9X,9(1H-),6X,8(1H-),6X,10(1H-),/)	MOM30238
	DO 399 I=1,NXORT	MOM30239
	XL=0.	MOM30240
	XR=0.	MOM30241
	DO 250 N=1,NSPAN	MOM30242
	XR=XR+SL(N)	MOM30243
	IF((X(I).LT.XL).OR.(X(I).GT.XR))GO TO 235	MOM30244
	SPANL=SL(N)	MOM30245
	SPANX=X(I)-XL	MOM30246
	ISUP=N-1	MOM30247
	KSUP=N	MOM30248
	NSP=N	MOM30249
	GO TO 255	MOM30250
235	XL=XL+SL(N)	MOM30251
250	CONTINUE	MOM30252
255	XLAMDA=(SSAM(NSP)*WF**2/(YM(NSP)*SI(NSP)))*.25	MOM30253
	B1=XLAMDA*(SPANL-SPANX)	MOM30254
	B2=XLAMDA*SPANX	MOM30255
	BETA=XLAMDA*SPANL	MOM30256
	V1=DSIN(B1)	MOM30257
	V2=DSIN(B2)	MOM30258
	V3=DSIN(BETA)	MOM30259
	V4=DSINH(B1)	MOM30260
	V5=DSINH(B2)	MOM30261
	V6=DSINH(BETA)	MOM30262
	V7=2*YM(NSP)*SI(NSP)*XLAMDA**2	MOM30263
	C1=DCOS(B1)	MOM30264
	C2=DCOS(B2)	MOM30265
	C4=DCOSH(B1)	MOM30266
	C5=DCOSH(B2)	MOM30267
	XX1=0.	MOM30268
	XX2=0.	MOM30269
	XX3=0.	MOM30270
	XX4=0.	MOM30271
	XM1=0.	MOM30272
	XM2=0.	MOM30273
	XM3=0.	MOM30274
	XM4=0.	MOM30275
	XQ1=0.	MOM30276
	XQ2=0.	MOM30277
	XQ3=0.	MOM30278
	XQ4=0.	MOM30279
	IF(NSP.EQ.1) GO TO 260	MOM30280
	XX1=TM(ISUP)*(V1/V3-V4/V6)/V7	MOM30281
	XX3=DELMAX(ISUP)*(V1/V3+V4/V6)/2.	MOM30282
	XM1=TM(ISUP)*(V1/V3+V4/V6)/2.	MOM30283
	XM3=DELMAX(ISUP)*XLAMDA**2*YM(NSP)*SI(NSP)*(V1/V3-V4/V6)/2.	MOM30284
	XQ1=TM(ISUP)*(-XLAMDA)*(C1/V3+C4/V6)/2.	MOM30285
	XQ3=-DELMAX(ISUP)*XLAMDA**3*YM(NSP)*SI(NSP)*(C1/V3-C4/V6)/2.	MOM30286
260	IF(NSP.EQ.NSPAN)GO TO 265	MOM30287
	XX2=TM(KSUP)*(V2/V3-V5/V6)/V7	MOM30288
	XX4=DELMAX(KSUP)*(V2/V3+V5/V6)/2.	MOM30289
	XM2=TM(KSUP)*(V2/V3+V5/V6)/2.	MOM30290
	XM4=DELMAX(KSUP)*XLAMDA**2*YM(NSP)*SI(NSP)*(V2/V3-V5/V6)/2.	MOM30291
	XQ2=TM(KSUP)*XLAMDA*(C2/V3+C5/V6)/2.	MOM30292
	XQ4=DELMAX(KSUP)*XLAMDA**3*YM(NSP)*SI(NSP)*(C2/V3-C5/V6)/2.	MOM30293
265	XX=XX1+XX2+XX3+XX4	MOM30294
	XM=XM1+XM2+XM3+XM4	MOM30295
	XQ=XQ1+XQ2+XQ3+XQ4	MOM30296
	WFT=WF*T(I)	MOM30297
	Y=XX*DSIN(WFT)	MOM30298
	RM=XM*DSIN(WFT)	MOM30299
	Q=XQ*DSIN(WFT)	MOM30300

270	IF(ITYPE.NE.4)HDIST)GO TO 300	MOM30301
	IF(DABS(YMAX).GT.DABS(Y))GO TO 275	MOM30302
	YMAX=Y	MOM30303
	XTYMAX=X(I)	MOM30304
275	IF(DABS(RMMAX).GT.DABS(RM))GO TO 280	MOM30305
	RMMAX=RM	MOM30306
	XTMMAX=X(I)	MOM30307
280	IF(DABS(QMAX).GT.DABS(Q))GO TO 285	MOM30308
	QMAX=Q	MOM30309
	XTQMAX=X(I)	MOM30310
285	WRITE(ID,609)X(I),Y,RM,Q	MOM30311
	GO TO 399	MOM30312
300	IF(DABS(YMAX).GT.DABS(Y))GO TO 305	MOM30313
	YMAX=Y	MOM30314
	XTYMAX=T(I)	MOM30315
305	IF(DABS(RMMAX).GT.DABS(RM))GO TO 310	MOM30316
	RMMAX=RM	MOM30317
	XTMMAX=T(I)	MOM30318
310	IF(DABS(QMAX).GT.DABS(Q))GO TO 315	MOM30319
	QMAX=Q	MOM30320
	XTQMAX=T(I)	MOM30321
315	WRITE(ID,609)T(I),Y,RM,Q	MOM30322
399	CONTINUE	MOM30323
400	CONTINUE	MOM30324
609	FORMAT(7X,4D15.5)	MOM30325
	WRITE(ID,610)YMAX,RMMAX,QMAX,ITYPE,XTYMAX,XTMMAX,XTQMAX	MOM30326
610	FORMAT(/// 'MAX VALUES'10X,3D15.5,/// ' AT 'A4,13X,3D15.5,///)	MOM30327
	GO TO 999	MOM30328
500	WRITE(ID,611)	MOM30329
611	FORMAT(/// 'SINGULAR MATRIX ENCOUNTERED'///)	MOM30330
999	RETURN	MOM30331
	END	MOM30332
	SUBROUTINE MOFREQ(FRQ,DET,ITN,ALPHA,VECTOR,WKAREA,SL,YM,SSAM,SI)	MOM30333
	IMPLICIT DOUBLE PRECISION(A-H,O-Z)	MOM30334
	DIMENSION DET(1),FRQ(1),ALPHA(NSUP,1),VECTOR(1),	MOM30335
	* WKAREA(1),SL(1),YM(1),SSAM(1),SI(1),ITN(1)	MOM30336
	* COMMON /SETUP/ WF, SXORT, XTINT, EPS1, PI, NSPAN, NSUP, NPOINT, NXORT,	MOM30337
	* NFRQ, IMAX, IN, IO, IOPT, ITYPE, IEQSP	MOM30338
	FMAX=1.D+4	MOM30339
	FNMIN=1.D+50	MOM30340
	DO 53 I=1,NFRQ	MOM30341
	FRQ(I)=0.0	MOM30342
	DET(I)=0.0	MOM30343
53	ITN(I)=0	MOM30344
	IF(IEQSP.EQ.0) GO TO 70	MOM30345
	TL=0.0	MOM30346
	DO 71 I=1,NFRQ	MOM30347
71	TL=TL+SL(I)	MOM30348
	DO 72 I=1,NFRQ	MOM30349
	JFRQ=1+(I-1)*NSPAN	MOM30350
	IF(JFRQ.GT.NFRQ) GO TO 73	MOM30351
72	FRQ(JFRQ)=(NSPAN*PI*I)**2*DSQRT(YM(1)*SI(1)/(SSAM(1)*TL**4))	MOM30352
73	CONTINUE	MOM30353
70	DO 74 I=1,NFRQ	MOM30354
	IF(FRQ(I).GT.0.0) GO TO 74	MOM30355
	IF(I.EQ.1) GO TO 75	MOM30356
	X1=FRQ(I-1)*1.05	MOM30357
	GO TO 76	MOM30358
75	DO 100 II=1,NSUP	MOM30359
	PN=DSQRT(YM(II)*SI(II)/(SSAM(II)*SL(II)**4))*PI*2	MOM30360
	IF(PN.GT.FNMIN)GO TO 100	MOM30361
	FNMIN=PN	MOM30362
100	CONTINUE	MOM30363
	X1=FNMIN*.5	MOM30364
76	DELX=X1/50.	MOM30365
	ITR=1	MOM30366
32	CALL MOFX(FX1,X1,ALPHA,VECTOR,WKAREA,SL,YM,SSAM,SI)	MOM30367
34	X2=X1+DELX	MOM30368
	CALL MOFX(FX2,X2,ALPHA,VECTOR,WKAREA,SL,YM,SSAM,SI)	MOM30369
	IF(DABS(FX2).LT.EPS1)GO TO 35	MOM30370
	IF(FX1*FX2) 37,35,38	MOM30371
35	XD=X2	MOM30372
	DET(I)=FX2	MOM30373
50	FRQ(I)=XD	MOM30374
	ITN(I)=ITR	MOM30375

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GO TO 74
38 ITR=ITR+1
IF(ITR.GT.IMAX) GO TO 901
X1=X2
FX1=FX2
GO TO 34
37 IF(DABS(FX2).GT.FMAX*DABS(FX1)) GO TO 60
IF(DABS(FX1).GT.FMAX*DABS(FX2)) GO TO 60
X3=(X1*FX2-X2*FX1)/(FX2-FX1)
GO TO 61
60 X3=(X1+X2)/2.
61 CALL MOFX(FX3,X3,ALPHA,VECTOR,WKAREA,SL,YM,SSAM,SI)
IF(DABS(FX3).GT.EPS1) GO TO 39
XD=X3
DET(I)=FX3
GO TO 50
39 IF(DABS(FX3).LT.FMAX) GO TO 40
WRITE(10,600) X3
600 FORMAT(/' ', 'DETERMINANT APPROACHING INFINITY FOR FREQ EQUAL TO
1 'D12.5)
X1=X3+DELX
ITR=0
GO TO 32
40 IF(FX1*FX3) 42,44,41
44 XD=X3
DET(I)=FX3
GO TO 50
41 ITR=ITR+1
IF(ITR.GT.IMAX) GO TO 901
X1=X3
FX1=FX3
GO TO 37
42 ITR=ITR+1
IF(ITR.GT.IMAX) GO TO 901
X2=X3
FX2=FX3
GO TO 37
901 WRITE(10,601) ITR, EPS1
601 FORMAT(/' ', 'NO CONVERGENCE INCREASE IMAX', I6, ' OR ',
1 'EPS1', D12.4)
74 CONTINUE
DO 79 I=1, NFRQ
79 WRITE(10,602) I, FRQ(I), ITN(I), DET(I)
602 FORMAT(/' ', 'FREQ (MODE', I2, ' ) = ', D9.4, ' AFTER', I5, ' ITRNS',
1 ' DET = ', D10.4)
RETURN
END
SUBROUTINE MOFX(DETER, WS, ALPHA, VECTOR, WKAREA, SL, YM, SSAM, SI)
IMPLICIT DOUBLE PRECISION(A-H, O-Z)
DIMENSION ALPHA(NSUP, 1), WKAREA(1), SL(1), YM(1), SI(1), SSAM(1),
1 VECTOR(1)
COMMON /SETUP/ WF, SXORT, XTINT, EPS1, PI, NSPAN, NSUP, NPOINT, NXORT,
* NFRQ, IMAX, IN, IO, IOPT, ITYPE, IEQSP
DO 125 I=1, NSUP
DO 125 J=1, NSUP
125 ALPHA(I, J)=0.
DO 150 IR=1, NSUP
IRR=IR+1
IRL=IR-1
BR=(SSAM(IR)*WS**2/(YM(IR)*SI(IR)))**.25*SL(IR)
H1BR=(DCOSH(BR)/DSINH(BR)-DCOS(BR)/DSIN(BR))/BR
BR1=(SSAM(IRR)*WS**2/(YM(IRR)*SI(IRR)))**.25*SL(IRR)
H1BR1=(DCOSH(BR1)/DSINH(BR1)-DCOS(BR1)/DSIN(BR1))/BR1
ALPHA(IR, IRL)=SL(IR)*H1BR/SI(IR)+SL(IRR)*H1BR1/SI(IRR)
IF(NSUP.EQ.1)GO TO 150
H2BR=(1./DSIN(BR)-1./DSINH(BR))/BR
H2BR1=(1./DSIN(BR1)-1./DSINH(BR1))/BR1
IF(IR.EQ.1)GO TO 135
130 ALPHA(IR, IRL)=H2BR*SL(IR)/SI(IR)
IF(IR.EQ.NSUP) GO TO 150
135 ALPHA(IR, IRR)=H2BR1*SL(IRR)/SI(IRR)
150 CONTINUE
B1=0.
CALL LINU3F(ALPHA, VECTOR, 4, NSUP, NSUP, D1, D2, WKAREA, IER)
IF(D2.LT.200)GO TO 12

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WRITE(10,600)WS
600 FORMAT('DETERMINANT APPROACHING INFINITY FOR FREQ EQUAL TO ',
1      D12.5)
GO TO 13
12 DETER=D1*2**D2
13 IF(IER.NE.65)GO TO 15
WRITE(10,601)
601 FORMAT('/ / A SINGULAR MATRIX WAS ENCOUNTERED. / /')
15 RETURN
END
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C MOM3ST
C A COMPUTER PROGRAM TO EVALUATE THE DISPLACEMENTS, M3ST0001
C MOMENTS AND SHEAR ALONG THE LENGTH OF A CONTINUOUS M3ST0002
C BEAM SUBJECTED TO STATIC SUPPORT SETTLEMENTS, USING M3ST0003
C THE STATIC THREE MOMENT EQUATION. M3ST0004
C M3ST0005
C***** M3ST0006
C PROGRAMMED SPRING 1980 FOR XEROX SIGMA 9 M3ST0007
C BY SAMIR ACRA M3ST0008
C AT MARQUETTE UNIVERSITY M3ST0009
C M3ST0010
C M3ST0011
C***** M3ST0012
C DEFINITION OF VARIABLES : M3ST0013
C M3ST0014
C HEAD = TITLE COMPOSED OF UP TO 80 ALPHANUMERIC CHARACTERS M3ST0015
C NSPAN = THE NO. OF SPANS M3ST0016
C TL = TOTAL SPAN LENGTH M3ST0017
C XINT = INTERVAL ALONG SPAN AT WHICH THE FUNCTIONS M3ST0018
C ARE TO BE EVALUATED M3ST0019
C SL(I) = SPAN LENGTH OF SPAN I M3ST0020
C YM(I) = YOUNG'S MODULUS OF SPAN I M3ST0021
C SI(I) = MOMENT OF INERTIA OF SPAN I M3ST0022
C SSAM(I) = MASS OF SPAN I M3ST0023
C DELMAX(J) = AMPLITUDE OF VIBRATION OF INTERIOR SUPPORT J M3ST0024
C THERE ARE NSPAN-1 INTERIOR SUPPORTS. M3ST0025
C M3ST0026
C***** M3ST0027
C INPUT CARDS SEQUENCE : M3ST0028
C M3ST0029
C NO. OF M3ST0030
C CARDS VARIABLES FORMAT M3ST0031
C M3ST0032
C 1 HEAD (20A4) M3ST0033
C 1 NSPAN,TL,XINT (I5,2G10.0) M3ST0034
C NSPAN SL(I),YM(I),SI(I),SSAM(I) (4G10.0) M3ST0035
C NSPAN-1 DELMAX(J) (10G10.0) M3ST0036
C M3ST0037
C***** M3ST0038
C M3ST0039
C M3ST0040
C IMPLICIT DOUBLE PRECISION(A-H,O-Z) M3ST0041
C REAL AA(2500) M3ST0042
C DIMENSION HEAD(20),NN(2500) M3ST0043
C COMMON A(1000) M3ST0044
C EQUIVALENCE(A(1),NN(1)),(A(1),AA(1)) M3ST0045
C IN=5 M3ST0046
C IO=6 M3ST0047
C 999 READ(IN,500,END=997) HEAD M3ST0048
C 500 FORMAT(20A4) M3ST0049
C WRITE(IO,600) HEAD M3ST0050
C 600 FORMAT(1H1,///',20A4) M3ST0051
C READ(IN,501) NSPAN,TL,XINT M3ST0052
C 501 FORMAT(I5,2G10.0) M3ST0053
C NSUP=NSPAN-1 M3ST0054
C NX=TL/XINT+1 M3ST0055
C L2=1+NSPAN M3ST0056
C L3=L2+NSPAN M3ST0057
C L4=L3+NSPAN M3ST0058
C L5=L4+NSPAN M3ST0059
C L6=L5+NSUP M3ST0060
C L7=L6+NSUP M3ST0061
C L8=L7+NSUP M3ST0062
C L9=L8+NX M3ST0063
C L10=(L9+NSUP**2)*2 M3ST0064
C L11=L10+NX M3ST0065
C L12=L11+NX M3ST0066
C L13=L12+NX M3ST0067
C L14=L13+NX M3ST0068
C NSIZE=L14+NSUP*2 M3ST0069
C WRITE(IO,602) NSIZE M3ST0070
C 602 FORMAT(///', 'BLANK COMMON SIZE',I6) M3ST0071
C IF(NSIZE.GT.2500) GO TO 998 M3ST0072
C CALL STMAIN(A(1),A(L2),A(L3),A(L4),A(L5),A(L6),A(L7),A(L8),A(L9), M3ST0073
C 1 TL,XINT,AA(L10),AA(L11),AA(L12),AA(L13),NN(L14),NSPAN,NSUP, M3ST0075

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2 NX,IN,IO)
GO TO 999
998 WRITE(ID,601) NSIZE
601 FORMAT(// ' ', 'INCREASE BLANK COMMON ARRAY UP TO ',IG)
997 STOP
END
SUBROUTINE STMAIN(SL,YM,SI,SSAM,DELMAX,TM,VECTOR,X,ALPHA,
1 TL,XINT,RX,XX,XM,XQ,INDEX,NSPAN,NSUP,NX,IN,IO)
IMPLICIT DOUBLE PRECISION(A-H,O-Z)
REAL RX(NX),XX(NX),XM(NX),XQ(NX)
DIMENSION SL(NSPAN),YM(NSPAN),SI(NSPAN),SSAM(NSPAN),
1 DELMAX(NSUP),TM(NSUP),VECTOR(NSUP),X(NX),
2 ALPHA(NSUP,NSUP),INDEX(NSUP,2)
MSING=1
C
C RAED WRITE BLOCK
C
DO 5 I=1,NSPAN
5 READ(IN,503)SL(I),YM(I),SI(I),SSAM(I)
READ(IN,503)(DELMAX(I),I=1,NSUP)
503 FORMAT(10G10.3)
WRITE(ID,606)NSPAN,TL,XINT
606 FORMAT(// ' NSPAN
* ' TOTAL SPAN LENGTH
* ' X INTERVAL ALONG THE SPAN
= 'I10.//,
= 'D10.4.//,
= 'D10.4//)
DO 201 I=1,NSPAN
201 WRITE(ID,607)I,SL(I),YM(I),SI(I),SSAM(I)
607 FORMAT(// ' FOR THE SPAN NO. 'I2//,
* ' THE SPAN LENGTH IS
* ' YOUNGS MODULUS IS
* ' THE MOMENT OF INERTIA IS
* ' THE MASS PER UNIT LENGTH IS
= 'D10.4.//,
= 'D10.4.//,
= 'D10.4.//,
= 'D10.4)
DO 202 I=1,NSUP
202 WRITE(ID,608)I,DELMAX(I)
608 FORMAT(// ' FOR THE SUPPORT NO. 'I2//,
* ' THE SETTLEMENT IS
= 'D10.4)
WRITE(ID,601)
601 FORMAT(1H1, //15X, 'DIST'11X, 'DISP',11X, 'MOM',10X, //,
* 'SHEAR', //,15X,4(1H-),11X,4(1H-),11X,3(1H-),11X,5(1H-),//)
C
C CALCULATION OF THE STATIC MOMENTS AT THE INTERIOR SUPPORTS
C
DO 25 I=1,NSUP
DO 25 J=1,NSUP
25 ALPHA(I,J)=0.
DO 100 IR=1,NSUP
GAMMA=0.
GAMMAL=0.
GAMMAR=0.
IRR=IR+1
IRL=IR-1
ALPHA(IR,IR)=2.*(SL(IR)/(YM(IR)*SI(IR))+SL(IRR)/(YM(IRR)*SI(IRR)))
GAMMA=(6./SL(IR)+6./SL(IRR))*DELMAX(IR)
IF(NSUP.EQ.1)GO TO 50
IF(IR.EQ.1)GO TO 45
40 ALPHA(IR,IRL)=SL(IR)/(YM(IR)*SI(IR))
GAMMAL=-6.*DELMAX(IRL)/SL(IR)
IF(IR.EQ.NSUP) GO TO 50
45 ALPHA(IR,IRR)=SL(IRR)/(YM(IRR)*SI(IRR))
GAMMAR=-6.*DELMAX(IRR)/SL(IRR)
50 VECTOR(IR)=GAMMAL+GAMMA+GAMMAR
100 CONTINUE
IF(NSUP.GT.1) GO TO 105
TM(1)=VECTOR(1)/ALPHA(1,1)
GO TO 109
105 CALL SIMULE(NSUP,ALPHA,VECTOR,INDEX,MSING)
IF(MSING.EQ.0) GO TO 400
DO 106 I=1,NSUP
106 TM(I)=0.
DO 107 I=1,NSUP
DO 107 J=1,NSUP
107 TM(I)=TM(I)+ALPHA(I,J)*VECTOR(J)
C
C CALCULATION OF THE DISP,MOM,SHEAR FUNCTIONS ALONG THE SPANS
C

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M3ST0080
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M3ST0125
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M3ST0148
M3ST0149
M3ST0150

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109  DO 210 I=1,NX
      X(I)=(I-1)*XINT
      XL=0.
      XR=0.
      DO 110 N=1,NSPAN
        XR=XR+SL(N)
        IF((X(I).LT.XL).OR.(X(I).GT.XR))GO TO 108
        SPANL=SL(N)
        SPANX=X(I)-XL
        ISUP=N-1
        KSUP=N
        NSP=N
        GO TO 120
108  XL=XL+SL(N)
110  CONTINUE
120  XX1=0.
      XX2=0.
      XX3=0.
      XX4=0.
      XM1=0.
      XM2=0.
      XQ1=0.
      XQ2=0.
      IF(NSP.EQ.1) GO TO 125
      XX1=(SPANX**3/(6.*SPANL)-SPANX**2/2.+SPANX*SPANL/3.)*
      *   TM(ISUP)/(YM(N)*SI(N))
      XX3=(1.-SPANX/SPANL)*DELMAX(ISUP)
      XM1=(1.-SPANX/SPANL)*TM(ISUP)
      XQ1=-TM(ISUP)/SPANL
125  IF(NSP.EQ.NSPAN)GO TO 130
      XX2=(-SPANX**3/(6.*SPANL)+SPANX*SPANL/6.)*TM(KSUP)/(YM(N)*SI(N))
      XX4=(SPANX/SPANL)*DELMAX(KSUP)
      XM2=TM(KSUP)*SPANX/SL(N)
      XQ2=TM(KSUP)/SPANL
130  XX(I)=XX1+XX2+XX3+XX4
      XM(I)=XM1+XM2
      XQ(I)=XQ1+XQ2
      WRITE(ID,602)X(I),XX(I),XM(I),XQ(I)
210  CONTINUE
602  FORMAT(7X,4E15.5)
      XXMAX=0.
      XMMAX=0.
      XQMAX=0.
      DO 220 I=1,NX
        IF(ABS(XX(I)).LT.ABS(XXMAX))GO TO 212
        XXMAX=XX(I)
        X1=X(I)
212  IF(ABS(XM(I)).LT.ABS(XMMAX))GO TO 214
        XMMAX=XM(I)
        X2=X(I)
214  IF(ABS(XQ(I)).LT.ABS(XQMAX))GO TO 220
        XQMAX=XQ(I)
        X3=X(I)
220  CONTINUE
      WRITE(ID,610)XXMAX,XMMAX,XQMAX,X1,X2,X3
610  FORMAT(// ' MAX VALUES '10X,3E15.5,/' AT DIST '13X,3E15.5,/)
      GO TO 401
400  WRITE(ID,402)
402  FORMAT(// ' ', 'SINGULAR MATRIX ENCOUNTERED'//)
401  RETURN
      END
      SUBROUTINE SIMULE(N,A,B,INDEX,MSING)
C
C
C      THIS IS AN INVERSION AND SIMULTANEOUS EQUATION SOLVER
      IMPLICIT DOUBLE PRECISION(A-H,O-Z)
      DIMENSION INDEX(N,2),A(N,N),B(1)
      DO 100 I=1,N
        INDEX(I,1)=0
100  CONTINUE
      II=0
109  AMAX=-1
      DO 110 I=1,N
        IF (INDEX(I,1))110,111,110
111  DO 112 J=1,N

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M3ST0200
M3ST0201
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M3ST0221
M3ST0222
M3ST0223
M3ST0224
M3ST0225

IF (INDEX(J,1))112,113,112	M3ST0226
113 TEMP=DABS(A(I,J))	M3ST0227
IF (TEMP-AMAX)112,112,114	M3ST0228
114 IROW=I	M3ST0229
ICOL=J	M3ST0230
AMAX=TEMP	M3ST0231
112 CONTINUE	M3ST0232
110 CONTINUE	M3ST0233
IF(AMAX)225,115,116	M3ST0234
118 INDEX(ICOL,1)=IROW	M3ST0235
IF (IROW-ICOL)119,118,119	M3ST0236
119 DO 120 J=1,N	M3ST0237
TEMP=A(IROW,J)	M3ST0238
A(IROW,J)=A(ICOL,J)	M3ST0239
120 A(ICOL,J)=TEMP	M3ST0240
II=II+1	M3ST0241
INDEX(II,2)=ICOL	M3ST0242
118 PIVOT=A(ICOL,ICOL)	M3ST0243
A(ICOL,ICOL)=1.0	M3ST0244
PIVOT=1./PIVOT	M3ST0245
DO 121 J=1,N	M3ST0246
121 A(ICOL,J)=A(ICOL,J)*PIVOT	M3ST0247
DO 122 I=1,N	M3ST0248
IF (I-ICOL)123,122,123	M3ST0249
123 TEMP=A(I,ICOL)	M3ST0250
A(I,ICOL)=0.0	M3ST0251
DO 124 J=1,N	M3ST0252
124 A(I,J)=A(I,J)-A(ICOL,J)*TEMP	M3ST0253
122 CONTINUE	M3ST0254
GO TO 108	M3ST0255
125 ICOL=INDEX (II,2)	M3ST0256
IROW= INDEX(ICOL,1)	M3ST0257
DO 126 I=1,N	M3ST0258
TEMP=A(I,IROW)	M3ST0259
A(I,IROW)=A(I,ICOL)	M3ST0260
126 A(I,ICOL)=TEMP	M3ST0261
II=II-1	M3ST0262
225 IF(II)125,127,125	M3ST0263
127 GO TO 130	M3ST0264
115 MSING=0	M3ST0265
130 RETURN	M3ST0266
END	M3ST0267

C	NO CARDS	VARIABLES	FORMAT	SAMI0001
C		INPUT SEQUENCE		SAMI0002
C	1	HEAD	20A4	SAMI0003
C	1	NSPAN, NTERM, NWS, NINT, IPLOT	5I5	SAMI0004
C	1	TL, SM, E, XI, XINT, WF1, WF2, WFINT, WSINT	9610.0	SAMI0005
C	1	(DEL(I), I=1, NSPAN-1)	10610.0	SAMI0006
C	1	(XL(I), I=1, NSPAN-1)	10610.0	SAMI0007
C	1	IF(NWS.EQ.0) SKIP THE FOLLOWING CARD		SAMI0008
C	1	(WS(I), I=1, NWS)	10610.0	SAMI0009
C	1	STDEF, STMOM, STSHR	3610.0	SAMI0010
C	1	TOPDEF, TOPMOM, TOPSHR	3610.0	SAMI0011
C				SAMI0012
C		IMPLICIT DOUBLE PRECISION(A-H, O-Z)		SAMI0013
C		REAL HEAD(20), AA(10000)		SAMI0014
C		DIMENSION A(5000), LKI(10000)		SAMI0015
C		LIMSIZ=10000		SAMI0016
C		COMMON AA		SAMI0017
C		EQUIVALENCE (AA(1), A(1)), (AA(1), LKI(1))		SAMI0018
C		COMMON /SETUP/ TL, SM, E, XI, EX, CX, PAI, WF1, WF2, WFINT, XINT, WSINT,		SAMI0019
C		1 NSPAN, NTERM, NS1, NOWF, NOWFT, IN, IO, IO2, NX, IPLOT,		SAMI0020
C		2 NWS, NINT		SAMI0021
C		IN=5		SAMI0022
C		IO=6		SAMI0023
C		IO2=3		SAMI0024
C		IDS=2		SAMI0025
C	998	READ(IN, 500, END=999) HEAD		SAMI0026
C	500	FORMAT(20A4)		SAMI0027
C		WRITE(IO, 600) HEAD		SAMI0028
C	600	FORMAT(1H1, 20A4)		SAMI0029
C		READ(IN, 501) NSPAN, NTERM, NWS, NINT, IPLOT		SAMI0030
C	501	FORMAT(5I5)		SAMI0031
C		READ(IN, 502) TL, SM, E, XI, XINT, WF1, WF2, WFINT, WSINT		SAMI0032
C	502	FORMAT(9610.0)		SAMI0033
C		NS1=NSPAN-1		SAMI0034
C		NS2=NS1-1		SAMI0035
C		NX=TL/XINT+1		SAMI0036
C		NOWF=(WF2-WF1)/WFINT+1		SAMI0037
C		NOWFT=NOWF+2*NINT*NWS		SAMI0038
C		IF(NS1.EQ.0) GO TO 996		SAMI0039
C		N2=1+NS1		SAMI0040
C		N3=N2+NS1		SAMI0041
C		N4=N3+NX		SAMI0042
C		N5=N4+NX		SAMI0043
C		N6=N5+NX		SAMI0044
C		N7=N6+NX		SAMI0045
C		N8=N7+NTERM		SAMI0046
C		N9=N8+NTERM		SAMI0047
C		N10=N9+(NS1*NTERM)		SAMI0048
C		N11=N10+NS1		SAMI0049
C		N12=N11+(NS1*NS1)		SAMI0050
C		N13=N12+NOWFT		SAMI0051
C		N14=N13+NOWFT		SAMI0052
C		N15=N14+NOWFT		SAMI0053
C		N16=N15+NOWFT		SAMI0054
C		N17=(N16+NWS)*IDS		SAMI0055
C		N18=N17+NOWFT		SAMI0056
C		N19=N18+NOWFT		SAMI0057
C		NSIZE=N19+2*NS1		SAMI0058
C		IF(NSIZE.GT.LIMSIZ) GO TO 997		SAMI0059
C		WRITE(IO, 601) NSIZE		SAMI0060
C	601	FORMAT(// ' ', 'BLANK COMMON ARRAY', I7)		SAMI0061
C		CALL MAIN(A(1), A(N2), A(N3), A(N4), A(N5), A(N6), A(N7),		SAMI0062
C		1 A(N8), A(N9), A(N10), A(N11), A(N12), A(N13), A(N14), A(N15),		SAMI0063
C		2 A(N16), AA(N17), AA(N18), LKI(N19))		SAMI0064
C		GO TO 998		SAMI0065
C	997	WRITE(IO, 602) NSIZE		SAMI0066
C	602	FORMAT(// ' ', 'YOUR BLANK COMMON ARRAY MUST BE', I7)		SAMI0067
C		GO TO 998		SAMI0068
C	996	WRITE(IO, 603) NSPAN		SAMI0069
C	603	FORMAT(// ' ', 'THIS IS NOT A MULTISPAN GIRDER', I5)		SAMI0070
C		GO TO 998		SAMI0071
C	999	STOP		SAMI0072
C		END		SAMI0073
C				SAMI0074

	SUBROUTINE MAIN(DEL,XL,X,PY,PM,PQ,PN,P,XN,RP,RCDEF,	SAMI0075
1	WF,DAFFY,DAFFM,DAFFQ,WS,RX,RY,INDEX)	SAMI0076
	IMPLICIT DOUBLE PRECISION(A-H,O-Z)	SAMI0077
	REAL RX(NOWFT),RY(NOWFT)	SAMI0078
	DIMENSION DEL(NS1),XL(NS1),X(NX),PY(NX),PM(NX),PQ(NX),	SAMI0079
	1 DN(NTERM),P(NTERM),XN(NS1,NTERM),RP(NS1),	SAMI0080
	2 RCOEF(NS1,NS1),WF(NOWFT),DAFFY(NOWFT),	SAMI0081
	3 DAFFM(NOWFT),DAFFQ(NOWFT),WS(NOWS),INDEX(2*NS1)	SAMI0082
	COMMON /SETUP/TL,SM,E,XI,EX,CX,PAI,WF1,WF2,WFINT,XINT,WSINT,	SAMI0083
	1 NSPAN,NTERM,NS1,NOWF,NOWFT,IN,IO,IO2,NX,IPLOT,	SAMI0084
	2 NWS,NINT	SAMI0085
C		SAMI0086
C	READ WRITE BLOCK	SAMI0087
C		SAMI0088
	READ(IN,500) (DEL(I),I=1,NS1)	SAMI0089
500	FORMAT(10G10.0)	SAMI0090
	READ(IN,500) (XL(I),I=1,NS1)	SAMI0091
	IF(NWS.EQ.0)GO TO 5	SAMI0092
	READ(IN,500) (WS(I),I=1,NWS)	SAMI0093
5	READ(IN,500)STDEF,STMOM,STSHR	SAMI0094
	READ(IN,500)TOPDEF,TPMOM,TPSHR	SAMI0095
	WRITE(IO,699) NSPAN,TL,SM,E,XI,NTERM	SAMI0096
699	FORMAT(//'0',NUMBER OF SPANS =',I10/	SAMI0097
8	' ',TOTAL SPAN LENGTH =',D10.4/	SAMI0098
1	' ',MASS DENSITY PER UNIT LENGTH =',D10.4/	SAMI0099
2	' ',YOUNG'S MODULUS =',D10.4/	SAMI0100
3	' ',MOMENT OF INERTIA =',D10.4/	SAMI0101
9	' ',NUMBER OF TERMS TO BE SUMMED =',I10/	SAMI0102
C	' ',SUPPORT DISTANCES FROM LEFT ABUTEMENT//)	SAMI0103
	DO 30 I=1,NS1	SAMI0104
30	WRITE(IO,698)I,XL(I)	SAMI0105
698	FORMAT(' ',SUPPORT NO,'I3,' =',D10.4)	SAMI0106
	WRITE(IO,697)	SAMI0107
697	FORMAT(/,' ',SUPPORT SETTLEMENTS//)	SAMI0108
	DO 35 I=1,NS1	SAMI0109
35	WRITE(IO,698)I,DEL(I)	SAMI0110
	MSING=1	SAMI0111
C		SAMI0112
C	COMMON FACTORS	SAMI0113
C		SAMI0114
	EX=E*XI/(SM*TL**4)	SAMI0115
	PAI=ATAN(1.D0)*4.	SAMI0116
	CX=DSQRT(2./(SM*TL))	SAMI0117
	DO 215 N=1,NTERM	SAMI0118
	PAIN=N*PAI	SAMI0119
	PAINTL=PAIN/TL	SAMI0120
	P(N)=DSQRT(PAIN**4*EX)	SAMI0121
	DO 215 I=1,NS1	SAMI0122
215	XN(I,N)=CX*DSIN(PAINTL*XL(I))	SAMI0123
C		SAMI0124
C	CALCULATION OF THE EXTRA FINE WF LOCATIONS NEAR THE SYSTEM	SAMI0125
C	FREQUENCIES	SAMI0126
C		SAMI0127
	DO 50 NWF=1,NOWF	SAMI0128
50	WF(NWF)=WF1+(NWF-1)*WFINT	SAMI0129
	IF(NWS.EQ.0)GO TO 35	SAMI0130
	DO 55 NWS=1,NWS	SAMI0131
	DO 55 I=1,NINT	SAMI0132
	INDX1=NOWF+(NWS-1)*NINT+I	SAMI0133
	WF(INDX1)=WS(NWS)+DFLOAT(I)*WSINT	SAMI0134
55	CONTINUE	SAMI0135
	DO 60 NWS=1,NWS	SAMI0136
	DO 60 I=1,NINT	SAMI0137
	INDX2=NOWF+NINT*NWS+(NWS-1)*NINT+I	SAMI0138
	WF(INDX2)=WS(NWS)-DFLOAT(I)*WSINT	SAMI0139
60	CONTINUE	SAMI0140
C		SAMI0141
C	REARRANGING THE WF ARRAY IN INCREASING ORDER	SAMI0142
C		SAMI0143
	DO 70 II=1,NOWFT-1	SAMI0144
65	DO 70 I=II+1,NOWFT	SAMI0145
	IF(WF(II).LE.WF(I)) GO TO 70	SAMI0146
	DUMMY=WF(II)	SAMI0147
	WF(II)=WF(I)	SAMI0148
	WF(I)=DUMMY	SAMI0149
70	CONTINUE	SAMI0150

WRITE(IO,602)	SAMI0225
602 FORMAT('1'//20X'MAXIMUM DYNAMIC DEFLECTIONS'//)	SAMI0226
WRITE(IO,605)	SAMI0227
605 FORMAT('//5X'FREQUENCY'//)	SAMI0228
DO 410 NWF=1,NOWFT	SAMI0229
WRITE(IO,604)WF(NWF),DAFFY(NWF)	SAMI0230
410 CONTINUE	SAMI0231
604 FORMAT(5D15.5)	SAMI0232
WRITE(IO,606)	SAMI0233
606 FORMAT('1'//20X'MAXIMUM DYNAMIC MOMENT VALUES'//)	SAMI0234
WRITE(IO,605)	SAMI0235
DO 420 NWF=1,NOWFT	SAMI0236
WRITE(IO,604)WF(NWF),DAFFM(NWF)	SAMI0237
420 CONTINUE	SAMI0238
WRITE(IO,608)	SAMI0239
608 FORMAT('1'//20X'MAXIMUM DYNAMIC SHEAR VALUES'//)	SAMI0240
WRITE(IO,605)	SAMI0241
DO 430 NWF=1,NOWFT	SAMI0242
WRITE(IO,604)WF(NWF),DAFFQ(NWF)	SAMI0243
430 CONTINUE	SAMI0244
WRITE(IO,610)	SAMI0245
610 FORMAT('1'//10X'MAXIMUM DYNAMIC AMPLIFICATION FACTORS FOR'	SAMI0246
1 ' DEFLECTION FUNCTIONS'//)	SAMI0247
WRITE(IO,605)	SAMI0248
DO 440 NWF=1,NOWFT	SAMI0249
440 DAFFY(NWF)=DABS(DAFFY(NWF)/STDEF)	SAMI0250
DO 450 NWF=1,NOWFT	SAMI0251
WRITE(IO,604)WF(NWF),DAFFY(NWF)	SAMI0252
450 CONTINUE	SAMI0253
WRITE(IO,612)	SAMI0254
612 FORMAT('1'//10X'MAXIMUM DYNAMIC AMPLIFICATION FACTORS FOR'	SAMI0255
1 ' MOMENT FUNCTIONS'//)	SAMI0256
WRITE(IO,605)	SAMI0257
DO 460 NWF=1,NOWFT	SAMI0258
460 DAFFM(NWF)=DABS(DAFFM(NWF)/STMOM)	SAMI0259
DO 470 NWF=1,NOWFT	SAMI0260
WRITE(IO,604)WF(NWF),DAFFM(NWF)	SAMI0261
470 CONTINUE	SAMI0262
WRITE(IO,614)	SAMI0263
614 FORMAT('1'//10X'MAXIMUM DYNAMIC AMPLIFICATION FACTORS FOR'	SAMI0264
1 ' SHEAR FUNCTIONS'//)	SAMI0265
WRITE(IO,605)	SAMI0266
DO 480 NWF=1,NOWFT	SAMI0267
480 DAFFQ(NWF)=DABS(DAFFQ(NWF)/STSHR)	SAMI0268
DO 490 NWF=1,NOWFT	SAMI0269
WRITE(IO,604)WF(NWF),DAFFQ(NWF)	SAMI0270
490 CONTINUE	SAMI0271
IF(IPLOT.NE.1)GO TO 1000	SAMI0272
C	SAMI0273
C	SAMI0274
C	SAMI0275
PLOTTING BLOCK	SAMI0276
CALL LIMITS(0.,11.,50.)	SAMI0277
DO 525 NWF=1,NOWFT	SAMI0278
RX(NWF)=WF(NWF)	SAMI0279
525 CONTINUE	SAMI0280
DO 540 NWF=1,NOWFT	SAMI0281
IF(DAFFY(NWF).GT.TOPDEF)DAFFY(NWF)=TOPDEF	SAMI0282
RY(NWF)=DAFFY(NWF)	SAMI0283
540 CONTINUE	SAMI0284
CALL EPLOT(RX,RY,NOWFT,0,1,5)	SAMI0285
DO 550 NWF=1,NOWFT	SAMI0286
IF(DAFFM(NWF).GT.TOPMOM)DAFFM(NWF)=TOPMOM	SAMI0287
RY(NWF)=DAFFM(NWF)	SAMI0288
550 CONTINUE	SAMI0289
CALL EPLOT(RX,RY,NOWFT,0,1,5)	SAMI0290
DO 560 NWF=1,NOWFT	SAMI0291
IF(DAFFQ(NWF).GT.TOPSHR)DAFFQ(NWF)=TOPSHR	SAMI0292
RY(NWF)=DAFFQ(NWF)	SAMI0293
560 CONTINUE	SAMI0294
CALL EPLOT(RX,RY,NOWFT,0,1,5)	SAMI0295
GO TO 1000	SAMI0296
998 WRITE(IO,620)	SAMI0297
620 FORMAT('/// A SINGULAR MATRIX WAS ENCOUNTERED')	SAMI0298
1000 RETURN	SAMI0299
END	

```

C
C   CALCULATION OF THE DEFLECTION, MOMENT, SHEAR ALONG THE SPAN
C
      DO 999 NWF=1, NWFNT
      WF2=WF(NWF)**2
      PYMAX=0.
      PMMAX=0.
      PQMAX=0.
      DO 105 I1=1, NS1
      DO 105 I2=1, NS1
105  RCOEF(I1, I2)=0.
      DO 150 N=1, NTERM
      PAIN=PAI*N
      PAINTL=PAIN/TL
      DEN=P(N)**2-WF2
      ADEN=DABS(DEN)
      IF(ADEN.LT..1D-50) GO TO 150
      DO 145 I1=1, NS1
      DO 145 I2=1, NS1
      RCOEF(I1, I2)=RCOEF(I1, I2)+XN(I1, N)*XN(I2, N)/DEN
145  CONTINUE
150  CONTINUE
      IF(NS1.GT.1)GO TO 170
      RP(1)=DEL(1)/RCOEF(1, 1)
      GO TO 185
170  CALL SIMULE(NS1, RCOEF, DEL, INDEX, MSING)
      IF(MSING.EQ.0)GO TO 998
      DO 175 I=1, NS1
175  RP(I)=0.
      DO 180 I=1, NS1
      DO 180 J=1, NS1
180  RP(I)=RP(I)+RCOEF(I, J)*DEL(J)
185  DO 777 K=1, NX
      X(K)=(K-1)*XINT
      PY(K)=0.
      PM(K)=0.
      PQ(K)=0.
      DO 45 N=1, NTERM
      PAIN=N*PAI
      PAINTL=PAIN/TL
      DEN=P(N)**2-WF2
      ADEN=DABS(DEN)
      IF(ADEN.LT..1D-50) GO TO 45
      DN(N)=0.
      DO 44 I=1, NS1
44  DN(N)=DN(N)+RP(I)*XN(I, N)/DEN
      XT=CX*DSIN(PAINTL*X(K))
      XTC=CX*DCOS(PAINTL*X(K))
      PY(K)=PY(K)+XT*DN(N)
      PM(K)=PM(K)+(PAINTL**2*XT*DN(N))*E*XI
      PQ(K)=PQ(K)+(PAINTL**3*XTC*DN(N))*E*XI
45  CONTINUE
      DO 190 I=1, NS1
      IF(X(K).NE.XL(I))GO TO 190
      PQ(K)=0.
190  CONTINUE
C
C   STORAGE OF MAXIMUM VALUES ALONG THE SPAN FOR EACH FORCING
C   FREQUENCY
C
      IF(DABS(PY(K)).LE.DABS(PYMAX))GO TO 305
      PYMAX=PY(K)
305  IF(DABS(PM(K)).LE.DABS(PMMAX))GO TO 310
      PMMAX=PM(K)
310  IF(DABS(PQ(K)).LE.DABS(PQMAX))GO TO 777
      PQMAX=PQ(K)
777  CONTINUE
      DAFFY(NWF)=PYMAX
      DAFFM(NWF)=PMMAX
      DAFFQ(NWF)=PQMAX
999  CONTINUE
C
C   OUTPUT OF RESULTS
C

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SAMI0151
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SAMI0210
SAMI0211
SAMI0212
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SAMI0215
SAMI0216
SAMI0217
SAMI0218
SAMI0219
SAMI0220
SAMI0221
SAMI0222
SAMI0223
SAMI0224

C	SUBROUTINE SIMULE(N,A,B,INDEX,MSING)	SAMI0300
C	THIS IS AN INVERSION AND SIMULTANEOUS EQUATION SOLVER	SAMI0301
C	IMPLICIT DOUBLE PRECISION(A-H,O-Z)	SAMI0302
	DIMENSION INDEX(N,2),A(N,N),B(1)	SAMI0303
	DO 100 I=1,N	SAMI0304
	INDEX(I,1)=0	SAMI0305
100	CONTINUE	SAMI0306
	II=0	SAMI0307
109	AMAX=-1	SAMI0308
	DO 110 I=1,N	SAMI0309
	IF (INDEX(I,1))110,111,110	SAMI0310
111	DO 112 J=1,N	SAMI0311
	IF (INDEX(J,1))112,113,112	SAMI0312
113	TEMP=0ABS(A(I,J))	SAMI0313
	IF (TEMP-AMAX)112,112,114	SAMI0314
114	IROW=I	SAMI0315
	ICOL=J	SAMI0316
	AMAX=TEMP	SAMI0317
112	CONTINUE	SAMI0318
110	CONTINUE	SAMI0319
	IF(AMAX)225,115,116	SAMI0320
116	INDEX(ICOL,1)=IROW	SAMI0321
	IF (IROW-ICOL)119,118,119	SAMI0322
119	DO 120 J=1,N	SAMI0323
	TEMP=A(IROW,J)	SAMI0324
	A(IROW,J)=A(ICOL,J)	SAMI0325
120	A(ICOL,J)=TEMP	SAMI0326
	II=II+1	SAMI0327
	INDEX(II,2)=ICOL	SAMI0328
118	PIVOT=A(ICOL,ICOL)	SAMI0329
	A(ICOL,ICOL)=1.0	SAMI0330
	PIVOT=1./PIVOT	SAMI0331
	DO 121 J=1,N	SAMI0332
121	A(ICOL,J)=A(ICOL,J)*PIVOT	SAMI0333
	DO 122 I=1,N	SAMI0334
	IF (I-ICOL)123,122,123	SAMI0335
123	TEMP=A(I,ICOL)	SAMI0336
	A(I,ICOL)=0.0	SAMI0337
	DO 124 J=1,N	SAMI0338
124	A(I,J)=A(I,J)-A(ICOL,J)*TEMP	SAMI0339
122	CONTINUE	SAMI0340
	GO TO 109	SAMI0341
125	ICOL=INDEX (II,2)	SAMI0342
	IROW= INDEX(ICOL,1)	SAMI0343
	DO 126 I=1,N	SAMI0344
	TEMP=A(I,IROW)	SAMI0345
	A(I,IROW)=A(I,ICOL)	SAMI0346
126	A(I,ICOL)=TEMP	SAMI0347
	II=II-1	SAMI0348
225	IF(II)125,127,125	SAMI0349
127	GO TO 130	SAMI0350
115	MSING=0	SAMI0351
130	RETURN	SAMI0352
	END	SAMI0353
		SAMI0354
		SAMI0355